Apollo 15 Flight Plan

Please note that most of the hand-written additions to this document were added during the compilation of the Apollo 15 Flight Journal in 1998 to 2000. To a large extent, they reflect changes read up to the crews during the course of the mission.

David Woods – Editor: Apollo Flight Journal



NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

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AS-510/CSM-112/LM-10 FINAL

FLIGHT PLAN

PREPARED BY

FLIGHT PLANNING SECTION
FLIGHT PLANNING BRANCH
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MANNED SPACECRAFT CENTER
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APOLLO 15

(JULY 26, 1971)

FLIGHT PLAN

FINAL

JUNE 21, 1971

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FLIGHT PLAN

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ABBREVIATIONS

ABB abbreviation or abbreviated AC alternating current

ACCEL accelerometer
ACN Ascension
ACT activation

ACQ acquisition or acquire
AEA abort electronics assembly
AGS abort guidance subsystem

AH ampere hours

ALSCC Apollo lunar surface close-up camera

ALSD Apollo lunar surface drill

ALSEP Apollo lunar surface experiment package

ALT altitude ALTM altimeter

AH amplitude modulation

AMP or amp amperes
AMPL amplifier
AMG Antigua
ANT antenna

AOH Apollo Operations Handbook

AOL Atlantic Ocean line

AOS acquisition of signal or acquisition of site

AOT alignment optical telescope AP alpha particle spectrometer APS ascent propulsion subsystem

ARIA Apollo range instrumentation aircraft ARS atmosphere revitalization system

ASC ascent

A/T alignment technique

ATT attitude
AUX auxiliary
AZ azimuth

BAT battery

BEF blunt end forward

BD band BDA Bermuda

BIOMED bio-medical data

BKWD backward

BMAG body mounted attitude gyro

BP barber pole BRKT bracket

BSLSS buddy secondary life support system

BT burn time BU backup

BW black and white (Film 3400)

```
BWD
            backward
            black and white (Film 3401)
BWl
            capsule communicator
CAP COM
CAL.
            calibration
CAMR or CAM camera
            circuit breaker
CB.
CCGE
            cold cathode gage experiment
CCIG
            cold cathode ion gage
CCW
            counter clockwise
CDH
            constant delta altitude
CDR
            Commander
            coupling data unit
CDU
            color exterior (S0368)
CEX
CIN
            color interior (S0168)
CIRC
            circulation
CK
            check
            circuit
CKT
C/L
            centerline or checklist
             command module
CM
CMC
             command module computer
CMD
             command
             Command Module Pilot
CMP.
CNTL
             control
0/0
             check out
             crew optical alignment sight
COAS
COMM
             communications
CONFIG
             configuration
COMP
             compare or compensate
             continue or contingency
CONT
CP
             control point
             charged particle lunar environment experiment
CPLEE
             Carnarvon, Australia
CRO
CRYO
             cryogenic
CS
             contingency sample
CSI
             coelliptic sequence initiation
CSM
             command and service modules
CST
             central standard time
C/S
             central station
CTR
             center
             caution and warning system
C&WS
CW
             clockwise
CWEA
             caution and warning electronics assembly
CWG
             constant wear garment
             Grand Canary Island -
CYI
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```
DAC
            data acquisition camera
            digital auto pilot
DAP
            deadband
DB
            direct current or data camera (70mm)
DC
            digital command assembly
DCA
            data entry and display assembly
DEDA
            degrees
DEG
            depletion
DEPL
            descent
DES
            digital event timer
DET
DIFF
            difference
DIR
            direct.
            docked
DK
DO
            detailed objective
            descent orbit insertion
DÛI
DPLY
            deployment
            descent propulsion system
DPS
DR
            dome removal tool
DRT
            documented sample
DS
            discriminator
DSCRM
            data storage equipment(CSM)
DSE
            data storage equipment assembly (LM)
DSEA
            display and keyboard
DSKY
            deep space measurement
DSM
            detailed test objective
DTO.
DUA
            digital uplink assembly
DMN
            dawn
Ē
            erasable or enter
ECS
            environmental control system
ED
            explosive device
EDT
             eastern daylight time
ËFH
             earth far horizon
             earth (atmosphere) interface and entry interface
ΕI
EKG
             electrocardiogram
             electric Hasselblad camera
EL
ELEY
             elevation
EMER
             emergency
EMS
             entry monitor system
             extravehicular mobility unit
EHU
ENG
             engine
ENH
             earth near horizon
ENT
             entry
E.O.
             earth orbit
EOM
             end of mission
EP0
             earth parking orbit
EPHEM
             Ephemeris
```

```
EPS.
              electrical power subsystem
 EQUIP
              equipment
 ERECT
              erectable
 ERR
              error
 EST
              eastern standard time
 ETB.
              equipment transfer bag
 E۷
              extravehicular
 EVA.
              extravehicular activity
 EVAP
              evaporator
 EVCS
             extravehicular communications system
 EVT
              extravehicular transfer
 EXP
              experiment
 EXT
             external or extend
 f
             f-stop
             familiarize or familiarization
 FAM
 FC.
             fuel cell
 FCS
             fecal containment system
             flight director attitude indicator
 FDAI
FLT
             f]iqht
FM
             frequency modulated
FOV
             field of view
FPS
             feet per second
fps
             frames per second
FR
             frame(s)
FREQ
             frequency
FT or ft
             feet
FTO
             flight test objective
             full throttle position
FTP
FTT
             fuel tranfer tool
FWD
             forward
G.A.
             gas analysis
GA
             gimbal angle
GBI
             Grand Bahama Islands
GBM
             Grand Bahama (MSFN)
GDC
             gyro display coupler
GDS
            Goldstone, California
GET
            ground elapsed time
            ground elapsed time of ignition
GETI
            ground elapsed time of landing for TIG time of abort burn
GETIL
GLY.
            glycol
GI/IT
            Greenwich mean time
G&C
            guidance and control
G&N
            guidance and navigation
GNCS
            guidance, navigation and control system (CSM)
GR
            gamma ray spectrometer
GMM
            Guam
GYM
            Guaymas, Mexico
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H2
              hydrogen
 HA
              apogee altitude
 HAW
              Hawaii
 HBR
             high bit rate (TLM)
 HD
             highly desirable
 HDC
              hasselblad data camera
 HFE
             heat flow experiment
 HGA.
             high-gain antenna
             high (switch position)
 ΗI
 HOR
             horizon
H20
             water
ΗP
             perioce altitude
HR
             hour(s)
HSK
             Homeysuckle (Camberra, Australia)
HTC
             hand tool carrier
HTR
             heater
HTV
             USNS Huntsville
ICDU
             inertial coupling data unit
ΙD
             identification
ICG
             inflight coverall garment
IGA
             inner gimbal angle
IGN
             ignition
IMU
             inertial measurement unit
INCR
             increase.
IND
             indicator
INIT
             initialization
TMI
             interval
ĬΡ
             initial point
ISA
             interim stowage assembly
IU
             instrumentation unit
IVC
             intervehicular communications
IVL
             intervalometer
IVI
             intravehicular transfer
iR.
            inclination of the ascending return
JETT
            jettison
KG.
            kilogram
KM.
            kilometer
kwh.
            kilowatt hour
LA
            launch azimuth or laser altimeter
LAT
            latitude
LBR
            low bit rate (TLM)
LB or 1b
            pound(s)
LCG
            liquid cooled garment
LCRU
            lunar communications relay unit
```

```
lift/drag
L/D
            lunar day (TV lens)
LD
            landing
LDG
            landmark
LDMK
LEB
            lower equipment bay
            lunar equipment conveyor
LEC
            lunar extravehicular visor assembly
LEVA
            lunar far horizon
LFH
            LM guidance computer
LGC
LH
            left-hand
            local horizontal
L/H
            left-hand equipment bay
LHEB
            left-hand forward equipment bay
LHFEB
            left-hand side storage container
LHSSC
            lithium hydroxide
LiOH
             lunar landing mission
LLM
             landmark line of sight
LLOS
             lunar module
LM
             Lunar Module Pilot
LMP
             lunar near horizon
LNH
             lift-off
L/0
             lunar orbit insertion
LOI
             longitude
LONG
             loss of signal or loss of site
LOS
             landing point designator
LPD
             lumar parking orbit
LP0
             lunar portable magnetometer
LPM
             landing radar
LR
LRRR or LR3 laser ranging retro-reflector
             lunar roving vehicle
             landing site or lunar surface
 L/S or LS
             lunar surface magnetometer
 LSM
 LT
             light
             lighting
 LTG
             lubrication
 LUB
             launch vehicle
 LV
             local vertical
 L/V
              launch vehicle pressure display
 LVPD
             mandatory
             Madrid, Spain
 MAD
              magazine (camera)
 MAG
              manual
 MAN
              maximum
 MAX
              maximum dynamic pressure
 MAX Q
              medium black and white film
 MBW
              mapping camera
 MC
              midcourse correction
 MCC
```

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44	
MCC-H	Mission Control Center - Houston
MDC	main display console
MEAS	measurement
MED	medical
MESA	modular experiment stowage assembly
MET	mission event timer
MGA	middle gimbal angle
M/I	minimum impulse
MIN	minimum or minutes(s)
MIR	mirror .
MLA	Merrit Island, Florida, launch area
nm or MM	
MINA or MNB	
MNVR	maneuver
MON	monitor
MONO	monaural
MPL	mid-Pacific line
MPS	main propulsion system
M/R	mixture ratio (fuel to oxidizer)
MS	mass spectrometer
MSFN	Manned Space Flight Network
MSO	mass spectrometer outgasing
MTVC	manual thrust vector control
MULT	multiple
N2	nitrogen
NAV	navigation
NK	nikon camera
NM	nautical miles
NO.	number
NOM	nominal
NXX	Noun XX
02	AVIAOD
	oxygen
OBS	observation
0/F	oxidizer to fuel ratio
OGA	outer gimbal angle
OID	octal identifier
OMNI	omnidirectional antenna
OPR	operate
OPS	oxygen purge system
	option
OPT	
ORB	orbital
ORDEAL	orbit rate display earth and lunar
ORIENT	orientation
OVBD	overboard
OVHD	overhead
~	erecter.

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pitch or program PAD voice update PAN panoramic PART particle PCM pulse code modulation PC plane change or chamber pressure PDI powered descent initiation PER or PC Pericynthion PGA pressure garment assembly **PGNCS** primary guidance, navigation and control system (LM) PGNS primary guidance navigation system (LM) PHOTO photograph PIPA pulse integrating pendulous accelerometer PKG package PLSS portable life support system PM phase modulated POL polarity or polarizing PRD passive radiation dosimeter PREF preferred PREP preparation PRESS pressure PRIM primary PROP proportional PRN pseudo random noise PRPLNT propellant PSE passive seismic experiment PSIA pounds per square inch absolute PSID pounds per square inch differential PSIG pounds per square inch gage PΤ point PTC passive thermal control рU propellant utilization PUGS propellant utilization gaging system PWR power PXX Program XX PYRO. pyrotechnic OTY quantity QUAD quadrant R roll or range R&B red and blue RAD radiator, radial, or radiation RCDR recorder RCS reaction control system RCU remote control unit

RCV

REACO

receiver

reacquire

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```
REFSMMAT
             reference stable member matrix
REG
             regulator
REL
             release
REOD
             required
RETR
             retract
REV
             revolution
RH
             right-hand
             rotational hand controller
RHC
RING
             ringsight
RLS
             radius of landing site
RMT
RNDZ

    rendezvous

RNG
             range or ranging
ROD
             rate of descent
RR
             rendezvous radar
RSI
             roll stability indicator
RSLV
             resolver
             realtime.
RT
RTC
             realtime command
             radioisotope thermoelectric generator
RTG
RXX
             Routine XX
SA
             shaft angle
SATT
             satellite
S~BD
             S-BAND
SC
             spacecraft
SCE
             signal conditioning equipment
SCS
             stabilization control system
SCT
             scanning telescope
             southeast or subearth
SE
SEC
             secondary
SECO
             S-IVB engine cutoff
SECS
             sequential events control system
SEF
             sharp end forward
$EL
             select
SEP
             separate
SEQ
             sequence
SEVA
             standup extravehicular activity
SIDE
             suprathermal ion detector experiment
SII
             Saturn II (second stage)
             scientific instrument module
Saturn IVB(third stage)
SIM
S-IVB
SLA
             service module LM adapter
SLOS
             star line-of-sight
SM
             service module
SPECT
             spectrometer
SPOT
             spot meter
```

SPS service propulsion system

SR sunrise

SRC sample return container SRX S-Band receiver mode no. X

SS sunset or subsolar

STBY standby

STX S-Band transmit mode no. X

SUBSAT subsateIlite s.v. state vector

SW switch

SWC solar wind composition SWE solar wind experiment

SXT sextant SYS system

T EPHEM time of Ephemeris update

TA trunnion angle

TAN Tananarive, Madagascar
TB time base or talkback
TCA time of closest approach

TD touchdown

T&D transposition and docking

TD&E transposition docking and LM ejection

TDS thermal degradation sample

TEC transearth coast

TECH technique

TEI transearth injection

TEMP temperature TERM terminate

TEX Corpus Christi, Texas

TGT target

THC translation hand controller

TIG time of ignition
TLC translunar coast
TLI translunar injection

TLM or TM telemetry

TPF terminal phase final TPI terminal phase initiation terminal phase midcourse T/R transmitter/receiver

TRANS translation

TRK track or tracking

TRUN trunnion television

TVC thrust vector control

TWR tower

```
UCTA
              urine collection transfer assembly
UHT
              universal hand tool
ULL
              ullage
UMB
              umbilical
UNBAL.
              unbalance (meter)
UNDK
              undock
US
              United States
U٧
              ultraviolet
٧
              velocity
{\rm VG}_{\rm IMU}
              velocity to be gained as related to IMU orientation
              velocity to be gained (X-body axis)
VGX.
VGY.
              velocity to be gained (Y-body axis) velocity to be gained (Z-body axis)
VGZ.
VR.
              resultant velocity
VΧ
             velocity along the X-axis velocity along the Y-axis
VY
٧Z
              velocity along the Z-axis
VAN
              USNS vanguard
             very high speed black and white film
VHBW
VHF
             very high frequency
VLV
             valve
VOX
             voice keying
XXV
             Verb XX
WRT
             with respect to
Х
             time of closest approach (symbol)
             rate of change along the X-axis
XDOT
XFER
             transfer
TIMX
             transmit or transmitter
XPNDER XPNDRtransponder
XR
             X-ray spectrometer
Υ
YDOT
             rate of change along the Y-axis
ZDOT
             rate of change along the Z-axis
ZPN
             impedance pneumogram
```

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ΔAz	azimuth change (difference)
ΔΗ	altitude change (difference)
ΔΡ	pressure change (difference)
ΔR	position change (difference)
ΔV	velocity change (difference)
ΔVC	velocity change at engine cutoff
ΔVT	velocity change loaded pre-burn
#	frame number(s) (for camera data)
ф	latitude
λ	longitude

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PHOTOGRAPHIC NOMENCLATURE

AAA/BBB/CCC/DDD - EEE, EEE, (fGG, HHH, III) (JJ fps or JJ FR) (KK MIN) (LL% MAG)

AAA - Location from which photography is to be accomplished

88B - Camera

CCC - Lens

DDD - Film Type

EEE - Photography aids (i.e., brackets, intervalometer, mirror, etc.)

fGG - Lens Aperture Setting

HHH - Shutter Speed

III - Focus Distance in Feet

JJ - Number of frames for DC, EL & NK cameras or

JJ - Frame Rate for the DAC only

KK - Magazine percent for the DAC only

KK - Operating time (minutes) for TV

LL - Magazine percent for the DAC only

CODE EXAMPLE:

CM4/DAC/18-CEX-BRKT, SPOT (S.1/250.∞)(12 fps)(4 min)(50% MAG)

Meaning: Photos are taken from CM right hand rendezvous window using the DAC with 18mm lens and SO368 film. The camera will be bracket mounted with the following camera settings: f-stop from spotmeter reading, shutter speed 1/250 of a second, focus at infinity, 12 frames per second, 50% MAG for 4 min to be used.

CM4/EL/80/BW-BRKT, IVL (f5.6,1/250,∞) 10 FR

Meaning: Photos are taken from CM right hand rendezvous window using the Electric Hasselblad camera with the 80mm lens and black & white film (3400). The camera will be bracket mounted with the following settings f-stop (aperture) f5.6, shutter speed 1/250, and focus at infinity. The operation of the shutter will be controlled by the intervelometer. Ten frames have been alloted for this sequence.

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PHOTOGRAPHIC NOMENCLATURE (CONT)

CAMERA LOCATIONS

COMMAND MODULE

CM-1	LH Side Window
CM-2	LH Rendezvous Window
CM-3	Hatch Window
CM-4	RH Rendezvous Window
CM-5	RH Side Window

LUNAR MODULE

LM-1	LH Window
LM-2	Docking Window
LM-3	RH Window

CAMERA MOUNTS

CSM

Electric Hasselblad (EL) +X axis +12°

Electric Hasselblad (EL) normal to RM Side Window

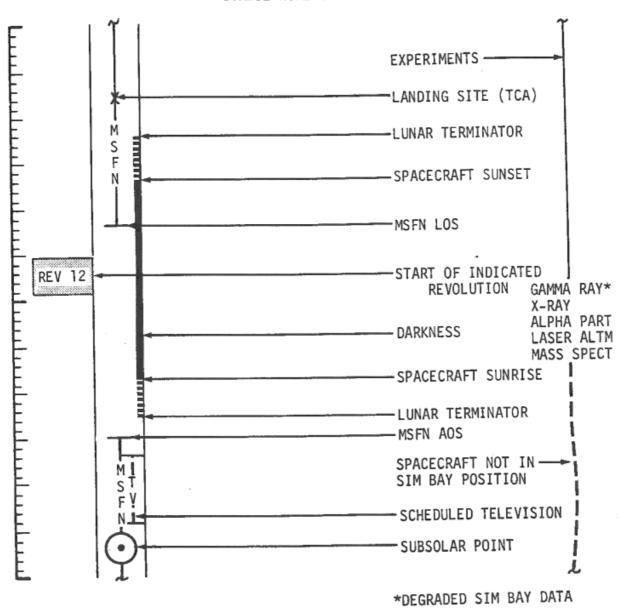
Data Acquisition Camera with right angle mirror (DAC) +X axis

Data Acquisition Camera with SXT Adapter - same as SXT shaft & trunnion

NIKON (NK) Two positions parallel to +X axis

+X axis +30°

SYMBOL NOMENCLATURE



FLIGHT PLAN NOTES

I. Crew

A. Crew designations are as follows:

Designation	Prime	Backup
Commander (CDR)	Scott	Gordon
Command Module Pilot (CMP)	Worden	Brand
Lunar Module Pilot (LMP)	Irwin	Schmitt

B. The nominal CM couch positions are:

Activity	Left	Center	<u>Right</u> LMP
Launch thru TLI	CDR	CMP	LMP
T&D thru Entry	CMP	CDR	LMP

- C. The PGA's are worn as shown in Table 2-1.
- D. The Crew Biomedical Harness and sensor wearing schedule is shown in Table 2-2.
- E. Crew status reports are voiced to MCC-H before and after crew sleep periods. After waking, the crew reports sleep obtained during the last 24 hours and personal dosimeter readings. Before going to sleep, the crew reports medication used and any other pertinent information on activities performed.
- F. Negative reporting is used in reporting completion of each checklist.
- G. All onboard gauge readings are read directly from the gauges with no calibration bias applied.

II. CSM Systems

A. Communications

- 1. The preferred S-Band communication modes are:
 - (a) Uplink Mode 6 (Voice, PRN, and Updata)
 - (b) Downlink Mode 2 (Voice, PRN, TLM-HBR)
- VHF Duplex B is used for launch, and Simplex A is used for earth-orbit operations.
- Table 2-3 summarizes the MSFN coverage available for the CSM.

- 4. Table 2-4 contains a summary of the scheduled CSM & LM TV transmissions.
- 5. MCC-H switches OMNI antennas during TLC and TEC rest periods and TLC PTC awake periods. MCC-H switches OMNI and HGA during TEC PTC awake periods. The crew manages antenna operations during all other TLC and TEC periods.
- To optimize the return of SIM Bay data, the crew will manually acquire with the HGA during awake period.

B. DSE

- During the earth-orbit phase, the CSM LBR data is recorded when the CSM is not within MSFN coverage. The DSE is dumped during the pass over the US prior to TLI if possible.
- 2. CSM HBR data will be recorded during all P24 landmark tracking.
- CSM HBR will be recorded during all CSM engine Burns, at Sim Door Jettison, TD&E, and during DAC/SXT Photo test.
- LM LBR data will be recorded during LOS periods between CSM/LM Separation and PDI.
- 5. Along with manually acquiring MSFN with the HGA, the crew, instead of the ground, will rewind the DSE prior to the manual HGA acquisition to optimize the return of SIM Bay data. The crew will rewind manually for playback at a specific time.
- 6. All entry data will be recorded in HBR during the blackout.

C. Electrical Power

- The CSM normally remains powered up throughout the mission.
- Table 2-5 lists the fuel cell purges.
- 3. Based on cryo purity and performance, the time between fuel cell 02 purges may be increased to coincide with water dump times. The 02 purge at 10.5 hours allows a judgement to be made on the defined purge schedule.

- 4. The cryogenic heaters are managed such that the planned usage is obtained out of each 0₂. The H₂ fans are operated manually for one minute before and after each sleep cycle, prior to SPS or S-IVB thrusting and pre-CSM/LM ejection.
- 5. Table 2-6 contains the battery charge schedule.
- D. ECS and Water Management
 - 1. Potable water is chlorinated once a day after the eat period prior to each sleep period.
 - Waste water dump and fuel cell purge criteria:
 - (a) Waste water dumps, fuel cell purges, and urine dumps in lunar orbit are scheduled at the following times: (see table 2-5 for the scheduled fuel cell purges and waste water dumps)
 - Once during each 24 hours, if possible, following the initial dump and purge
 - (2) ${\rm H_2}$ fuel cell purges are scheduled at every other ${\rm O_2}$ fuel cell purge after the first O2 fuel cell purge
 - (b) The most opportune time to perform waste water dumps and fuel cell purges are as follows:
 - (1) Immediately <u>after</u> the sextant star check in maneuver preparation or cislunar navigation
 - (2) Behind the moon, with completion of dump or purge before AOS
 - (c) If possible, dumps and purges are not scheduled during the following periods:
 - Ten hours before MCC-2 or a TLC P23 or until just before the midcourse change only
 - (2) Eight hours before MCC-5

- (d) Dumps and purges are <u>not</u> scheduled during the following MSFN tracking periods:
 - (1) Between MCC-4 and LOI
 - (2) Four hours before DOI
 - (3) Ten hours before MCC-7 until entry
- (e) All waste water dumps are manual.
- 3. Only one CO2 absorber filter (LiOH canister) is changed at a time. Table 2-7 lists the LiOH canister change schedule. There are 30 filters on board, with 28 stowed at launch, only 23 are required.
- 4. At lift-off, the cabin contains 60% 0₂ and 40% N₂. The CM ispurged after launch. The purge is terminated prior to LM pressurization after TLI. After the LM is configured for ejection, it is isolated and the CM is purged for eight more hours.
- 5. CSM $\rm O_2$ pressurizes the LM after transposition and docking; and repressurizes the LM before TLC LM entry, Sim Door Jett and LM activation.
- E. Guidance and Navigation
 - REFSMMAT Definitions
 - (a) The "Launch Pad" REFSMMAT is used for launch, TLI, and TD&E. This REFSMMAT places the IMU X-axis along the launch azimuth at the pad and the Z-axis along the negative radius vector. The FDAI, at launch, will display roll 170° (launch azimuth +90°), pitch 90°, and yaw 0°.

- (b) The "PTC" REFSMMAT is used for all midcourse maneuvers (except MCC-7) and for other operations during TLC and TEC. This REFSMMAT places the X-axis in the ecliptic plane and perpendicular to the earth-moon line projection in the ecliptic plane at the average time of transearth injection for the monthly launch window and azimuth range. The Z-axis is then perpendicular to the ecliptic and directed south. At the beginning of the PTC Mode the spacecraft maneuvers to an FDAI display of pitch 90°.
- (c) A "Preferred" REFSMMAT is used by the CSM for LOI, Lunar-Orbit Plane Change, and TEI. The CSM IMU X-axis aligns normally with the spacecraft X-body axis at the vehicle attitude for ignition with the thrust directed through the center of gravity. At burn ignition, the FDAI displays roll 0°, pitch 0°, and yaw 0°. In the case of the DOI TRIM burn, the IMU X-axis may be aligned 45° from the spacecraft body axis at ignition attitude. The Z-axis is in the plane formed by the X-axis and the position vector and directed down toward the moon.
- (d) The "Landing Site" REFSMMAT is used for DOI, PDI, landing, and CSM lunar orbit activities up to the second plane change. This REFSMMAT places the CSM IMU X-axis along the positive lunar radius vector at the landing site at the predicted landing time and places the Z-axis in the direction of flight parallel to the CSM orbital plane. At nominal touchdown, the LM FDAI displays roll 0°, pitch 0°, and yaw 0°.
- (e) The "Lift-Off" REFSMMAT is used for all lunar activities after plane change 2, through rendezvous, and LM jettison. This REFSMMAT places the CSM IMU X-axis along the positive lunar radius vector at the landing site at predicted lift-off time, with the Z-axis down range parallel to the CSM orbital plane. At nominal lift-off time, the LM FDAI displays roll 0°, pitch 0°, and yaw 0° with slight differences reflecting actual touchdown yaw and slope tilt angles.
- (f) The "Entry" REFSMMAT aligns the IMU X-axis in the local horizontal plane in the direction of flight at entry interface. The entry REFSMMAT is used for MCC-7 and all remaining activities. The Z-axis is down along the negative radius at entry interface. At entry interface, with wings level, local horizontal, heat shield for-

ward, inplane, lift up, heads down, the FDAI displays roll 0°, pitch 180°, and yaw 0°.

- 2. The CSM external lighting is operated during the rendezvous from lift-off to docking. The running lights only are on from CSM/LM separation to PDI.
- 3. The time tags on maneuvers in Section 3 indicate the completion time of the maneuvers unless otherwise stated. All maneuver angles are the angles read on the FDAI after the maneuver has been completed.
- 4. CSM/LM and CSM attitude maneuvers are normally performed at the rate of 0.2°/sec (0.5°/sec after rendezvous and docking) unless other rates are required. LM maneuvers are normally performed at 2°/sec unless otherwise specified.
- 5. The SIM Bay jett configuration provides single jet control authority in each axis to eliminate RCS contamination of the SIM experiments. See Table 2-8 for the period in lunar orbit when the CSM is in a non-SIM Bay configuration.
- 6. Undocking is done radially, CSM below, using the soft undocking procedure. The probe is extended its full length with the LM held on by the capture latches. When the rates are nulled, the CSM releases the LM. The separation maneuver is then performed immediately.
- 7. LM jettison is done radially, CSM below, with final sep pyros providing approximately 0.4 foot per second thrust radial. The separation burn is performed five minutes after jettison, providing one foot per second thrust retrograde.

F. Propulsion Systems

- In order to conserve SM RCS, the SPS engine is used to "back-up" all nominal LM rendezvous burns. The SPS gimbal motors are not turned on during the normal maneuver preparation.
- The SPS is always started using a single bank, however, the other bank will be opened 2 to 5 seconds after ignition for burns longer than 6 seconds.
- Table 2-9 lists the CSM propulsion burns.

III. LM Systems

A. Communications

The preferred S-Band communications are:

(a) Uplink Mode 7 (Voice, Updata)

(b) Downlink Mode 2 (Voice, TLM-HBR, PRN, BIOMED)

The LM DSEA schedule is shown in Table 2-10.

B. ECS

- 1. The LM contains ambient air at lift-off. During launch the pressure bleeds to zero psia. CSM 02 pressurizes the LM after T&D. After T&D, the LM is isolated and allowed to bleed down via leakage. After the first LM egress, the LM is isolated and allowed to leak down. For the entry into the LM before undocking, the CSM 02 is used to pressurize the LM. This procedure insures a higher percentage of oxygen in the LM at the first EVA.
- LM 02 is used to pressurize the LM five times; after the SEVA, EVA-1, EVA-2, EVA-3, and after equipment jettison.

C. Guidance Systems

- The LGC and CMC use the same landing site and lift-off REFSMMATS.
- The AGS is placed in standby after the "GO" is given for lunar stay at T3.
- 3. The IMU is powered down and the LGC placed in standby approximately 1 hour after TD until prior to Post EVA-3 cabin cleanup. The LGC is placed in operate several times to update the computer clock.
- 4. To prevent overheading of the antenna, the rendezvous radar is pointed away from the sun and turned off when no functional use is required.
- 5. Lunar gravity measurement data are provided by putting the LGC in POO for 10 minutes prior to surface powerdown, and for approximately 45 minutes prior to the P22 Lunar Surface Navigation before ascent.
- The LM tracking light is operated continuously in the S/C dark period during rendezvous.

D. Propulsion Systems

- The APS/RCS interconnect is used during the lunar liftoff and ascent only.
- 2. Table 2-11 lists the LM propulsion burns.

E. Electrical Power System

- The LM is powered down to a minimum level to conserve battery consumables on the lunar surface from PDI +1:15 to lift-off -1:15.
- 2. LM battery management is scheduled on the lunar surface to equalize the usage of the five descent stage batteries. Battery management periods are at 105:35, 118:10, 127:45, 148:50, 160:15 and 169:40.

IV. Procedures

- A. CSM Crew procedures called out in the Flight Plan may be found in the following documents:
 - Apollo Operations Handbook CSM 112 (AOH), Volume 2
 - Crew Checklists
 - CSM Rendezvous Procedures
 - Photographic and TV Procedures
 - Lunar Landmark Tracking Attitude Studies
 - Lunar Orbit Attitude Sequence for Mission J-1
- B. LM Crew procedures called out in the flight plan may be found in the following documents:
 - Apollo Operations Handbook LM-10, Volume 2
 - Crew Checklists
 - LM Rendezvous Procedures
 - LM Descent/Ascent Procedures
 - 5. Photographic and TV Procedures
 - EVA Procedures
 - 7. Lunar Surface Procedures

V. <u>Synchronization of Ground Elapsed Time (GET)</u>

The realtime GET is synchronized with the Flight Plan GET. In TLC, the GET is synchronized at 53:00 if the difference is more than ±1 minute. In lunar orbit the GET is synchronized at 96:50 and at 150:20 if the difference is more than ±2 minutes. The time changes are based on the expected difference between realtime and flight plan GET's at the start of lunar orbit revs. The synchronization is performed by a V70 uplink from the ground followed by the crew synchronizing the mission timer to the CMC clock.

VI. <u>Miscellaneous</u>

- A. Table 2-12 contains a summary of the expected block data update times.
- B. Table 2-13 contains landmark tracking and landing site data.
- C. Table 2-14 is a schedule of the P23 cisiunar navigation sightings.
- D. Table 2-15 contains a film budget for the Mapping Camera.
- E. Table 2-16 contains a film budget for the Panoramic Camera.

TABLE 2-1 SUIT WEARING SCHEDULE

ACTIVITY	PRESSURIZED (HARD SUIT)	SUITED (SOFT: SUIT)	PARTIAL SUIT WITH- OUT HELMET & GLOVES	SHIRTSLEEVES (ICG)
LAUNCH		ALL		
EARTH ORBIT THRU S-IVB EVASIVE MNVR			ALL	
SIM DOOR JETTISON	ALL			
TLC & TEC EXCEPT TEC EVA				ALL
LM ACTIVATION			ALL*	
UNDOCKING -40MIN TO UNDOCKING +5MIN		CDR & LMP*	CWb**	
UNDOCKING +5MIN THRU CIRC			ALL*	
CIRC TO TD - 1 HR			CDR & LMP*	СМР
TD - 1 HR TO +15MIN		CDR & LMP*		СМР
LUNAR STAY EXCEPT SEVA & EVA				ALL
SEVA* & SURFACE EVA's	CDR & LMP			CMP
LIFTOFF -25MIN THRU DOCKING		CDR & LMP	CMP	
LM JETT THRU TEI				ALL
TEC EVA	ALL			
ENTRY				ALL

^{*}The CDR & do not wear the LCG during these phases. **The CMP dons helmet and gloves for latch cocking and then doffs.

TABLE 2-2 CREW BIOMED HARNESS WEARING SCHEDULE (7/26)

Time, g.e.t.	CDR	CMP	<u>LMP</u>
Prelaunch	on	on	on
13:30	off	off	
26:10	on		off
37:30	off	on	
49:00		off	on
60:00	on		off
71:15	off	on	
85:50		off	on
97:05	on		
97:30		on	
107:19			off*
117:50			oп*
127:50	off*		
139:18	on*		
149:30			off*
159:45			on*
178:05	off	off	
189:00	on		off
202:14	off	on	
211:42		aff	on
225:40		on	off
240:50	òn		ón
243:25		off	off
249:30	off		on
261:38		on	off
274:41	on	off	
288:30	off		on

 $[\]mbox{*Crew option}$ - the CDR & LMP may elect to wear their biosensors throughout the period of lunar surface activities.

TABLE 2-3 (7/26) SC COVERAGE BY MSFN STATIONS USING 85FT/210FT DISH ANTENNA

	GOLDSTONE (GDS)	OME	PARKS (PKS)	KS S)	HONEYSUCKLE (HSK)	UCKLE K)	MADRID (MAD)	01 D)
	AOS	L05	AOS	507	AOS	507	AOS	LOS
EARTH ORBIT	01:28	01:33						
	02:57	16:09						
							03:16	08:01
			13:15	21:21				
					10:45	23:49		
TRANSLUNAR							22:27	32:47
COAST	30:00	40:31						
					34:59	48:19		
			37:45	45:52				
							46:52	56:54
	54:17	64:36						
					59:03	72:29		
			61:35	70:03				
							71:05	78:23
101 (78:31)	78:20	78:23						
TEI (233-46)							223:58	229:18
75	227:58	237:05						
					231:34	246:11		
			234:18	243:30			L	0.1
TRANSEARTH							244:55	253: 3
COAST	252:13	261:04						
					255:35	270:35	•	
			258:24	267:47				
							269:05	277:13
	276:56	285:12						
					279:55	299:52		
EI (294:58)			282:56	292:49				publican and an addition

TABLE 2-3 (CONT)
SC COVERAGE BY MSFN STATIONS USING 85FT/210FT DISH ANTENNA

DEM	GET AT END	GOLDS1 (GDS)		PAI (Pi	RKS KS)		(SUCKLE HSK)		ORID NAD)
REV	OF REV 78:31	AOS	LOS	AOS	LOS	AOS	LOS	AOS	LOS
LOI	80:41	78:55	80:20	7100				78:56	80:20
2	82:49	81:03	82:28						
3	84:43	83:16	84:21			83:16	84:21		
4	86:39	85:09	86:14	85:43	86:14	85:09	86:14		
5	88:30	87:03	88:08	87:03	88:08	87:03	88:08		
6	90:23	57.55	3,1,22	88:57	90:02	88:57	90:02		
7	92:17			90:50	91:56	90:50	91:56		
8	94:11			92:44	93:49	92:44	93:49		
9	96:04					94:38	95:43	95:42	95:42
10	97:58					96:31	97:10	96:31	97:36
H	99:52							98:24	99:30
12	101:45							100:18	101:23
13	103:43	103:19	103:21					102:10	103:22
14	105:41	104:08	105:20					104:08	105:20
15	107:40	106:06	107:18						
16	109:38	108:04	109:16			108:04	109:16		
17	111:36	110:02	111:14	110:18	111:14	110:02	111:14		
18	113:34	112:01	113:13	112:00	113:13	112:00	113:13		
19	115:32			113:59	115:11	113:59	115:11		
20	117:30		T	115:56	117:09	135:56	117:09		
21	119:28			117:55	119:07	117:55	119:07	100.46	101.05
22	121:27					119:53	121:06	120:46	121:05
23	123:25		<u> </u>			121:51	122:09	121:51	123:03
24	125:23							123:49	125:01
25	127:21					ļ		125:47	126;59 128:58
26	129:19	128;18	128:58				ļ <u>.</u>	127:46	129:57
27	131:17	129:43	130:56	l			<u> </u>	129:44	129:5/

TABLE 2-3 (CONT)

148:39 150:37 152:35 154:33 146:40 74:14 72:16 76:13 79:29 178:1 20.2 MADRID (MAD) 149:25 173:00 175:00 176:58 178:57 51:23 145:48 171:04 53:21 A0S SC COVERAGE BY MSFN STATIONS USING 85FT/210FT DISH ANTENNA HONEYSUCKLE (HSK) L0S 132:54 134:52 136:50 142:45 138:48 144:43 158:29 160:28 162:26 168:21 172:14 146:41 164:24 184:05 186:03 187:59 166:22 82:07 A0S 132:15 133:40 135:38 137:36 39:34 145:28 141:32 57:17 59:15 143:31 167:08 81:53 182:52 84:50 86:48 [6]:13163:12 165:10 90:69 71:04 36:50 38:48 140:47 142:45 144:28 160:28 162:26 164:24 166:22 69:24 87:59 168:21 186:03 108 PARKS (PKS) 135:38 137:36 139:34 141:33 [6]:13 59:50 165:10 167:08 163:12 90:69186:48 84:50 AOS 132:54 134:52 136:50 154:33 156:31 158:29 160:28 37:57 180:09 184:05 186:03 187:40 182:07 GOLDSTONE (GDS) 103 137:26 31:42 133:40 35:38 | 53:21 | 55:19 | 57:17 159:15 61:14 180:54 182:53 186:49 178:56 184:51 AOS GET AT END OF REV 33:15 35:14 37:12 39:10 143:06 145:04 147:03 52:57 41:08 54:55 160:50 162:48 64:46 168:42 70:40 72:39 74:37 176:35 178:33 180:31 182:29 56:53 66:44 186:25 188:23 84:27 [49:0] 58:51 REV 임임임의의의 8889 42 43 文 45 46 4 49 S 왕동 7 53 120 귝 2

TABLE 2-3 (CONT)
SC COVERAGE BY MSFN STATIONS USING 85FT/210FT DISH ANTENNA

DEM	GET AT END OF REV	GOLDSTO (GDS)		PARI (PKS			SUCKLE SK)	MADRI (MAS	
REV	OF REV	AOS	LOS	AOS	LOS	AOS	LOS	AOS	LOS
57	190:22			188:47	189:59	188:47	189:59		
58	192:20			190:45	191:57	190:45	191:57		
59	194:17			192:42	193:55	192:43	195:55		
60	196:16					194:41	195:54	195:49	195:53
61	198:14					196:39	197:09	196:39	197:31
62	200:12							198:37	199:49
63 .	202:10							200:35	201:47
64	204:08	203:16	203:45					202:33	203:46
65	206:06	204:3]	205:43	ļ			L		
66	208:04	206:29	207:41			206:53	207:41		ļ
67	210:02	208:27	209:39			208:27	209:39	ļ	<u> </u>
68	212:00	210:26	211:38	210:25	211:37	210:25	211:37	ļ	ļ
69	213:58	212:23	212:41	. 212:23	213:35	212:23	213:35	 	
70	215:56			214:21	215:33	214:21	215:33		ļ
71	217:54			216:19	217:32	216:19	217:32		
72	219:52			218:18	219:18	218:18	219:30	000 07	001.00
73	221:51					220:15	221:28	220:37	221:28
TEI	223:46		i					222:13	223:25

TABLE 2-4
APOLLO 15 TV SCHEDULE
(7/26)

DAY	DATE	CDT	GET HR:MIN	DURATION HR:MIN	ACTIVITY SUBJECT	VEHICLE	STATION
Monday	26 July	11:59 AM	03:25	00:25	TRANSPOSITION & DOCKING	CSM	GDS
Tuesday	27 July	6:19 PM	33:45	00:45	INTERIOR & IVT TO LM	CSM	GDS
Friday	30 July	7:32 AM	94:58	00:10	LANDING SITE	CSM	HSK
Saturday	31 July	8:34 AM	120:00	06:40*	LUNAR SURFACE EVA-1	LM/LRV	HSK/MAD
Sunday	1 Aug	6:04 AM	141:30	06:30*	LUNAR SURFACE EVA-2	LRV	PKS/HSK/MAD
Monday	2 Aug	2:44 AM	162:10	05:40 *	LUNAR SURFACE EVA-3	LRY	PKS
Monday	2 Aug	12:04 PM	171:30	00:15	LM LIFTOFF	LRV	MAD
Monday	2 Aug	1:44 PM	173:10	00:06	RENDEZVOUS	CSM	MAD
Monday	2 Aug	2:05 PM	373:31	00:05	DOCKING	CSM	MAD
Thursday	5 Aug	10:41 AM	242:07	00:30	TRANSEARTH EVA	CSM	HSK/PKS
Friday	6 Aug	2:54 PM	270:20	00:30	PRESS CONFERENCE	CSM	MAD

^{*}SEE SECTION 3 FOR DETAILED START/STOP TIMES DURING EVA

TABLE 2-5
FUEL CELL PURGE, URINE DUMP AND WASTE WATER DUMP SCHEDULE
(7/26)

	O ₂ FUEL (CELL PURGE		:		
GET	& WASTE !	1 ₂ 0 DUMP	URINE	DUMP	H ₂ FUEL CE	LL PURGE
(HR:MIN)	NUMBER	ATIME (HR:MIN)	NUMBER	ΔTIME (HR:MIN)	NUMBER	ATIME (HR:MIN)
11:40	1	11:40	1	11:40*	:	İ
30:40	2	19:00	2	19:00*	1	30:40
56:15	3	25:35	3	25:35*		
73:15	4	17:00	4	17:00*	2	42:35
97:38	5	24:23	5	24:23		
125:20	6	27:42	6	27:42	3	52:05
146:59	7	21:39	7	21:39		
170:25	8	23:26	8	23:26	4	45:05
193:58	9	23:33	9	23:33		
221:41	10	27:43	10	27:43	5	51:16
238:18			11	16:37		
244:00	11	22:19				
272:27			12	32:54		
272:40	12	28:40			6	50:59

^{*}URINE DUMP NOT CRITICAL PERFORM AT THE SAME TIME AS WASTE $\mathrm{H_{2}O}$ DUMP

TABLE 2-6

BATTERY CHARGE SCHEDULE (7/26)

GET	····	
HR:MIN		BATTERY
4:30		В
25:10		Α
31:00 (IF MCC 2 IS PERFORMED)	Α
50:00		В
120:30		В
141:51		A
189:02		В
214:24		A
239:02	,	В
270:30		A

TABLE 2-7
CSM Light Canister Change Schedule (7/26)

CHANGE	PPROXIMATE	APPROXIMATE	INS	TALL	REMOVE	& STOW
NC.	GET, HR:MIN	∆T, HR	CANISTER	POSITION	CANISTER NO.	STOWAGE LOCATION
Ţ	12:10		3	A	1	B 5
2	26:00	14	4	В	22	В5
3	38:00	12	5	A	3	B5
4	51:00	13	6	В	4	B5
5	61:45	11	7	A	5	B6
6	75:55	14	8	В	6	В6
7	85:55	10	9	А	7	B6
8	98:08	12	10	В	8	B6
9	122:00	24	11	A	9	A9
10	146:00		12	В	10	A9
11	169:00	23	13	А	11	A9
12	179:38	11	15	В	12	А3
13	191:59	12	16	A	13	А3
14	203:30	12	17	В	15	A3
15	215:27	12	18	A	16	А3
16	226:35	11	19	В	17	A4
17	238:57	12	20	A	18	A4
18	251:00	12	21	В	19	A4
19	263:02	12	22	А	20	A4
20	274:57	12	23	В	21	A5
21	288:35	14	24	А	22	A5

LM Lioh CARTRIDGE CHANGE:GET (HR:MIN) 161:30 & 151:10

TABLE 2-8

LUNAR ORBIT NON-SIM BAY RCS JET CONFIGURATION PERIODS

(7/26)

REASON	FROM	<u>T0</u>	TOTAL	ULLAGE
LOI TRACKING & LOI THRU BAILOUT	74:00	83:55	9:55	16 SEC (DOI), 16 SEC (BAILOUT)
LM ACT DAY*	94:58	105:40	10:42	14 SEC (CIRC)
UV & DIM LIGHT PHOTOGRAPHY	123:15	125:30	2:15	NONE
LOPC I**, RNDZ, & DOCKING	164:08	174:00	9:52	13 SEC (LOPC1)
LM JETT & CSM SEP	176:16	177:45	1:29	NONE
SHAPING BURN & TEI	222:42	224:03	1:21	13 SEC

NOTES:

SIM BAY SINGLE JET AUTHORITY WILL BE USED IN ALL LUNAR ORBIT PERIODS EXCEPT AS IDENTIFIED ABOVE SIM BAY SINGLE JET AUTHORITY INHIBITS ALL, JETS EXCEPT AT & C2 or B2 & D1 FOR ROLL, A3 & C4 FOR PITCH AND B3 & D4 FOR YAW CONTROL

*AN SPS DOI TRIM MAY BE REQUIRED WHICH WILL REQUIRE AN ULLAGE FOR ${\sim}14$ SEC.

**IF Y&Z TRIMS ARE REQUIRED FOR LOPC 1, THE CMP WILL ROLL THE CSM SO THAT JETS B1 AND A2 ARE NOT USED.

TABLE 2-9 CSM BURN/EVENT SCHEDULE (7/26)

			(1/20/			
BURN/ MNVR	GETI/ BURN TIME	ΔV _T (FPS)	ULLAGE/ Δ7 (FPS)	REFSMMAT	RESULTANT HA/HP(NM)	REMARKS
TLI	2:49:58 5 MIN 56 SEC	10421.0		LAUNCH		S-IVB BURN
CSM EJECTION	4:16:00 3 SEC	0.4		LAUNCH		RCS BURN
MCC-1	11:55:54			PTC	T	NOM ZERO
MCC-2	30:55:54			PTC		NOM ZERO
MCC-3	56:31:14.7			PTC		NOM ZERO
MCC-4	73:31:14.7			PTC		NOM ZERO
L01	78:31:14.7 6 MIN 32 SEC	2997.9	NOT REQ'D	LOI	HA 170.0 HP 58.3	SPS BURN
DOI	82:39:32.6 22.9 SEC	207.6	4 JET 16 SEC	LDG SITE	HA 58.4 HP 9.6	SPS BURN
BAILOUT BURN	82:26:16.9 10.2 SEC	94.0	4 JET 16 SEC	LDG SITE	HA 71.0 HP 3.9	SPS BURN
DOI TRIM	96:17:00			LS OR PC AS REQ'D		SPS BURN NOM ZERO
UNDOCK & SEP	100:13:56.1 3.3 SEC	1.0		LDG SITE	HA 59.8 HP 8.4	RCS BURN
CSM CIRC	101:34:55.1 3.9 SEC	70.8	4 JET 14 SEC	LDG SITE	HA 64.7 HP 54.5	SPS BURN
LOPC	165:12:50.6 16.5 SEC	308.6	4 JET 13 SEC	PLANE CHANGE	HA 59.82 HP 59.2	SPS BURN
LM JETT	177:20:45	~0.4		LIFTOFF		NO BURN
CSM SEP	177:25:45 6.4 SEC	1.0		LIFTOFF	HA 59.9. HP 58.7	RCS BURN
CSM SHAPE	221:25:52 3.4 SEC	64.2	4 JET 13 SEC	TEI	HA 77.6 HP 57.5	SPS BURN
TEI	223:46:06 2MIN 17.8 SEC	3049.7	4 JET 13 SEC	TEI		SPS BURN
MCC-5	238:46:06			PTC		NOM ZERO
MCC-6	272:58:20			PTC		NOM ZERO
MCC-7	291:58:20			ENTRY		NOM ZERO
EI	294:58:20			ENTRY		NO BURN

TABLE 2-10

APOLLO 15/LM - 10 DSEA SCHEDULE (7/26)

MODE	GET (HR:MIN)	RECORD TIME X DUTY CYCLE = TAPE USED (HR:MIN)	ACCUM. TAPE USED (HR:MIN)
ICS/PTT (VOX)	98;24 104:20	5:56 X 100% = 5:56	5 :56
VOX OFF	104:20 104:49	0:29 X 63% = 0:17	6:13
VOX OFF	119:10 120:00	0:50 X 63% = 0:32	6:45
VOX OFF	140:28 141:18	0:50 X 63% = 0:32	7:17
VOX OFF	161:07 161:57	0:50 X 63% = 0:32	7:49
ICS/PTT (VOX)	171:20 171:35	0:15 X 100% = 0:15	8:04
VOX (ICS/PTT)	171:35 171:45	0:10 X 63% = 0:06	8:10
ICS/PTT OFF	171:45 173:35	1:50 X 100% = 1:50	10:00*
	ICS/PTT (VOX) VOX OFF VOX OFF VOX OFF VOX OFF ICS/PTT (VOX) VOX (ICS/PTT)	MODE (HR:MIN) ICS/PTT 98:24 (VOX) 104:20 VOX 104:20 VOX 104:49 VOX 119:10 OFF 120:00 VOX 140:28 OFF 141:18 VOX 161:07 OFF 161:57 ICS/PTT 171:20 (VOX) 171:35 VOX 171:35 ICS/PTT 171:45 ICS/PTT 171:45	MODE

*END OF TAPE

TABLE 2-11 LM BURN SCHEDULE (7/26)

BURN/ MNVR	GETI/ BURN TIME	ΔV _T (FPS)	ULLAGE/ ΔV (FPS)	REFSMMAT	RESULTANT HA & HP	REMARKS
PDI	104:28:54.8 12 MIN 2.1 SEC	6700.0	4 JET 7.5 SEC	LDG SITE		DPS BURN
ASCENT	171:37:23.9 7 MIN 15.2 SEC	6055.5	NONE	LIFT-OFF	HA 45.6 HP 9.0	APS BURN
TPI	172:29:39.1 2.6 SEC	73.6	4 JET 10.0 SEC	LIFT-OFF	HA 61.5 HP 43.9	APS BURN
LM DEORBIT	179.06:22.7 1 MIN 22.3 SEC	195.4	N/A	LIFT-OFF	HA 62.0 HP-52.8	RCS BURN

TABLE 2-12 APOLLO 15 RETURN TO EARTH BLOCK DATA SCHEDULE (7/26)

DATA	GET UPDATE	GETI	PAD TYPE
TLI+90 L0+8 L0+15 L0+25 L0+35 L0+45 L0+60 L0I-5 FLYBY PER+2	1:30 1:30 6:00 13:30 13:30 13:30 13:30 35:00	4:16:34.32 8:00:00 15:00:00 25:00:00 35:00:00 45:00:00 60:00:00 73:29:25.00	Complete P-30 P-37 P-37 P-37 P-37 P-37 P-37 Complete P-30 (docked) ABBRR-30 (docked) ABBRR-30
TEI-4 TEI-5 TEI-12 TEI-19 TEI-26 TEI-37 TEI-45 TEI-52 TEI-58 TEI-60 TEI-62 TEI-64 TEI-69 TEI-71 TEI-73	76:50 81:40 85:38 95:25 108:57 126:44 150:20 163:35 179:36 189:05 193:42 196:50 203:20 213:28 216:54	87:12:24.45 88:25:54.66 101:47:06.57 115:28:46.69 129:21:36.43 150:57:57.96 166:38:52.24 180:20:15.47 192:13:36.08 196:11:15.16 200:08:54.90 204:06:34.11 213:49:29.41 217:47:07.83 221:43:59.16	ABB P-30
PREL. TEI-74	220:22	223:43:57.86	Complete P-30
<u>NOM</u> TEI-74 TEI-75	222:43	223:43:57.86	Complete P-30 ABB P-30

Data are based on MCS IN71-FM112,S/C OT, Vol II., dated March 26, 1970. NOTES:

- 1. All block data maneuvers are to the MPL line except:
 - a. $TLI+90^{M}$ abort is to the AOL.
 - b. Nominal TEI (TEI $_{74})$ is to EOM target ($_{\lambda} =$ 158°W).
 - c. TEI_{75} is also to EOM target (λ = 158°W).

TABLE 2-12 (CONT)

- 2. Pass flyby early if pericynthion is not clear of moon.
- 3. The flyby and PER+2 maneuvers are \underline{docked} . All other aborts are undocked.
- 4. PER+2 fast return to MPL assumes \mbox{MCC}_4 .
- 5. TEI, assumes LOI and \underline{no} DOI.
- 6. ${\sf TEI}_5$ assumes DOI.
- 7. TEI₁₂ assumes <u>no</u> circ.
- 8. TEI $_{19}$ assumes circ but \underline{no} LOPC.
- 9. TEI_{45} assumes <u>no</u> LOPC.
- 10. TEI₅₂ assumes LOPC.
- 11. TEI_{73} assumes <u>no</u> shaping maneuver.
- 12. All TEI's are constrained to 40° inclination.

TABLE 2-13 LANDMARK AND LANDING SITE DATA

SITE	REV	LATITUDE DEG	LONGITUDE DEG	ALTITUDE**
J-1	3 & 10	25.958 N	11.3278 E	0.000
15-1	12* 13	26.065 N	3.671 E	-1.935
15-2	12* 47	25.991 N	3.613 E	-1.744
15-3	12	26.039 N	3.496 E	-1,795
15-4	12*	26.352 N	3.323 E	-1.867
15-5	12*	25.809 N	2.808 E	-1.854
15-6	12*	25.541 N	2.818 E	-1.877
HADLEY		26.074 N	3.654 E	-11.917

^{**} Difference between landmark radius vector and 938.4935 NM (Mean Lunar Radius)

 $[\]star$ Landing Site Landmark to be determined in real-time

TABLE 2-14
P23 CISLUNAR NAVIGATION (7/26)

GET	STAR/HORIZON	COMMON NAME
8:00	1/ENH 45/EFH 131/ENH 224/EFH	ALPHERATZ FOMALHAUT BETA ARIETIS MARKAB
236:20	57/EFH 50/ENH 13/EFH	BELLA TRIX POLLUX CAPELLA
251:10	50/ENH 11/EFH 13/EFH	POLLUX ALDEBARAN CAPELLA
263:15	142/EFH 11/EFH 13/EFH	ZETA TAURI ALDEBARAN CAPELLA
267:48	60/EFK 13/EFH 21/ENH	EL NATĤ CAPELLA ALPHARD
273:05	60/EFH 13/EFH 21/ENH	EL NATH CAPELLA ALPHARD
276:00	60/EFH 13/EFH 21/ENH	EL NATH CAPELLA ALPHARD
289:00	40/MFH 221/MFH 223/MNH 224/MNH 126/MNH	ALTAIR DELTA CAPRICORNI BETA PEGASI MARKAB GAMMA PEGASI
292:02	126/MNH 120/MFH 40/MFH	GAMMA PEGASI ALPHA GRUIS ALTAIR

TABLE 2-15 MAPPING CAMERA FILM BUDGET (7/26)

REV	STA	RT	ST	DP	REMARKS	FRAMES USED	FILM USED FT	FILM REMAINING FT
	GET (H:M:S)	LONG(DEG)	GET (H:M:S)	LONG(DEG)				
3/4	84:42:35	179.6W (E TERM)	84:54:15	142.5°E		36	15	1445.0
15	105:52:20	144.5E	106:17:20	74.0°E		59	24.6	1420.4
15/16	106:56:15	49,8₩	108:54:40	50.8W	FULL REV	295	122.9	1297.5
22/23	119:33:35	162.7E (& TERM)	121:32:02	161.7E (E TERM)	FULL REV	297	123.8	1173.7
23	121:37:04	143.7E	122:31:28	18,8W (W TERM)	FORWARD OBLIQUES	118	49.2	1124.5
27	129: 2 6:27	157.7E (E TERM)	130:26:00	22.8W (W TERM)	LIGHTSIDE PASS	152	63.4	1051.1
33/34	141:16:34	151.6E (E TERM)	143:15:02	150.6E (E TERM)	FULL REV	298	124.2	936.9
34	143:15:02	150.6E (E TERM)	144:09:00	11.8W	BKWD OBLIQUES	118	49.2	887.7
35	145:13:30	149.6E (E TERM)	146:12:55	30,8W (W TERM)	NORTH OBLIQUES	130	54.2	833.5
37/38	150:08:58	32.9W (W TERM)	152:08:20	33.9W (W TERM)	FULL REV	303	126.4	707.1
44	162:59:44	140.6E (E TERM)	163:59:10	39,9W (W TERM)	LIGHTSIDE PASS	152	63.4	643.7
50	174:50:06	134,6E (E TERM)	175:49:22	45.9W (W TERM)	LIGHTSIDE PASS	152	63 .4	580.3
58/59	190:37:25	126.6E (E TERM)	192:35:51	125.6E (E TERM)	FULL REV	303	126,4	453.9
62/63	199:30:22	57.9W (W TERM)	201:28:11	58.9W (W TERM)	FULL REV	303	126.4	327.5
69	212:19:43	115.6E (E DURM)	214:18:10	114.6E (E TERM)	FULL REV	313	130.5	197.0
71	216:15:37	113.6E (E TERM)	217:15:47	66.9W (W TERM)	SOUTH OBLIQUES	130	54.2	142.8
71/72	217:20:00	82.9W (W TERM)	219:14:14	67.9W (W TERM)	02214022	292	121.7	21.1

TABLE 2-17

CR	YO MANAGEMENT S	CHEDUL	E.				
CET	(TANK NUMBERS)						
GET	OXYGEN HE		EATERS	HYDROGEN	HEATERS		
(HRS:MIN)	A	UTO	OFF	AUTO	OFF		
0:00	1	,2	3	1,2	3*		
13:25	3	-	1,2	1,2,3			
25:18		İ		3	1,2		
74:20	1	,2	3	1,2	⁻ 3		
95:00	3		1,2				
109:00	1	,2	3				
190:00	3	.	1,2				
194:00	1	,2	3				
218:00	3		1,2				
222:00	1	,2	3				
241:25	1	,2,3					
243:30	1	,2	3				

^{*} Manual Mode

PDI 2 ABORT PADS

1	PURPOSE			P	Dl2	PA	<u> </u>	
	GETI	HRS	+	0	0			
T	PDI	MIN	+	0	0	0		
		SEC	+	0				

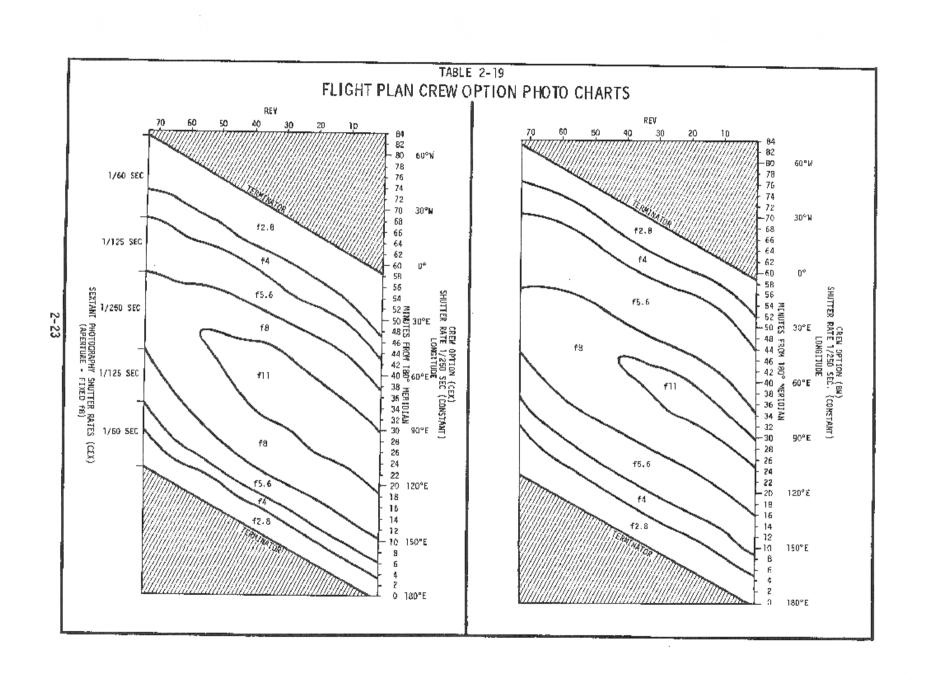
PURPO)SE	PDI	AB	ORT	EAI	SFA (PAD
GETI	HRS	+	0	0			
TPI	MIN	+	0	0	0		
173	SEC	+	0				

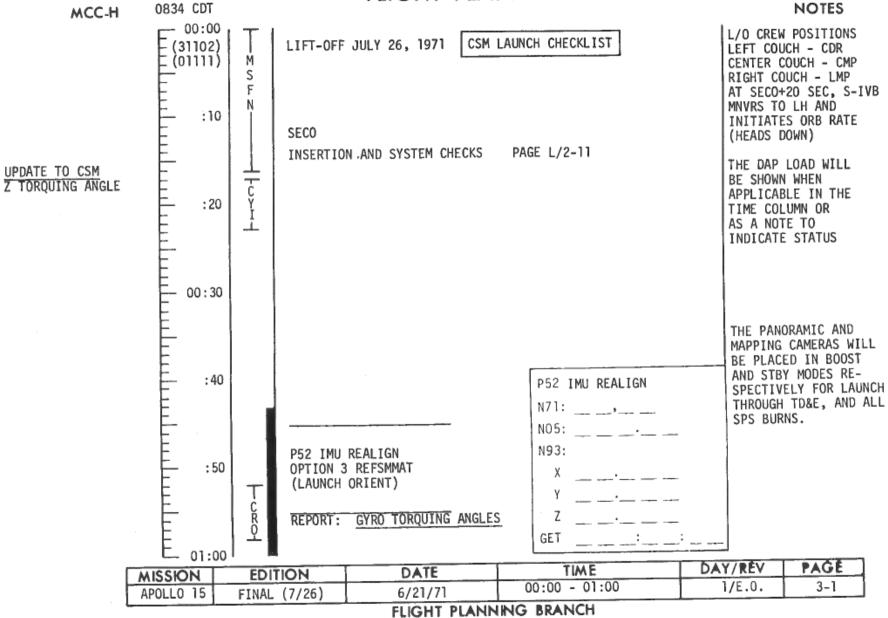
ſ	PURPO	OSE	PDI	2 AB	OR1	LA	TE	'n
-	GETI	HRS	+	0	0		L	
ĸ	TPI	MIN	+	0	0	0		
	N37	SEC	+	0				

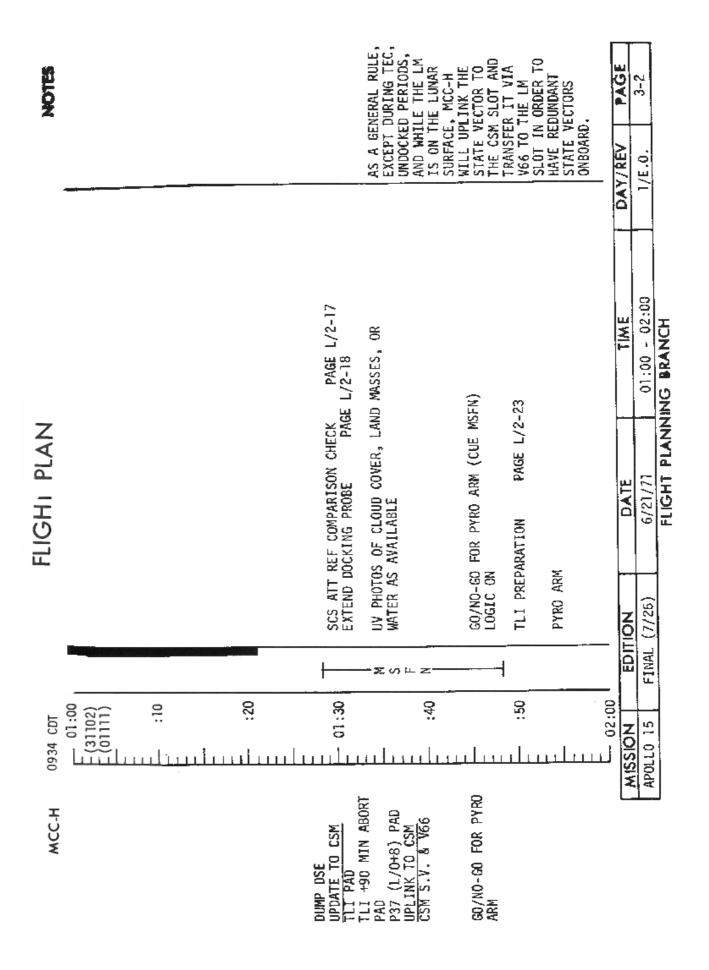
F	PURPO	SE	NO	PD	12+	12 /	ABC	RT
Ì	GETI	HRS	+	0	0			
E	N33	MIN	+	0	0	0		
١		SEC	+	0				
	N84	ΔVX						
F	LOCAL	ΔΥΥ		L			_	<u> </u>
'	[* - **	ΔVZ						
	GETI	HRS	+	0	0			
G	CSI N11	MIN	+	0	0	0	L	1_
_	 ~	SEC	+	0			•	1.
	GETI	HR5	+	0	0	L	\perp	1
Н	TPI N37	MIN	•	0	0	0	_	
•	737	SEC	+	0			1	

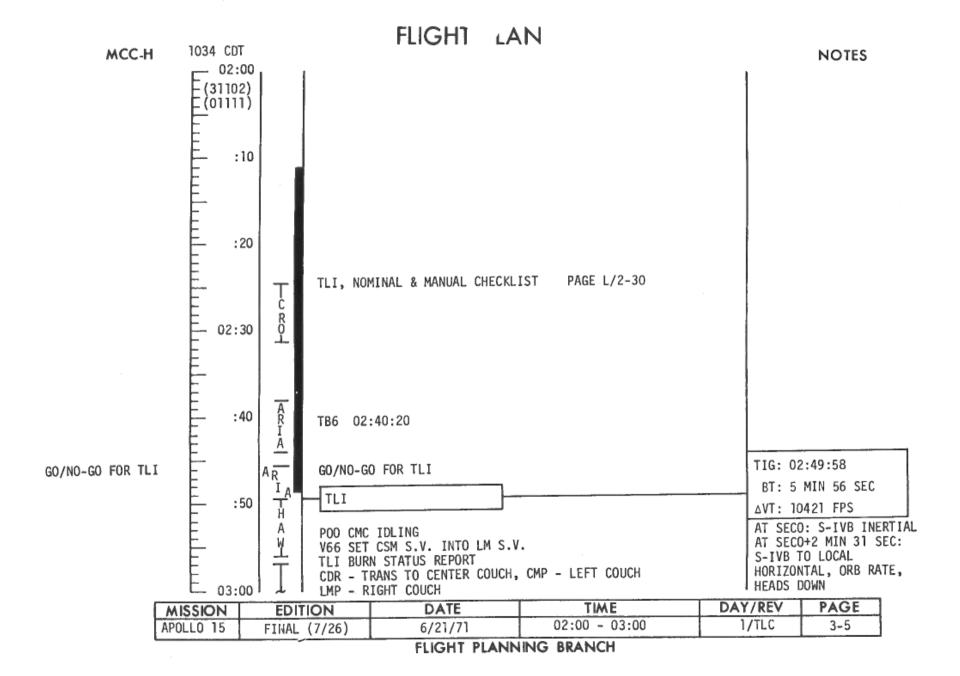
ſ	PURPOSE		T2-2 ABORT PA					
Ì	GETI	HRS	+	0	0			
L	T2	MIN	+	0	0	0		
٦		SEC	+	0			_	_
GETI TPI N37	HRS	+	0	0			L	
		MIN	+	0	0	0		_
•	173/	SEC	+	0				L

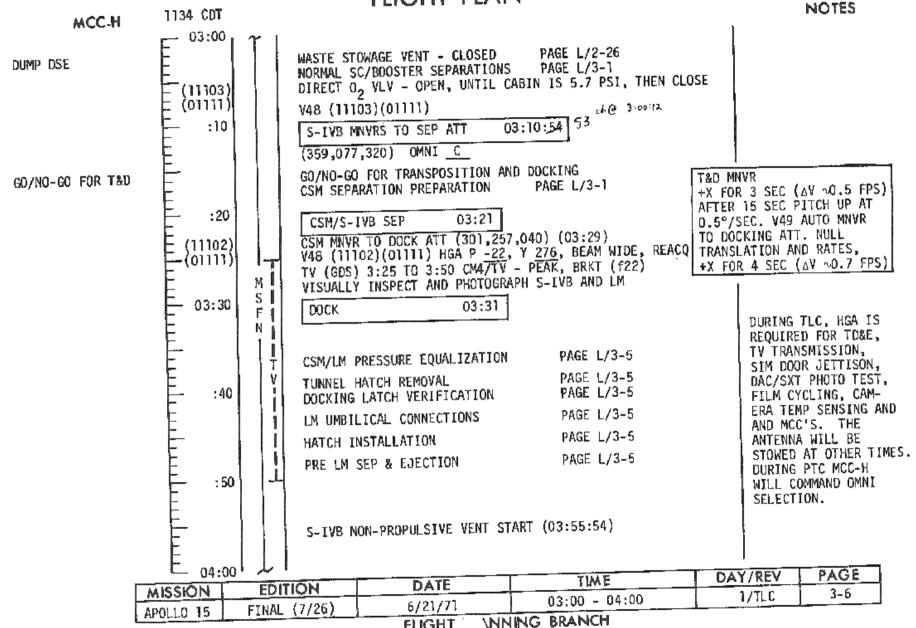
PUF	PURPOSE			T3 ABORT PAD					
GET	HRS	+	0	0					
1 T3	MIN	•	0	0	0				
	SEC	+	0						





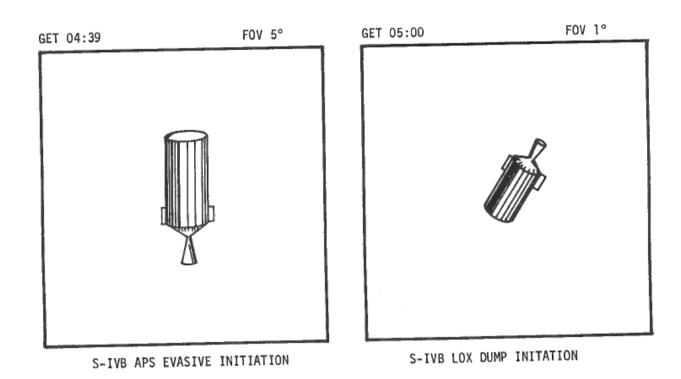


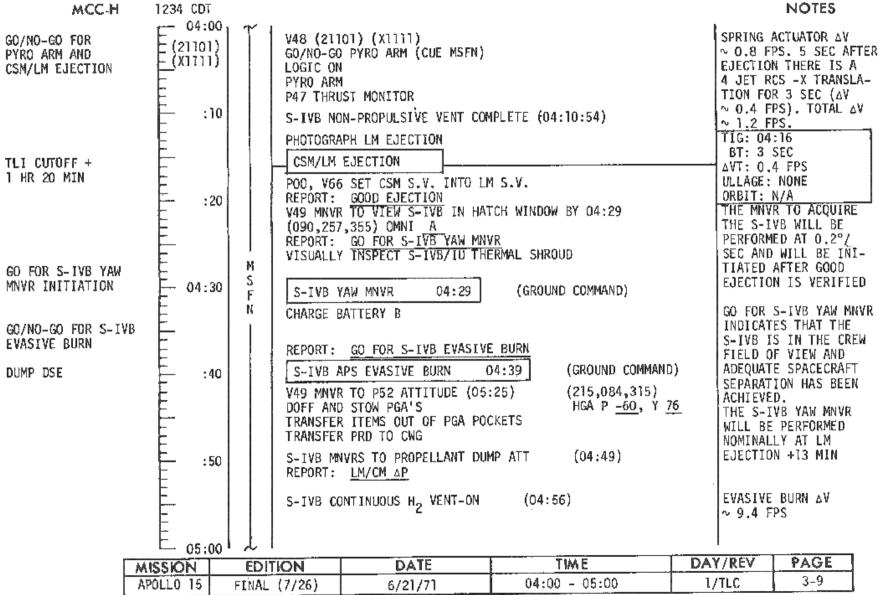




INNING BRANCH FLIGHT

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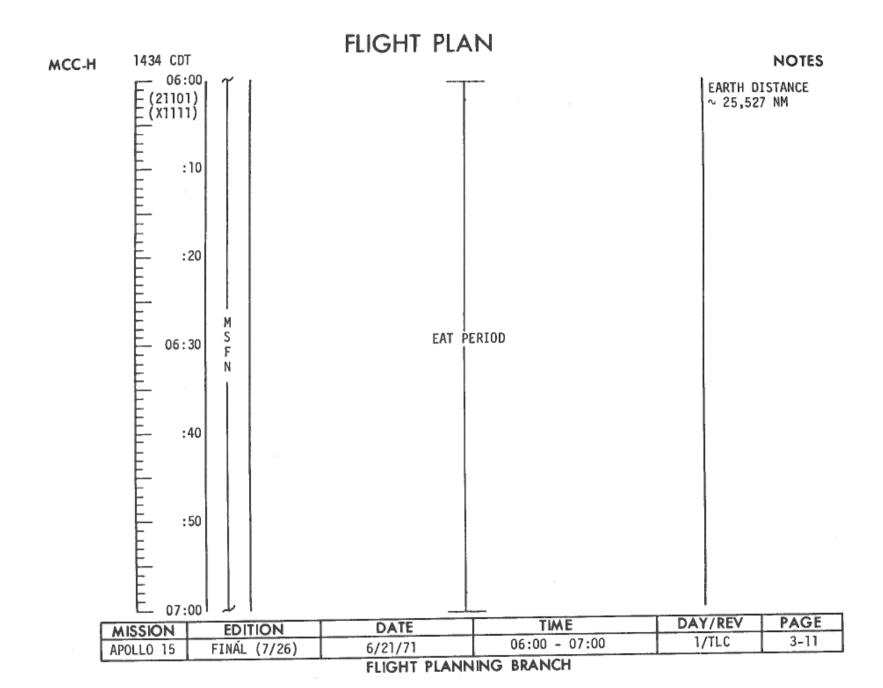


FLIGHT PLANNING BRANCH

FLIGHT PLAN NOTES 1334 CDT MCC-H LOX DUMP ∆V ~ 28 FPS 05:00, S-IVB LOX DUMP (05:00) (21101) SC INTERIOR PHOTOG-(X1111)RAPHY AT CREW OPTION CM/DAC/10/CIN- SPOT (T2.8,1/60,3) 6 fps :10 (87% MAG) UPLINK TO CSM MAG () ___, FR # ___ DESIRED ORIENTATION (PTC) ZERO TRUNNION BIAS ATT DEADBAND - MIN :20 RATE - LOW BMAG (3) - ATT 1/RATE 2 SC CONT - SCS P52 IMU REALIGN STARS N71: ___,__ P52 IMU REALIGN 05:30 OPTION 3 REFSMMAT NO5: ______ (LAUNCH ORIENT) N93: REPORT: GYRO TORQUING ANGLES :40 P52 IMU REALIGN OPTION 1 PREFERRED GET (PTC ORIENT) S-IVB APS MCC-1 GET ~05:45 SC CONT - CMC :50 ΔV ~30 FPS BMAG (3) - RATE 2 P37 PAD ASSUMES UPDATE TO CSM NO MCC-1 P37 PAD (L/0+15) VHF A SIMPLEX - OFF VERIFY WASTE STOWAGE VENT VALVE - VENT 06:00

			DATE	TIME	DAY/REV_	PAGE
- (MISSION	EDITION	DATE		1/TLC	3-10
- 1	APOLLO 15	FINAL (7/26)	6/21/71	05:00 - 06:00	1/120	
	WLOFFO 12	1 THAT (1/20)		INIC PRANCH		

FLIGHT PLANNING BRANCH



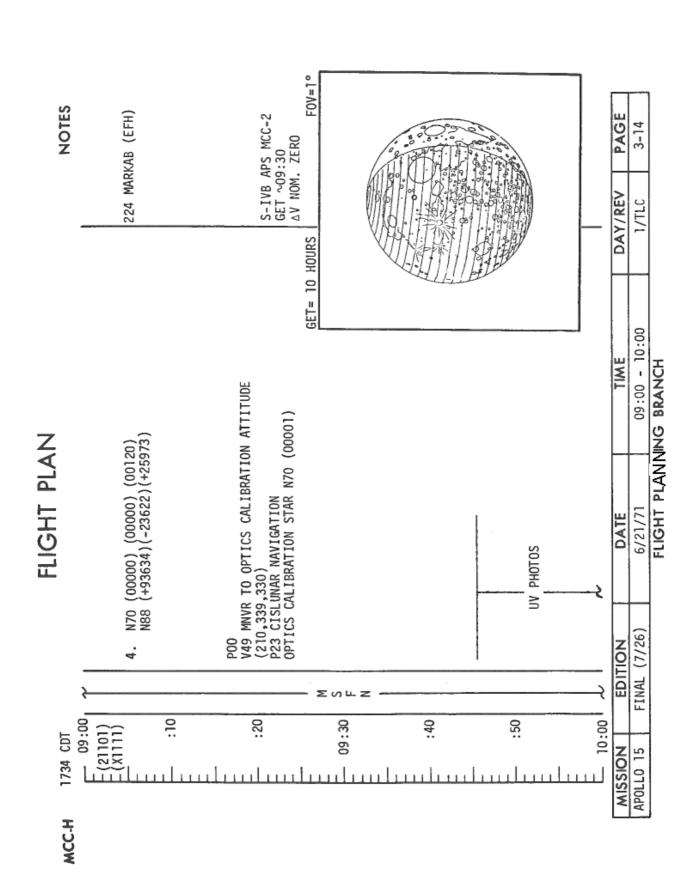
FLIGHT PLAN MCC-H 1534 CDT NOTES 07:00 CSM SYSTEMS CHECKLIST (21101) (X1111) DEACTIVATE PRIMARY EVAP PAGE S/1-13 :10 :20 CSM G&C CHECKLIST 07:30 GET 8:00 F0V 12° ΔV TEST & NULL BIAS CHECK REPORT: <u>BIAS</u> PAGE G/2-5 :40 :50 08:00 DAY/REV PAGE TIME DATE MISSION EDITION 3-12 07:00 - 08:00 1/TLC APOLLO 15 6/21/71 FINAL (7/26)

FLIGHT PLANNING BRANCH

MCC-H	1634 CDT		FLIGHT PLA	N	NOTES
MCC41	08:00 E(21101) E(X1111)	7 V49 MNVR (210,339	TO OPTICS CALIBRATIO	N ATTITUDE (08:06)	EARTH DISTANCE ~ 37,581 NM
	:10	OPTICS C	UNAR NAVIGATION ALIBRATION STAR N7O (TO SIGHTING ATTITUDE		
		(221,327	,330) 000) (+00070) (+00003		LOAD W MATRIX
	:20	5 MARKS	UNAR NAVIGATION ON EACH STAR, UPDATE (00001) (00000) (0011	STATE VECTOR 0)	001 ALPHERATZ (ENH)
	08:30	M S F N			
	:40	2. N70	(00045) (00000) (0012	0)	045 FOMALHAUT (EFH)
	:50	3. N70 N88	(00000) (00000) (0011 (+82401)(+44316)(+353	0) 01)	131 BETA ARIETIS (ENH)
٦	09:00	EDITION	DATE	TIME	DAY/REV PAGE

MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
APOLLO 15	FINAL (7/26)	6/21/71	08:00 - 09:00	1/TLC	3-13

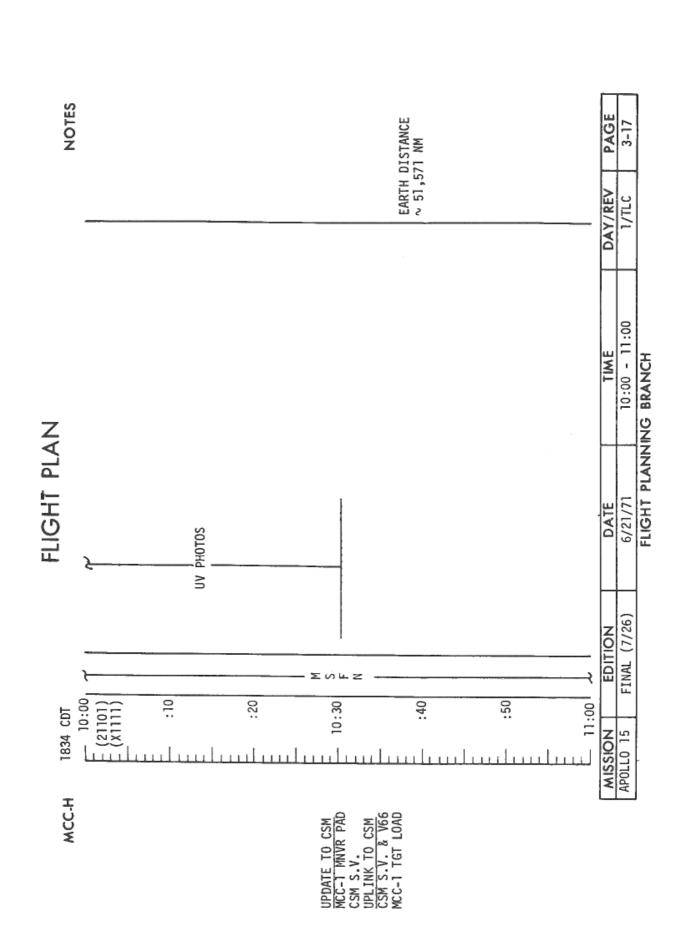
FLIGHT PLANNING BRANCH



UV PHOTOS - CISLUNAR COAST

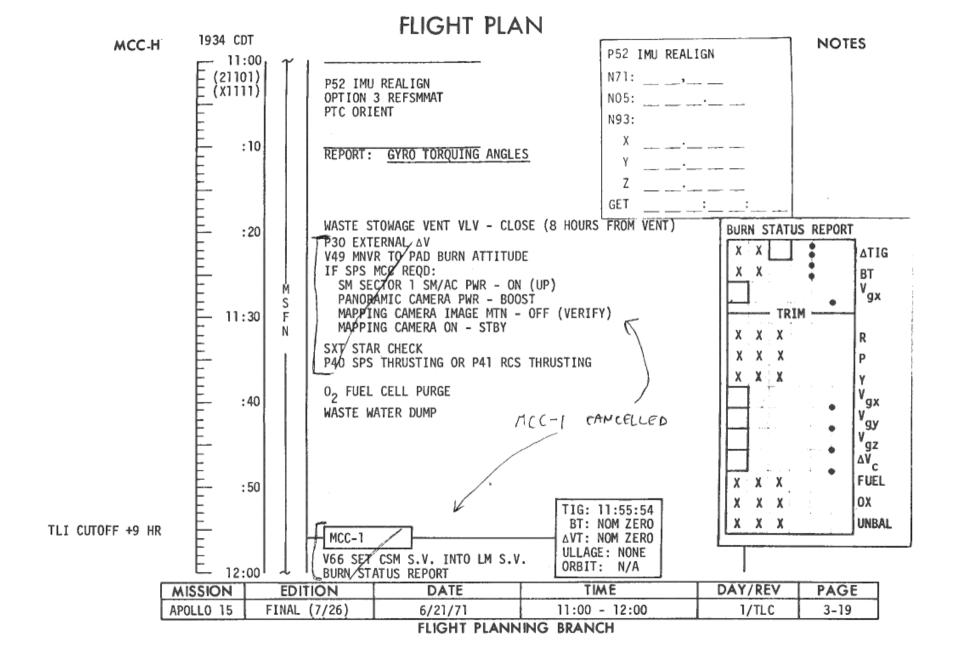
```
V49 MNVR TO MOON UV PHOTO CAL ATT (169,223,014) (10:20)
V49 MNVR TO EARTH UV PHOTO ATT (335,319,331) (09:55)
                                                         HGA P -33, Y 213
 OMNI D
                                                        CONFIGURE CAMERA: (UV CALIBRATION)
CONFIGURE CAMERA: (UV)
                                                         CM5/EL/105/UV-BRKT, CONT (f4.3,1/60,∞) (8 FR)
  CM5/EL/105/UV-BRKT, CONT (f4.3,1/60,∞) (8 FR)
                                                          RINGSLIDE
  RINGSLIDE
                                                         MAG (N) , FR # _____
            , FR #
  MAG (N)
  REMOVE RT2 FLIGHT DATA FILE STOWAGE BOX
                                                        DAMP RATES
                                                         VERIFY FDAI SCALE - 5/1
DAMP RATES
                                                          WAIT 5 MIN FOR RATES TO DAMP
  INHIBIT ALL JETS EXCEPT A1 & C2 OR B2 & D1,
                                                          VERIFY RATES ON FDAI ARE <0.2°/SEC IN ALL AXES
    A4, C3, B4, D3
                                                          2 FRAMES, FILTER 1, CHANGE SHUTTER TO B
  VERIFY FDAI SCALE - 5/1
                                                          2 FRAMES, FILTER 2, EXP TIME 20 SEC
  WAIT 5 MIN FOR RATES TO DAMP
                                                          CHANGE SHUTTER TO 1/250
  VERIFY RATES ON FDAI ARE <0.2°/SEC IN ALL AXES
                                                          2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500
REMOVE CM5 WINDOW COVER, MOUNT UV CARDBOARD SHADE
                                                          2 FRAMES, FILTER 4
  AND CAMERA
                                                          RECORD FR #
  2 FRAMES, FILTER 1, CHANGE SHUTTER TO B
                                                        REMOVE CAMERA AND UV CARDBOARD SHADE
1 2 FRAMES, FILTER 2, EXP TIME 20 SEC (Amorther
                                                        COMMENT - CONDITION OF CM5 WINDOW
                                                2 SECS
  CHANGE SHUTTER TO 1/250
                                                        REPLACE CM5 WINDOW COVER
  2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500
                                                        ENABLE ALL JETS
  2 FRAMES, FILTER 4
  RECORD FR #
CONFIGURE CAMERA: (UV COLOR)
  CM5/EL/105/CEX- CONT (f8,1/250,∞) (1 FR)
   RINGSLIDE
  MAG (M) , FR #
  I FRAME, FILTER 4
  RECORD FR #
```

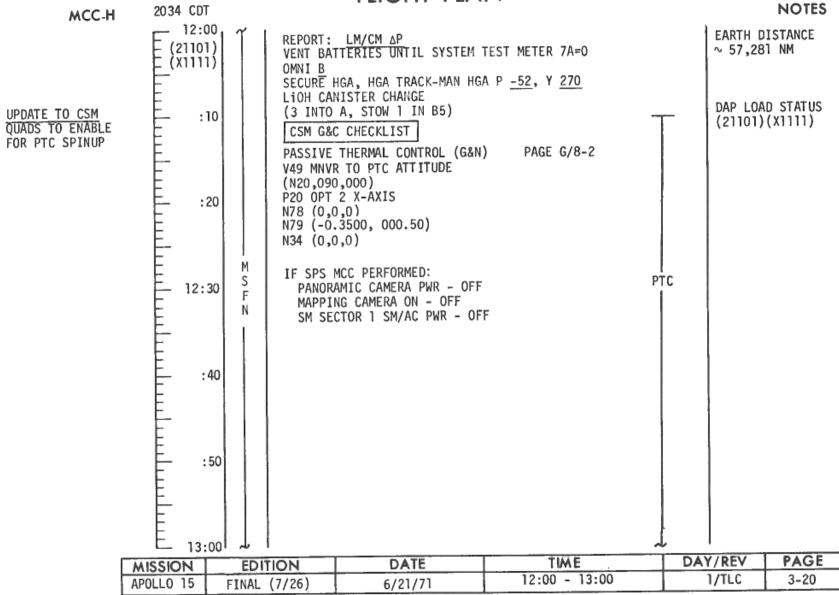
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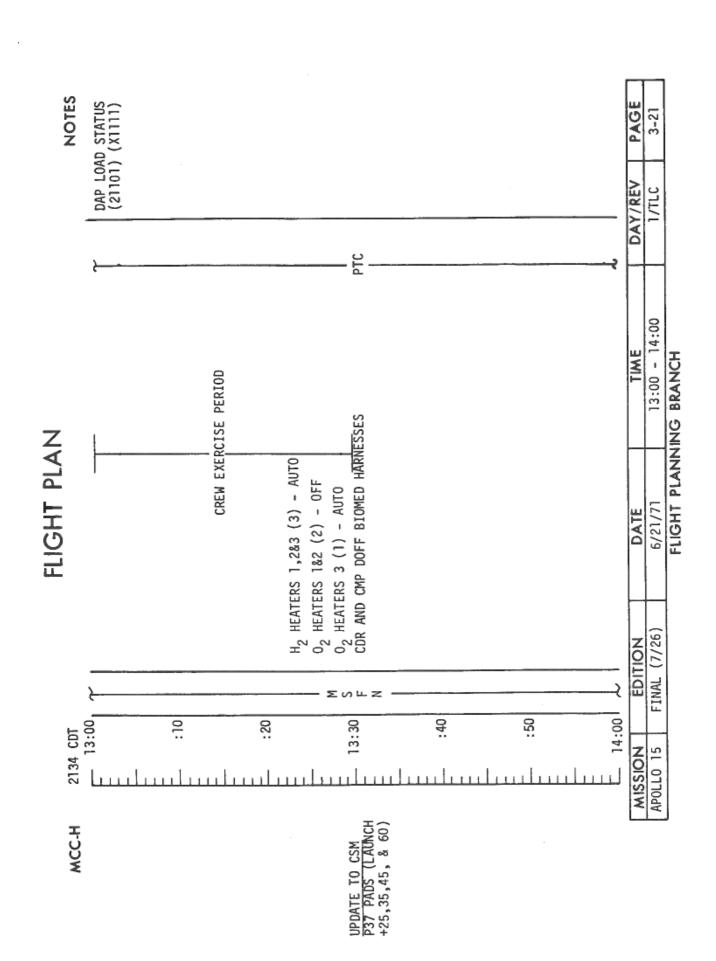
MCC-1 BURN TABLE

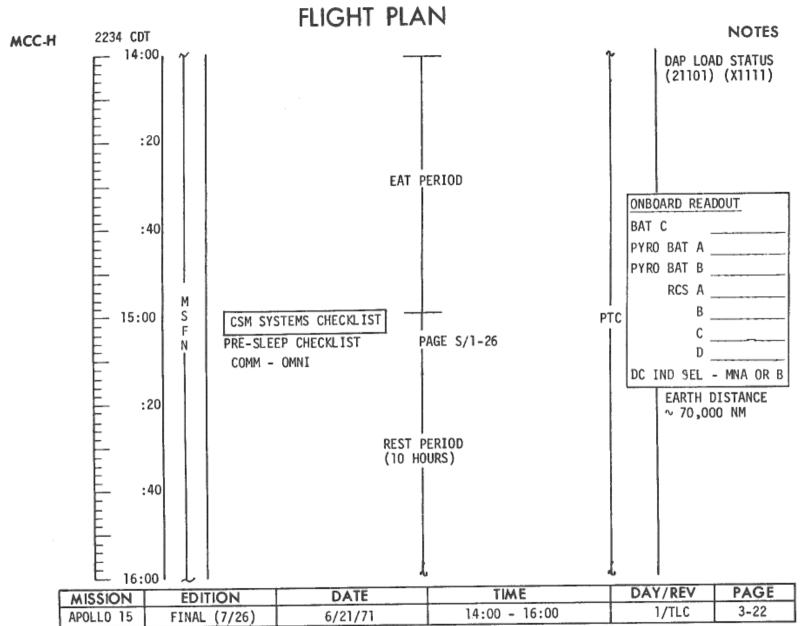
RESIDUALS	IF <2 FPS, TRIM X AXIS TO 0.2 FPS IF >2 FPS, NO TRIM	
SHUTDOWN	BT + 1 SEC	
ATT DEVIATION	±10° TERMINATE	
P OR Y RATES	10°/SEC TERMINATE	



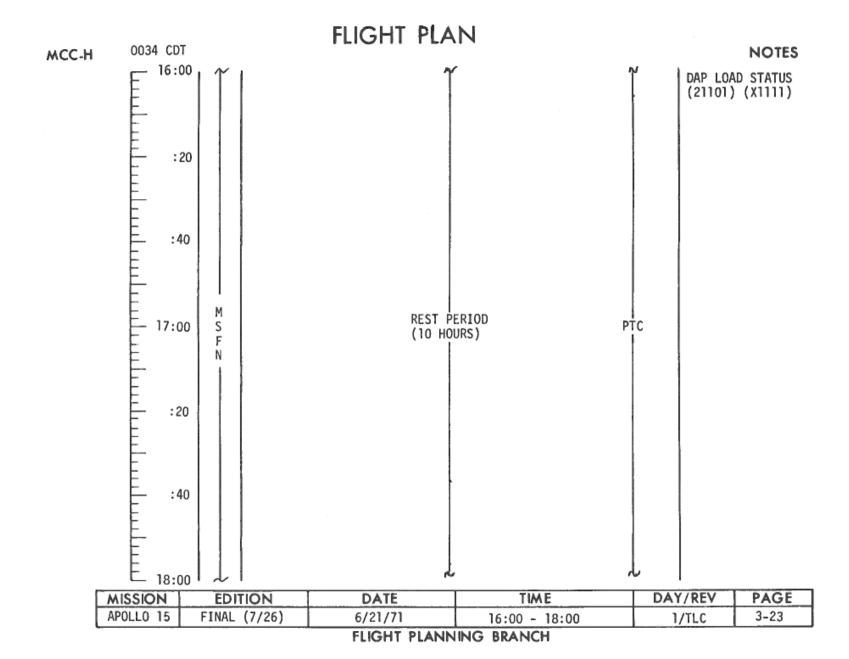


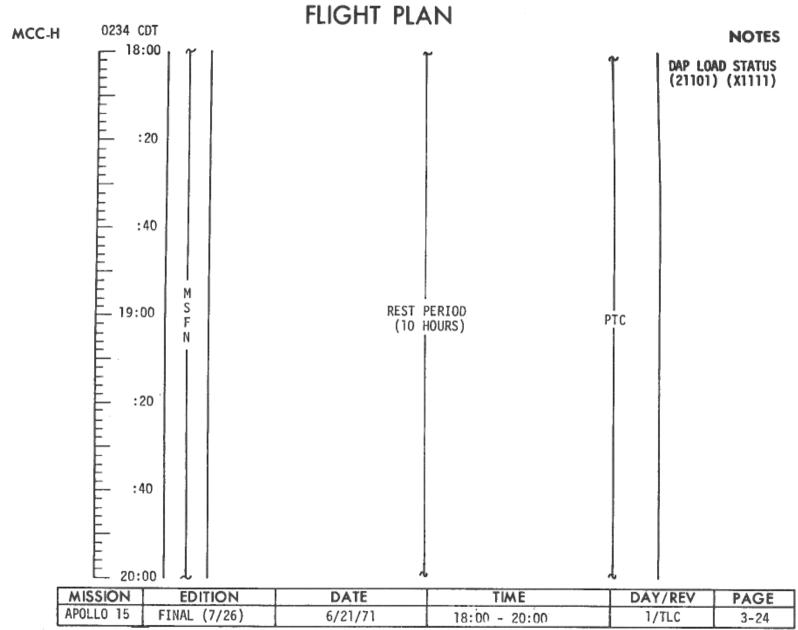
FLIGHT PLANNING BRANCH



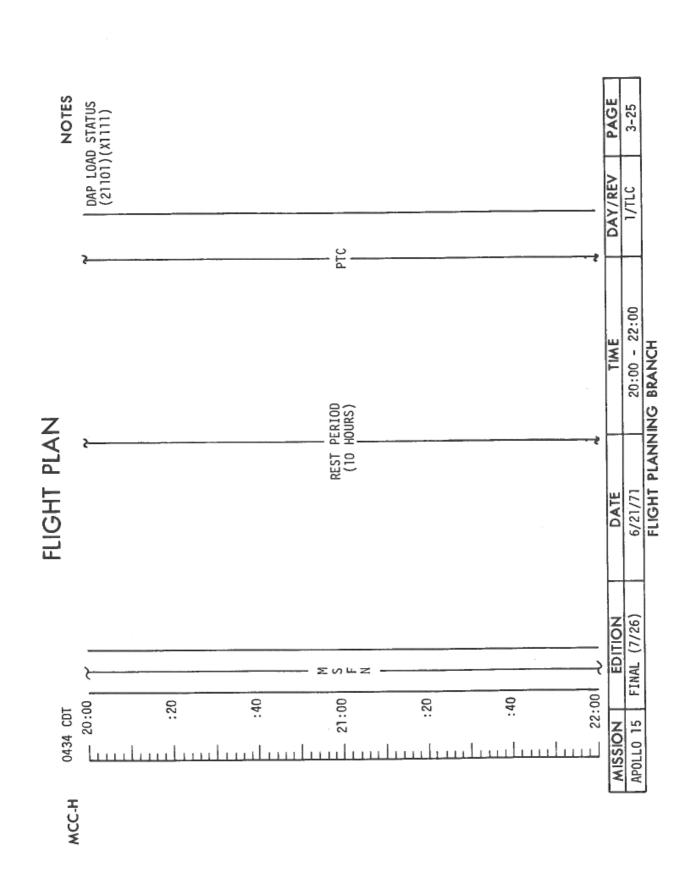


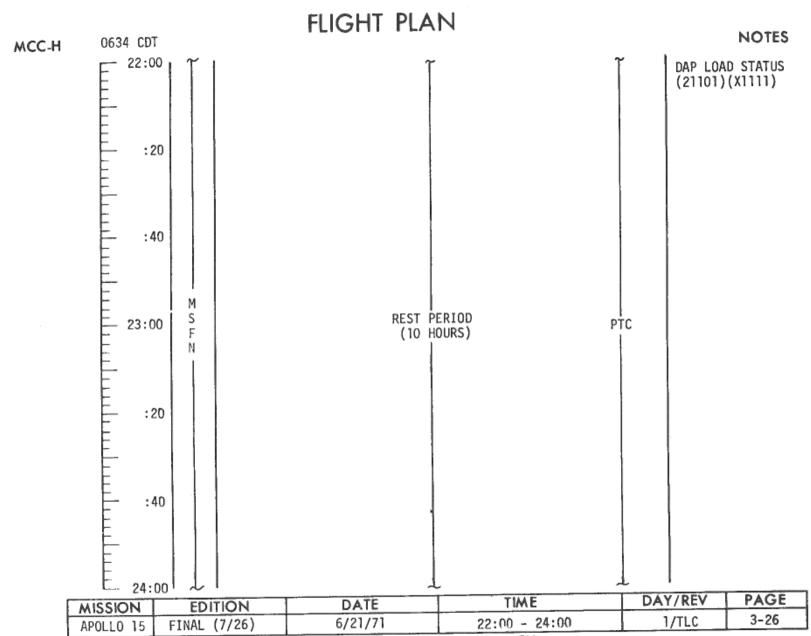
FLIGHT PLANNING BRANCH



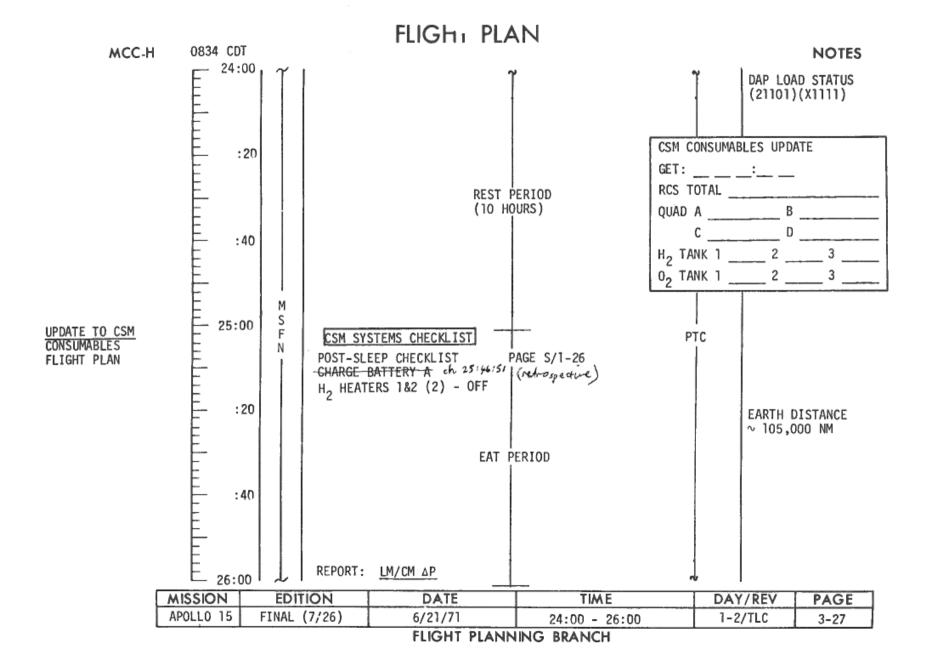


FLIGHT PLANNING BRANCH





FLIGHT PLANNING BRANCH



CYCLE FILM IN PANORAMIC AND MAPPING CAMERAS

ACQ MSFN, HGA: REACQ, NARROW (CUE MSFN FOR PITCH AND YAW)

VERIFY:

cb SCI EQUIP SEB 1 & 2 - CLOSE NON ESS BUS - MNA cb SM SECTOR 1 AC2 (3) - CLOSE MAPPING CAMERA IMAGE MTN - OFF

PCM BIT RATE - HIGH S-BD AUX TV - SCI SM SECTOR 1 SM/AC PWR - ON (UP) DATA SYS ON - ON

MAPPING CAMERA ON - STBY/tb-GRAY
PANORAMIC CAMERA MODE - STBY
PANORAMIC CAMERA POWER - ON (UP)/tb-BP (2 SEC), then GRAY

CONFIRM MSFN HAS TLM, CYCLE FILM ON MSFN CUE

MAPPING CAMERA ON - ON (2 MIN) then OFF
PANORAMIC CAMERA SELF TEST - ON (UP)/tb-BP (5 FR), then GRAY
PANORAMIC CAMERA SELF TEST - OFF (CTR)
PANORAMIC CAMERA POWER - OFF

AFTER MAPPING CAMERA OFF FOR 1 MIN: SM SECTOR 1 SM/AC PWR - OFF DATA SYS ON - OFF

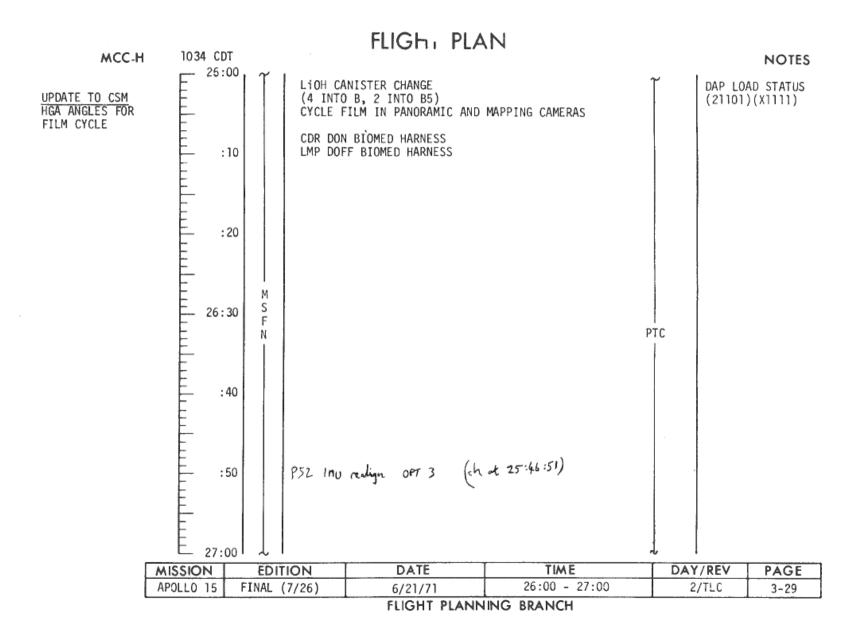
OMNI B

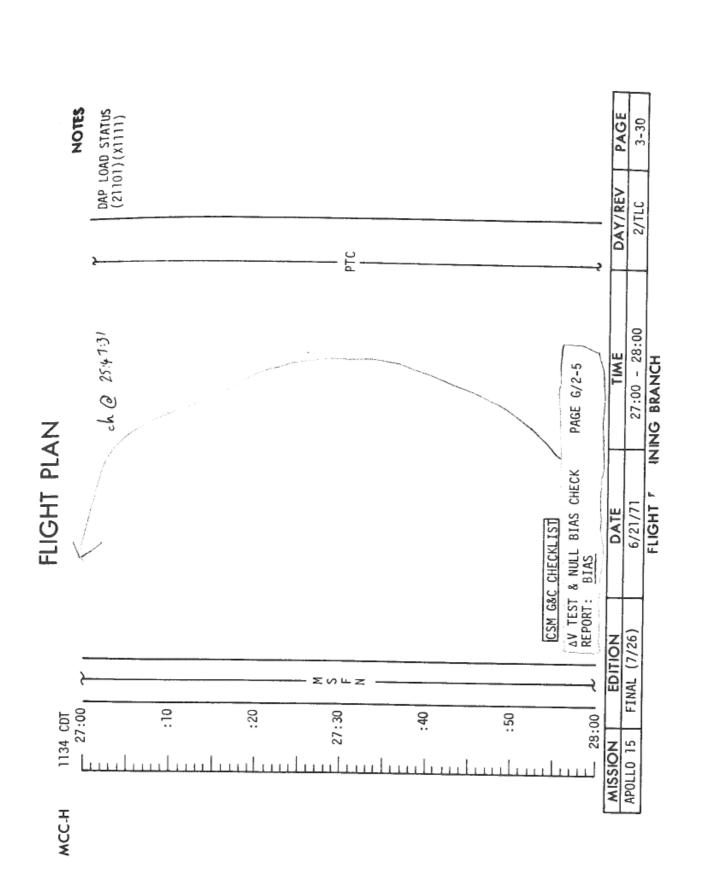
SECURE HGA, TRACK MAN HGA P -52, Y 270 S-BD AUX TV - OFF (CTR) PCM BIT RATE - LOW

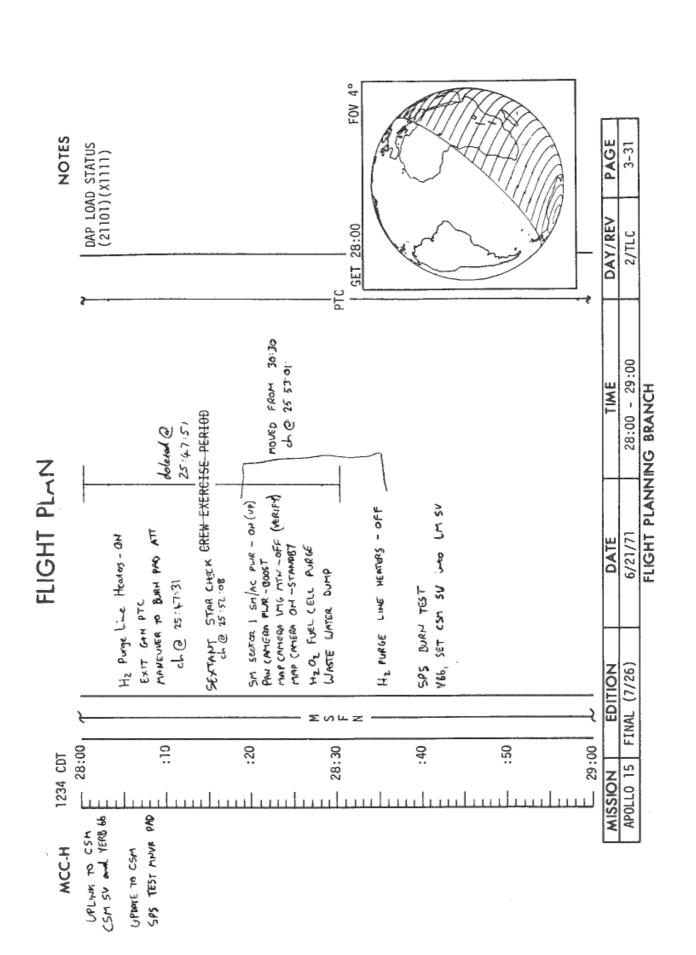
6/21/71

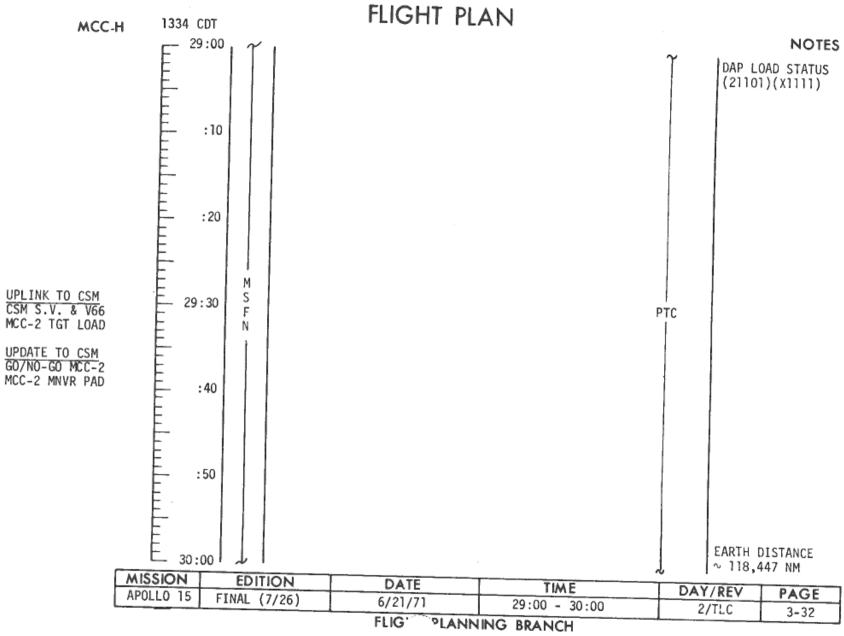
FINAL (7/26)

3-28





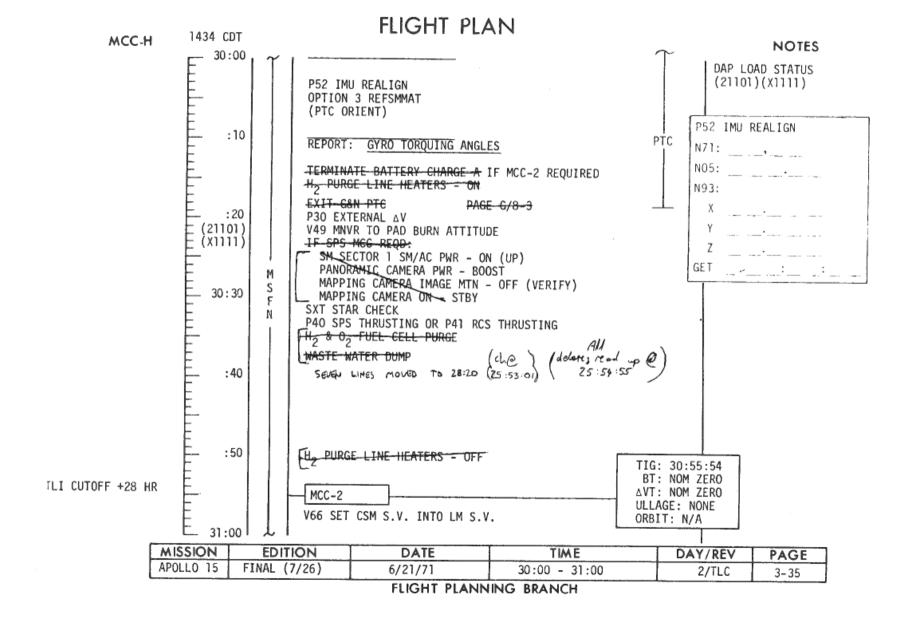


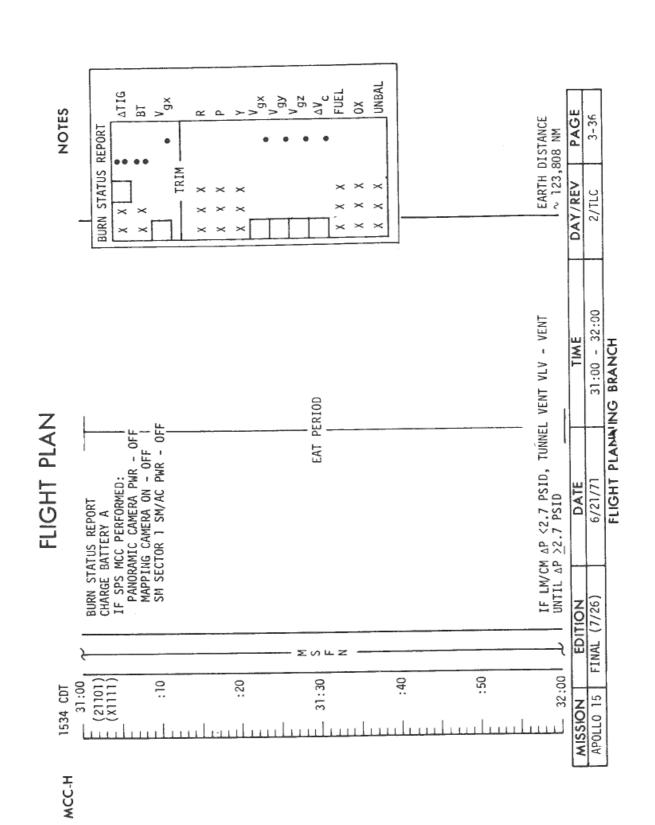


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MCC-2 BURN TABLE

RESIDUALS	AND HOLE THE
SHUTDOWN	15.1.11
ATT DEVIATION	110" ILIMINATI
P OR Y RATES	10"7"R





3-37

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FINAL (7/26)

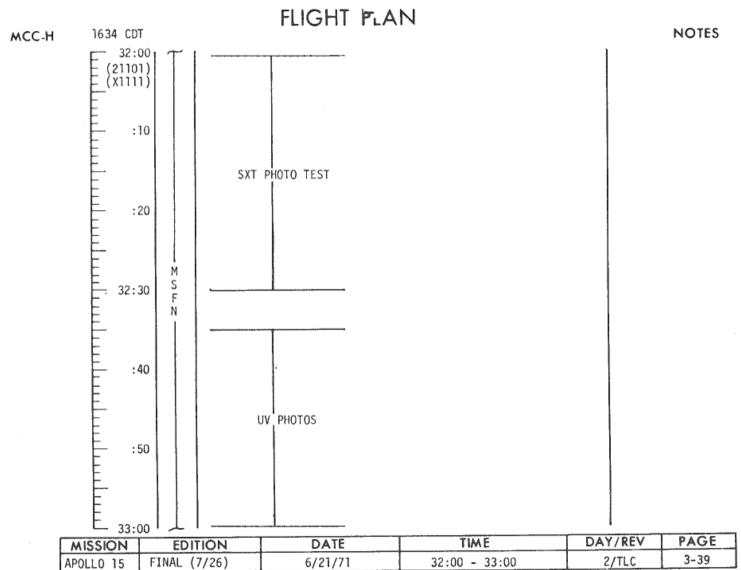
6/21/71

UV PHOTOS - CISLUNAR COAST SXT PHOTO TEST V49 MNVR TO EARTH UV PHOTO ATT (150,052,033) (32:50) V49 MNVR TO SXT PHOTO TEST ATT (32:15) OMNI D (040,012,050) HGA P -38, Y 318 2. CONFIGURE CAMERA CONFIGURE CAMERA: (UV) CM5/EL/105/UV-BRKT, CONT (f8,1/60,∞) (8 FR) CM/DAC/SXT/VHBW-PCM CABLE (EXP 1/500) 24 fps (5% MAG) MAG (H) , MAG % _____ UTILITY POWER - ON RINGSLIDE MAG (N) , FR # REMOVE R12 FLIGHT DATA FILE STOWAGE BOX DISABLE JETS A3, B3, C4, D4 P52 (NO MARKS) DAMP RATES INHIBIT ALL JETS EXCEPT A1 & C2 OR B2 & D1, A4, N70 (00000) N88 (+78675)(+47868)(+38950) C3, B4, D3 VERIFY THRU SXT THAT OPTICS BORESIGHTED ON STAR VERIFY FDAI SCALE - 5/1 WAIT 5 MIN FOR RATES TO DAMP G&N PWR (AC-PNL 5) - OFF VERIFY RATES ON FDAI ARE <0.2°/SEC IN ALL AXES MOUNT DAC ON SXT DIM INTERIOR LIGHTS, DAC - ON, 24 fps FOR 2 SEC REMOVE CM5 WINDOW COVER, MOUNT UV CARDBOARD SHADE AND CAMERA CHANGE TO TIME & 1/60 2 FRAMES, FILTER 1, CHANGE SHUTTER TO B 1 FRAME - 60 SEC EXP TIME I & FRAMES. FILTER 2, EXP TIME 20 SEC 1@ 2 sec exp. 1 FRAME - 20 SEC EXP TIME CHANGE SHUTTER TO 1/250 1 FRAME - 5 SEC EXP TIME ch @ 11:40:01 2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500 1 FRAME - 1 SEC EXP TIME 2 FRAMES, FILTER 4 CHANGE TO 24 fps & 1/500 SEC RECORD FR # RUN DAC FOR 2 SEC, LIGHTS UP CONFIGURE CAMERA ENABLE JETS CM5/EL/105/CEX- CONT (f8,1/250,∞) (1 FR) RINGSLIDE V49 MNVR TO SXT PHOTO TEST ATT (32:25) MAG (M) , FR # (090,000,043) HGA P -47, Y 353 1 FRAME, FILTER 4 RECORD FR # REPEAT STEP 3 REMOVE CAMERA AND UV CARDBOARD SHADE COMMENT - CONDITION OF CM5 WINDOW RECORD MAG % REPLACE CM5 WINDOW COVER REMOVE AND STOW DAC ENABLE ALL JETS

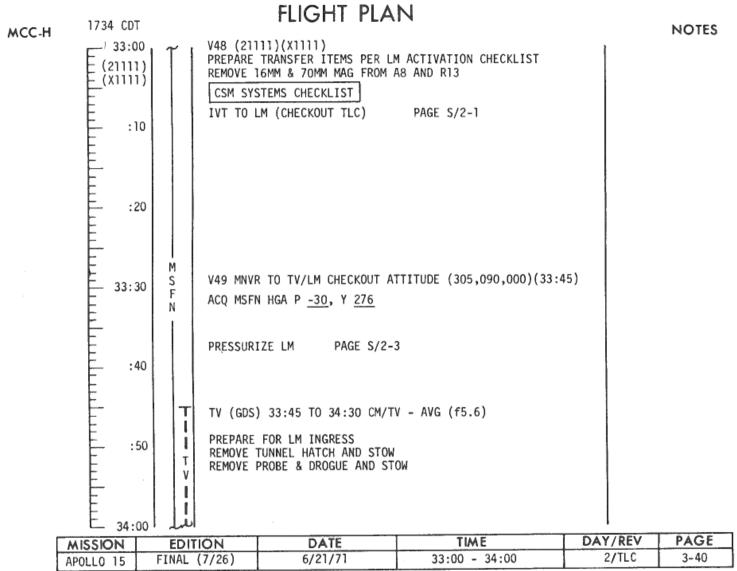
3-38

6/21/71

FINAL (7/26)



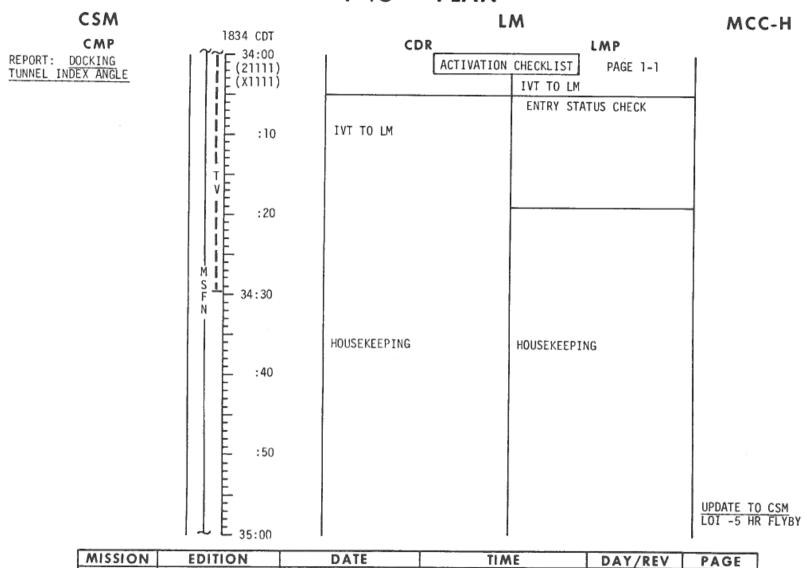
FLIGHT PLANNING BRANCH



FLIGHT P' 'INING BRANCH

FIIG TPIAN

F'IG ' PLAN



6/21/71

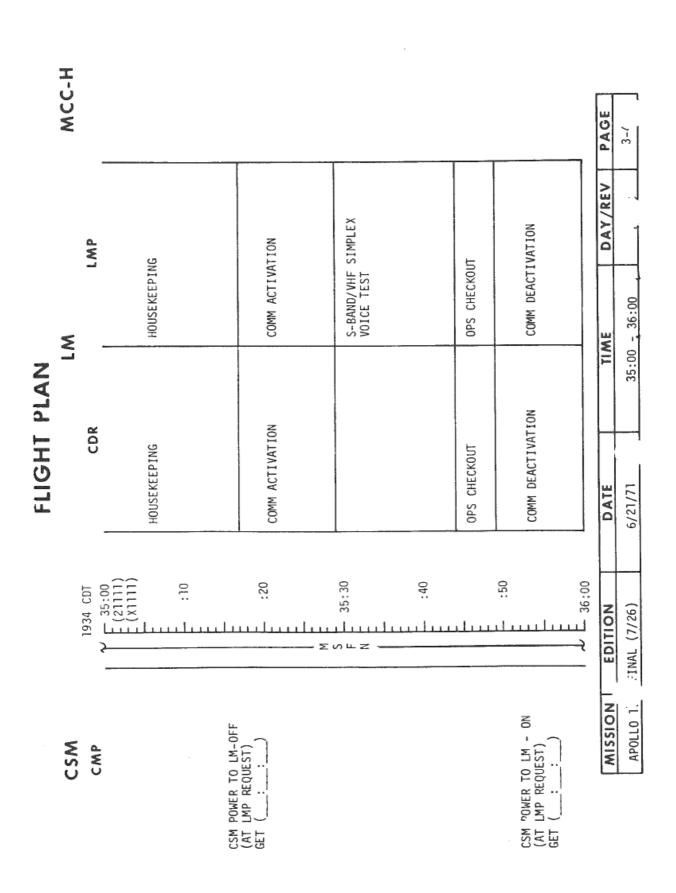
34:00 - 35:00

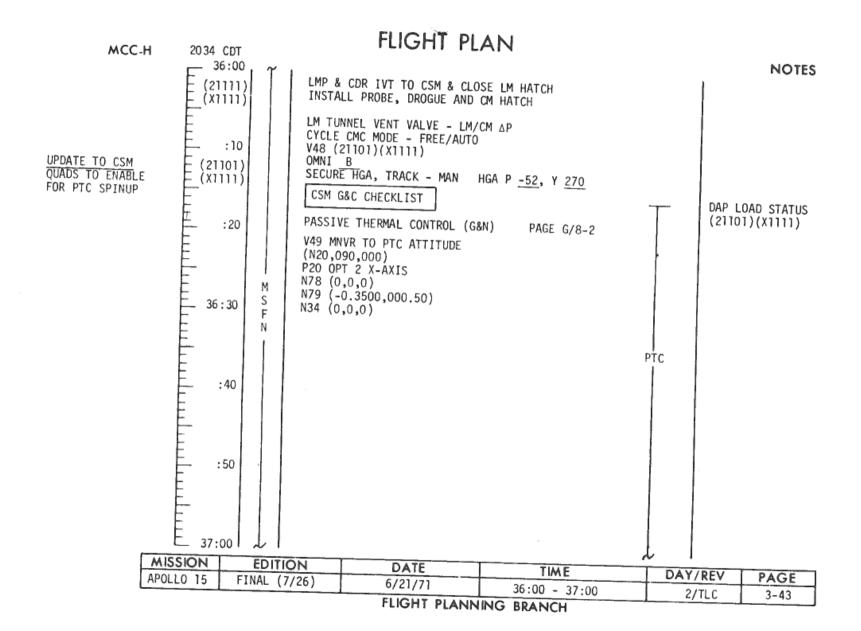
2/TLC

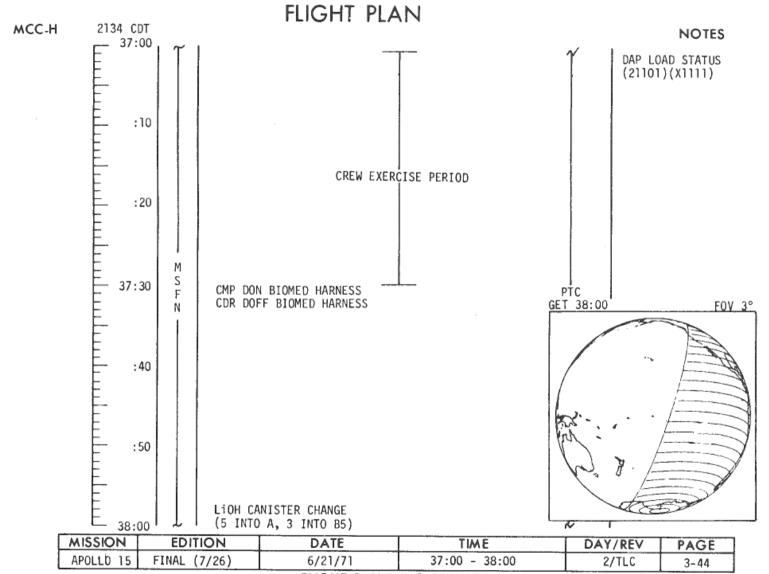
3-41

APOLLO 15

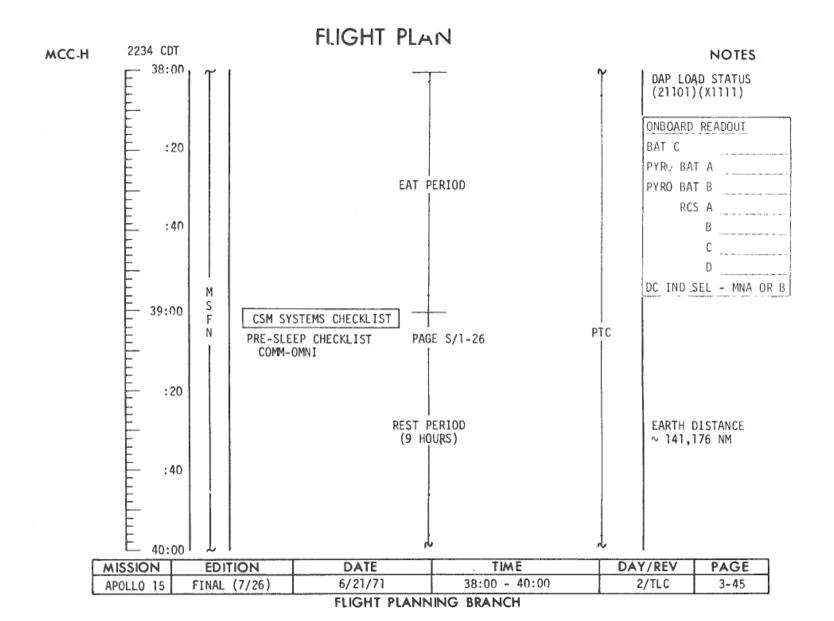
FINAL (7/26)

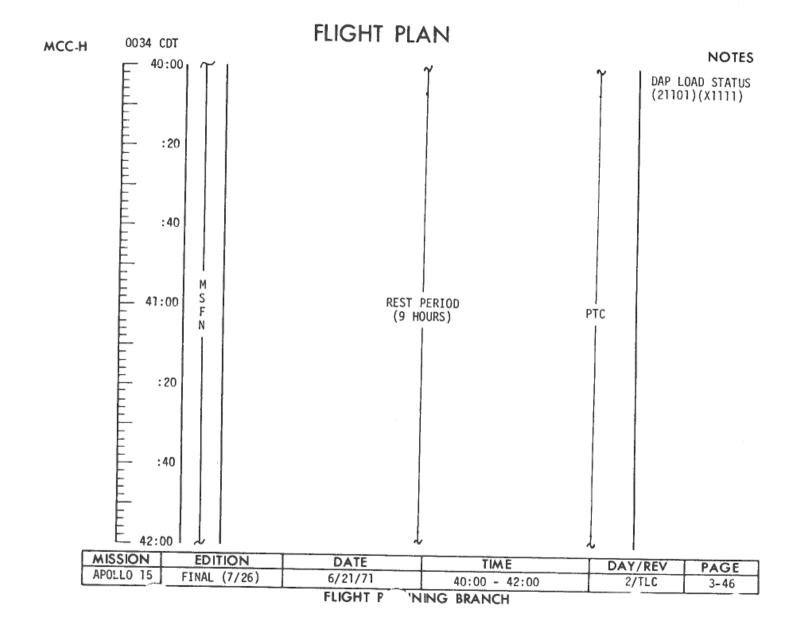


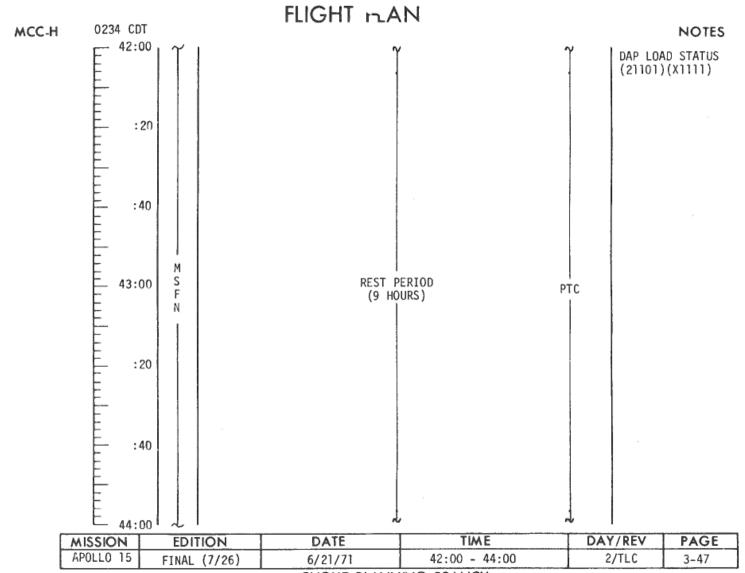




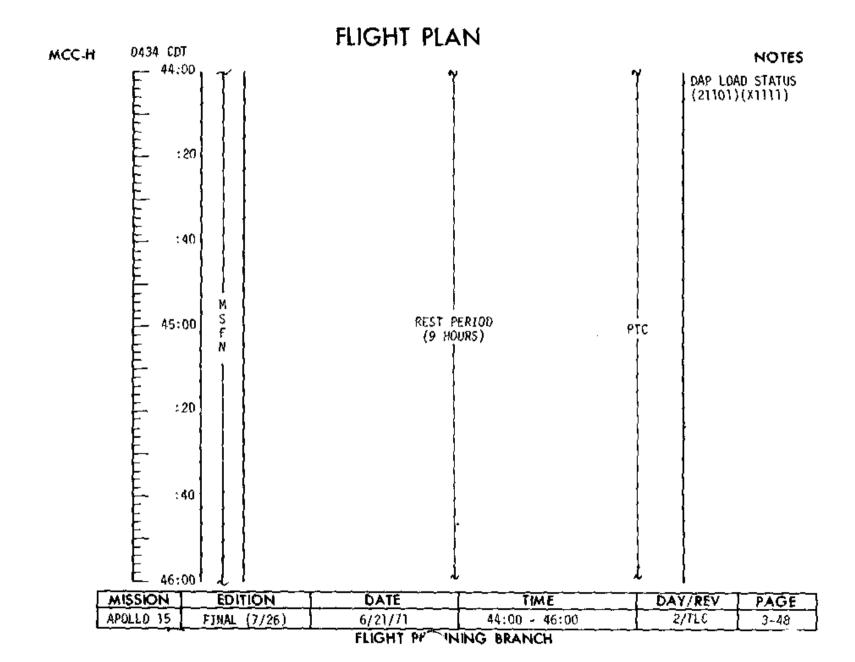
FLIGHT PLA' "ING BRANCH

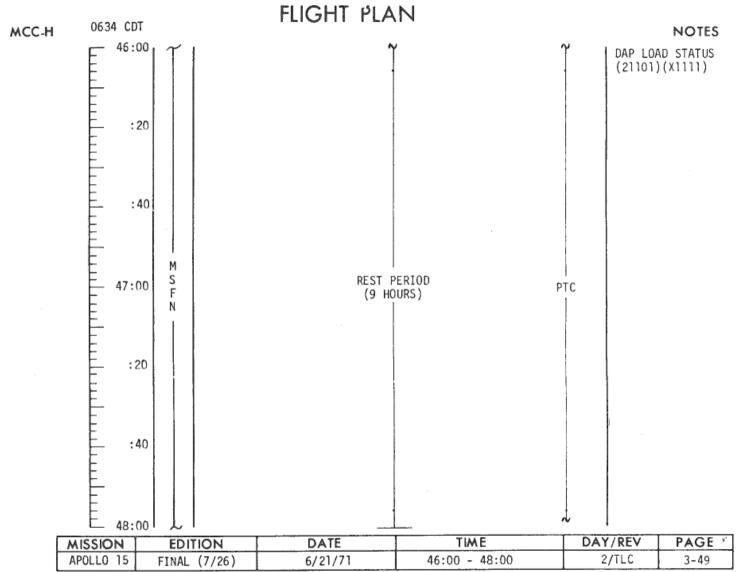




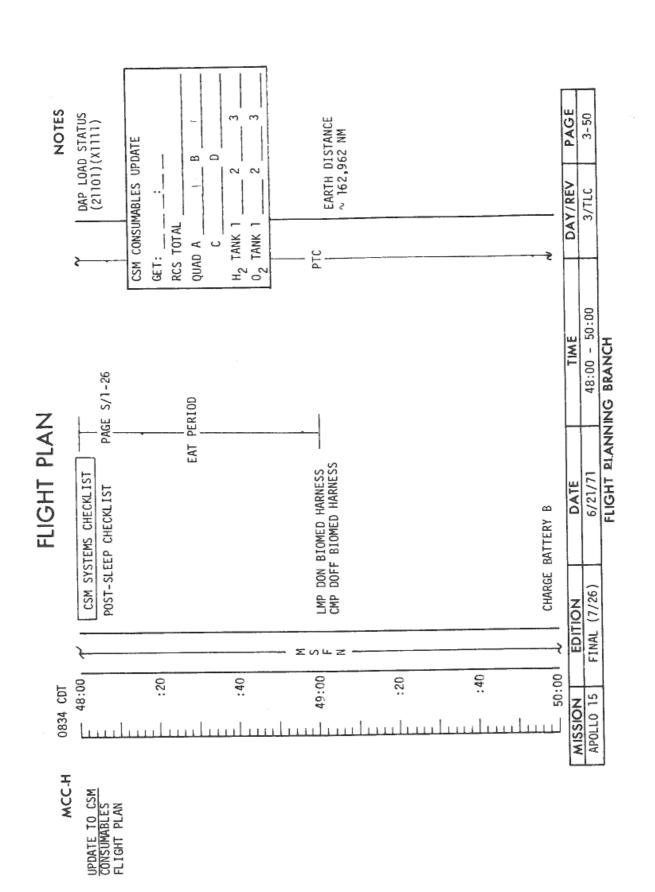


FLIGHT PLANNING BRANCH



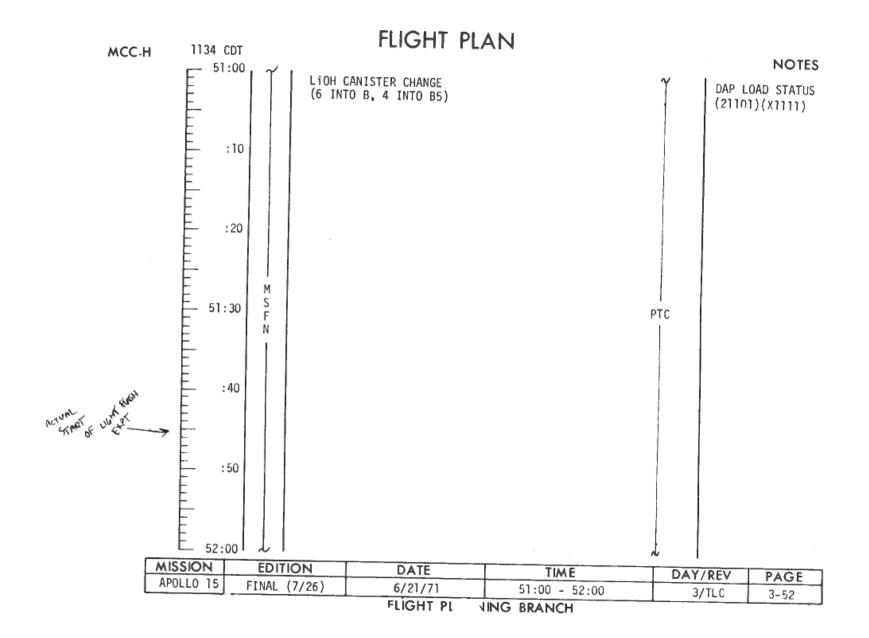


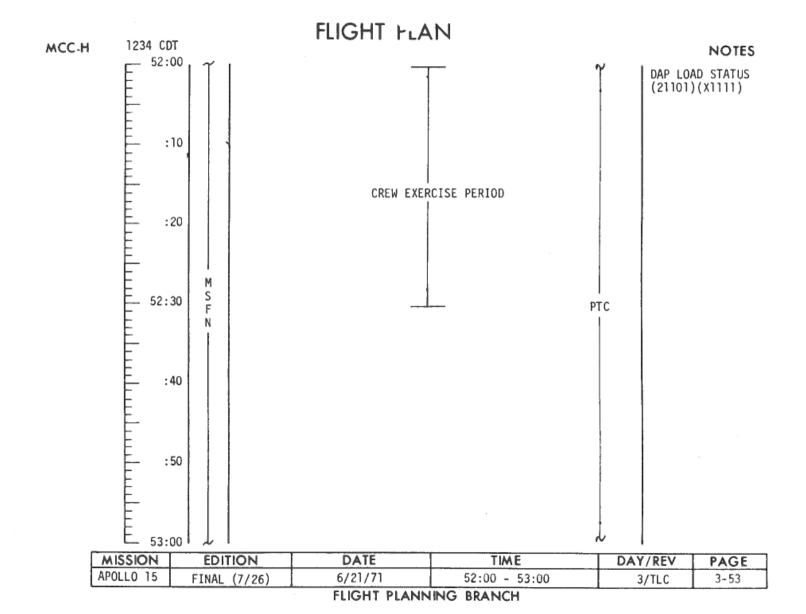
FLIGHT PLANNING BRANCH



FLIGHT PLAN 1034 CDT MCC-H NOTES 50:00 VISUAL LIGHT FLASH PHENOMENON EXPERIMENT DAP LOAD STATUS SET KITCHEN TIMER TO 50 MIN (21101)(X1111) ALL CREWMEN DON EYESHIELDS REPORT: DON EYESHIELDS (GET : : FLASH FREQUENCY DURING FIRST 50 FLASH DESCRIPTION MINUTES ONE CREW-:10 CREW POSITION W.R.T. X, Y, AND Z AXES MAN WILL FACE THE AND LOCATION IN CM DIRECTION OF THE -X AXIS. DURING THE LAST 10 MINUTES HE WILL FACE THE DIRECTION OF THE :20 +X AXIS. EXPERI-MENT WILL BE TER-MINATED ON MSFN CUE. 50:30 PTC :40 :50 REPORT: DOFF EYESHIELDS (GET ___:__:__) 51:00 MISSION TIME **EDITION** DATE DAY/REV PAGE APOLLO 15 FINAL (7/26) 6/21/71 50:00 - 51:00 3/TLC 3-51

FLIGHT PLANNING BRANCH





CYCLE FILM IN PANORAMIC AND MAPPING CAMERAS

```
ACQ MSFN, HGA: REACQ, NARROW (CUE MSFN FOR PITCH AND YAW)
VERIFY:
```

cb SCI EQUIP SEB 1 & 2 - CLOSE
NON ESS BUS - MNA
cb SM SECTOR 1 AC2 (3) - CLOSE
MAPPING CAMERA IMAGE MTN - OFF

PCM BIT RATE - HIGH
S-BD AUX TV - SCI
SM SECTOR 1 SM/AC PWR - ON (UP)
DATA SYS ON - ON
MAPPING CAMERA ON - STBY/tb-GRAY
PANORAMIC CAMERA MODE - STBY
PANORAMIC CAMERA POWER - ON (UP)/tb-BP (2 SEC), then GRAY

CONFIRM MSFN HAS TLM, CYCLE FILM ON MSFN CUE

MAPPING CAMERA ON - ON (2 MIN), then OFF
PANORAMIC CAMERA SELF TEST - ON (UP)/tb-BP (5 FR), then GRAY
PANORAMIC CAMERA SELF TEST - OFF (CTR)
PANORAMIC CAMERA POWER - OFF

MAPPING CAMERA ON - STBY/16-GRAY AFTER MAPPING CAMERA OFF FOR 1 MIN: SM-SECTOR 1 SM/AC PWR - OFF

CAN 6 49:35:59

SECURE HGA, TRACK-MAN HGA P -52, Y 270 S-BD AUX TV - OFF (CTR) PCM BIT RATE - LOW

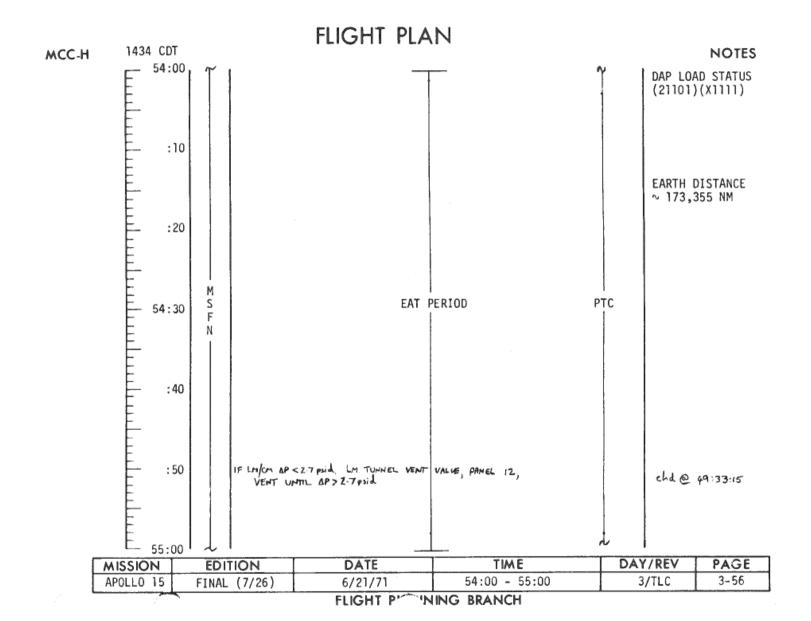
FINAL (7/26)

6/21/71

3-54

FLIGHT PLAN NOTES 1334 CDT MCC-H SYNCHRONIZE MISSION TIMER TO CMC CLOCK (IF REQUIRED) DAP LOAD STATUS UPLINK TO CSM NOT LIFT-OFF TIME (S2:46) (21101)(X11111)VO5NO1E, 1706E T EPHEM UPDATE LOAD B OID UPDATE TO CSM T EPHEM :10 (IF REQUIRED) 03 :20 LIFT-OFF TIME WILL BE UPDATED IF THE TIME PROPAGATED AHEAD TO START OF PTC REV 2 DIFFERS FROM 53:30 80:40:37.6 BY MORE THAN 1 MIN :40 :50 CSM G&C CHECKLIST AV TEST & NULL BIAS CHECK PAGE G/2-5 REPORT: BIAS UPDATE TO CSM CYCLE FILM IN PANORAMIC AND MAPPING CAMERAS HGA ANGLES FOR FILM CYCLE 54:00 PAGE DAY/REV TIME DATE EDITION MISSION 3-55 3/TLC 53:00 - 54:00 6/21/71 FINAL (7/26) APOLLO 15

FLIGHT PLANNING BRANCH



UV PHOTOS - CISLUNAR COAST

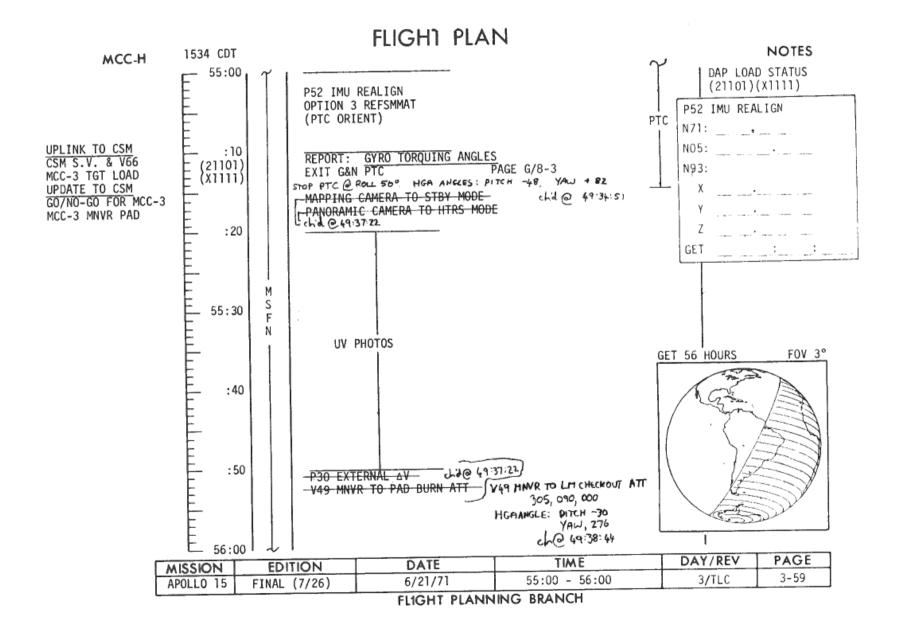
```
V49 MNVR TO EARTH UV PHOTO ATT (153,050,030) (55:35)
  OMNI D
CONFIGURE CAMERA: (UV)
  CM5/EL/105/UV-BRKT, CONT (f4.3,1/60,∞) (8 FR)
  RINGSLIDE
  MAG (N) , FR #
  REMOVE RT2 FLIGHT DATA FILE STOWAGE BOX
DAMP RATES
  INHIBIT ALL JETS EXCEPT AT & C2 OR B2 & D1, A4, C3, B4, D3
  VERIFY FDAI SCALE - 5/1
  WAIT 5 MIN FOR RATES TO DAMP
  VERIFY RATES ON FDAI ARE <0.2°/SEC IN ALL AXES
REMOVE CM5 WINDOW COVER, MOUNT UV CARDBOARD SHADE AND CAMERA
  2 FRAMES, FILTER 1, CHANGE SHUTTER TO B
1 # FRAMES, FILTER 2, EXP TIME 20 SEC and 1 @ 2 sec exp
 CHANGE SHUTTER TO 1/250
                                                     de@ 11:40:01
  2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500
  2 FRAMES, FILTER 4
  RECORD FR #
CONFIGURE CAMERA
 CM5/EL/105/CEX-CONT (f8,1/250,∞) (1 FR) (factor ?)
  RINGSLIDE
 MAG (M) ____, FR # ___
 1 FRAME, FILTER 4
 RECORD FR #
REMOVE CAMERA AND UV CARDBOARD SHADE
COMMENT - CONDITION OF CM5 WINDOW
REPLACE CM5 WINDOW COVER
ENABLE ALL JETS
```

MAPPING CAMERA TO STBY MODE

cb SCI EQUIP SEB 1 & 2 - CLOSE (VERIFY) NON ESS BUS - MNA (VERIFY) SM SECTOR 1 SM/AC PWR - ON (UP) MAPPING CAMERA IMAGE MTN - OFF (VERIFY) MAPPING CAMERA ON - STBY

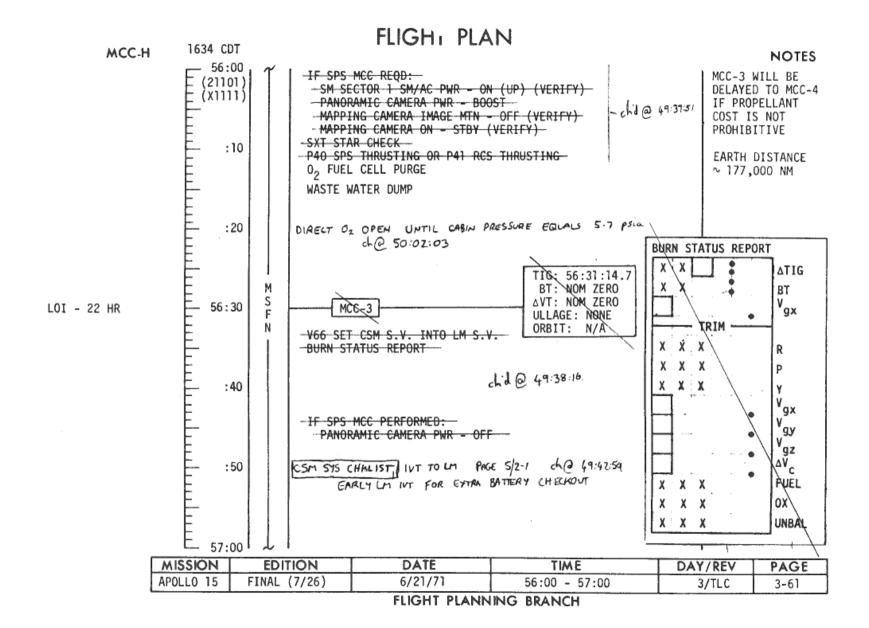
PANORAMIC CAMERA TO HTRS MODE

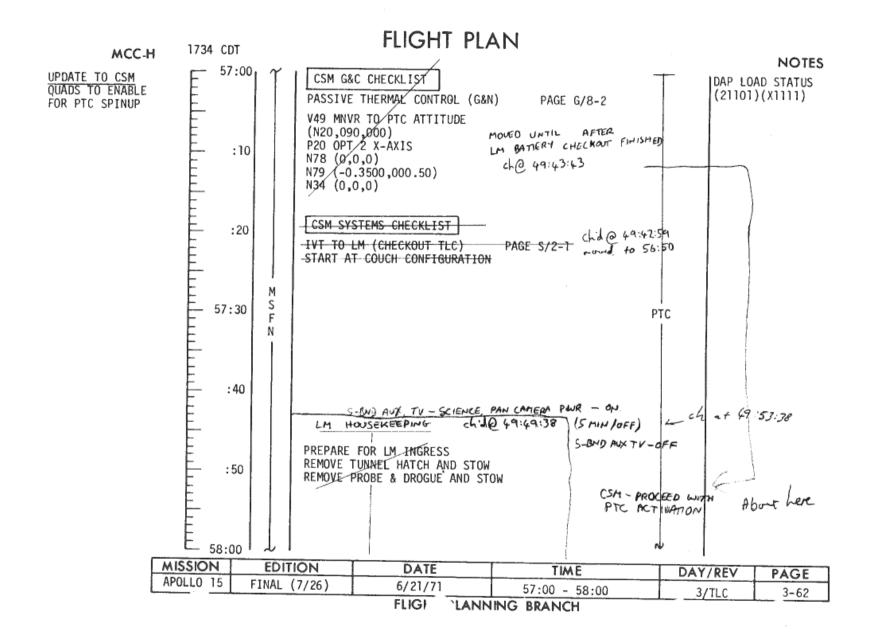
cb SCI EQUIP SEB 1 & 2 - CLOSE (VERIFY)
NON ESS BUS - MNA (VERIFY)
cb SM SECTOR 1 SM/AC2 (3) - CLOSE (VERIFY)
SM SECTOR 1 SM/AC PWR - ON (UP)
PANORAMIC CAMERA POWER - OFF (VERIFY)
PANORAMIC CAMERA MODE - STBY
PANORAMIC CAMERA SELF TEST - HTRS

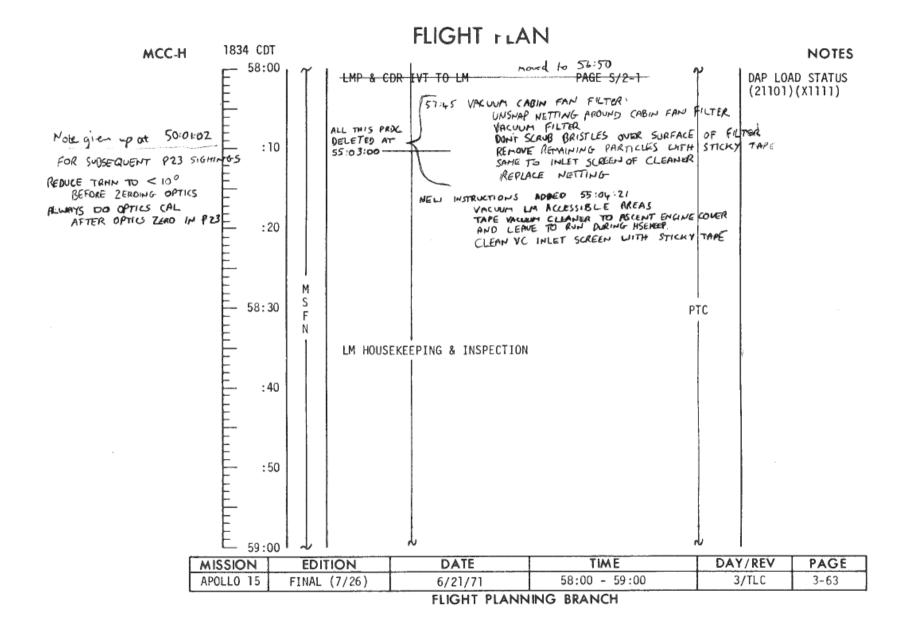


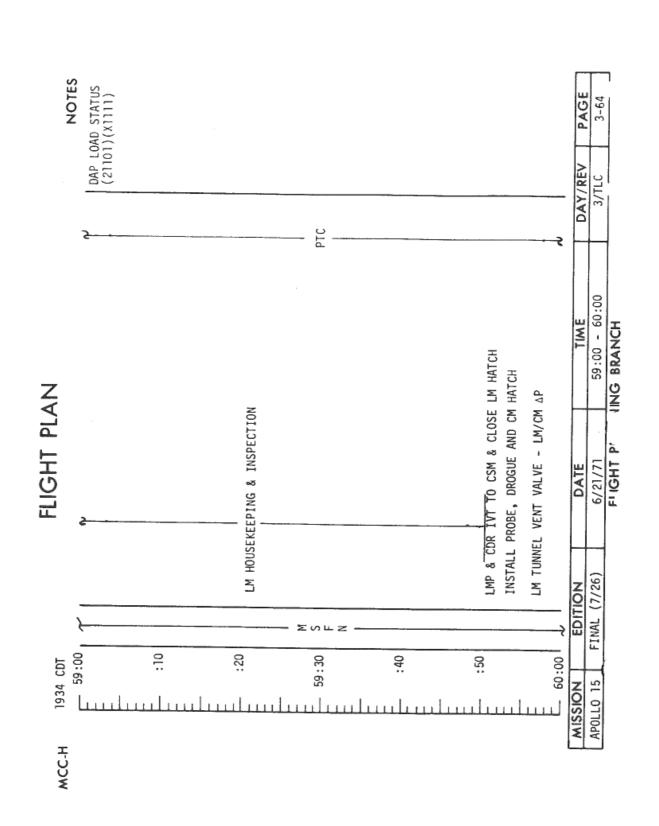
MCC-3 BURN TABLE

P OR Y	ATT	SHUTDOWN	RESIDUALS
RATES	DEVIATION	TIME	
10°/SEC TERMINATE	±10° TERMINATE	BT + 1 SEC	IF <2 FPS, TRIM X AXIS TO 0.2 FPS IF >2 FPS, NO TRIM









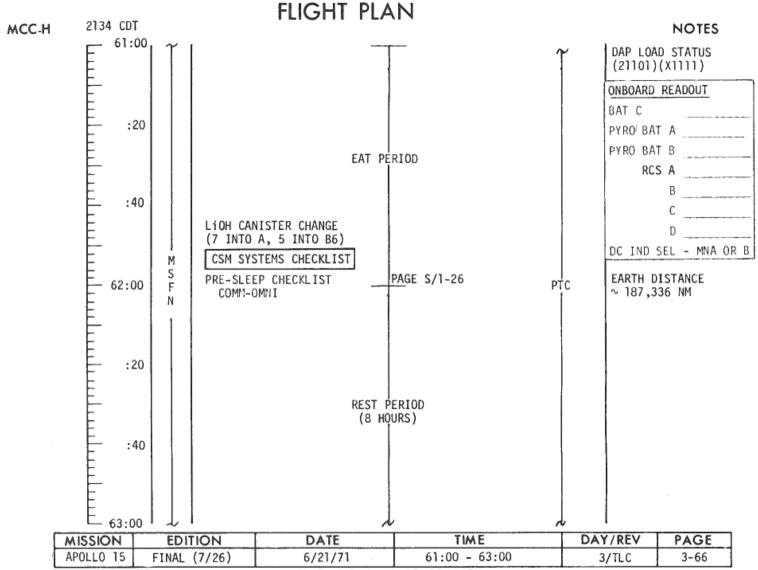
FLIGHI PLAN 2034 CDT MCC-H NOTES 60:00 UPDATE TO CSM PANORAMIC CAMERA POWER - ON (FOR 5 MIN)/OFF DAP LOAD STATUS HGA ANGLES FOR CDR DON BIOMED HARNESS (21101)(X1111) LMP DOFF BIOMED HARNESS CAM TEMP MONITOR MCC-H MONITOR DATA SYSTEM further d @ 59 48:56 che 49:58:10 FOR CAMERA LENS :10 GO:00 ofter S-BND AUX TV to OFF 5-BND AUX TV - SCIENCE **TEMPERATURES** PAN (AH PLUR - ON FOR 5 MM) JOFF added. WAIT FOR MSFN CE PAN CAMERA SELF TEST TO OFF PAN CAM SELFTEST - OFF MAP CAMERA ON to OFF MAP CAM, ON -DEF 3-BND AUX, TV -OFF CREW EXERCISE PERIOD :20 FOR THIS TASK HGA P,-30 4,90 PTC 60:30 S-170 BISTATIC RADAR FREQUENCY CHECK VHF AM B - DUPLEX VHF RANGING - ON :40 VHF ANT - LEFT (VERIFY) NOTE: MSFN WILL TURN OFF S-BAND UPLINK FOR APPROXIMATELY 5 MIN WHILE S-BAND DOWNLINK FREQUENCY IS MEASURED :50 ON GROUND CUE:

MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
APOLLO 15	FINAL (7/26)	6/21/71	60:00 - 61:00	3/TLC	3-65

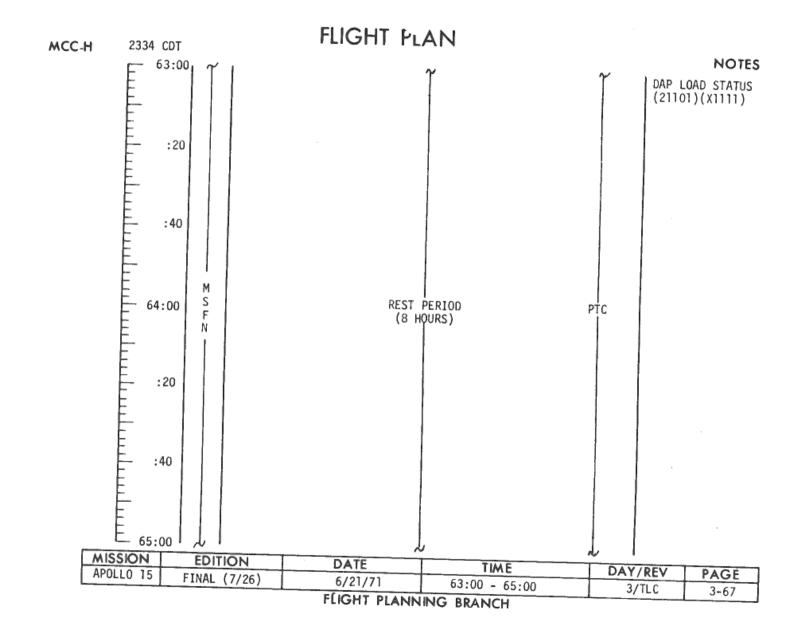
VHF AM B - OFF VHF RANGING - OFF

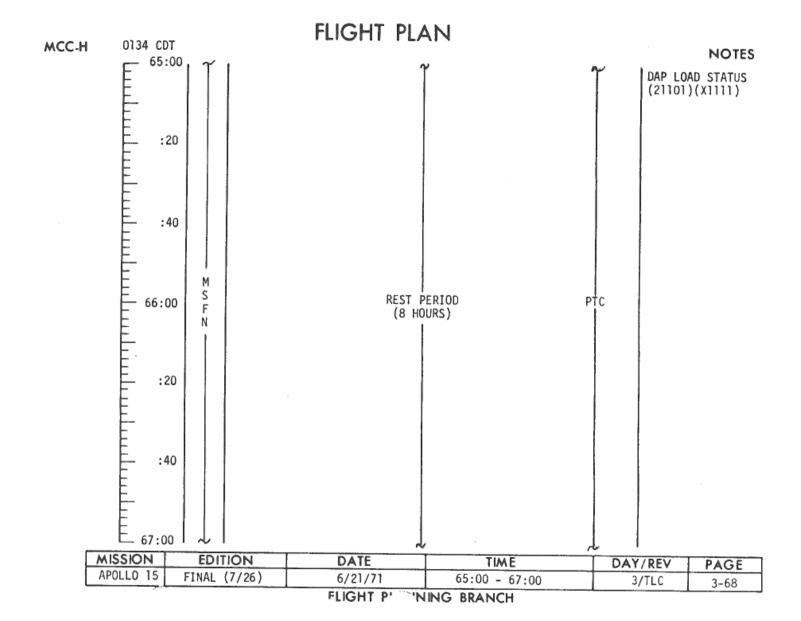
61:00

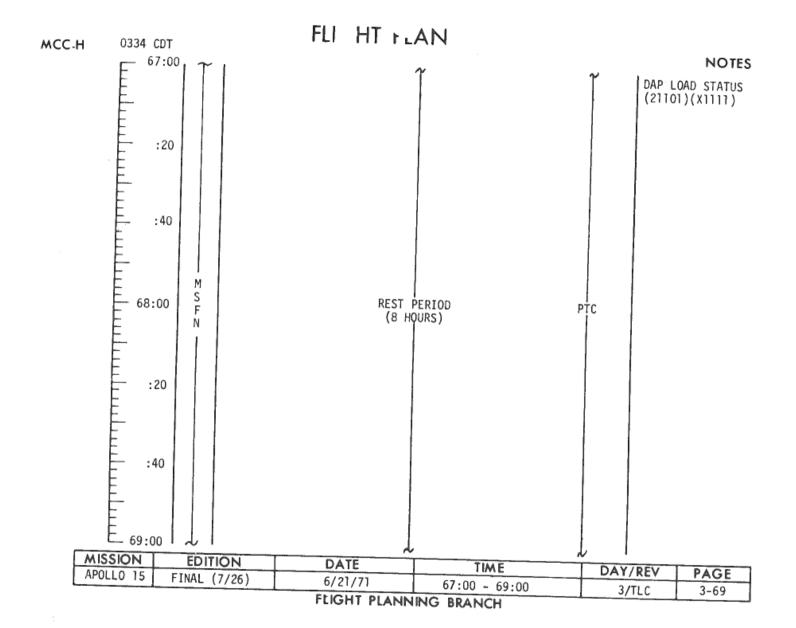
FLIGHT PLANNING BRANCH

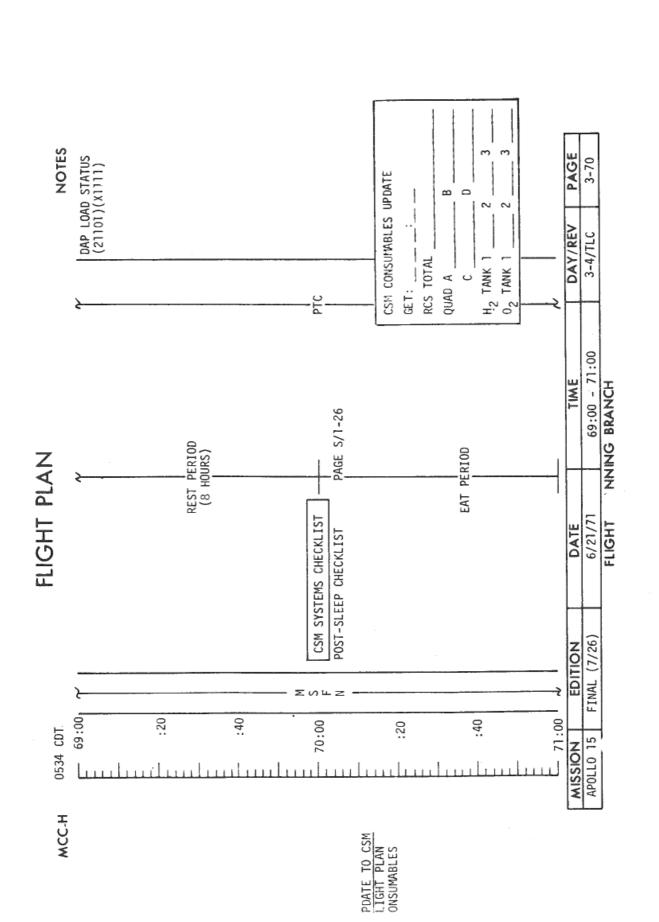


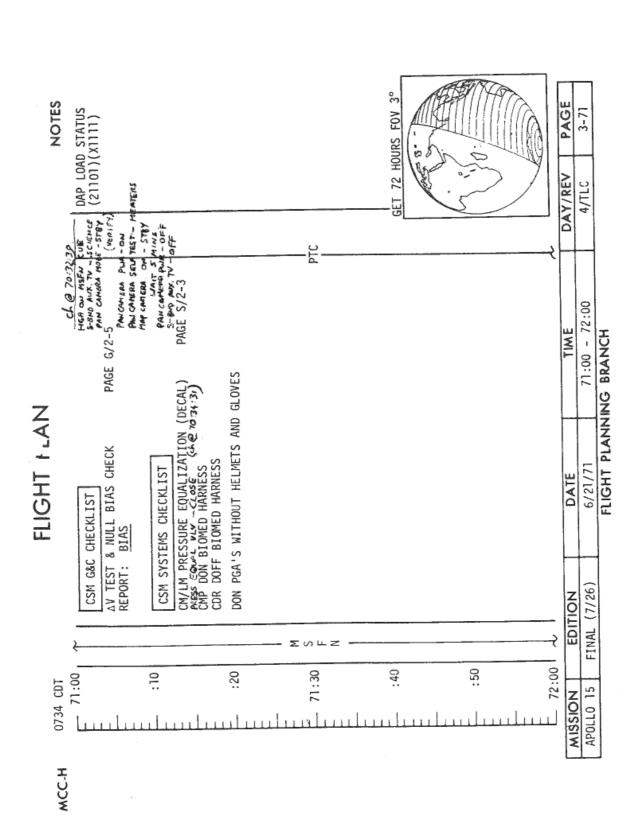
FLIGHT PL' VING BRANCH

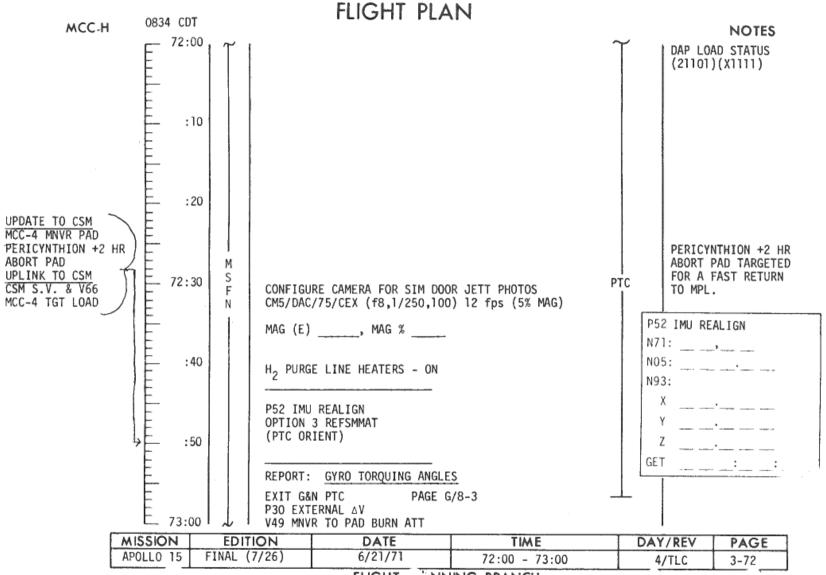












FLIGHT 'NNING BRANCH

OPTICS CAL PROCEDURES

(READ UP AT 70:38:11)

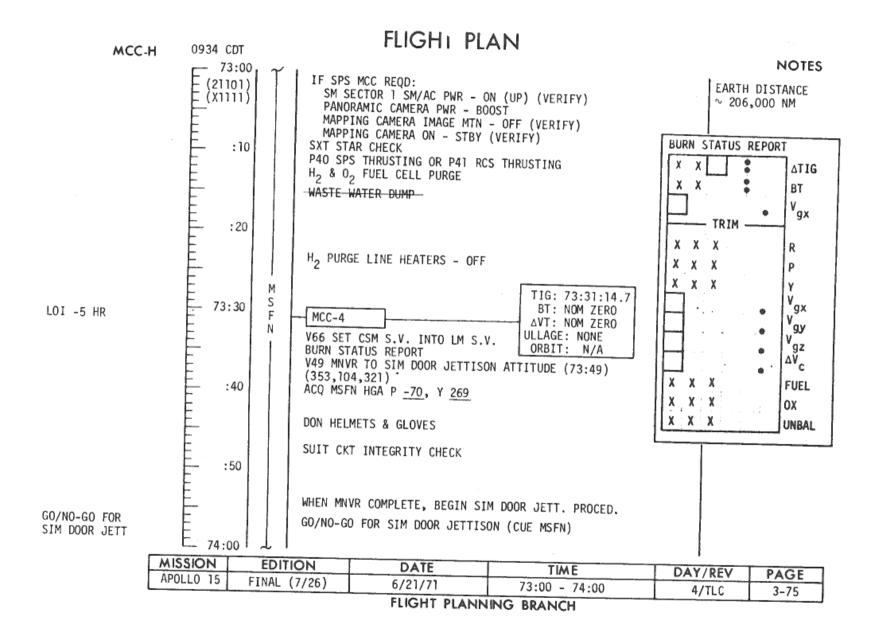
IF OPTICS POWER IS OFF
PLACE 'ZERO' SWITCH-OFF
BEFORE OPTICS POWER - DN

APTER OPTICS, POWER ON
DRIVE OPTICS HANVALLY
TO TRUNNION OF < 10°
BEFORE 'ZERO' SWITCH - ON

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MCC-4 BURN TABLE

P OR Y	ATT	SHUTDOWN	RESIDUALS
RATE	DEVIATION	TIME	
10°/SEC	±10°	BT + 1 SEC	TRIM ONLY X AXIS
TERMINATE	TERMINATE		TO 0.2 FPS



SIM DOOR JETTISON PROCEDURE

1. VERIFY:

MAPPING CAMERA IMAGE MTN - OFF
MAPPING CAMERA TRACK - OFF/TB-GRAY

RAY/X DR - \alpha OFF

SUBSAT EXTEND/LAUNCH - OFF/TB-GRAY

DATA SYS ON - ON

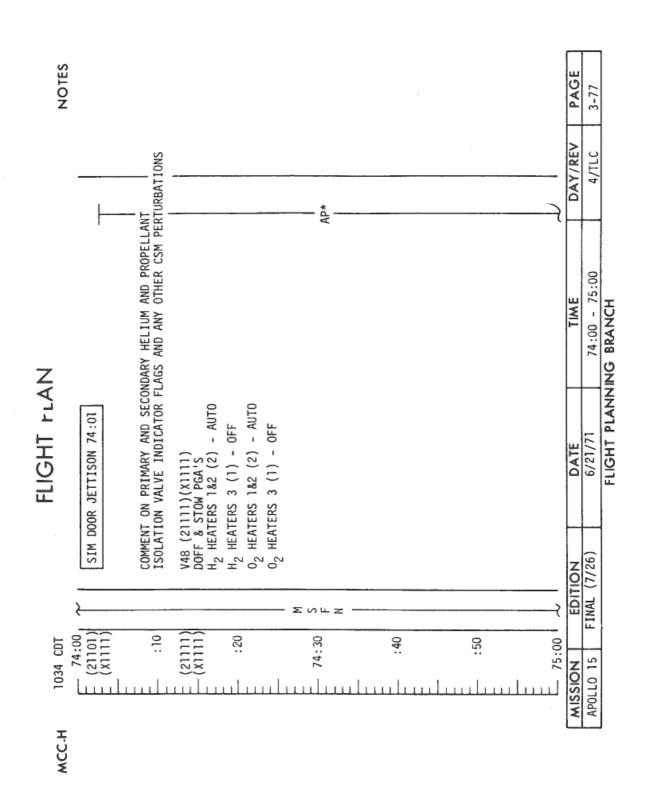
- PANORAMIC CAMERA PWR ON (FOR 2 MIN ON MSFN CUE) PANORAMIC CAMERA PWR - BOOST
- FC REACS VALVES LATCH S-BD AUX TV - SCI
- 4. SIM DOOR JETTISON SM SECTOR 1 SM/AC PWR - OFF cb LOGIC POWER (2) - CLOSE LOGIC POWER (2) - JETT GO/NO-GO FOR SIM DOOR JETTISON (CUE MSFN) DOOR JETT - DOOR JETT 74:01 DAC-ON (FOR 20 SEC) LEXAN SHIELD IN CMS OBSERVE AND PHOTOGRAPH SIM DOOR THROUGH CM5 WINDOW DOOR JETT - OFF LOGIC POWER (2) - OFF (CTR) SM SECTOR 1 SM/AC PWR - ON FC REACS VALVES - NORM SM RCS PRPLNT TB(8)-GRAY (VERIFY) SM RCS He TB(8)-GRAY (VERIFY) SM RCS SEC PRPLNT FUEL PRESS (4) - CLOSE MAPPING CAMERA ON - STBY X-RAY - STBY α RAY/X DR - α ON MASS SPECT EXP - STBY

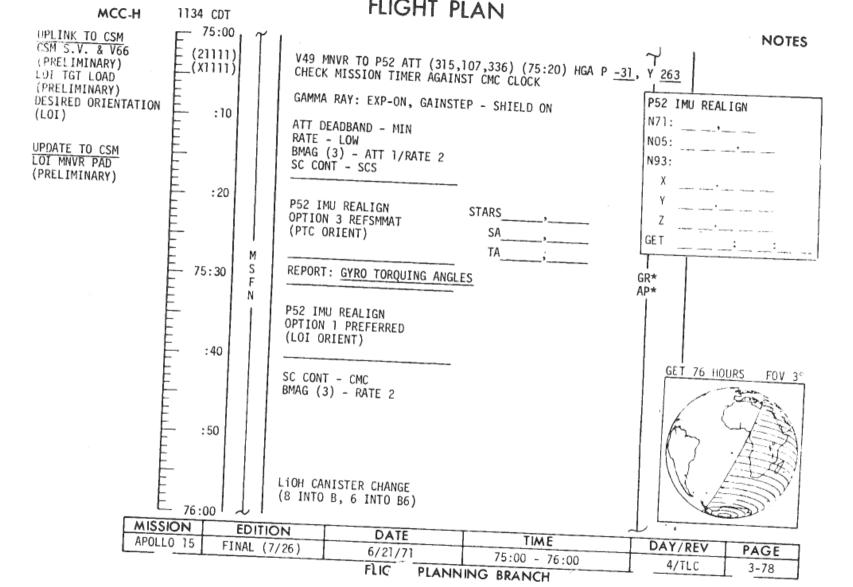
FINAL (7/26)

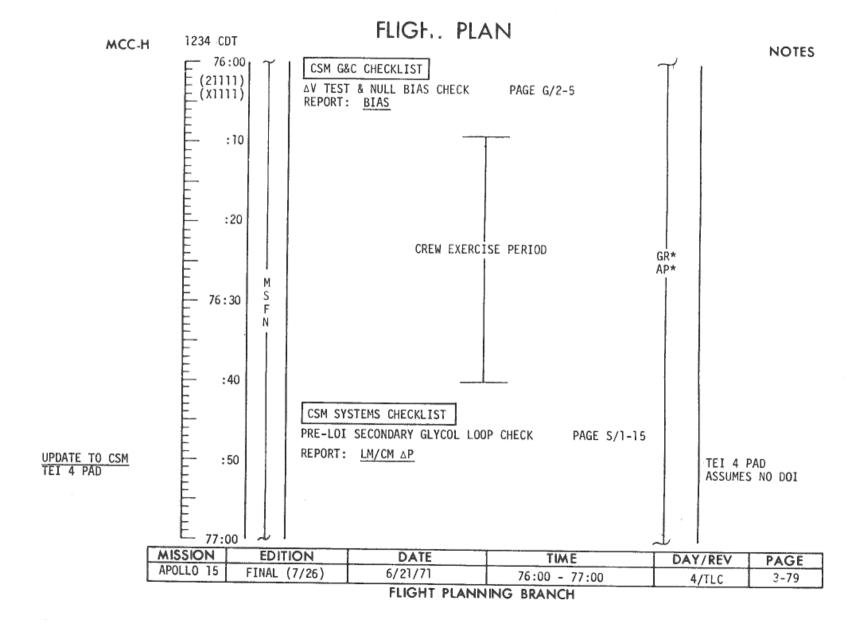
6/21/71

4/TLC

3-76







FLIGHT PLAN 1334 CDT MCC-H NOTES r 77:00 MAP UPDATE REV 1 (21111) <code>ᡛ(x1111)</code> LOS: 180°: CSM SYSTEMS CHECKLIST AOS WITH LOI: PAGE S/1-17 :10 C&W SYSTEM OPERATION CHECK SPS MONITORING CHECK PAGE S/1-1 AOS WITHOUT LOI: SM RCS MONITORING CHECK PAGE S/1-1 CM RCS MONITORING CHECK PAGE S/1-1 THE PU VALVE SHOULD ECS MONITORING CHECK PAGE S/1-5 BE USED TO MAINTAIN GR* OXIDIZER FLOW VALVE INCR - NORM (VERIFY) THE INDICATED AP* :20 UNBALANCE TO WITHIN +50 LBS OF THE STABILIZED READING S-BD AUX TV - OFF (TIG +25 SEC) UNTIL DATA SYS ON - OFF CROSSOVER. AFTER GAMMA RAY EXP - OFF CROSSOVER THE VALVE S X-RAY - OFF UPDATE TO CSM 77:30 SHOULD BE USED TO LOI MNVR PAD α RAY/X DR - α OFF CONTROL THE MASS SPECT EXP - OFF MAP UPDATE REV 1 THE GREEN BAND (0 + 100 LBS).UPLINK TO CSM (21101) CYCLE CMC MODE - FREE/AUTO THE APPROXIMATE TIME CSM S.V. & V66 (X1111) V48 (21101)(X1111) OF CROSSOVER IS LOI TGT LOAD :40 P30 EXTERNAL AV 04:22 TO 04:29 MINUTES INTO THE LOI BURN. V49 MNVR TO PAD BURN ATTITUDE (78:00) (000,000,000) OMNI C :50 78:00 MISSION EDITION DATE TIME DAY/REV PAGE

FLIGHT ANNING BRANCH

77:00 - 78:00

4/TLC

3+80

6/21/71

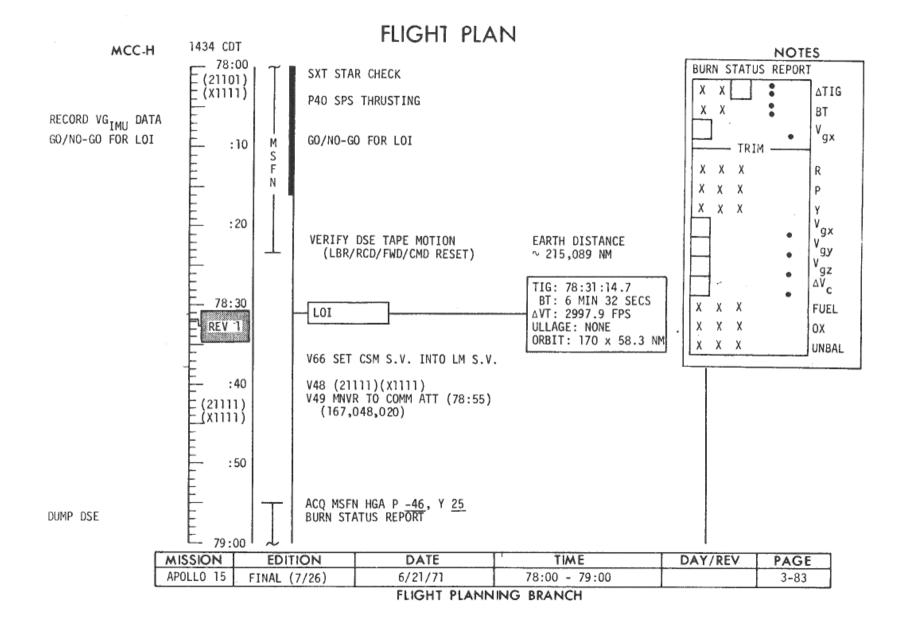
APOLLO 15

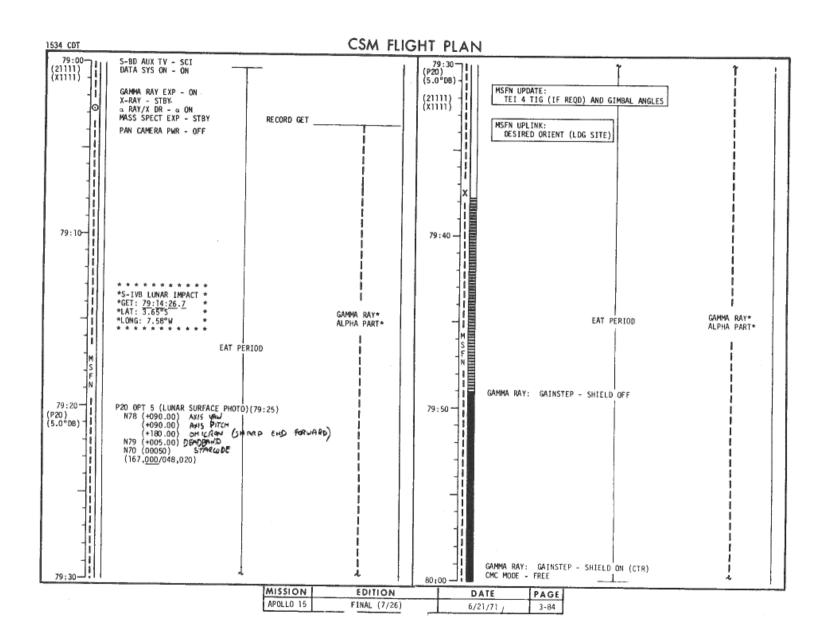
FINAL (7/26)

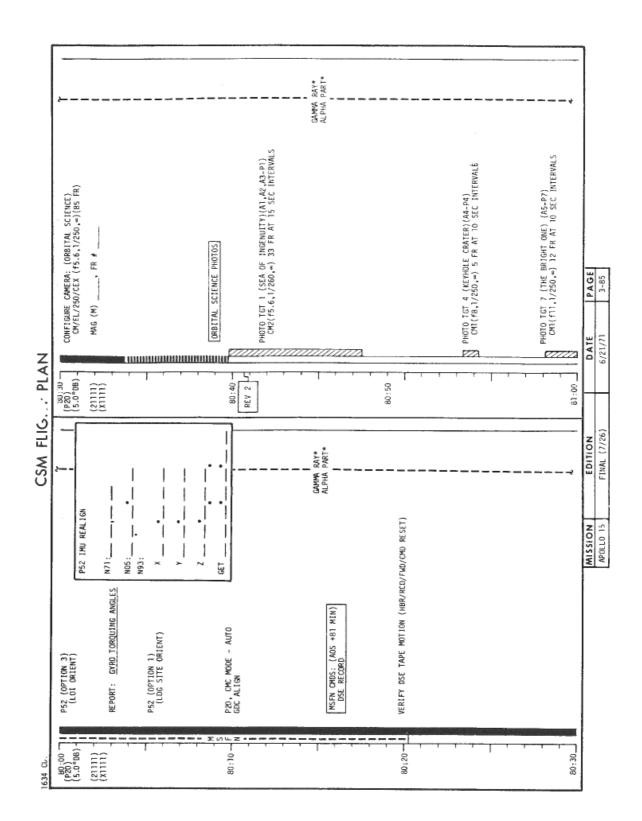
Prellating (74:51:56) P30 MANEUVE. LOI VEGA DENEB PURPOSE Set stars 5P5/G+N PROP/GUID SET STARS VERA DENEB 66244 +1.21 RALIGN 2 6 4 GDC ALIGN PALIGN 0 9 0 -012 R 264 YALIGN 3 4 9 78 GETI P 90 31 / MIN N33 Y 349 NO ULLAGE 34.48 N81 2894.5 766.4 ULLAGE NONE 112.3 0 0 169.5 58.3 2996.6 6:40 2990.2 25 SXTS 267.1 O TRN 22.8 x x BSS SPA 3-81

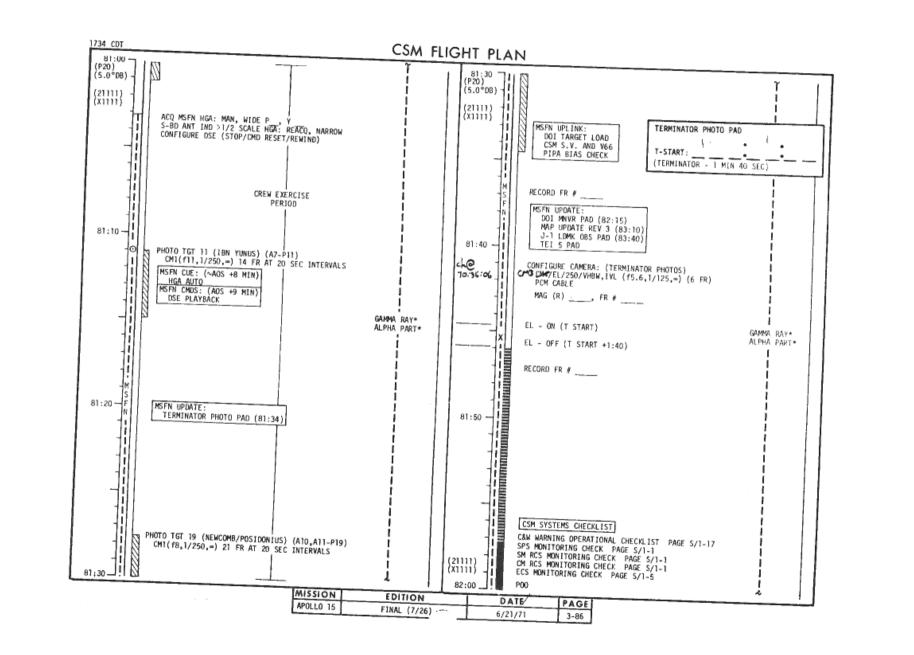
LOI BURN TABLE

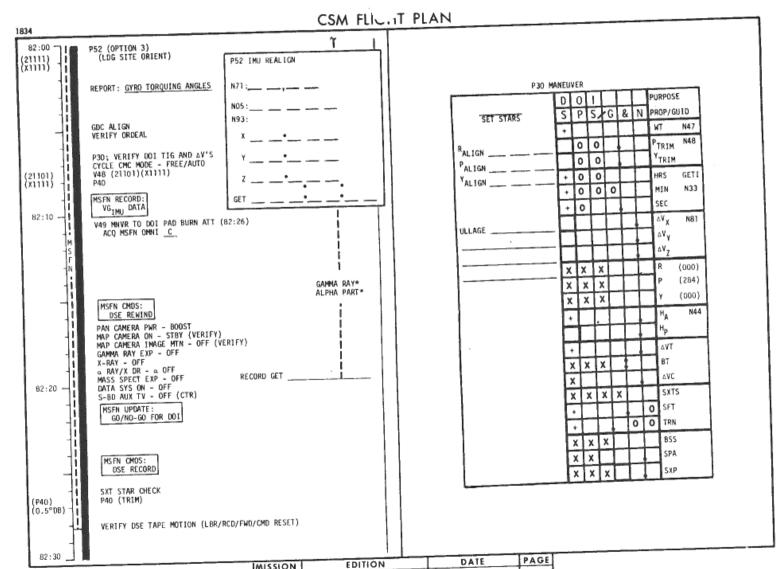
P OR Y	ATT	SHUTDOWN	RESIDUALS
RATES	DEVIATION	TIME	
10°/SEC TAKEOVER & COMPLETE	±10° TAKEOVER & COMPLETE	BT + 10 SEC	DO NOT TRIM



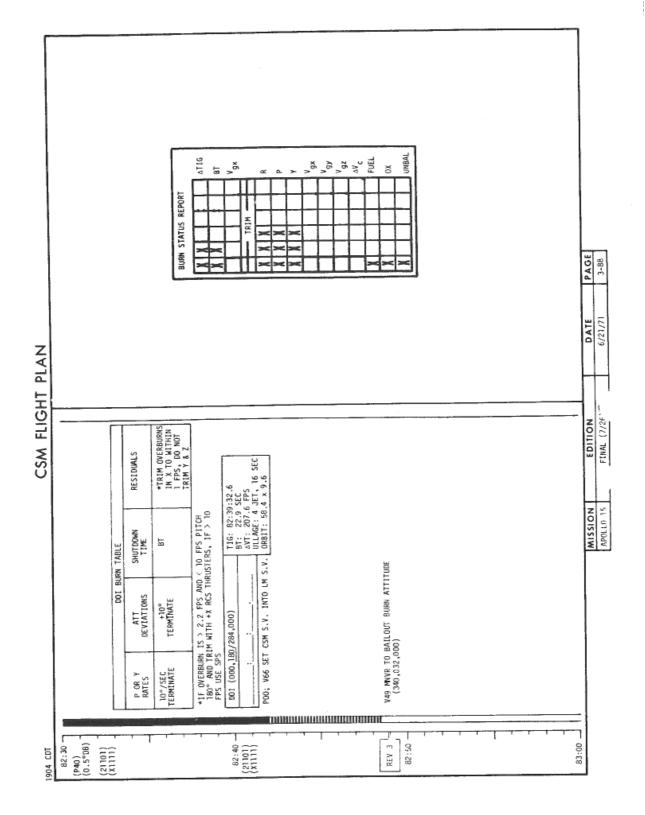


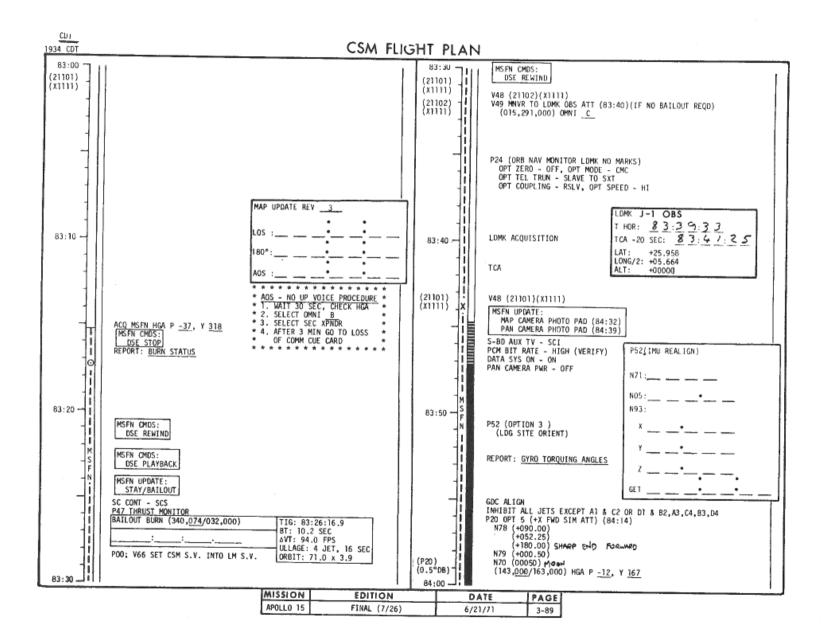


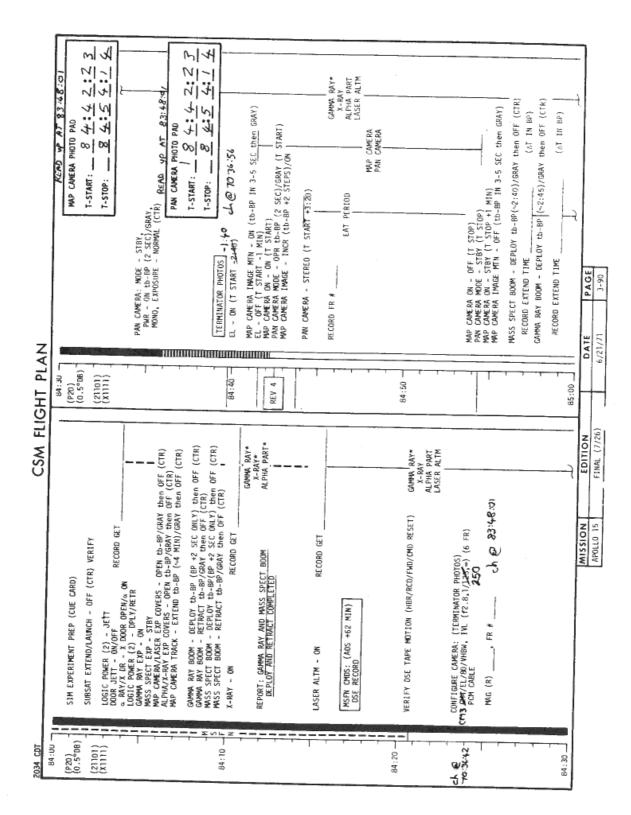


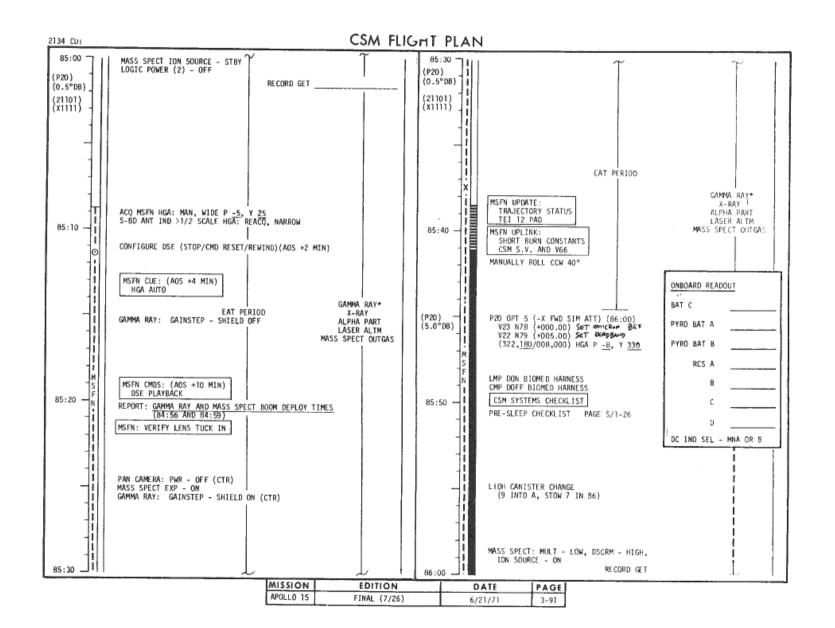


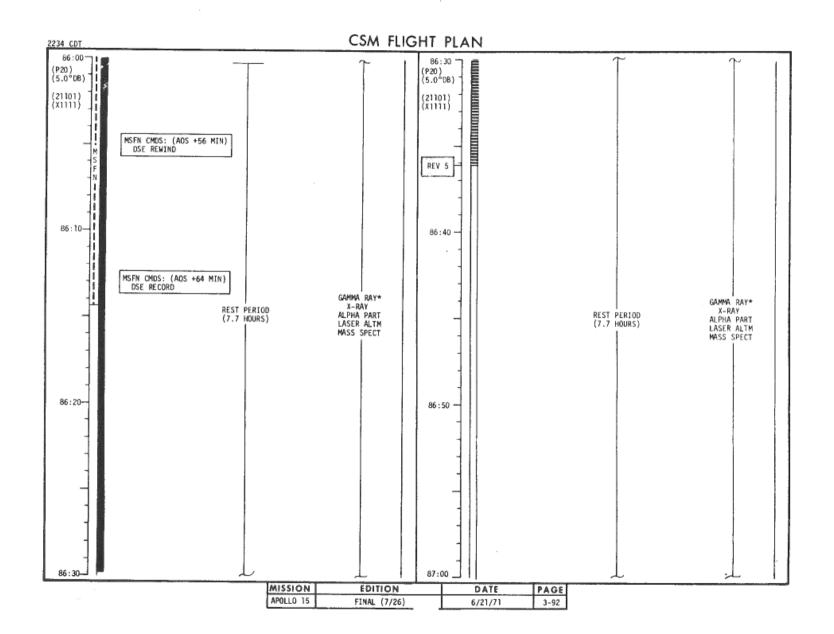
IMISSION	EDITION	DATE	PAGE
	FINAL (7/26)	6/21/74	3-87
APOLLO 15	THING (TYPES)		

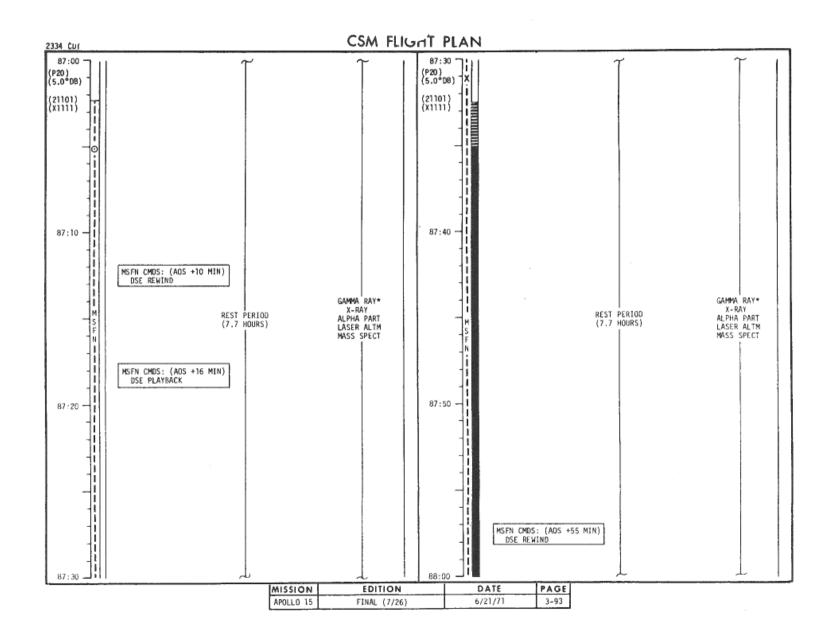


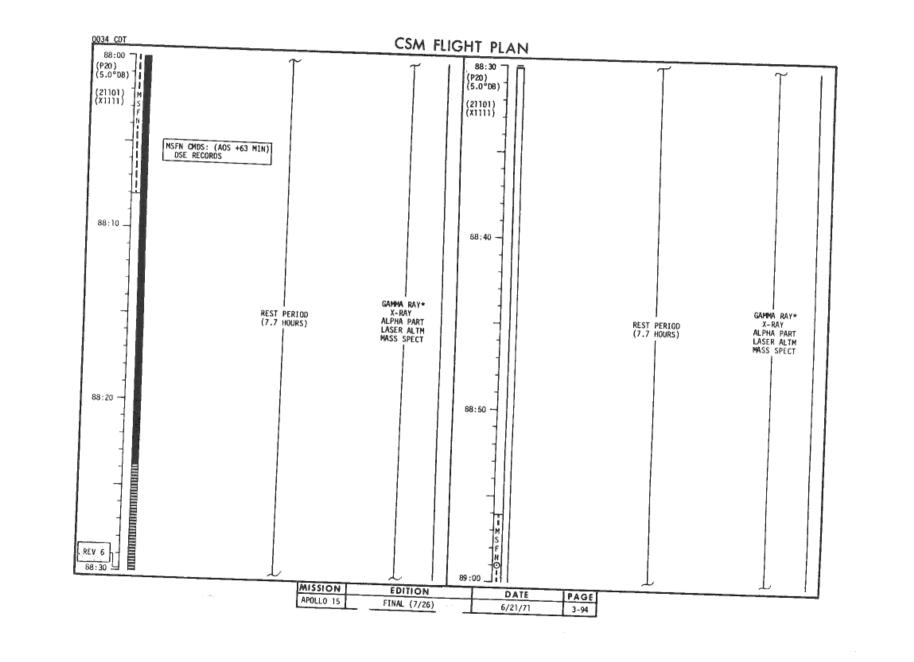


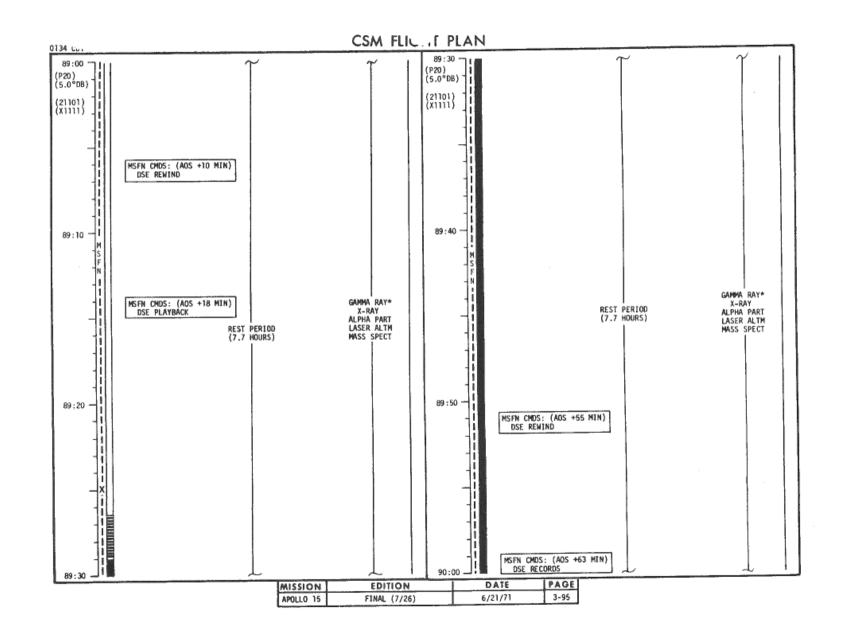


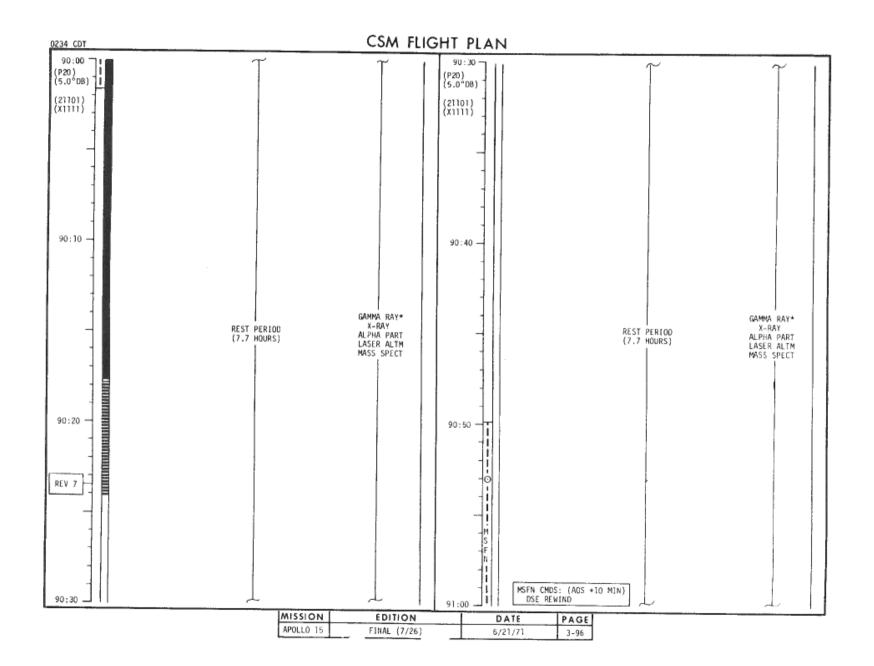


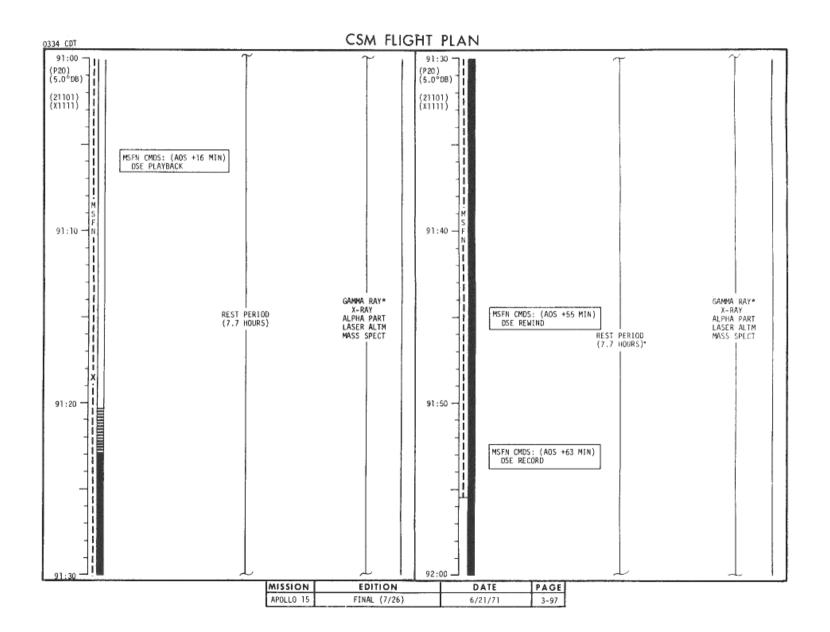


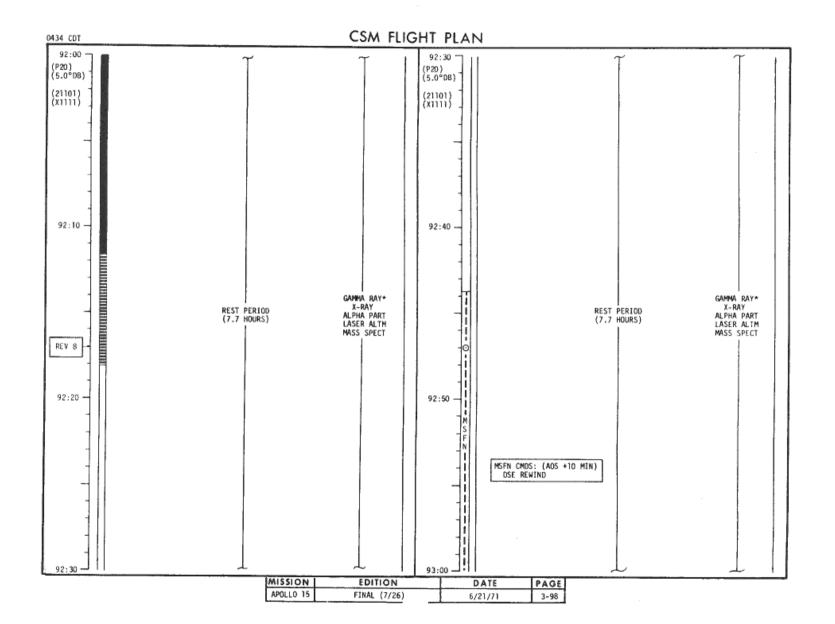


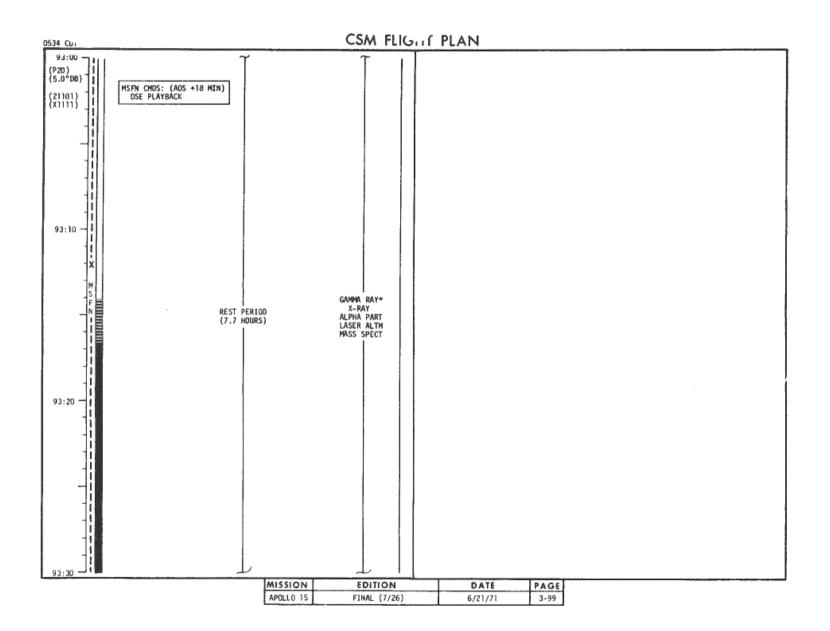


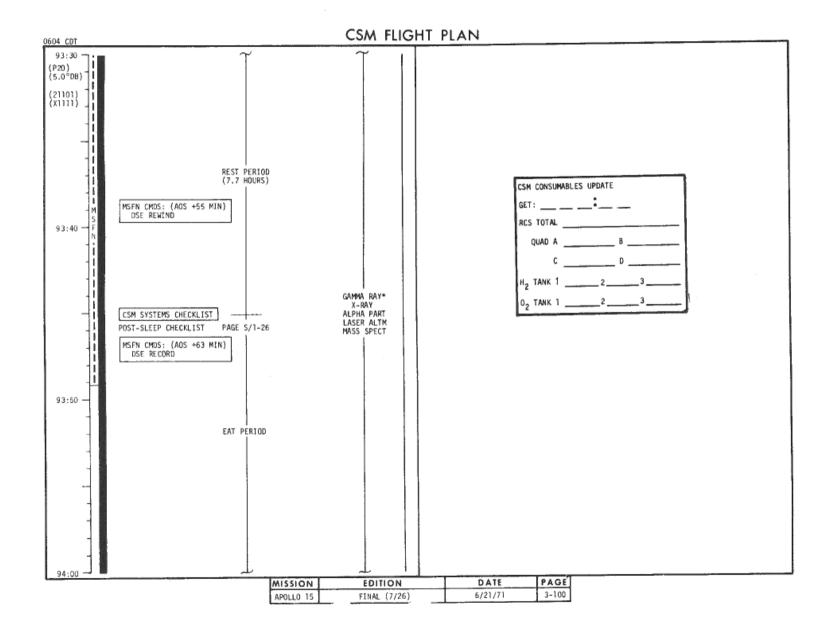


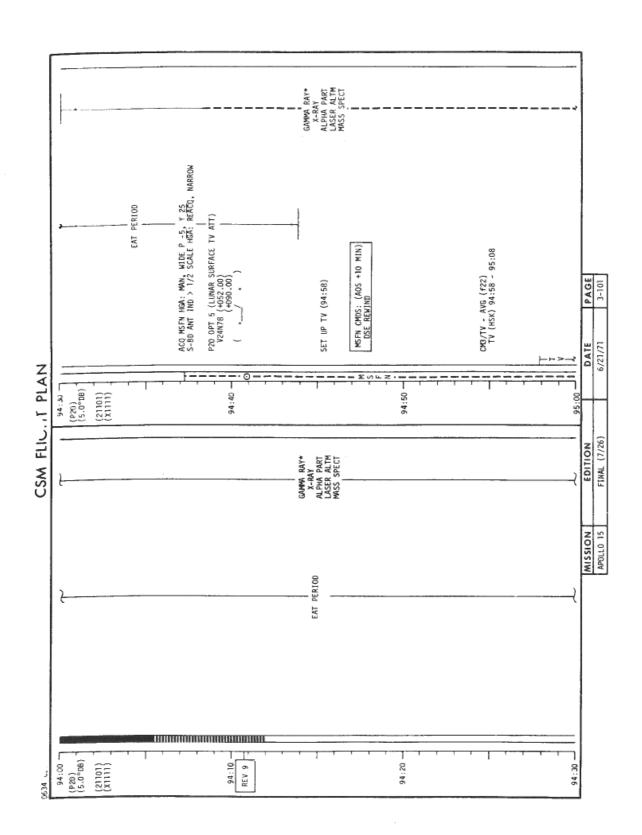


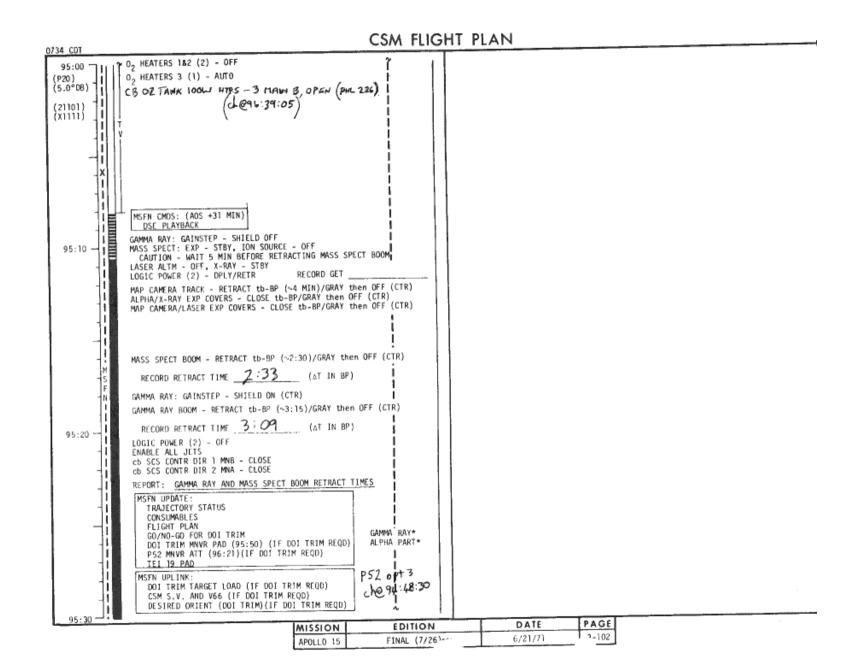


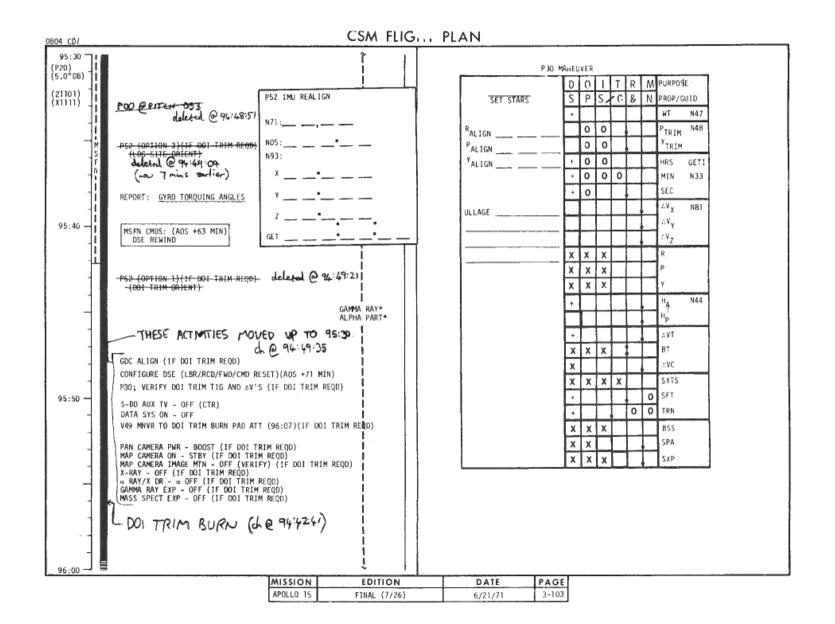


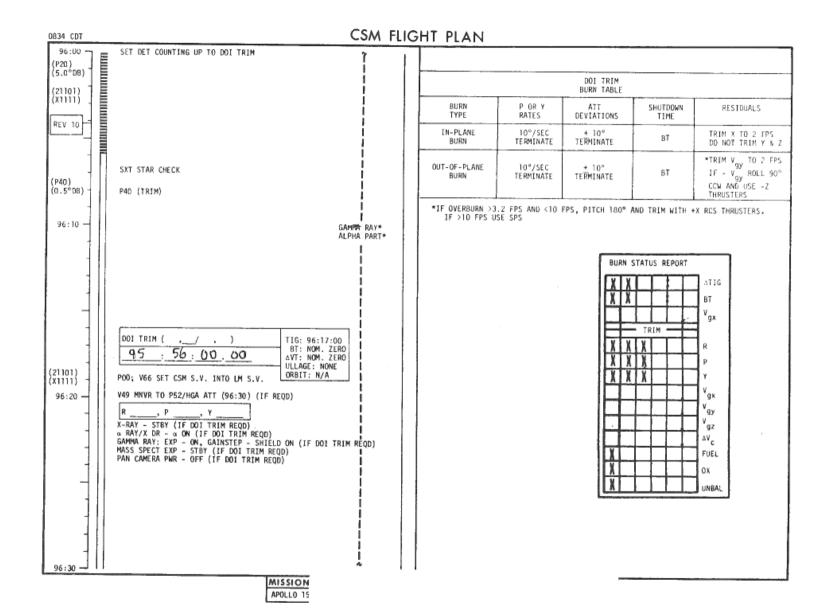


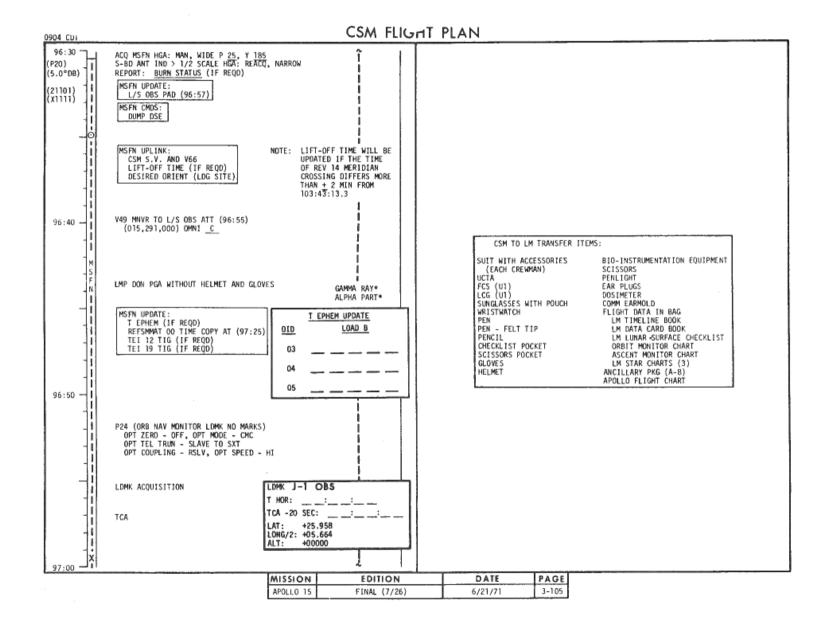


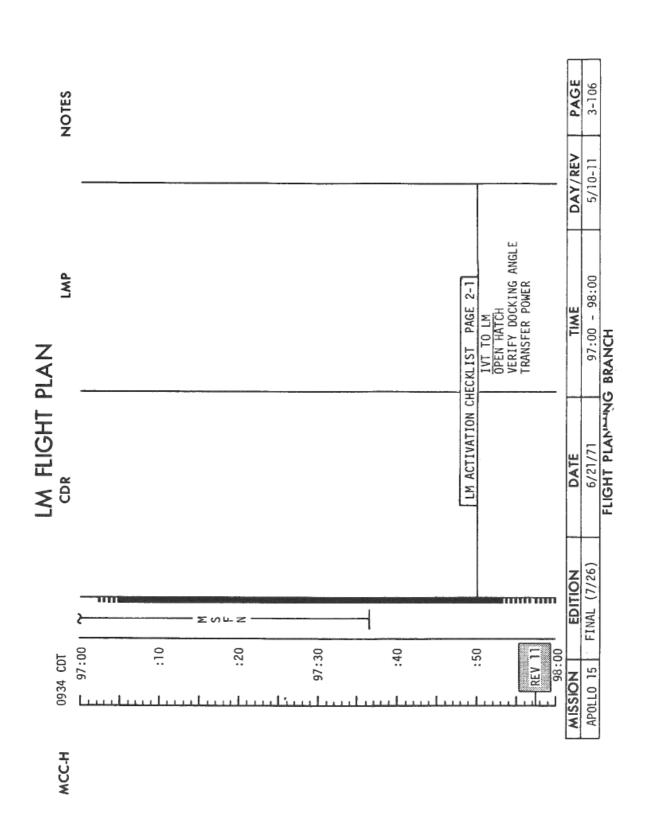


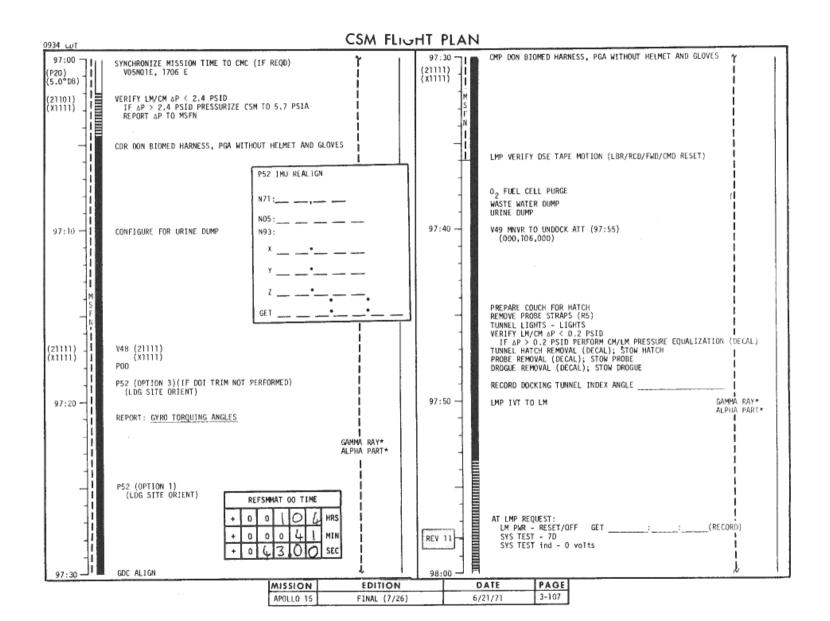




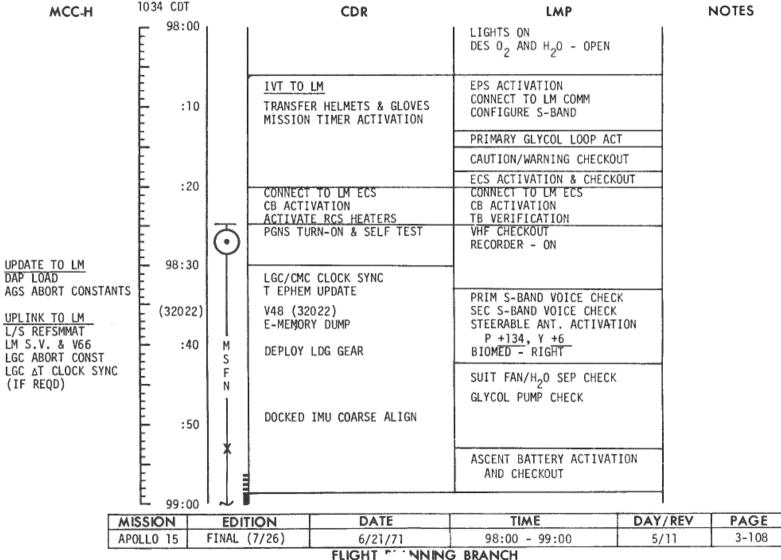


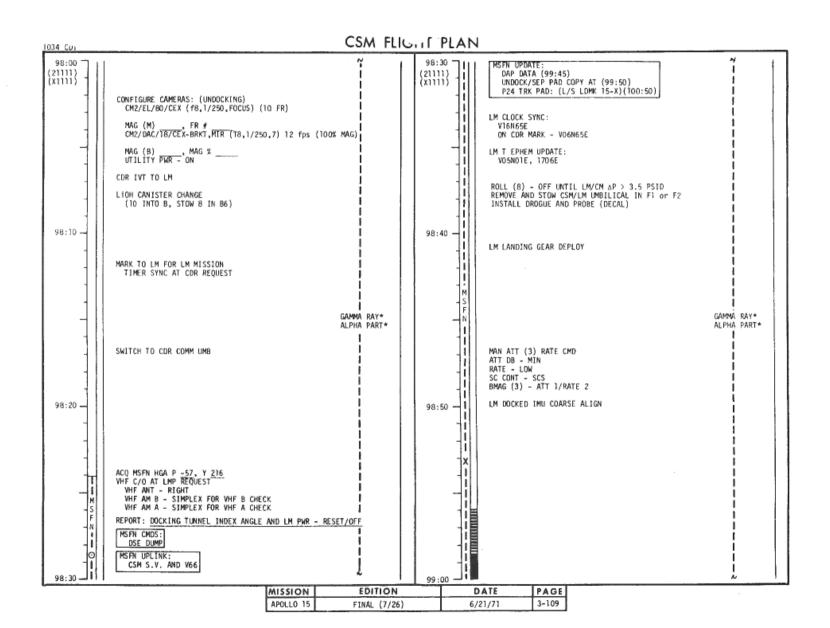


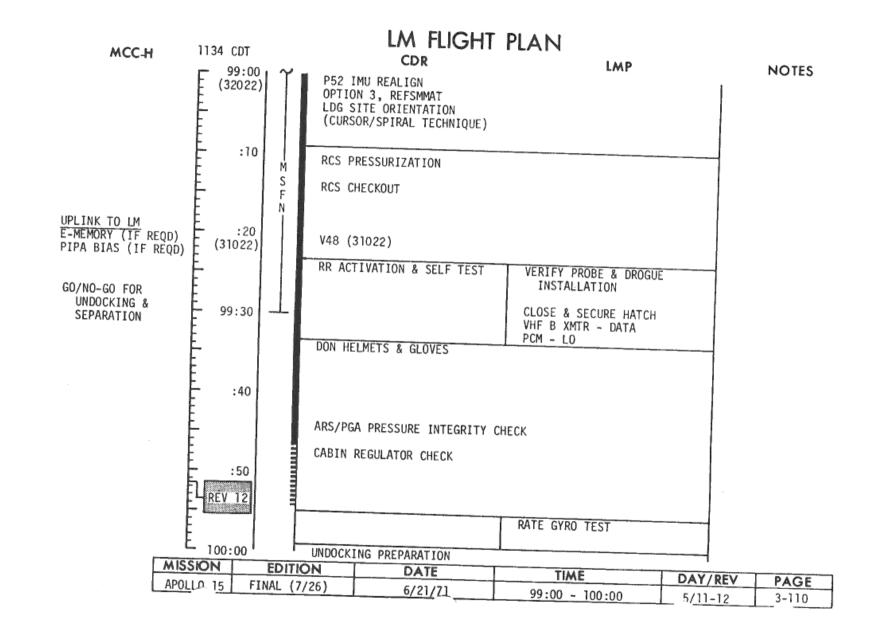


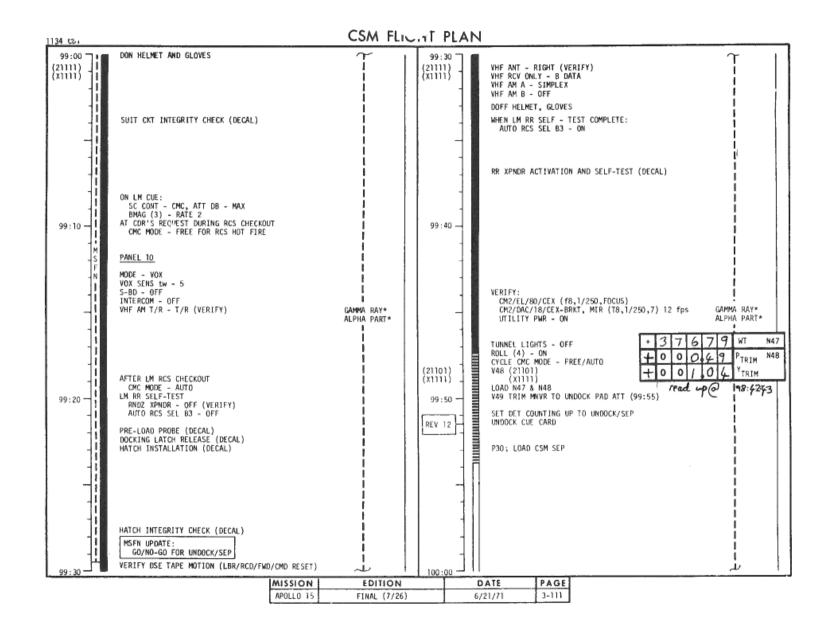


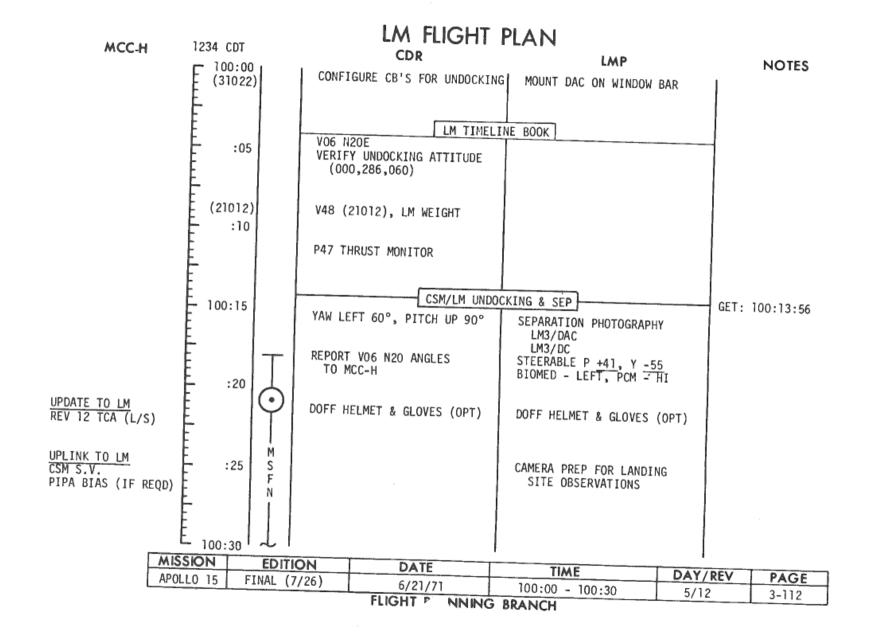
LM FLIGHT PLAN

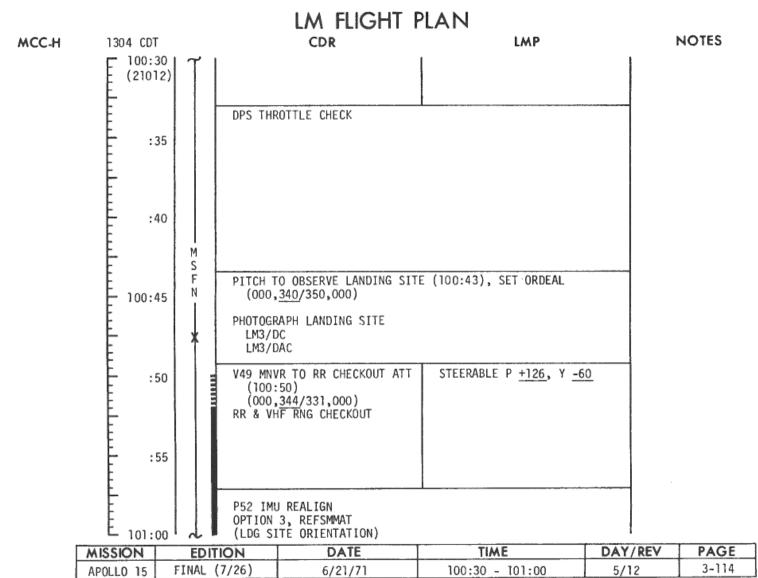




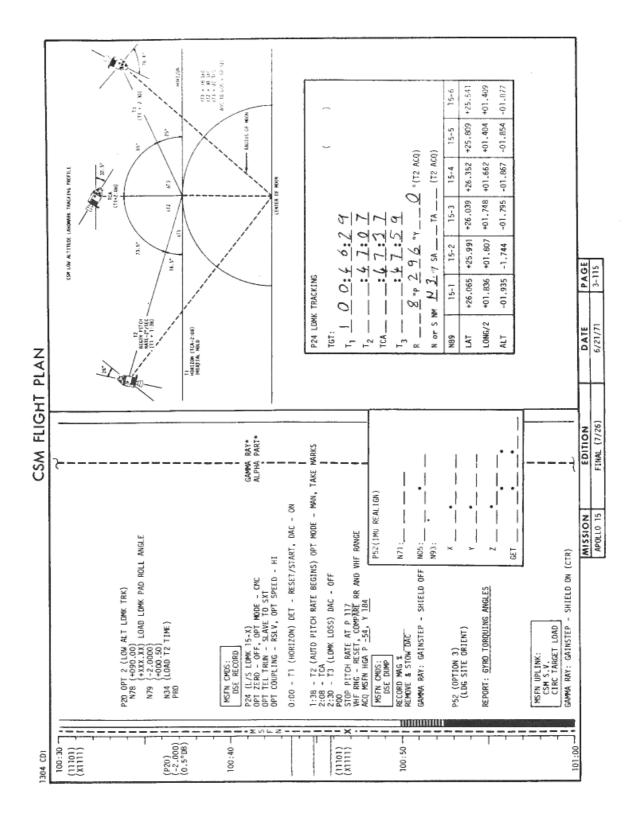


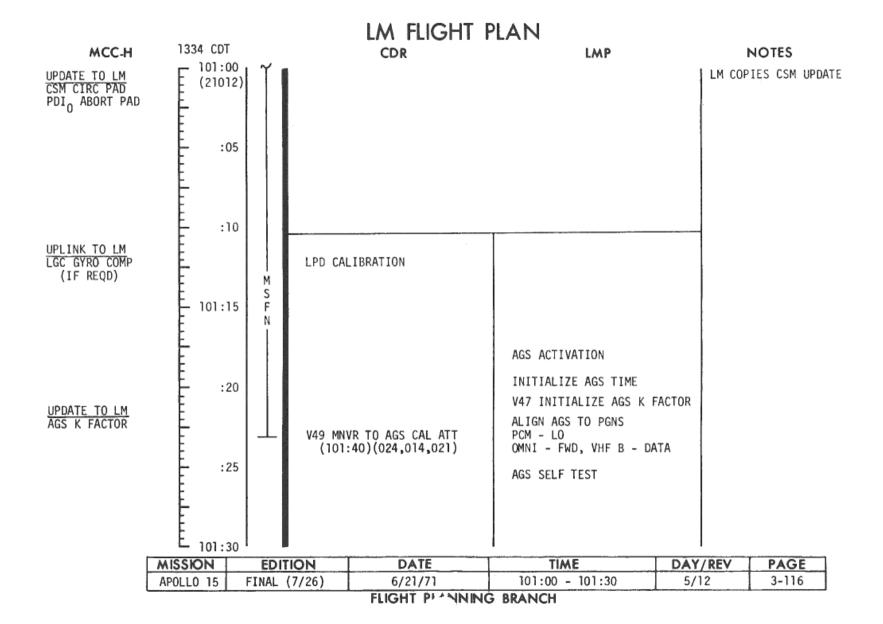




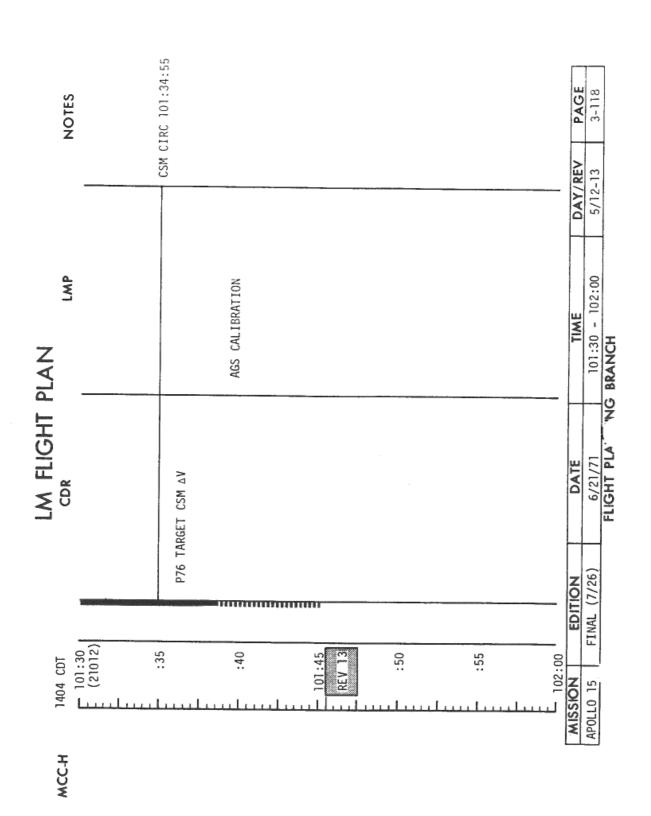


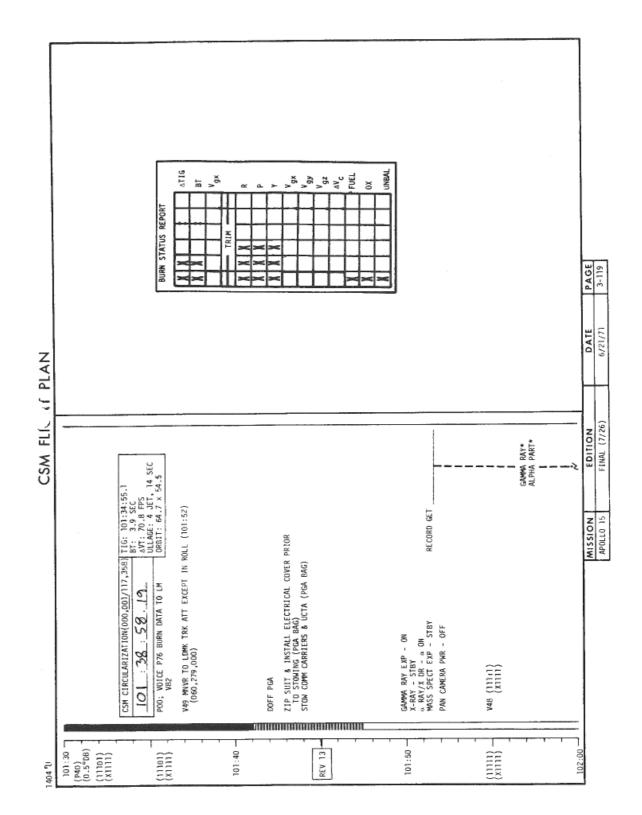
FLIGHT PLANNING BRANCH

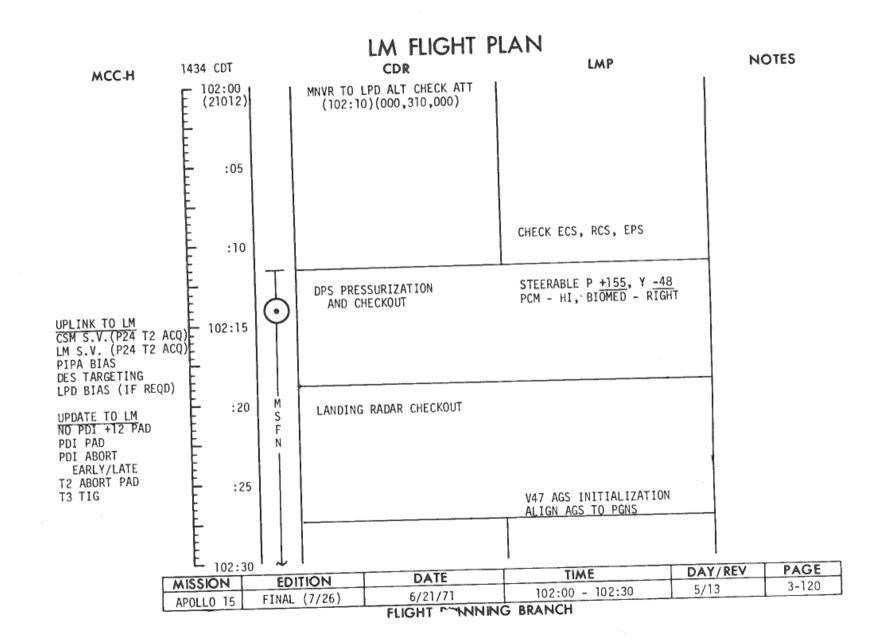


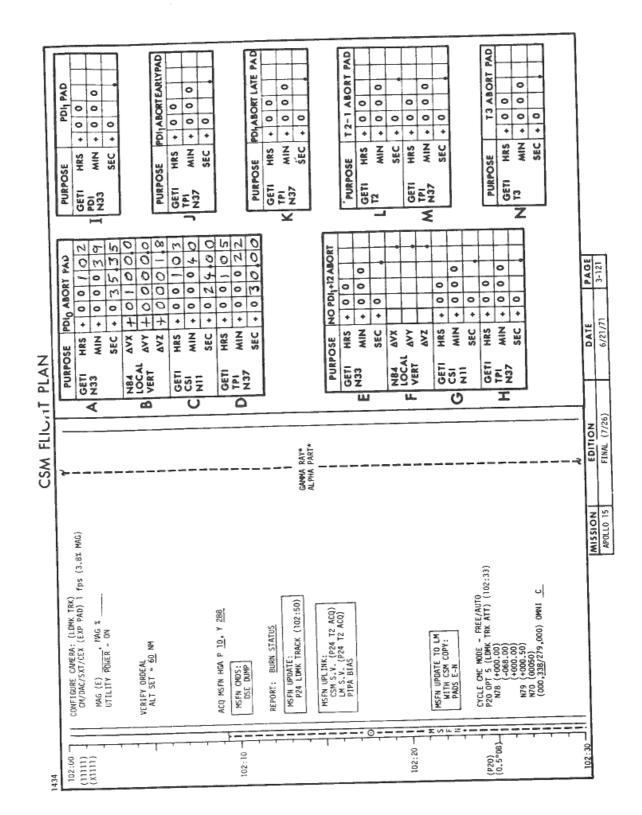


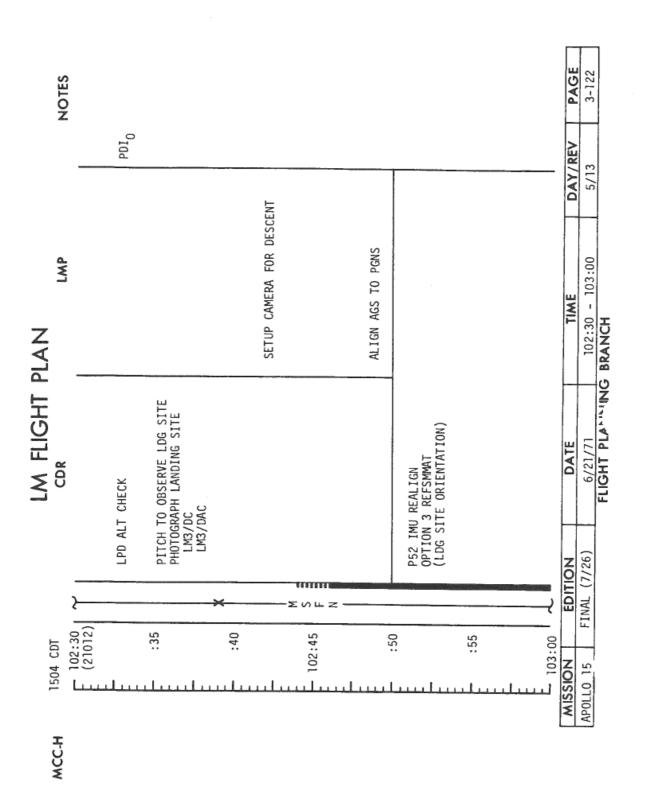
CSM FLIGHT PLAN 1334 Cb. 101:00 -MSFN UPDATE: CIRC PAD PADS A-D COPY AT (102:05) P30 MANEUVER (11101) (xiiiii) CIRC PURPOSE SET STARS S P S G & N PROP/GUID P30; VERIFY CIRC TIG AND AV'S V49 TRIM TO CIRC BURN PAD ATT R_{ALIGN} 0 0 PTRIM 0 0 P_{ALIGN} YTRIM SXT STAR CHECK GAMMA RAY* 0 0 Y ALIGN HRS ALPHA PART* GETI (P40) (0.5°DB) P40 (TRIM) 0 0 MIN **K33** 0 SEC ΔVχ N81 ULLAGE GDC ALIGN VERIFY ORDEAL 101:10 ALT SET = 50 NM ΔV, GAMMA RAY EXP - OFF X-RAY - OFF a RAY/X DR - a OFF (000) (117) MASS SPECT EXP - OFF RECORD GET x x x (358)PAN CAMERA PWR - BOOST MAP CAMERA ON - STBY (VERIFY)
MAP CAMERA IMAGE MTN - OFF (VERIFY) N44 ΔVT MSFN UPDATE: GO/NO GO FOR CIRC ΔVC SET DET COUNTING UP TO CIRC XXXX sxts O SFT 101:20 VHF AM B - OFF VHF AM A - SIMPLEX O O TRN x x x VHF RCV ONLY - B DATA 855 VHF RANGING - OFF SPA VERIFY DSE TAPE MOTION (LBR/RCD/FWD/CMD RESET) SXP CIRC BURN TABLE P OR Y ATT SHUTDOWN RESIDUALS RATES DEVIATION TIME 10°/SEC +10° BT +1 SEC TRIM X&Y TO < 0.2 FPS DO NOT TRIM Z TERMINATE TERMINATE 101:30 MISSION EDITION DATE PAGE APOLLO 15 FINAL (7/26) 6/21/71 3-117

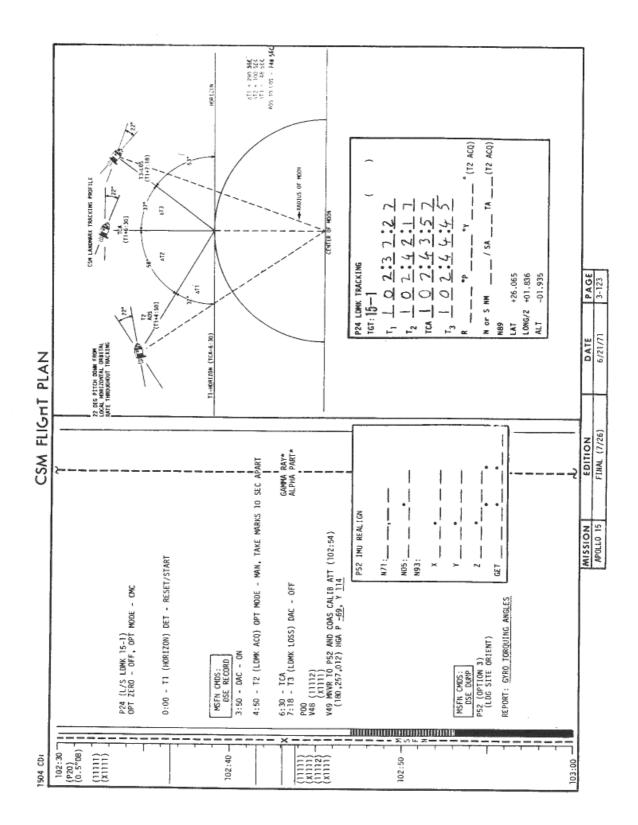


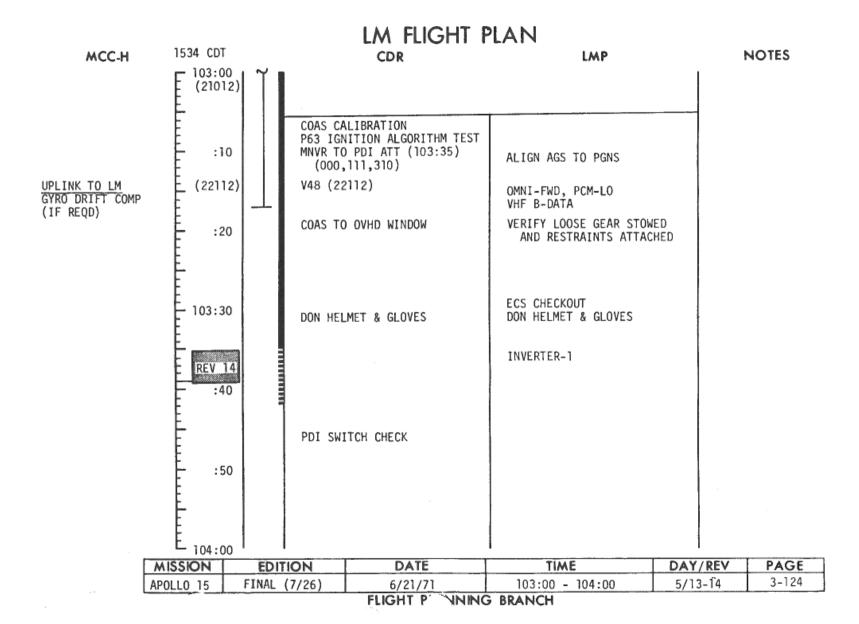


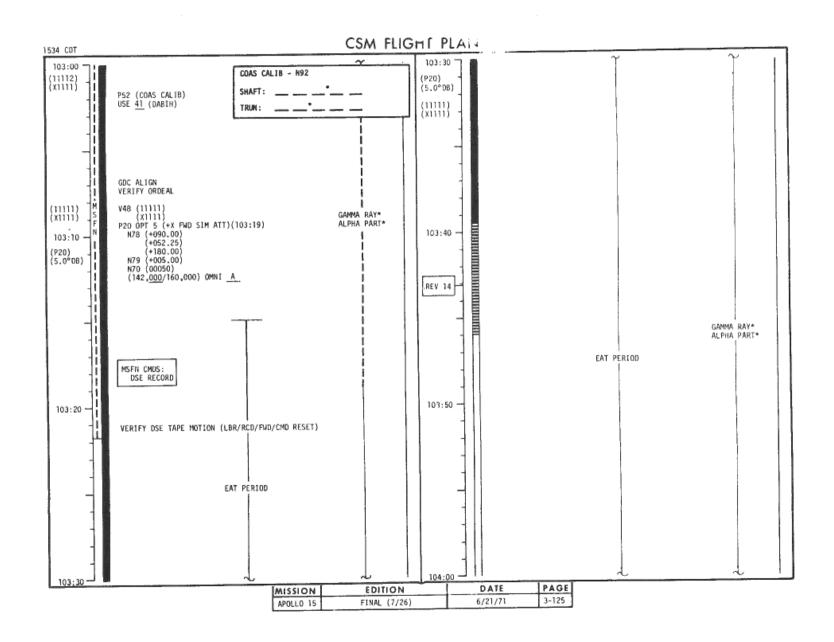


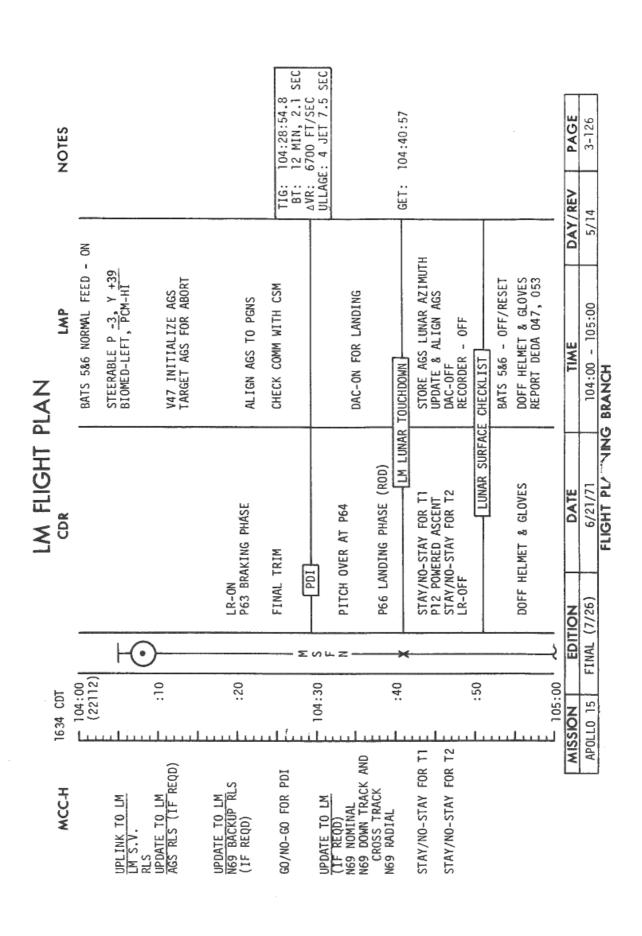


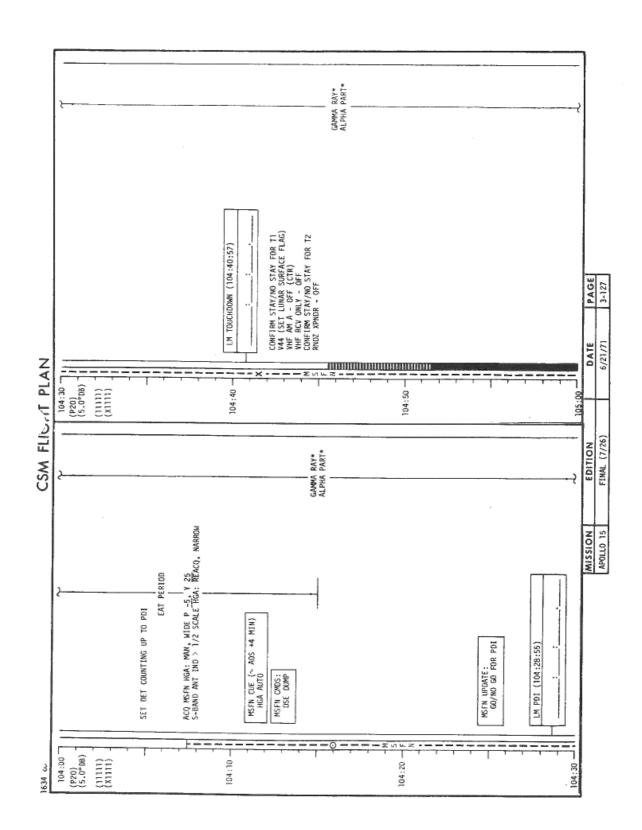










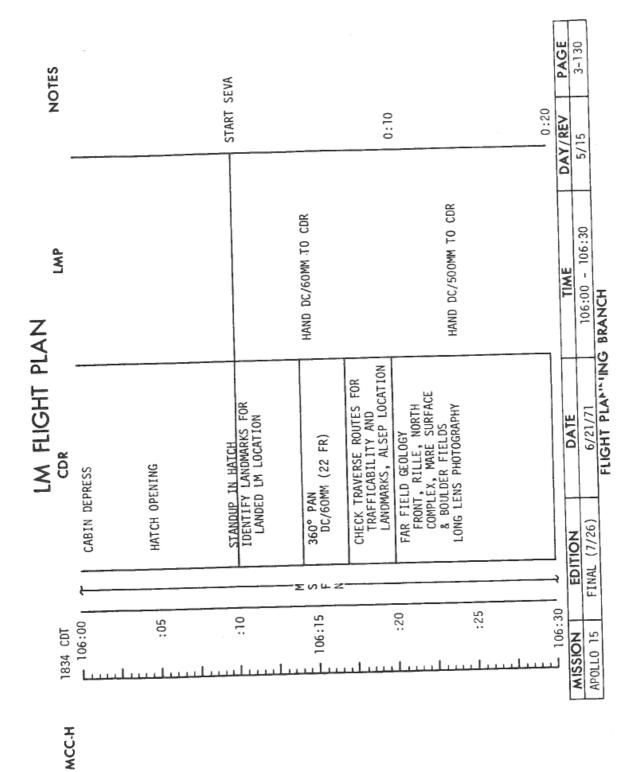


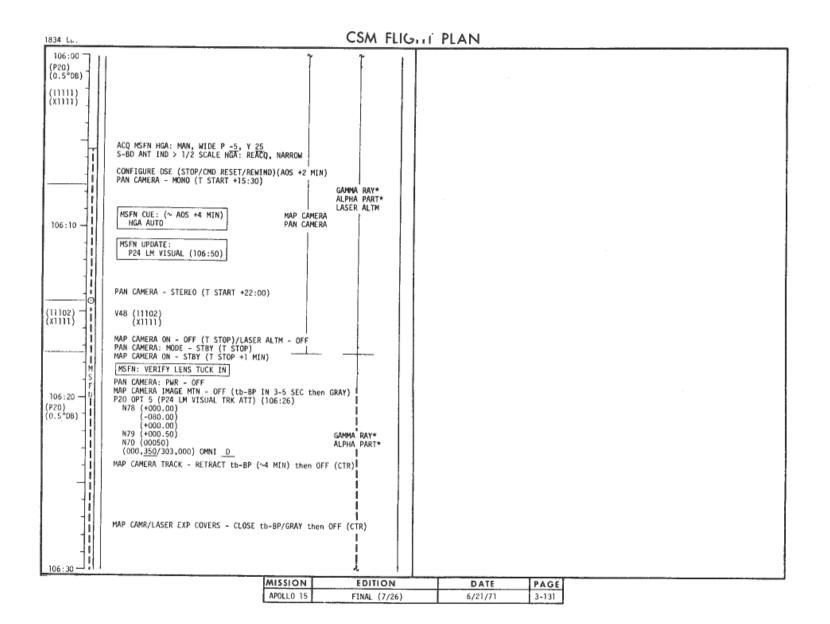
LM FLIGHT PLAN MCC-H 1734 CDT CDR LMP NOTES 105:00 (22112)P57 LUNAR SURFACE ALIGN AGS LUNAR SURFACE GYRO OPTION 3 REFSMMAT CALIBRATION A/T-1 GRAVITY & LM Z-AXIS (LANDING SITE ORIENT) STEERABLE P +73, Y -62 INSTALL WINDOW SHADES :10 PARTIAL POWER DOWN P57 LUNAR SURFACE ALIGN OPTION 3, REFSMMAT A/T-2, TWO CELESTIAL BODIES :20 (LANDING SITE ORIENT) P57 LUNAR SURFACE ALIGN OPTION 3, REFSMMAT A/T-2, TWO CELESTIAL BODIES (LANDING SITE ORIENT) STAY/NO-STAY FOR T3 105:30 E-MEMORY DUMP, POO AGS/PGNS ALIGN AGS STANDBY CONFIGURE TO POWER DOWN BATS 3&4 - OFF/RESET UPDATE TO LM LUNAR BAT (CDR) - ON. LIFT-OFF TIMES FOR CHECK BUS VOLTS :40 LGC TO STANDBY, IMU PWR DN REVS 16 THRU 20 CSM REV 15 SEVA PREP EQUIPMENT PREP FOR SEVA :50 HELMET/GLOVE DONNING PRESSURE INTEGRITY CHECK 106:00 LIADISSIAA

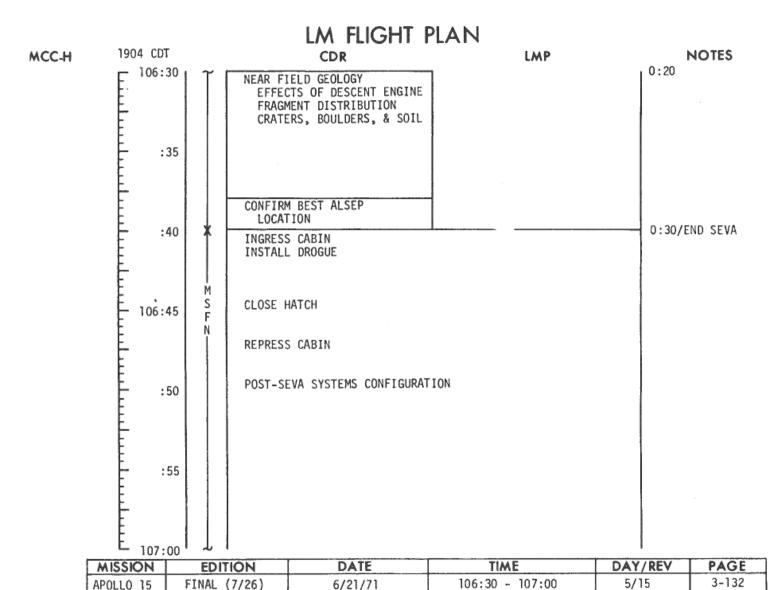
WISSIGN	EDITION	DATE	TIME	DAY/REV	2405
APOLLO 15	FINAL (7/26)			DAT/REV	PAGE
- 11 0220 13	TIMAL (7/20)	6/21/71	105:00 - 106:00	5/14-15	3-128
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FLIGHT " NNING BRANCH

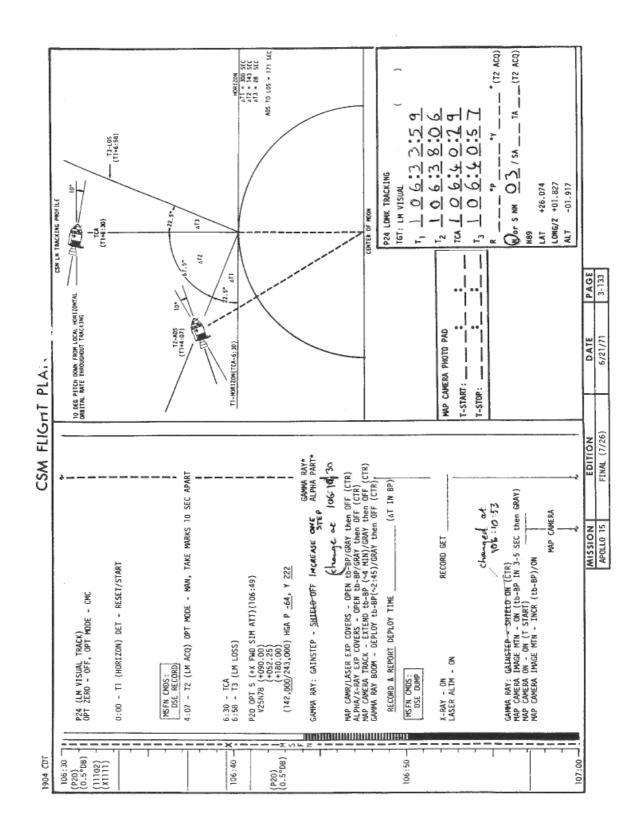
CSM FLIGHT PLAN 1734 Cu. 105:30 MAP CAMERA PHOTO PAD CMC MODE - FREE 105:00 71 P52 (OPTION 3) (LDG SITE ORIENT) (P20) (P20) (5.0°DB) T-START: (5.0°D8) (11111) (X1111) T-STOP: ____ REPORT: GYRO TORQUING ANGLES (11111)(x1111) PAN CAMERA PHOTO PAD S-BD AUX TV - SCI T-START: ____ P20, OMC MODE - AUTO DATA SYS ON - ON GDC ALIGN T-STOP: ____ VERIFY ORDEAL CMC MODE - FREE (P20) V22N79 (+000.50) CMC MODE - AUTO (0.5°DB) MSFN UPDATE: MAP CAMERA PHOTO PADS (105:30 AND 106:50) PAN CAMERA PHOTO PAD (105:35) SIM EXPERIMENT PREP (CUE CARD) INHIBIT ALL JETS EXCEPT AT & C2 OR DT & B2, A3,C4.B3,D4 P52 IMU REALIGN 105:40 -LOGIC POWER (2) - DPLY/RETR
MAP CAMR/LASER EXP COVERS - OPEN tb-BP/GRAY then OFF (CTR)
MAP CAMERA TRACK - EXTEND tb-BP (~4 MIN)/GRAY 105:10 MSFN UPLINK: REV 15 CSM S.V. N93: PAN CAMERA: MODE - STBY, PWR - ON tb-BP (2 SEC)/GRAY, STEREO, EXPOSURE - NORMAL (CTR)
PCM BIT RATE HIGH/CMD RESET RECORD GET LASER ALTM - ON MSFN CMDS: (AOS +69 MIN) DSE RECORD ORBITAL SCIENCE PHOTOS PHOTO TGT 2 (GAGARIN) (A3-P2) CM3 (f5.6,1/250, w) 36 FR AT 5 SEC INTERVALS VERIFY DSE TAPE MOTION (LBR/RCD/FWD/CMD RESET) 105:50 105:20 MAP CAMERA IMAGE MTN - ON (tb-8P IN 3-5 SEC then GRAY) MAP CAMERA ON - ON (T START)
PAN CAMERA MODE - OPR tb-BP (2 SEC)/GRAY (T START)
MAP CAMERA IMAGE MTN - INCR (tb-BP +4 STEPS)/CN GAMMA RAY* ALPHA PART* GAMMA RAY* ALPHA PART* MAP CAMERA LASER ALTM CONFIGURE CAMERA: (ORBITAL SCIENCE) PAN CAMERA CM/EL/250/CEX (5.6,1/250,=) (64 FR) REMOVE CM5 WINDOW COVER MAG (0) ____, FR # ____ PHOTO TGT 6 (N.W. TSIOLKOVSKY) (A4,A5-P6) CM5 (f11,1/250,∞) 28 FR AT 5 SEC INTERVALS REPLACE CM5 WINDOW COVER RECORD FR # 105:30 PAGE DATE EDITION MISSION 6/21/71 3-129 FINAL (7/26) APOLLO 15

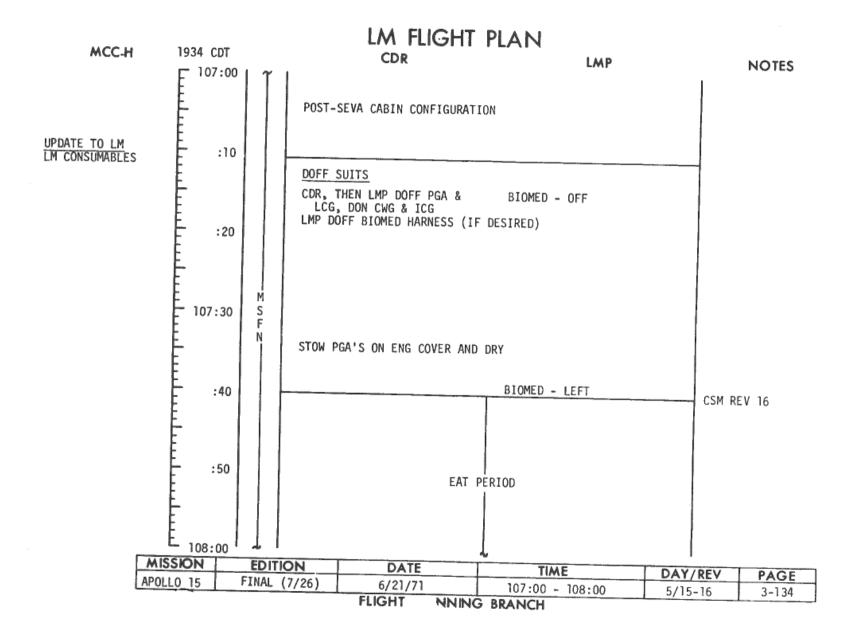


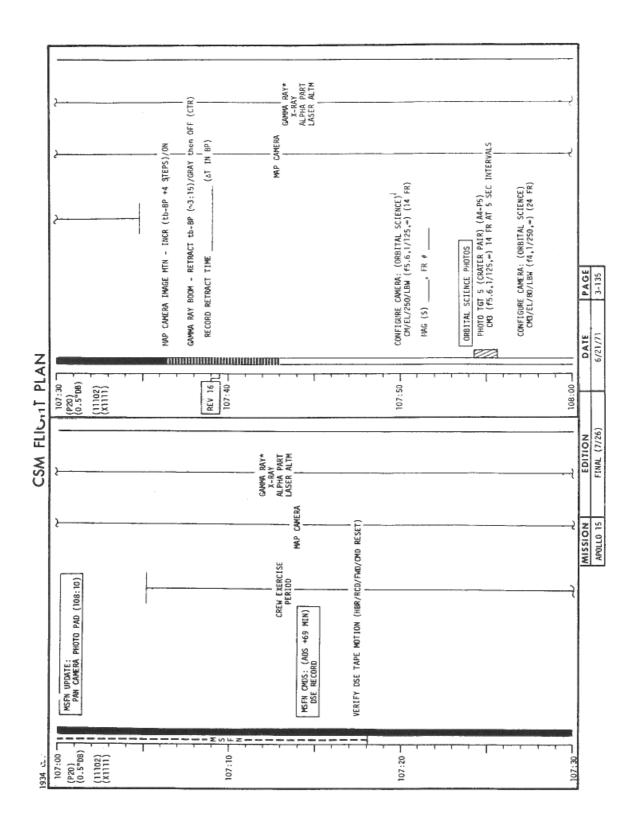


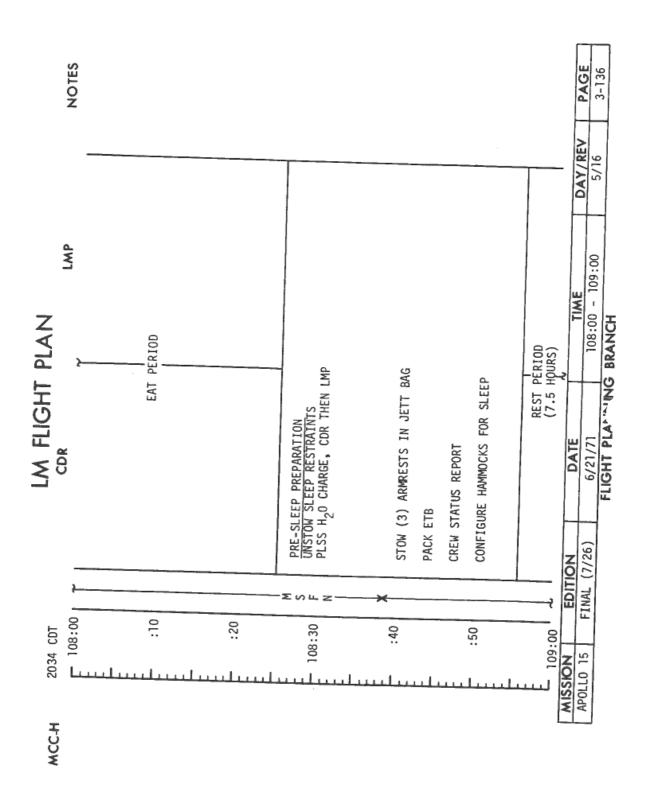


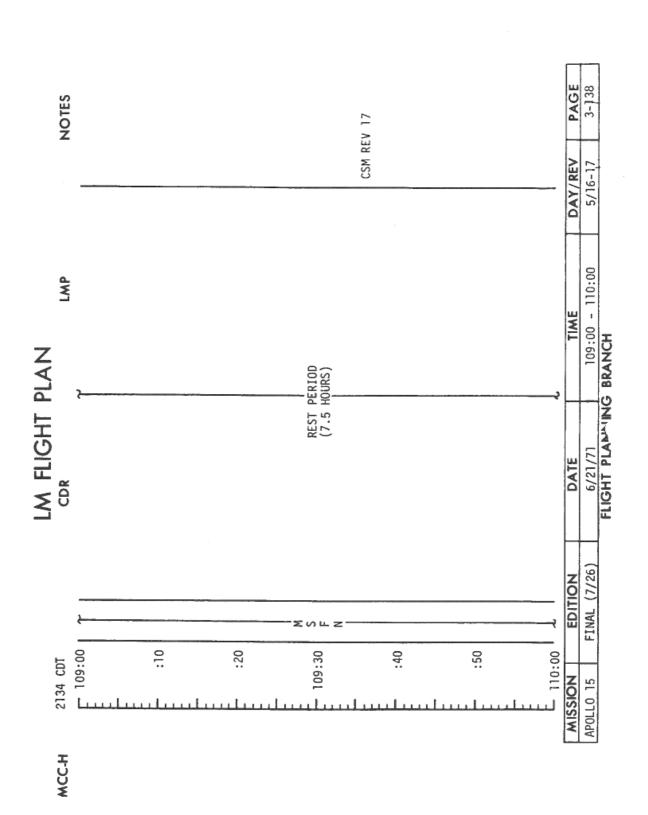
FLIGHT PLANT NG BRANCH

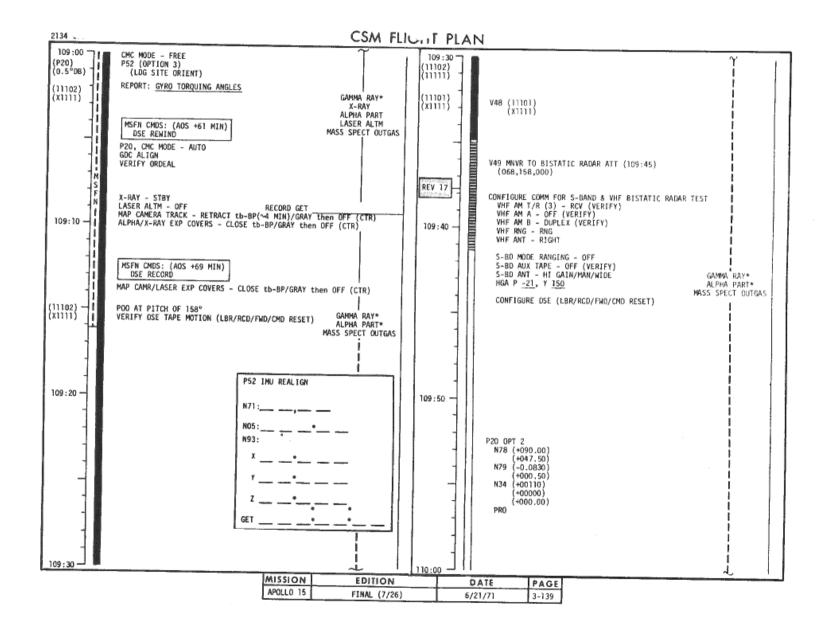


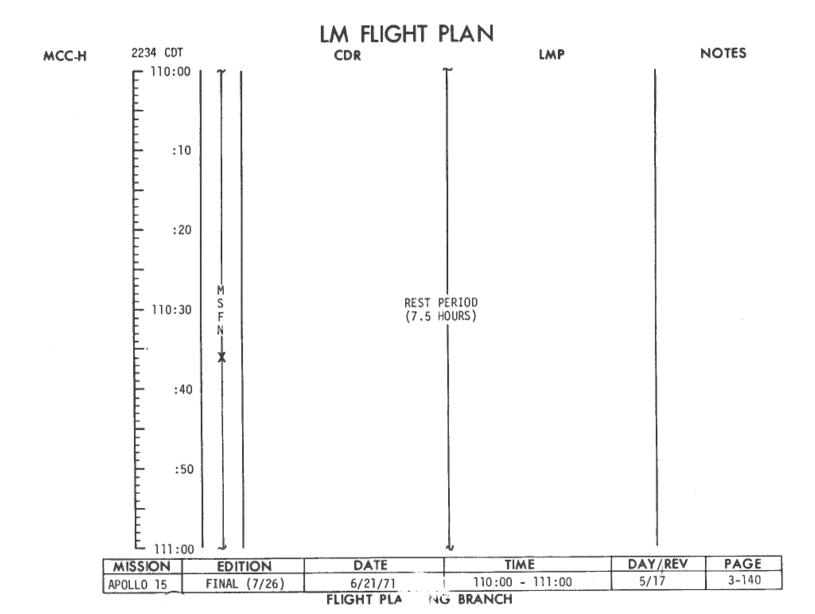


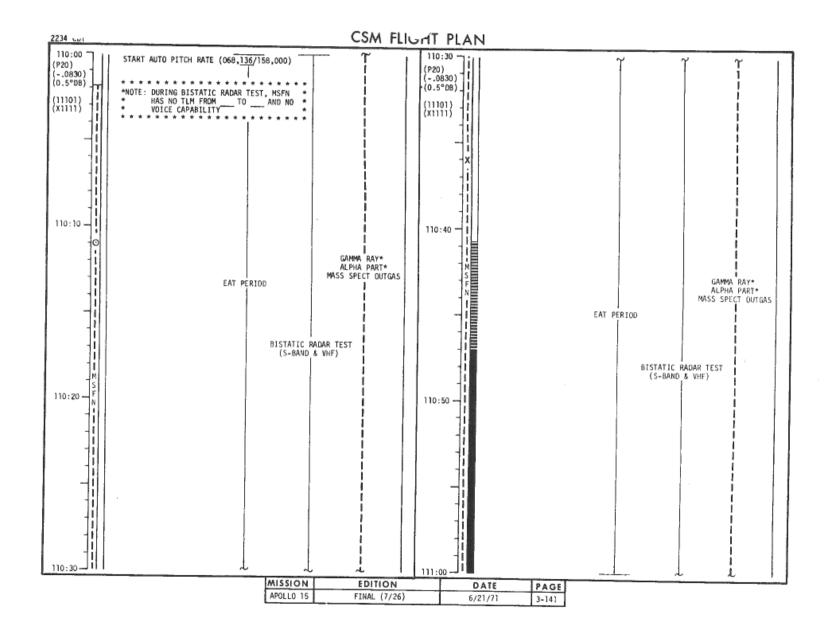


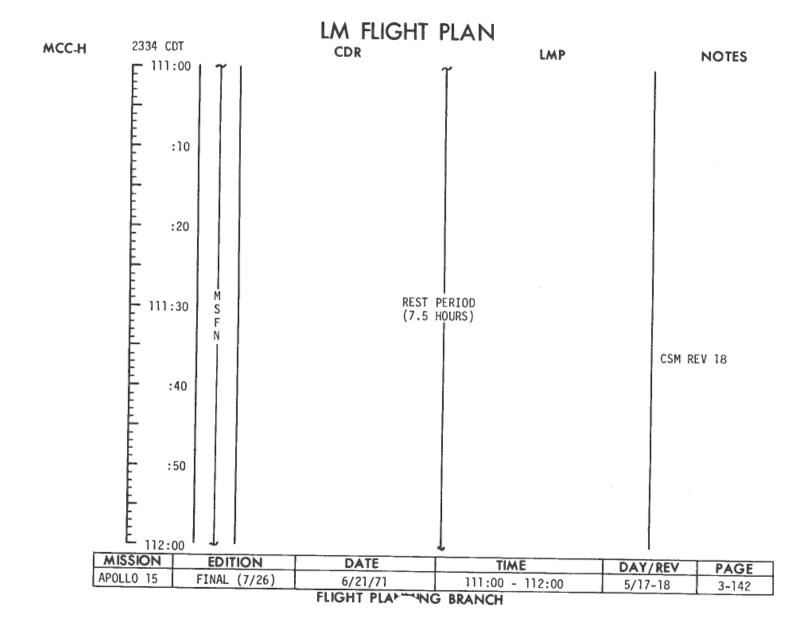


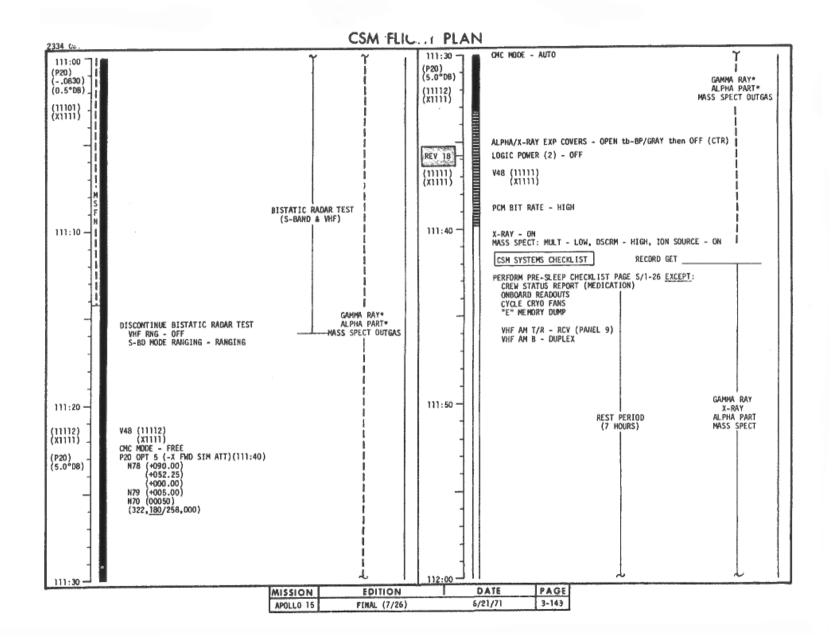


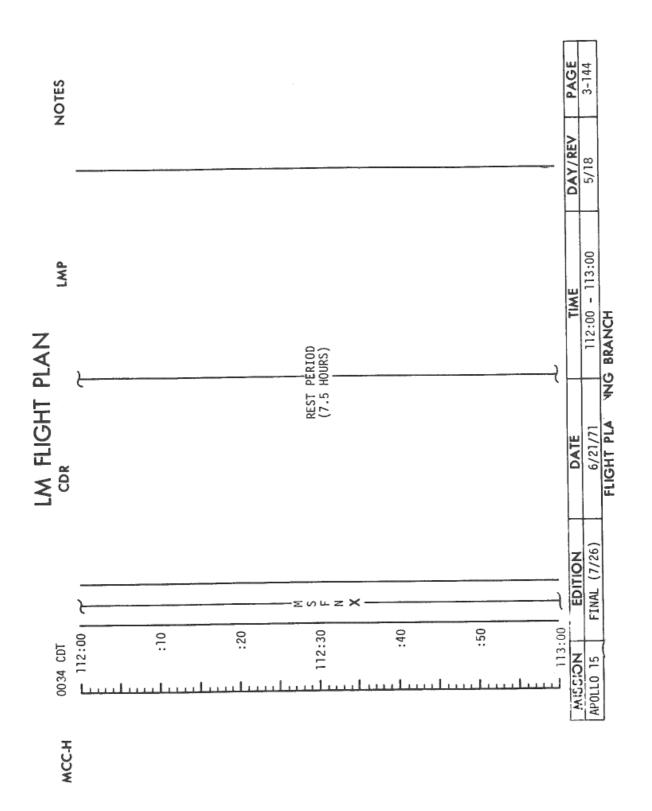


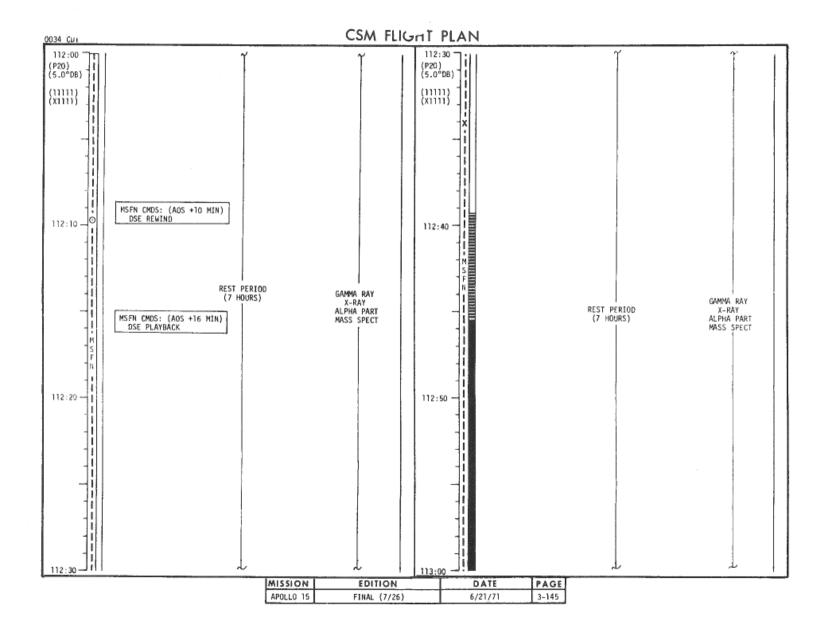


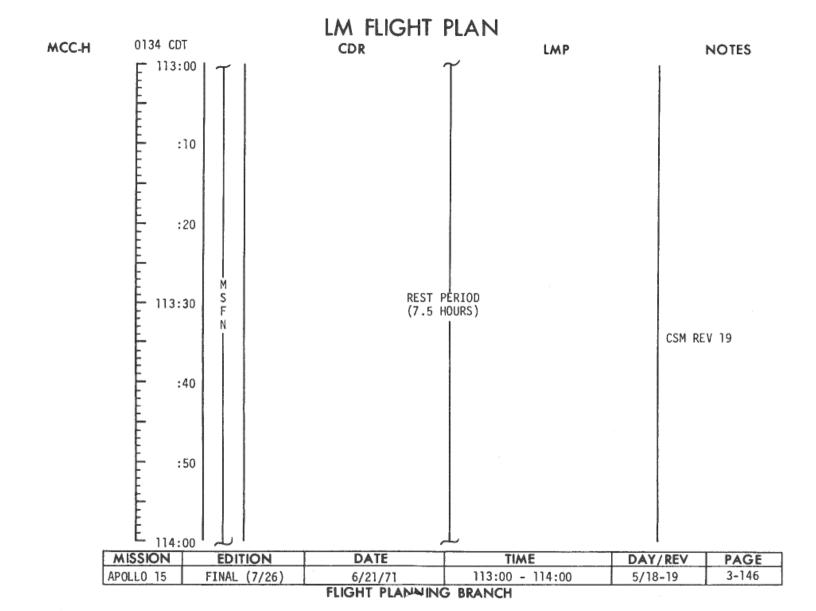


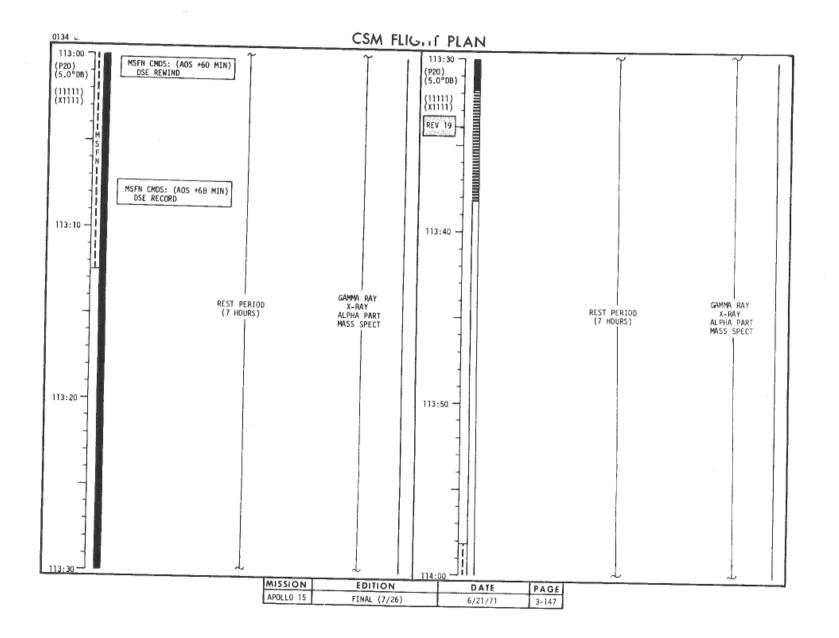


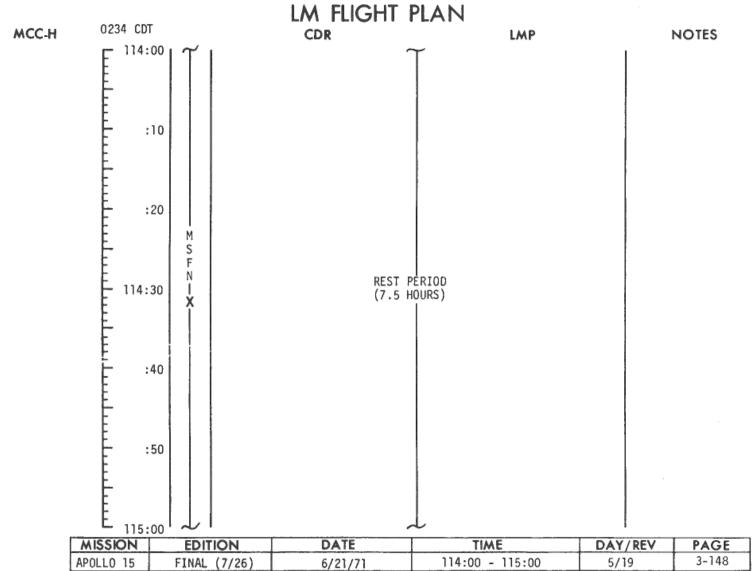




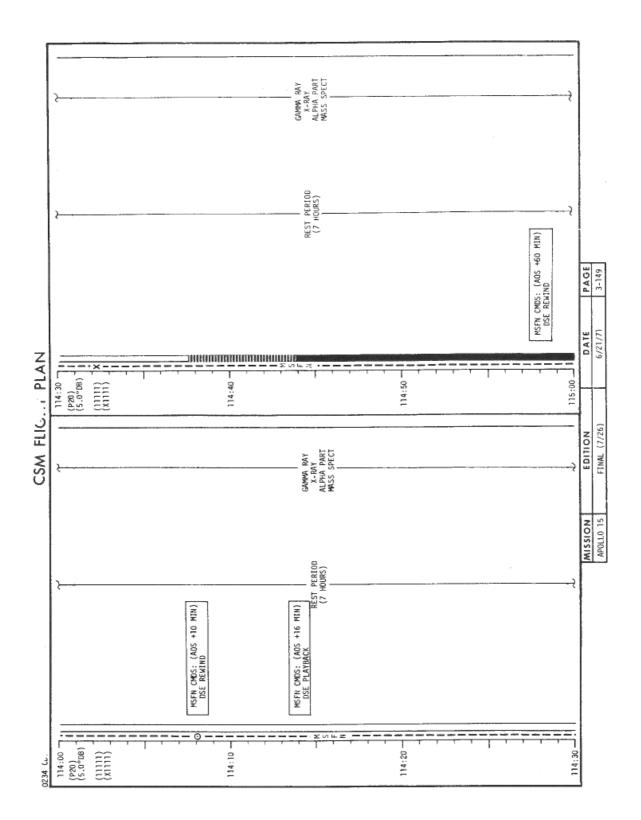


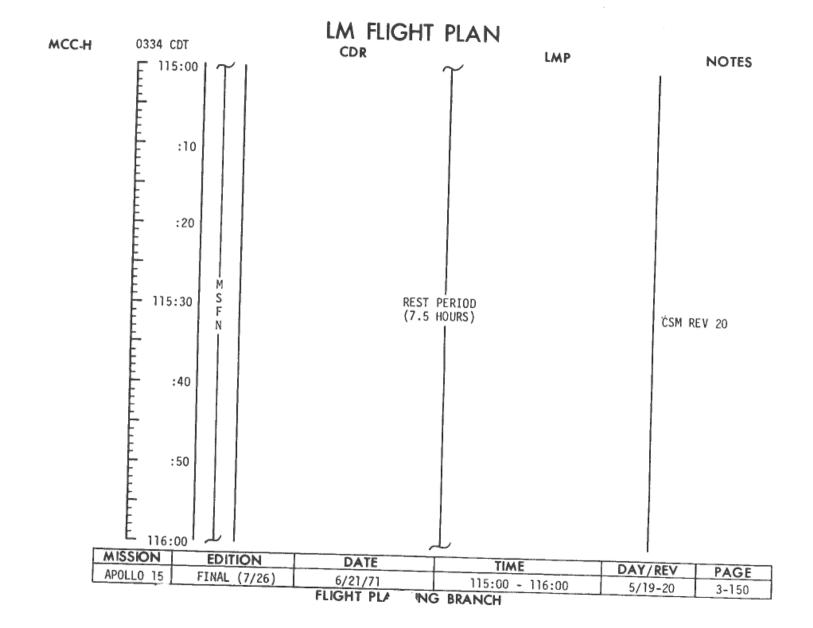


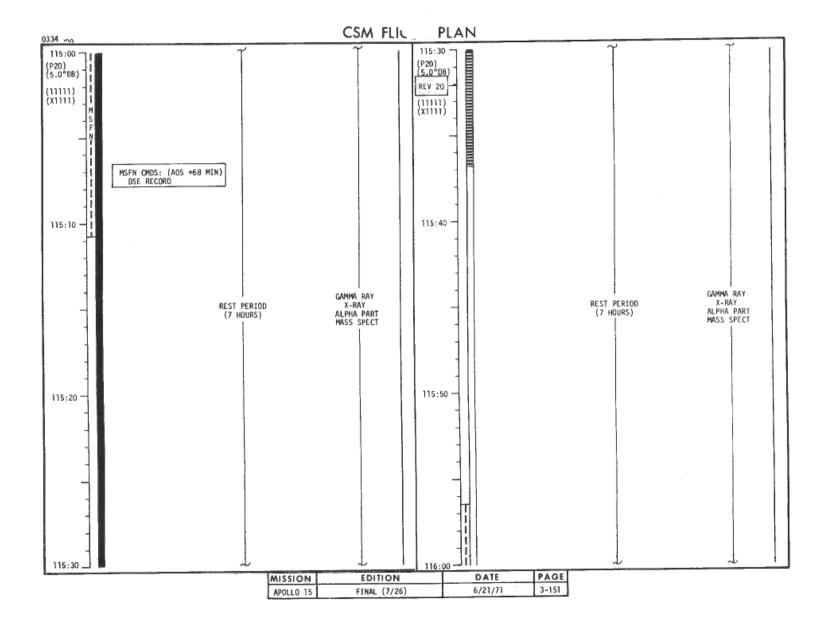


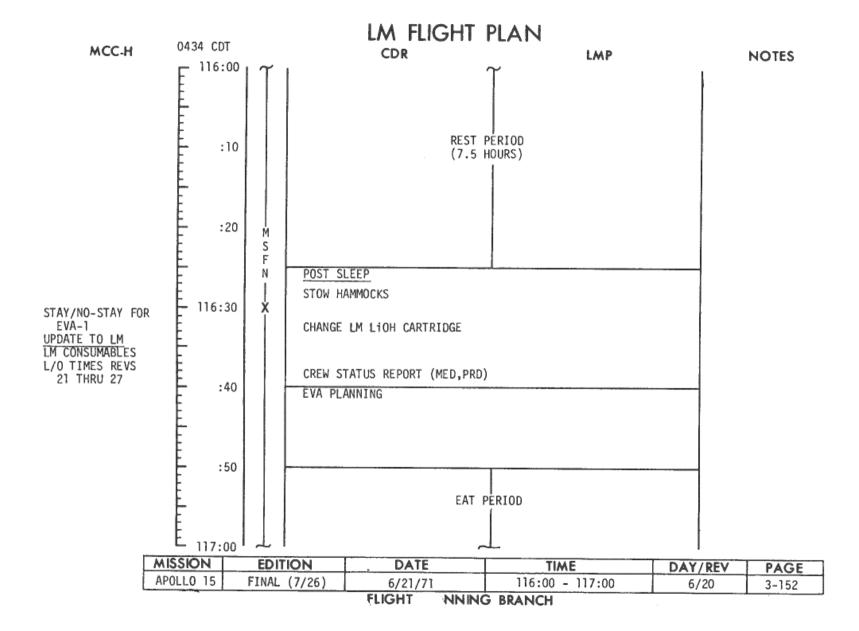


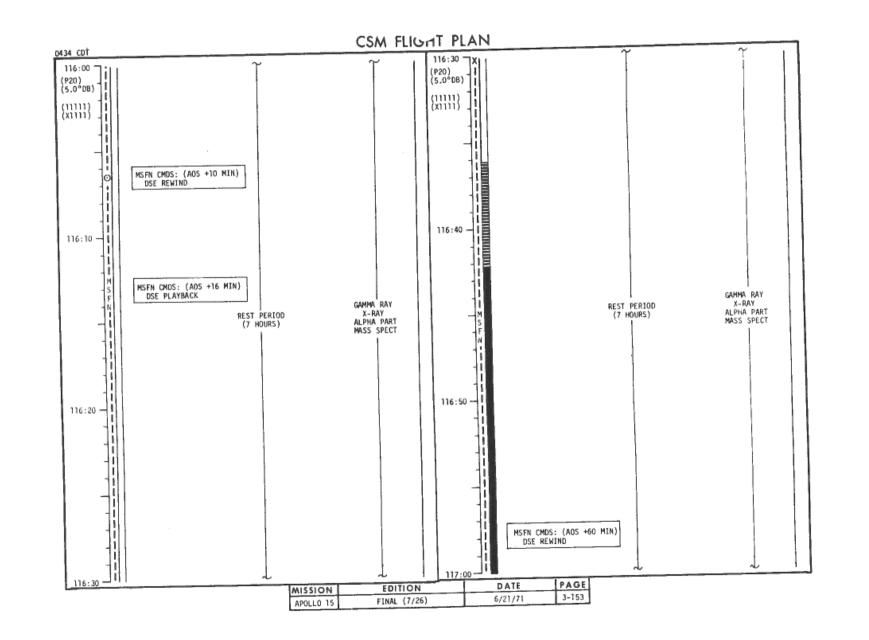
FLIGHT PLANNING BRANCH

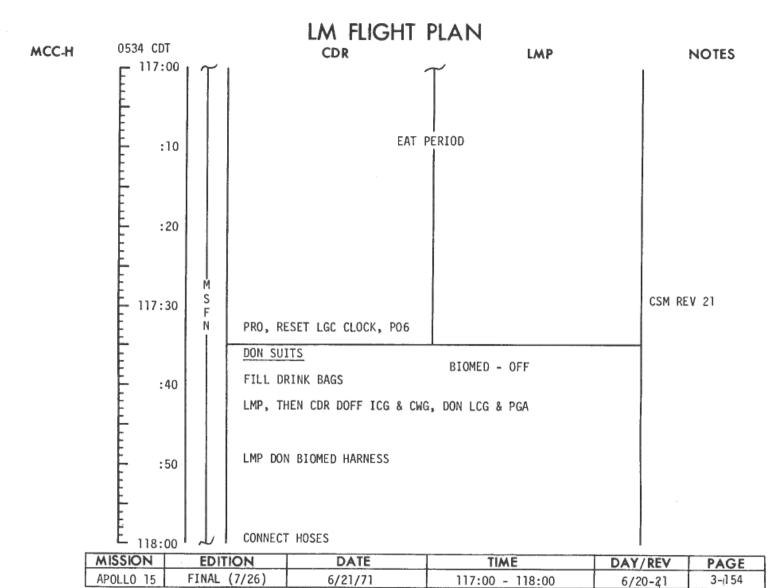




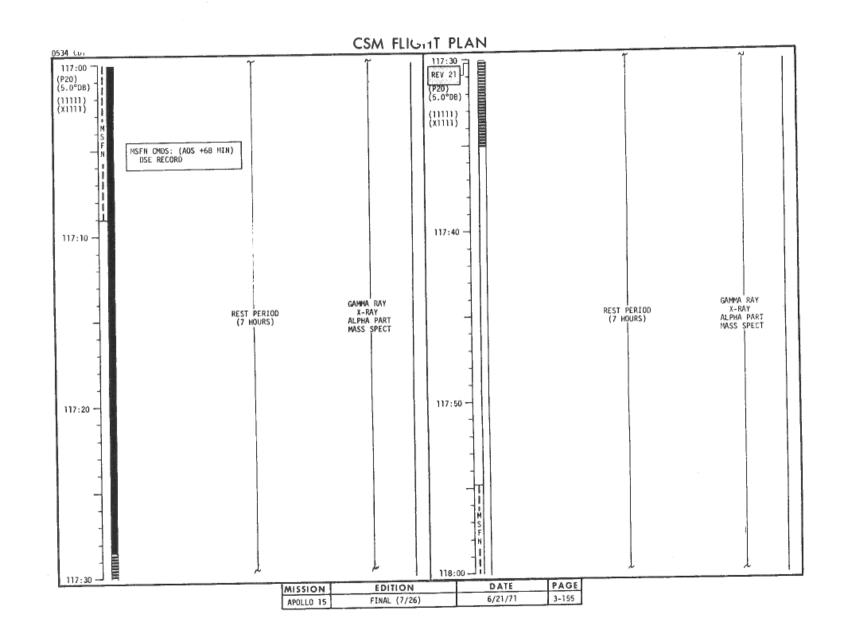


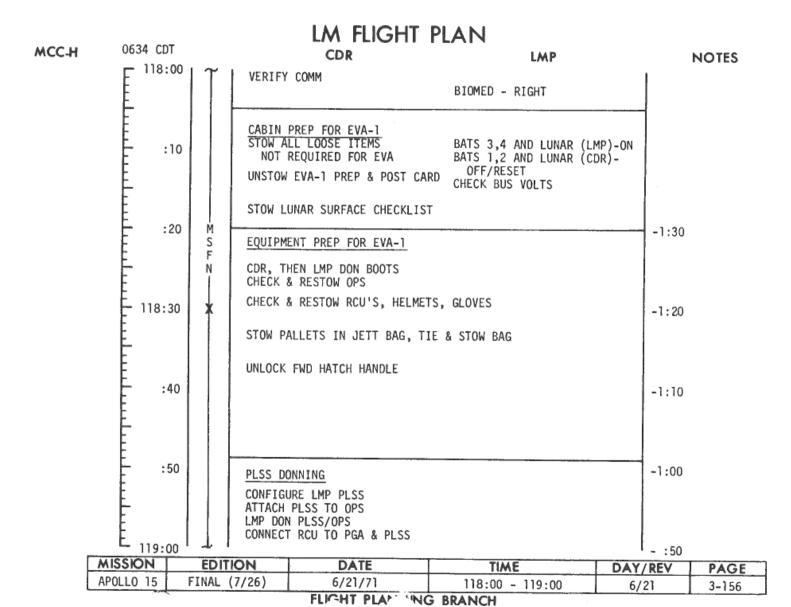


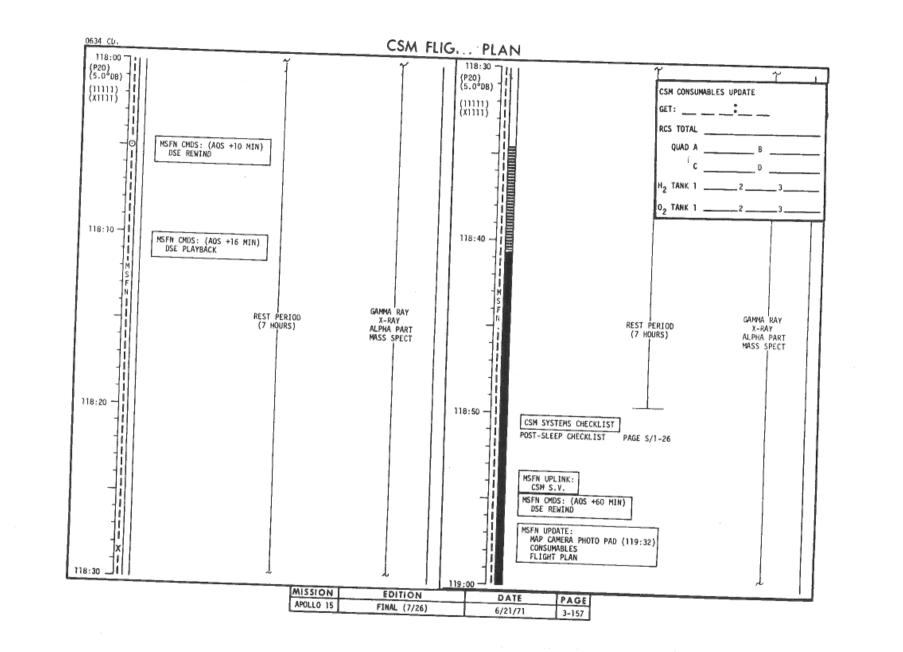




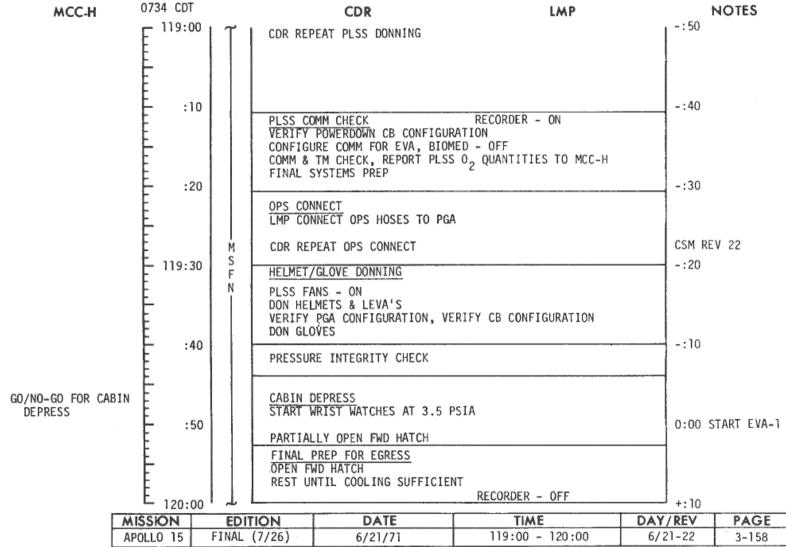
FLIGHT PLAPTING BRANCH



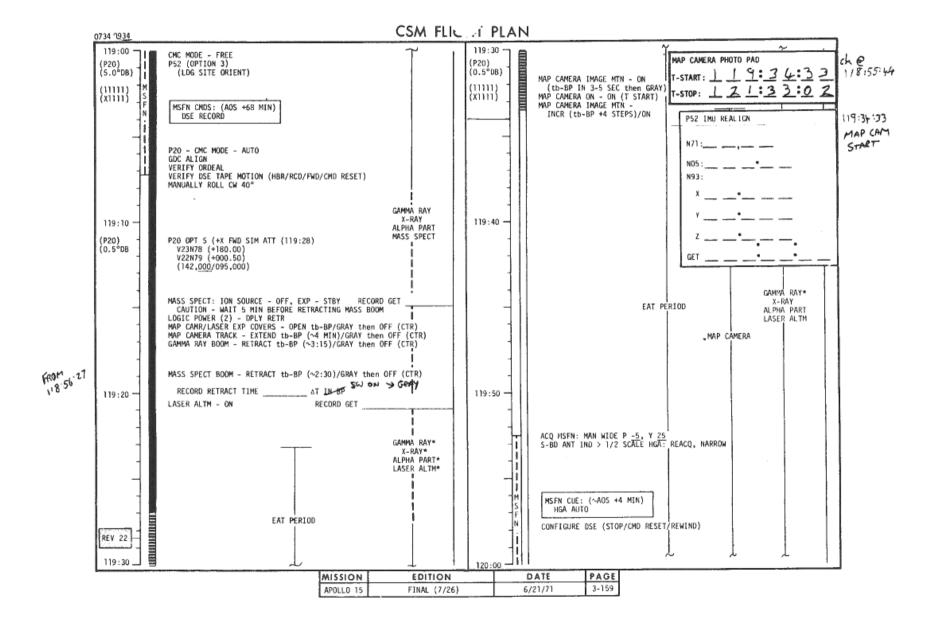




LM FLIGHT PLAN

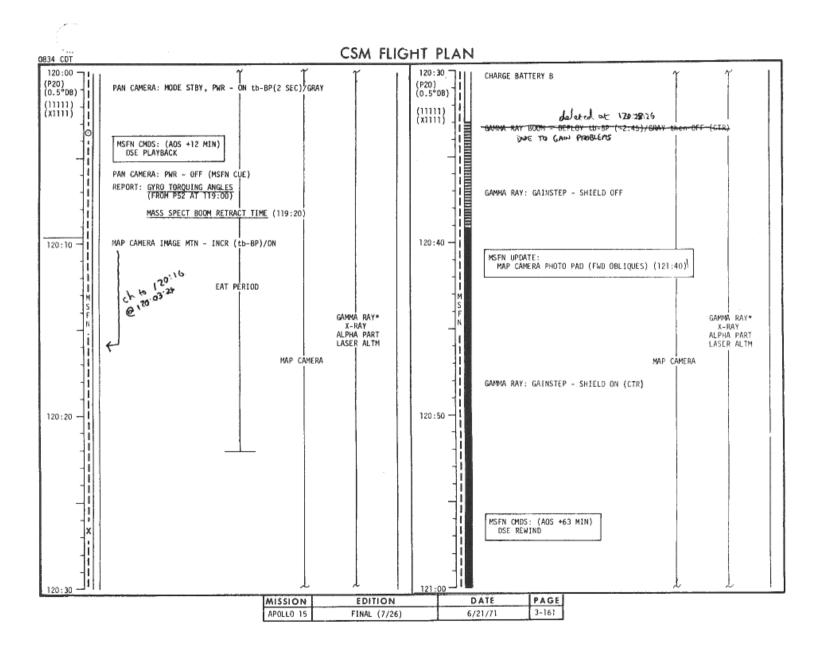


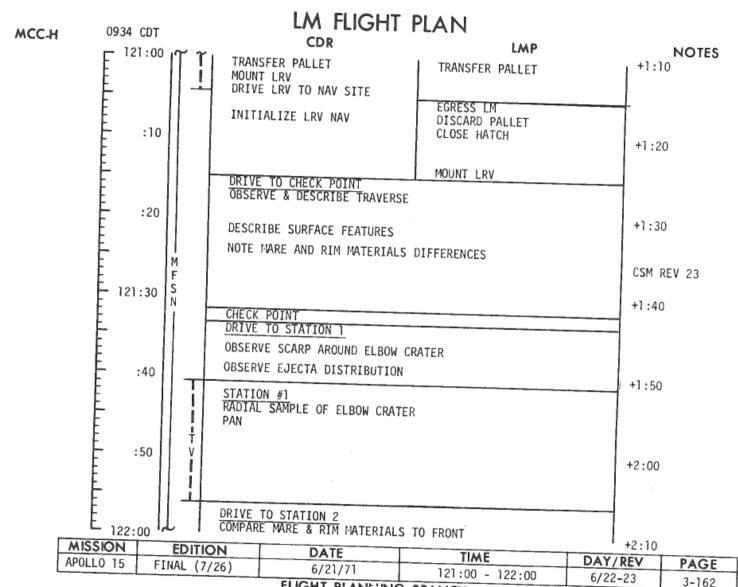
FLIGHT " NNING BRANCH



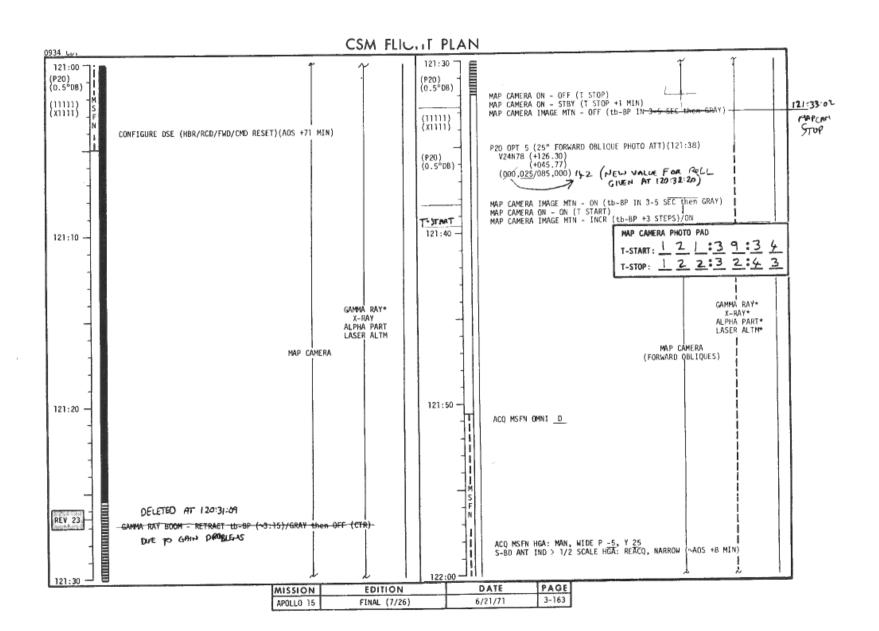
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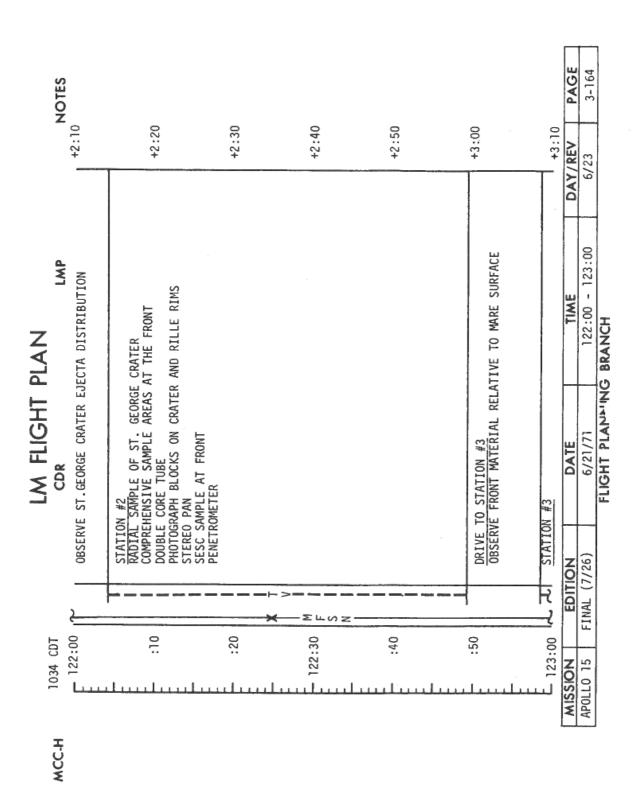
	LW TEIGHT TEATS					
MCC-H	0834 CDT		CDR	LMP		OTES
	E 120:00	T DEPLOY		ASSIST CDR EGRESS RELEASE CDR PLSS ANTENNA LEC TO CDR RECORDER - OFF VERIFY CB & SW CONFIGURA	ATION	
	:10	TRANSFE	R ETB	TRANSFER ETB	+:20	
	:20	TV DEPL	.OY E LMP PLSS ANTENNA	EGRESS, CLOSE HATCH DESCEND TO SURFACE	+:30	
	F .20 1	T LRV DEI	PLOY	COLLECT CONTINGENCY SAMPLE	1.50	
	-	LRV SET	ГИР	ASSIST CDR		
	120:30	M I ERV CHE	CKOUT		+:40	
	:40	N LCRU DI	PLOY	LRV EQUIPMENT PREP	+:50	
	121:00		ER TV TO LRV		+1:00	
				INGRESS LM WITH CONTINGENCY SAMPLE CONFIGURE COMM	+1:10	
	MISSION	EDITION	DATE	TIME	6/22	PAGE
						3-160
FLIGHT PLA "NG BRANCH						

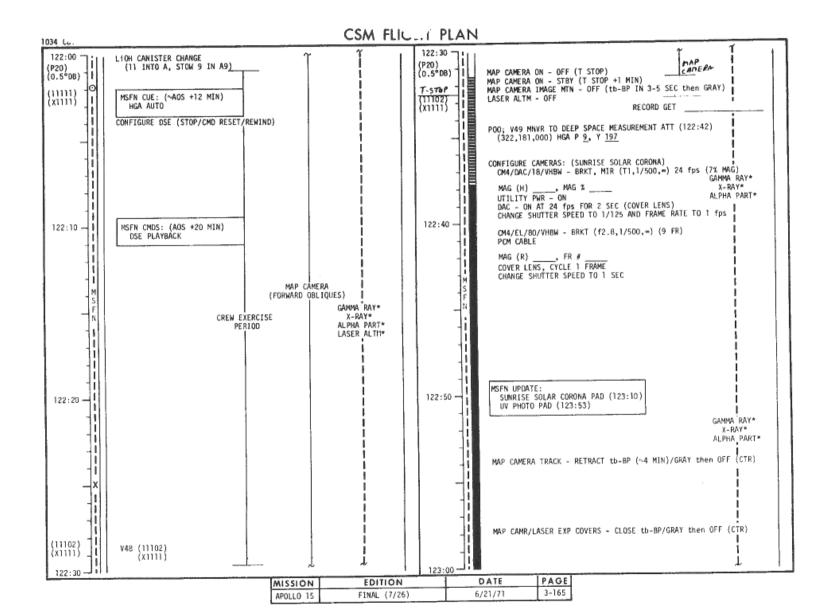




FLIGHT PLANTING BRANCH





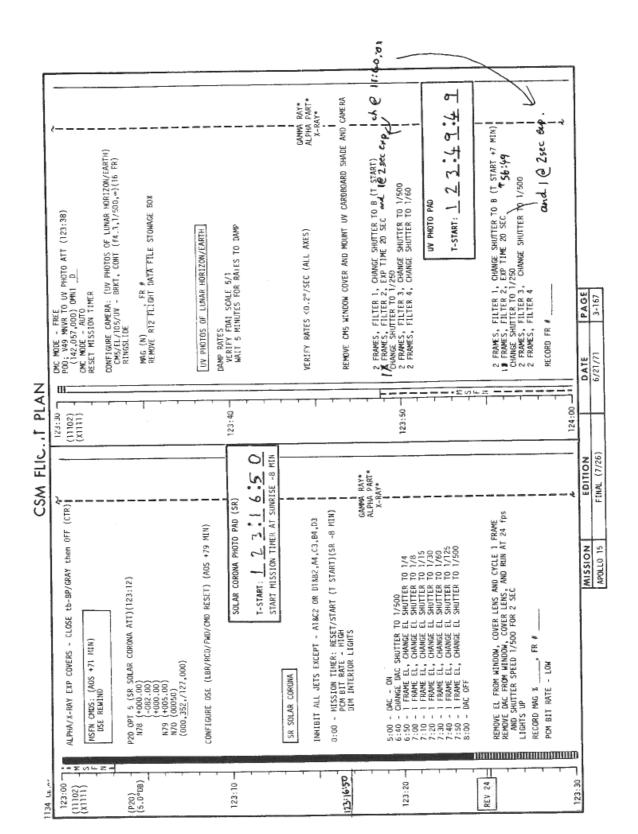


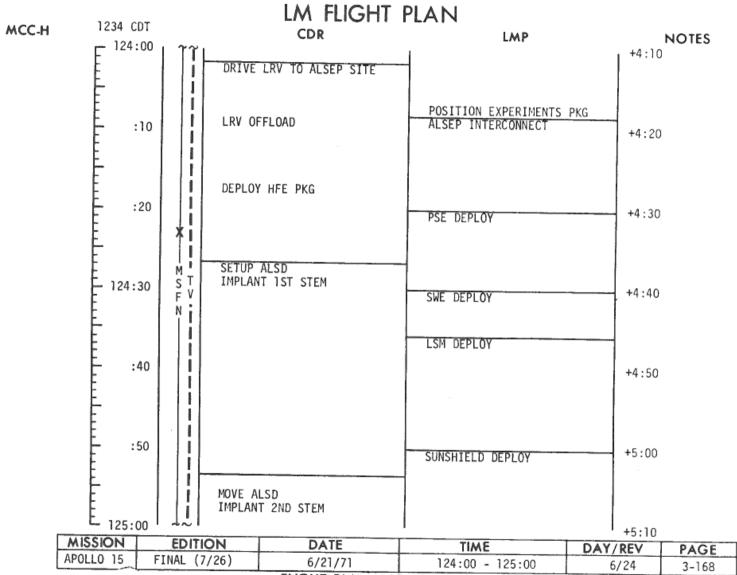
LM FLIGHT PLAN 1134 CDT MCC-H CDR LMP **NOTES —** 123:00 EXAMINE FLOW, COMPARE TO MARE AND FRONT +3:10 TAKE DOC. SAMPLES OF FRONT & FLOW MATERIAL OBSERVE VERTICAL & LATERAL CHANGES, COMPARE TO STA. #2 PREVIEW EVA-2 ROUTE :10 +3:20 DRIVE TO LM OBSERVE CHARACTERISTICS OF DEBRIS FLOW PREVIEW EVA-2 ROUTE COMPARE MARE MATERIAL TO FRONT AND RILLE RIM OBSERVE POSSIBLE RAY MATERIAL :20 +3:30 CSM REV 24 123:30 +3:40 :40 +3:50 ALSEP OFFLOAD ALSEP OFFLOAD :50 +4:00 FUEL RTG 124:00 ALSEP TRAVERSE +4:10 MISSION

 MISSION
 EDITION
 DATE
 TIME
 DAY/REV
 PAGE

 APOLLO 15
 FINAL (7/26)
 6/21/71
 123:00 - 124:00
 6/23-24
 3-166

FLIGHT PLAP" 'NG BRANCH





FLIGHT PLA 'NG BRANCH

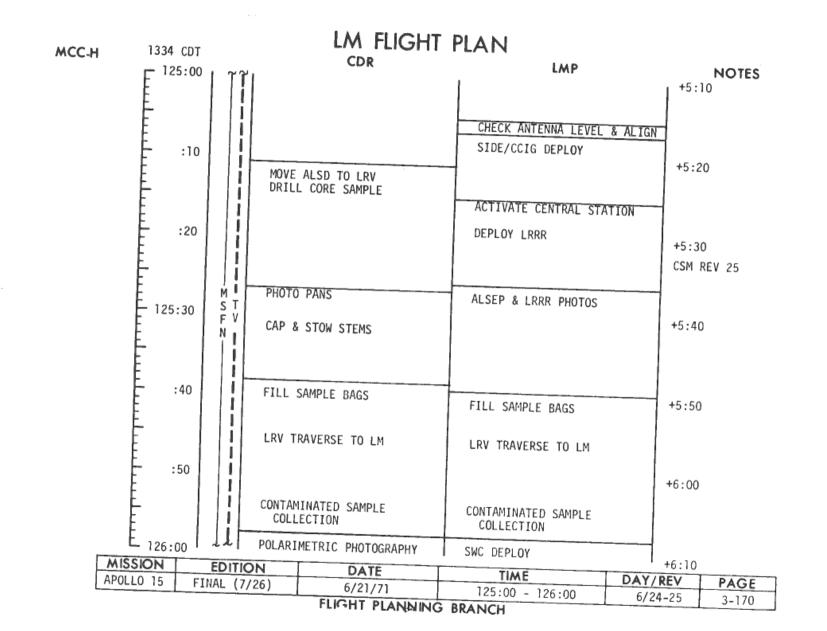
CSM FLIGHT PLAN 1234 CDT 124:30 124:00 (P20) (5.0°DB) (11102) (X1111) CONFIGURE CAMERA (UV COLOR PHOTOS OF EARTH) SUNSET SOLAR CORONA (11102) CM5/EL/105/CEX-BRKT, CONT (f8,1/250,∞) (1 FR) 0:00 - MISSION TIMER: RESET/START/(T START)(SS -5 MIN) (X1111) RINGSLIDE MAG (M) , FR # _____ VERIFY PCM BIT RATE HIGH 124:32:22 T-START (X1111)-DIM INTERIOR LIGHTS REMOVE CAMERA & UV CARDBOARD SHADE, NOTE COMMENTS AS TO SOLAR CORONA PHOTO PAD (SS) CONDITION OF CM5, REPLACE WINDOW COVER RECORD FR # T-START: 1 2 4:3 2:2 2 P20 OPT 5 (SS SOLAR CORONA ATT)(124:12) (P20) START MISSION TIMER AT SUMSET +5 MIN (5.0°DB) N78 (+000.00) 5:00 - DAC - ON (+090.00) 5:10 - 1 FRAME EL, CHANGE EL SHUTTER TO 1/60 5:20 - 1 FRAME EL, CHANGE EL SHUTTER TO 1/30 (+000.00 N79 (+005.00) 5:30 - 1 FRAME EL, CHANGE EL SHUTTER TO 1/15 N70 (00050) 5:40 - 1 FRAME EL, CHANGE EL SHUTTER TO 1/8 (000,180/133,000) HGA P -45, Y 208 5:50 - 1 FRAME EL, CHANGE EL SHUTTER TO 1/4 6:00 - 1 FRAME EL, CHANGE EL SHUTTER TO 1.0 6:10 - 1 FRAME EL 124:10 6:20 - CHANGE DAC SHUTTER TO 1/125 124:40 -8:00 - DAC - OFF, CHANGE EL SHUTTER TO 1/500 REMOVE EL FROM WINDOW, COVER LENS, AND CYCLE 1 FRAME REMOVE DAC FROM WINDOW, COVER LENS, AND RUN AT 24 fps, AND SHUTTER SPEED 1/500 FOR 2 SEC MSFN CMDS: DSE DUMP LIGHTS - UP RECORD MAG % _____, FR # _____ GAMMA RAY* MSFN UPDATE: GAMMA RAY* ALPHA PART* SUNSET SOLAR CORONA PAD (124:35) ALPHA PART* X-RAY* X-RAY* (11101) V48 (11101) Dinini (x1111) CONFIGURE CAMERAS: (SUNSET SOLAR CORONA) CM4/DAC/18/VHBW - BRKT, MIR (T1,1/500,~) 24 fps (7% MAG) RESET MISSION TIMER MAG (H) ____, MAG % ____ CONFIGURE FOR URINE DUMP UTILITY PWR - ON 124:20 -124:50 -CMC MODE - FREE DAC - ON AT 24 fps FOR 2 SEC (COVER LENS) POO; V49 MNVR TO L4 PHOTO ATT (125:00) (11101) CHANGE FRAME RATE TO 1 fps (328,293,354) OMNI D (x1111) CMC MODE - AUTO CM4/EL/80/VHBW - BRKT (f2.8,1/500, w)(9 FR) PCM CABLE H₂ PURGE LINE HEATERS - ON MAG (R) , FR # COVER LENS, CYCLE 1 FRAME CHANGE SHUTTER TO 1/125 CONFIGURE CAMERA: (LUNAR LIBRATION) INSTALL CAMERA SHIELD CM4/NK/55/VHBW - BRKT, (fl.2,1/500,*) (6 FR) MAG () ____, FR # ___ X-RAY EXP - STBY RECORD GET 125:00 MISSION EDITION DATE PAGE

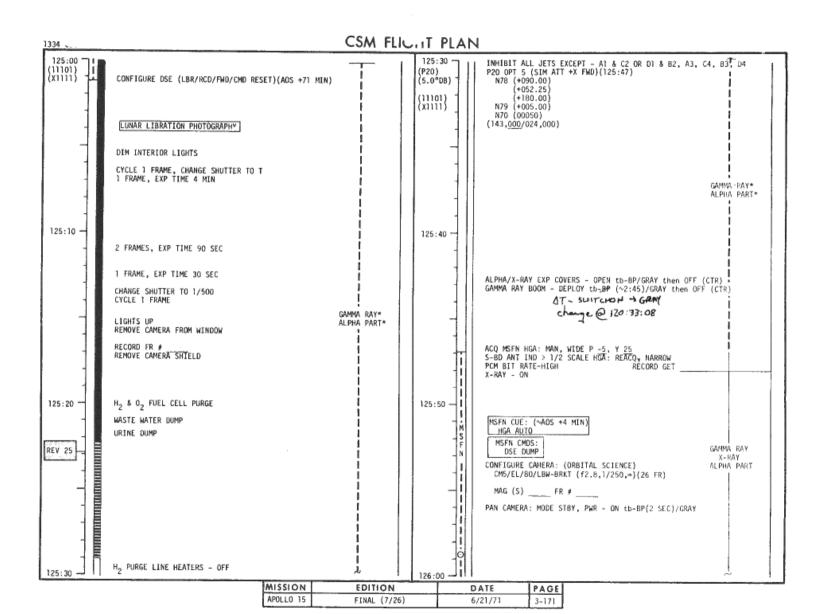
6/21/71

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APOLLO 15

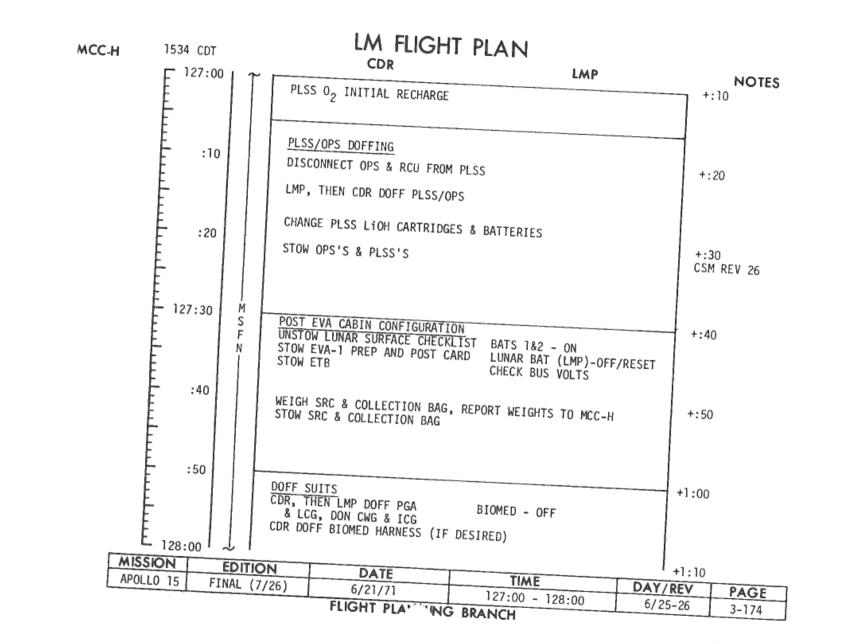
FINAL (7/26)

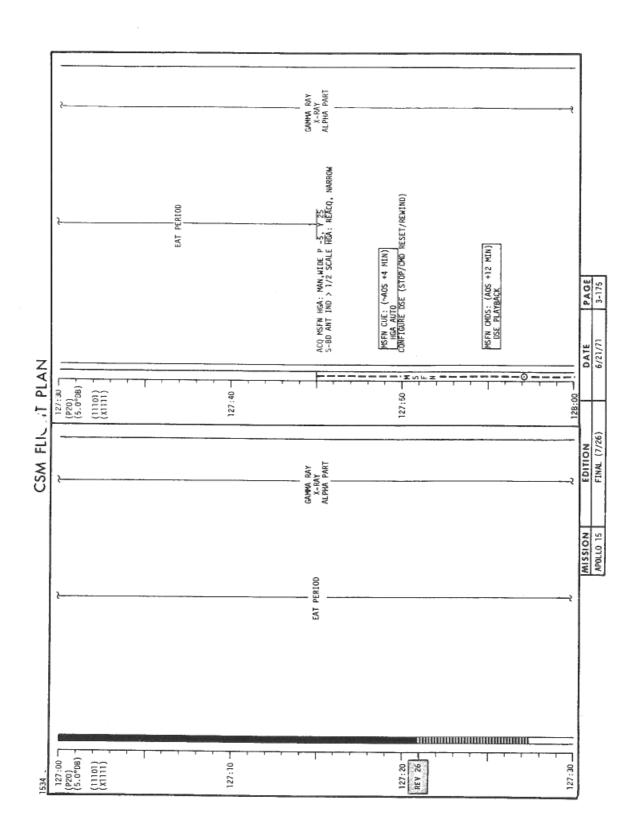


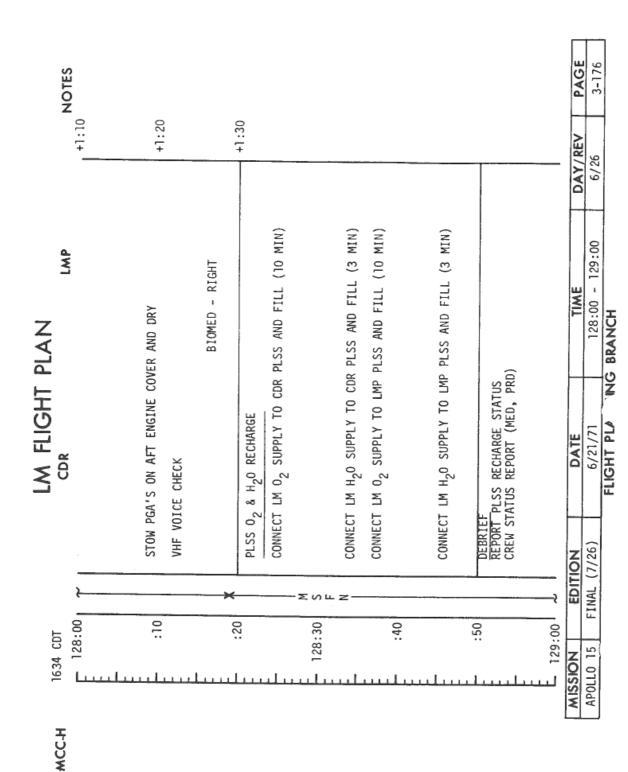


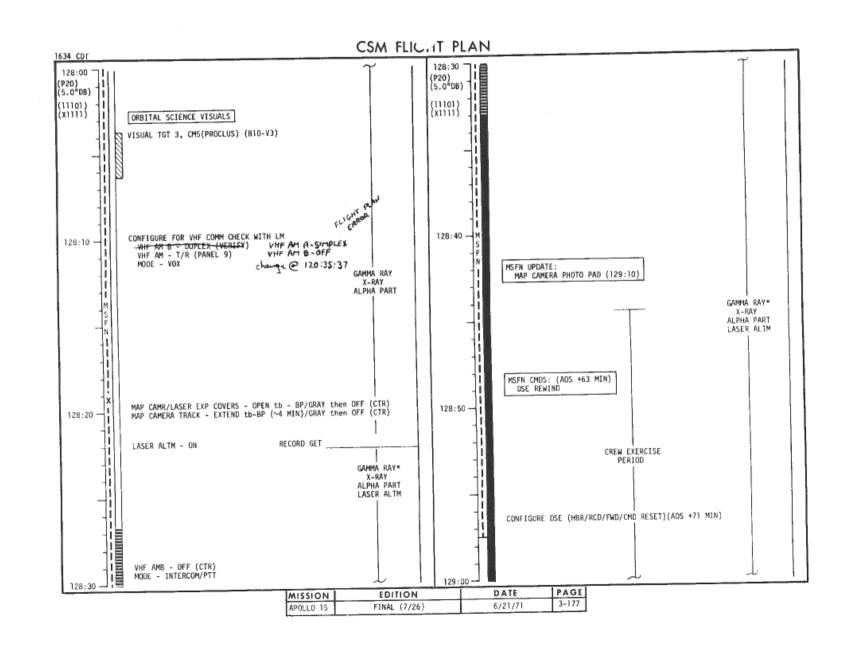
LM FLIGHT PLAN 1434 CDT MCC-H CDR LMP NOTES +6:10 - 126:00 FAR-FIELD POLARIMETRIC PHOTOGRAPHY NEAR-FIELD POLARIMETRIC LM INSPECTION & PHOTOS PHOTOGRAPHY +6:20 :10 FLAG DEPLOY FLAG DEPLOY LOAD ETB :20 +6:30 CLEAN EMU'S STOW LMP PLSS ANTENNA STOW BSLSS & 500mm LENS INGRESS CAMERA ON LRV 126:30 +6:40 STOW SAMPLE BAG TRANSFER ETB TRANSFER ETB :40 +6:50 INGRESS CLOSE HATCH REPRESS LM CABIN :50 +7:00/END EVA POST EVA SYSTEMS CONFIGURATION CONFIGURE LM ECS, DOFF HELMETS & GLOVES CONNECT LM ECS HOSES TO SUITS CONNECT TO LM COMM, BIOMED LEFT L 127:00 +:10 MISSION EDITION DATE DAY/REV TIME PAGE APOLLO 15 6/21/71 126:00 - 127:00 6/25 FINAL (7/26) 3-172

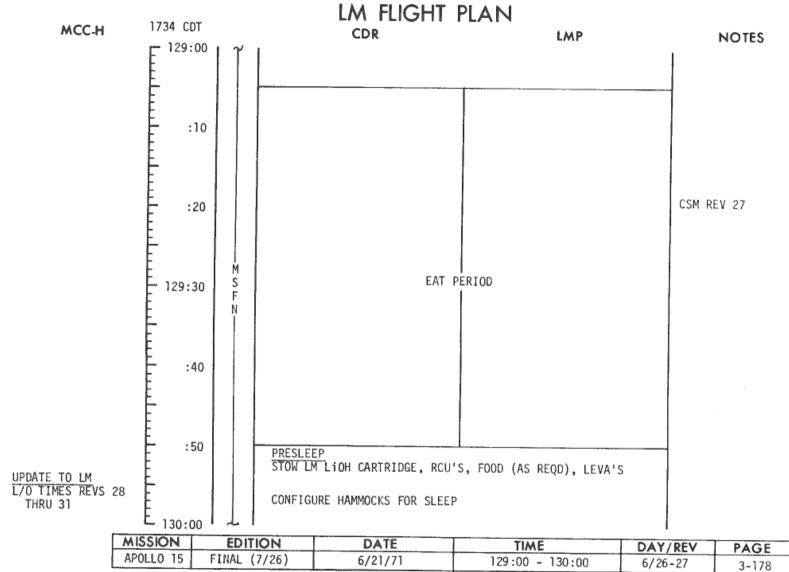
FLIGHT PLANNING BRANCH



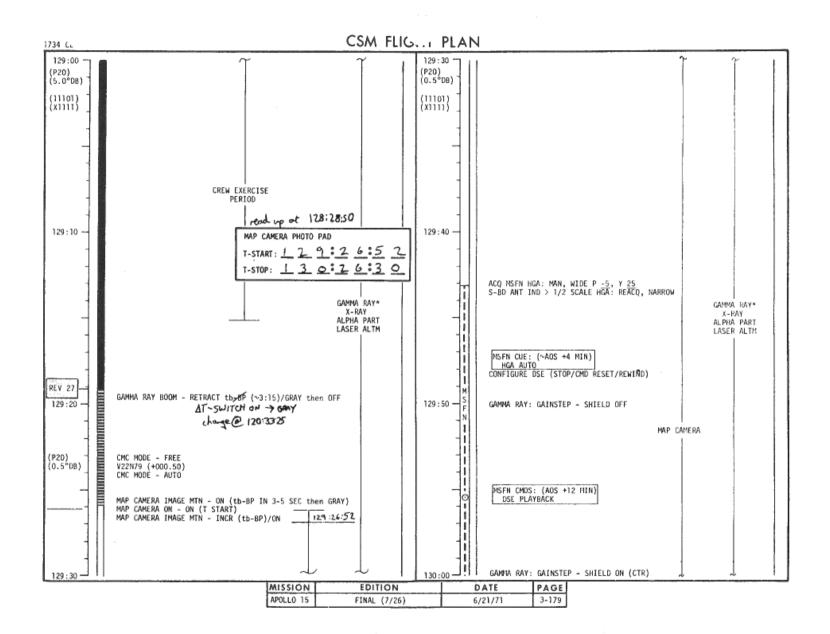


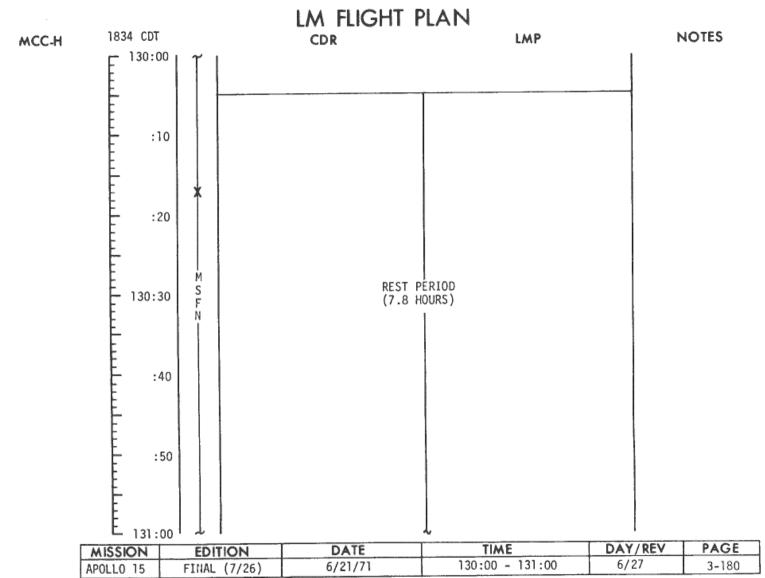




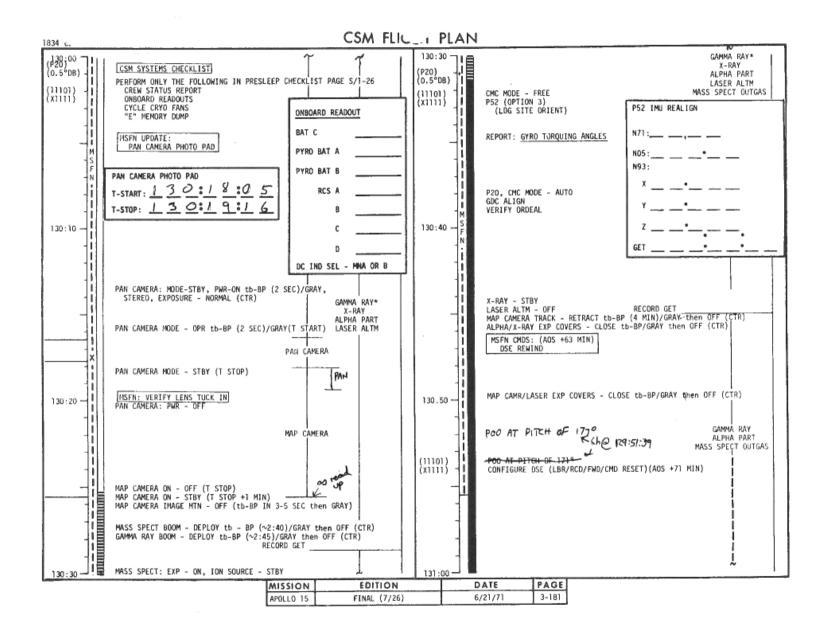


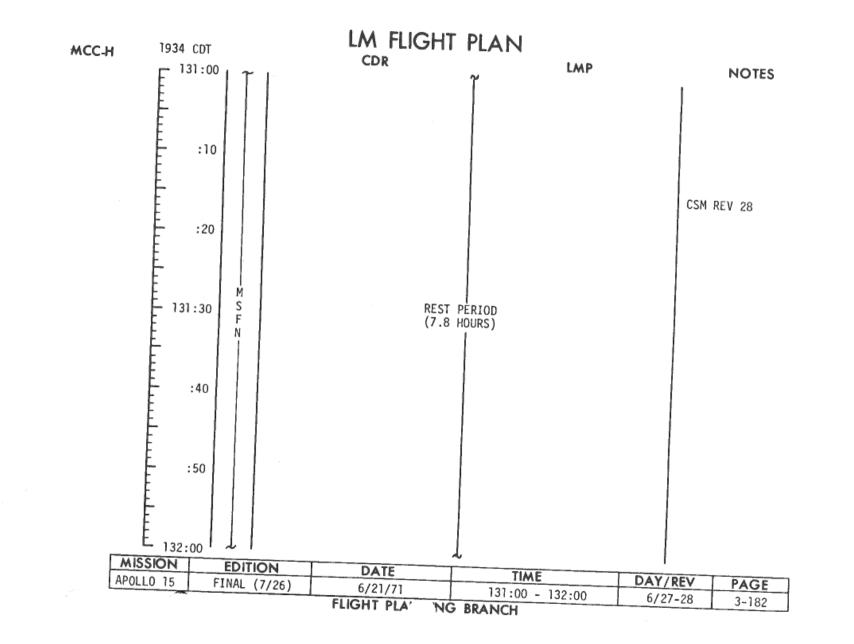
FLIGHT P* "NNING BRANCH

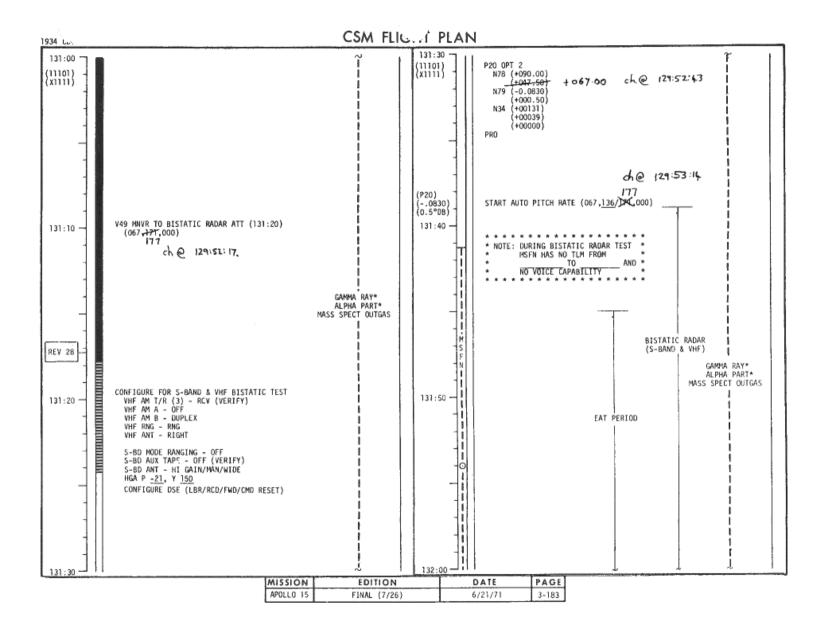


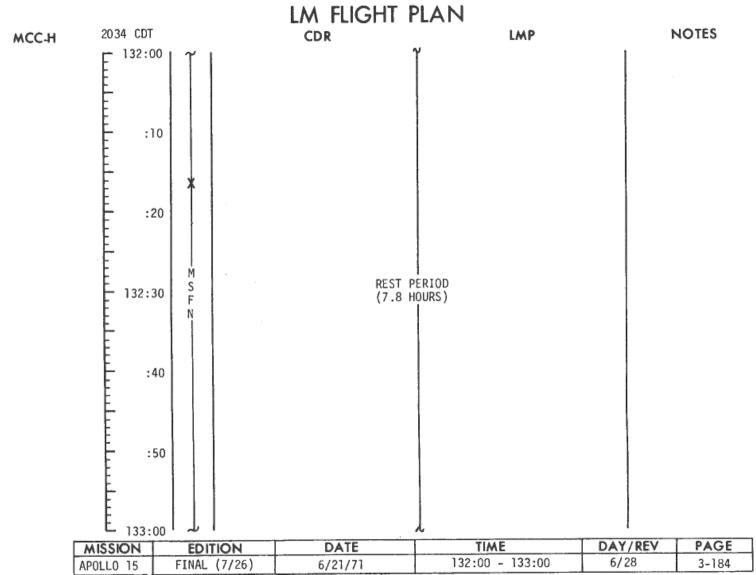


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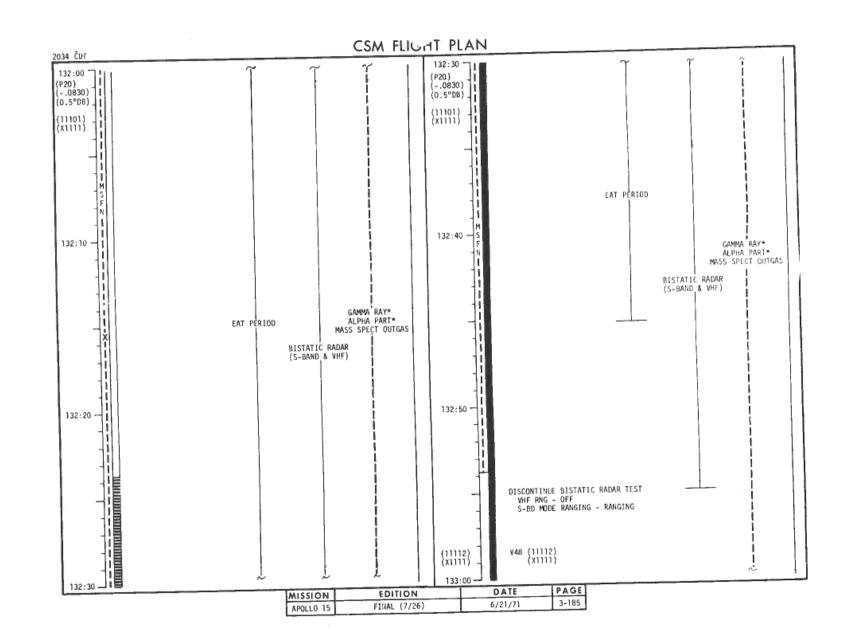


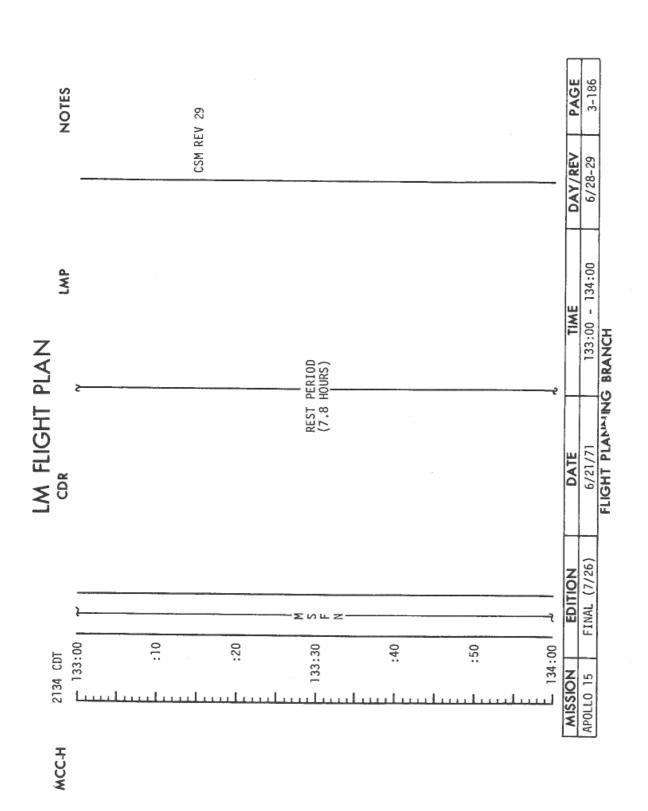


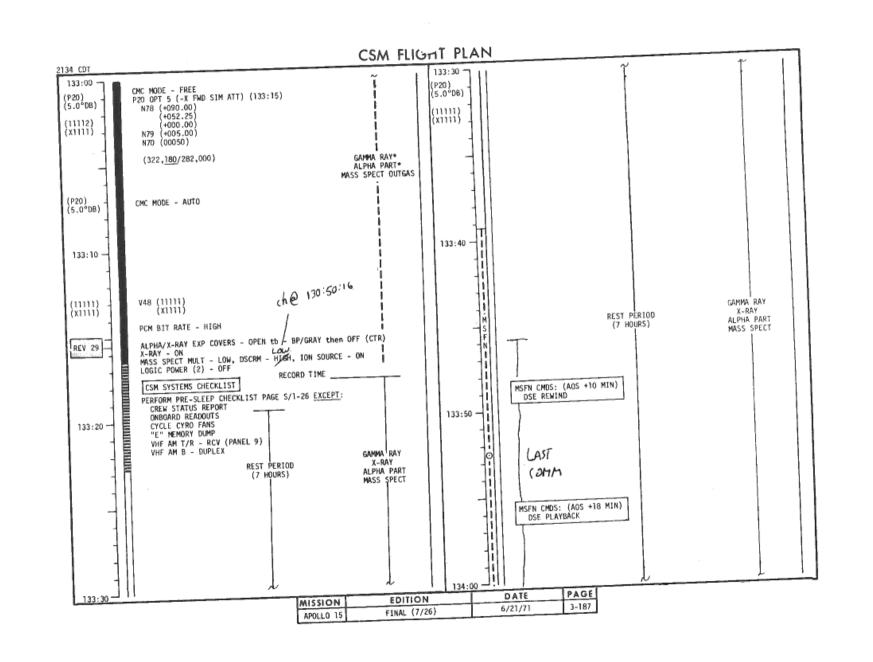


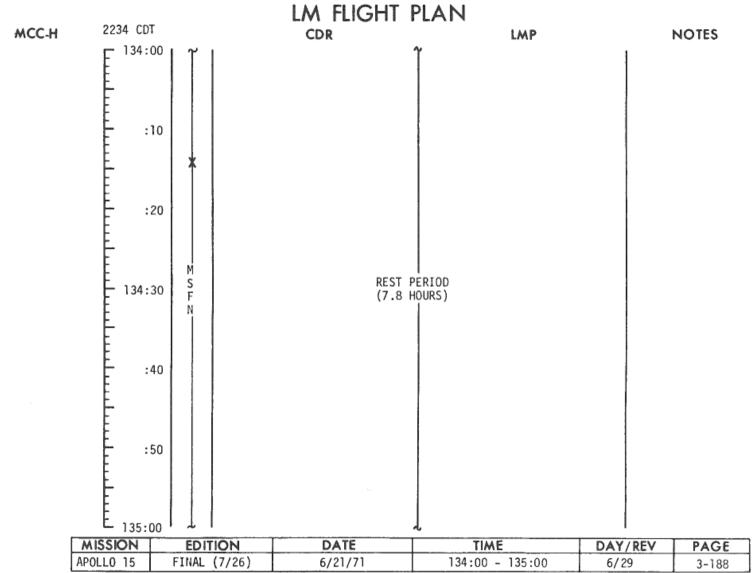


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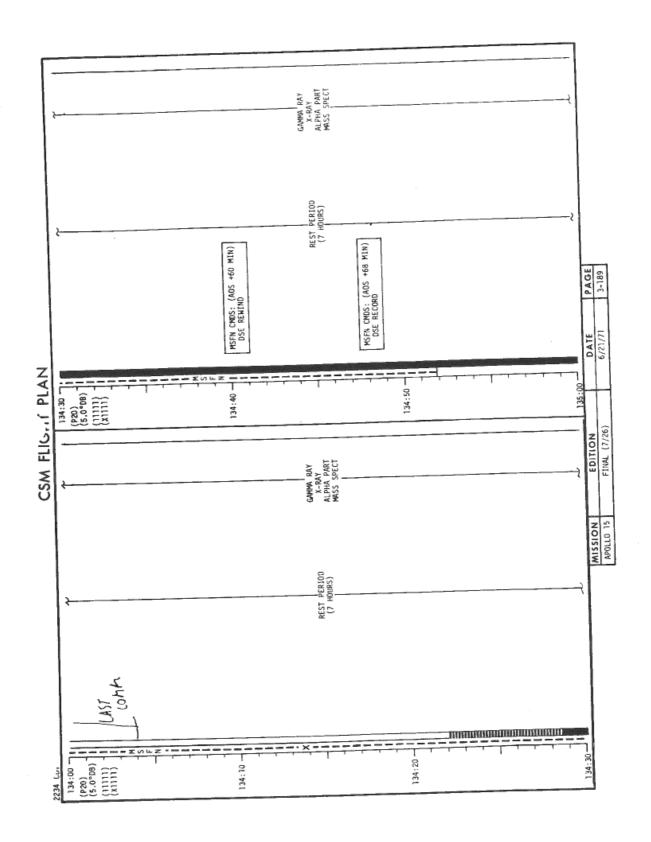


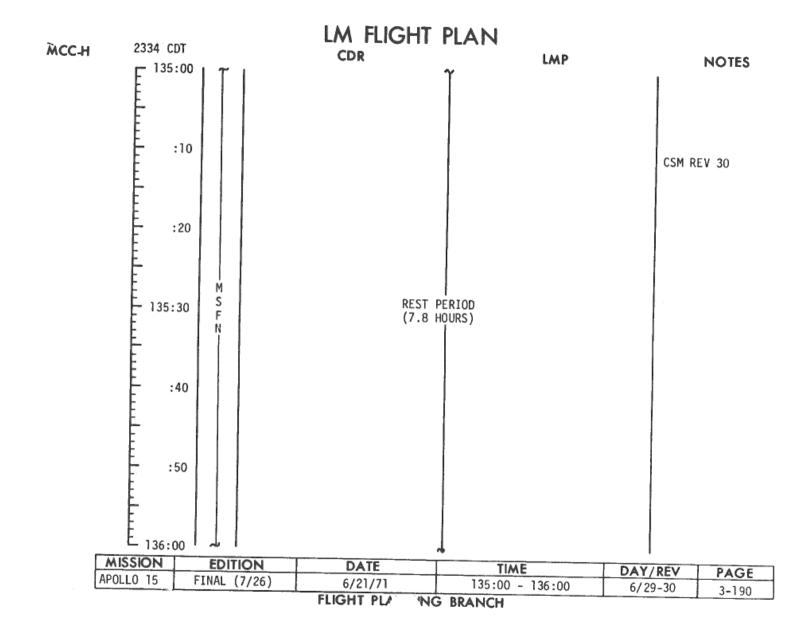


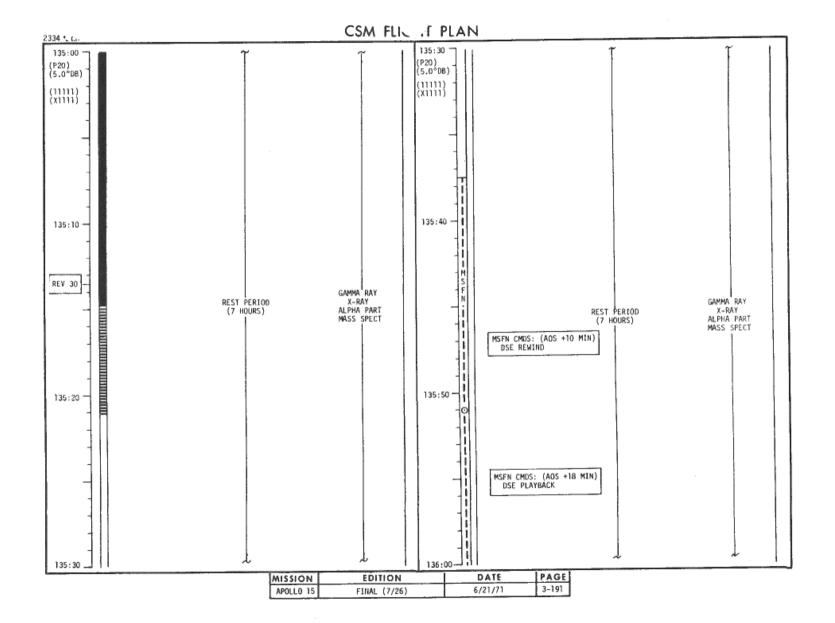


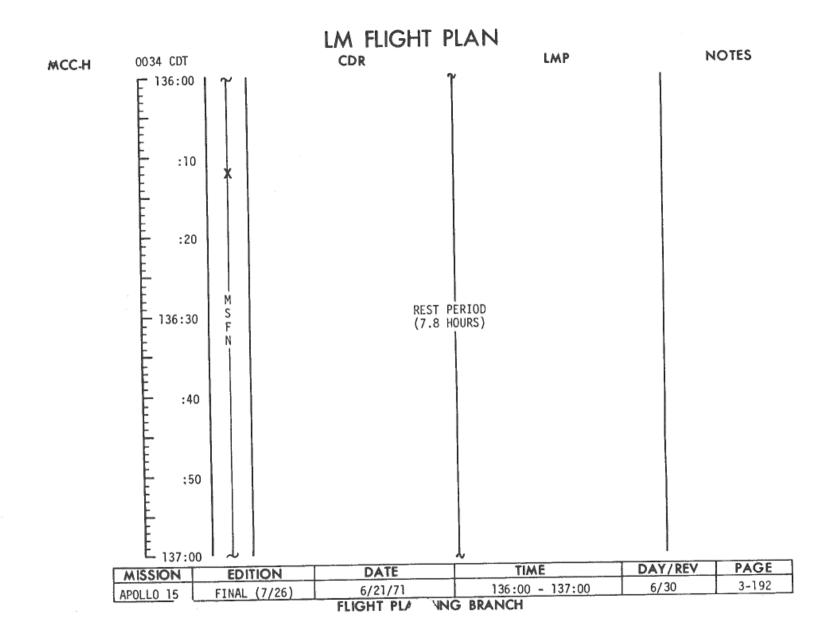


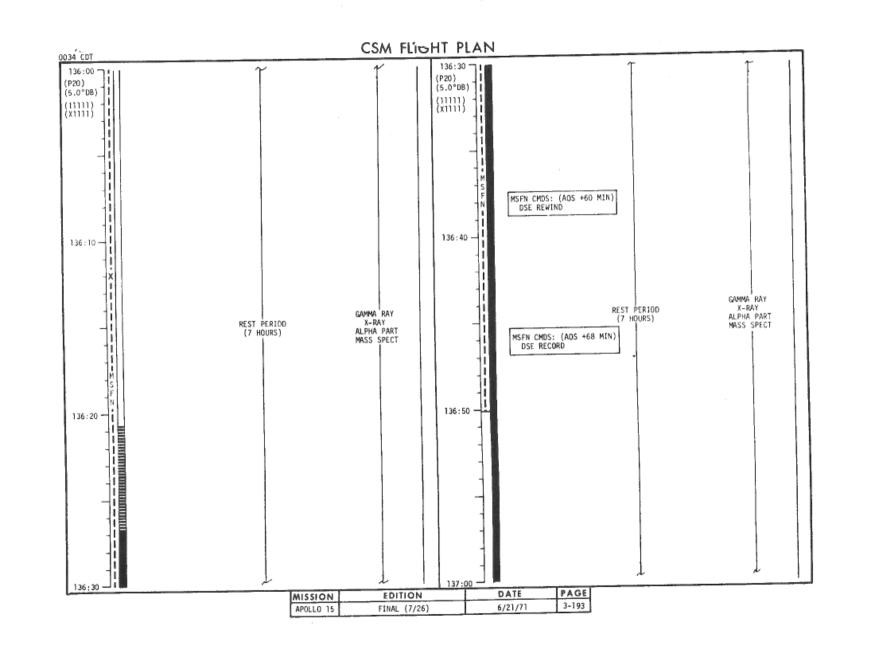
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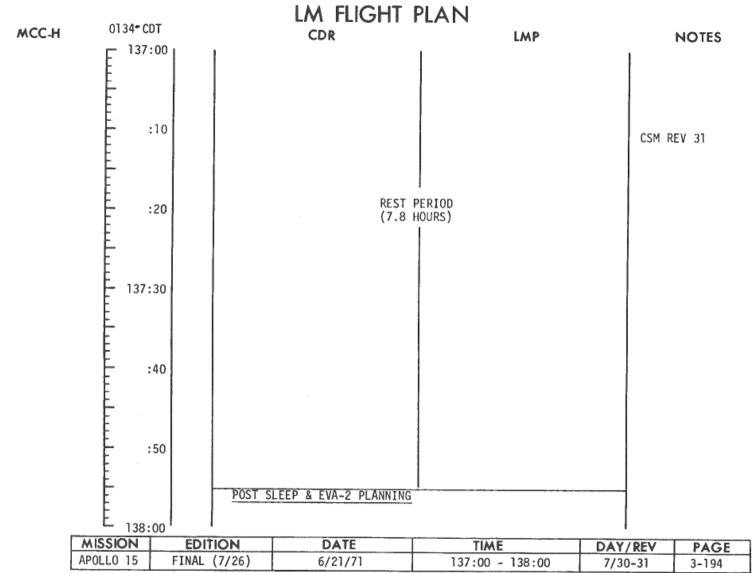




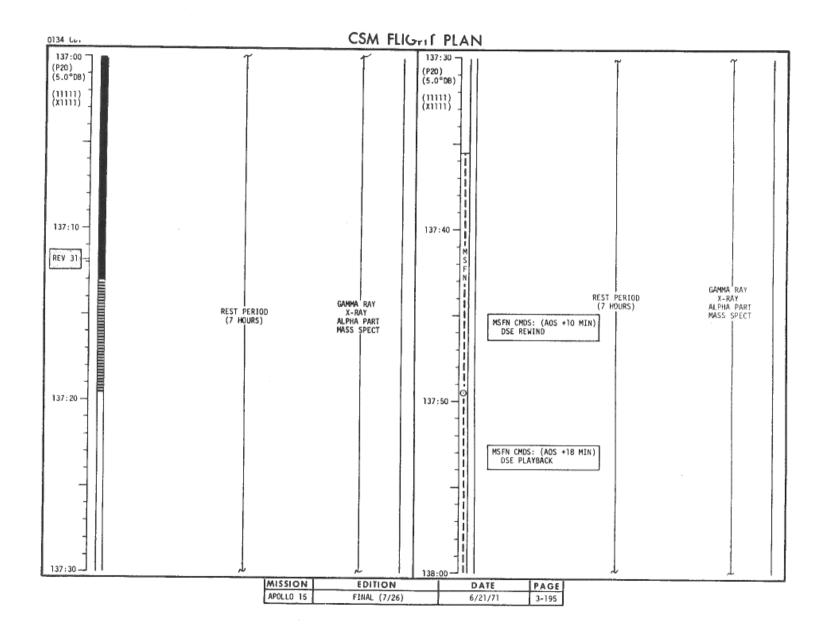




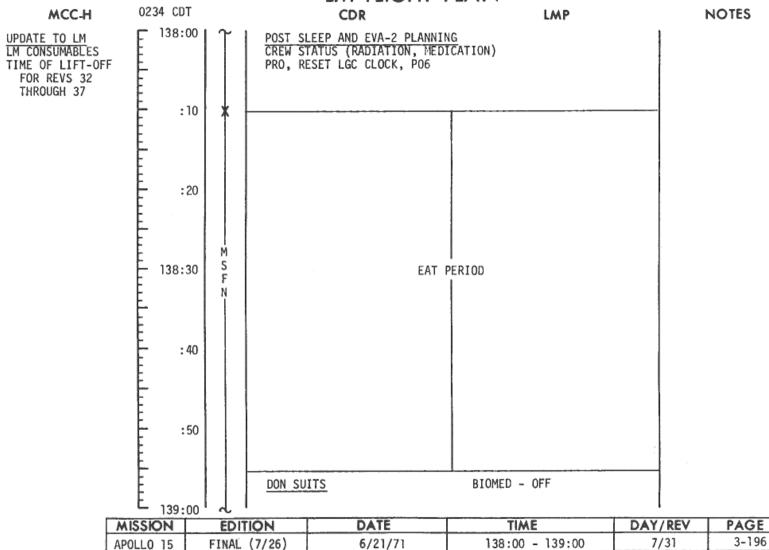




FLIGHT PLATING BRANCH



LM FLIGHT PLAN

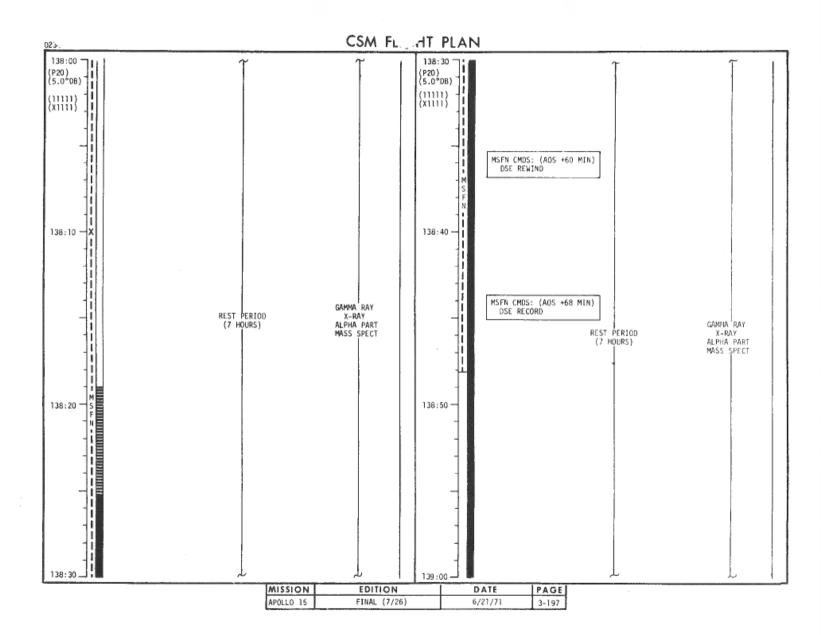


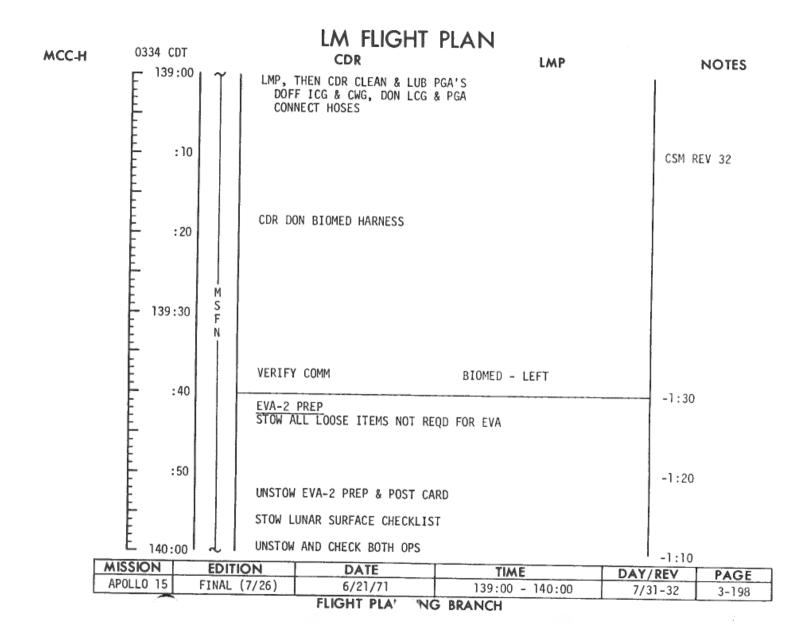
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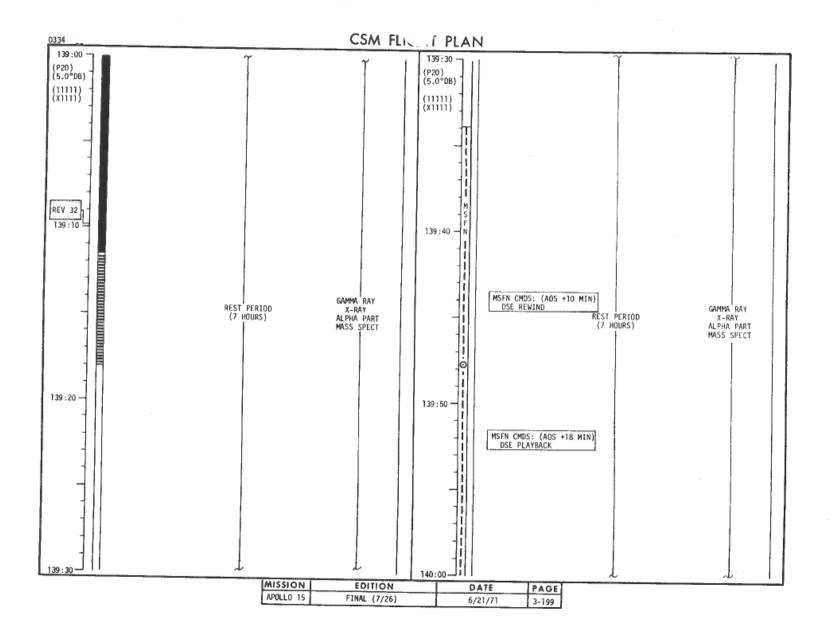
6/21/71

APOLLO 15

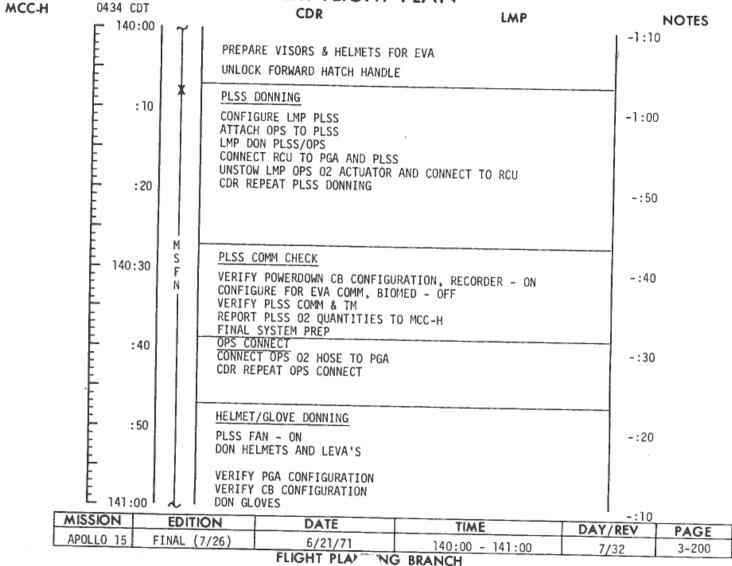
FINAL (7/26)

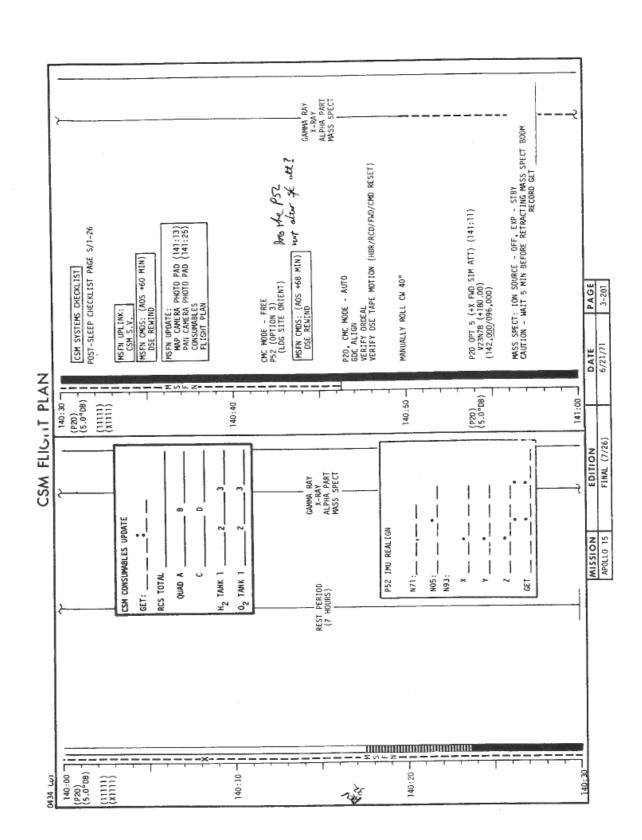


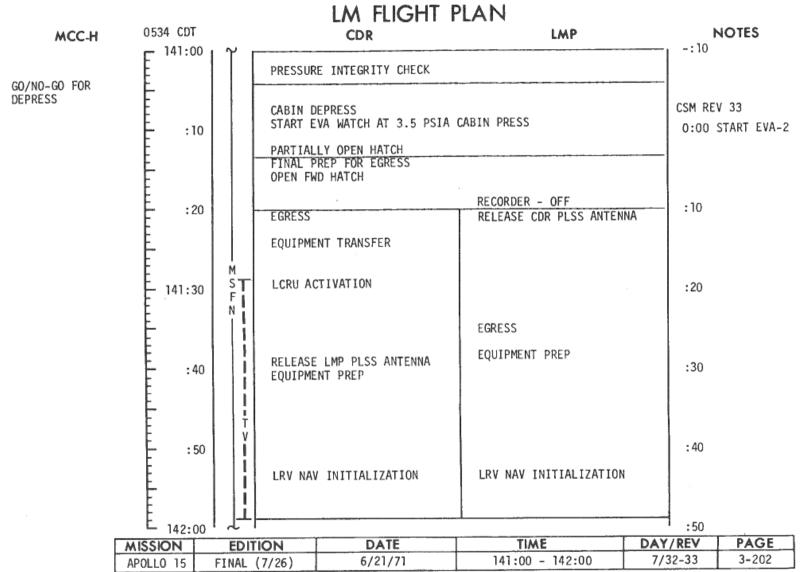




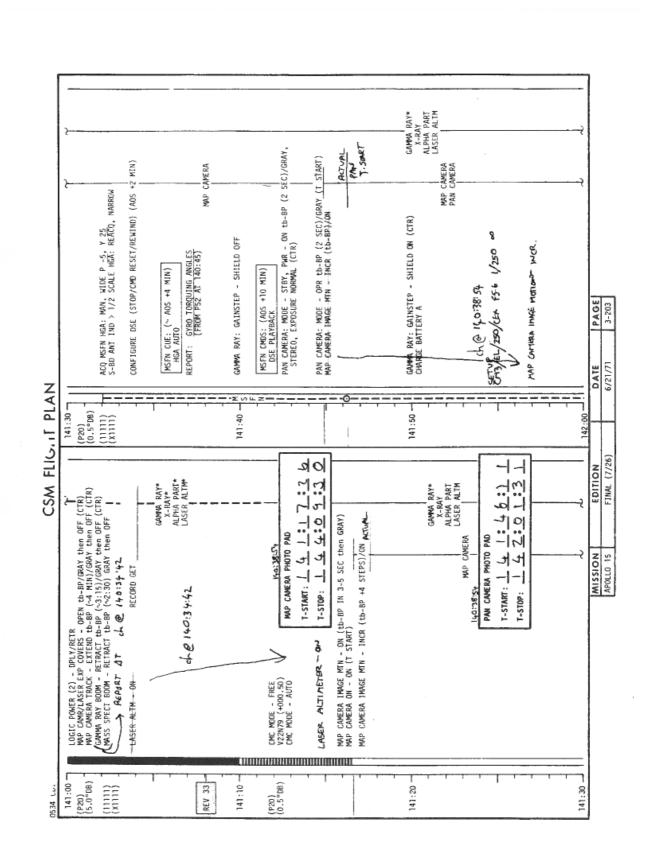
LM FLIGHT PLAN

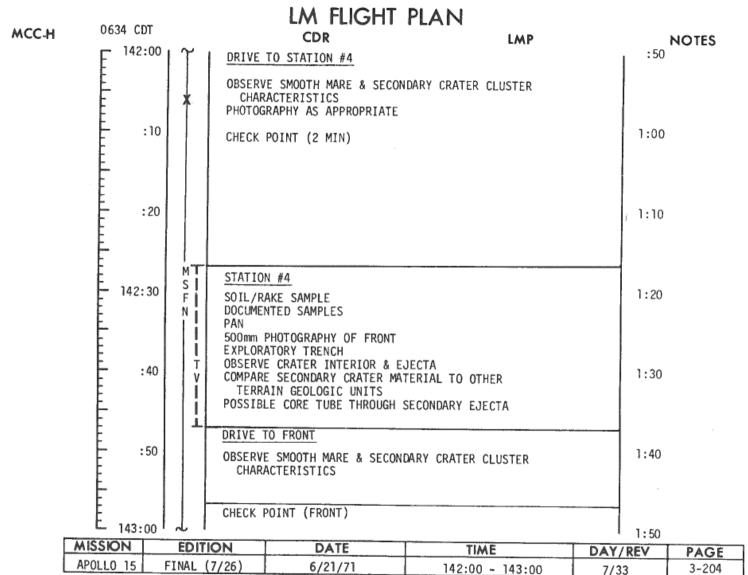




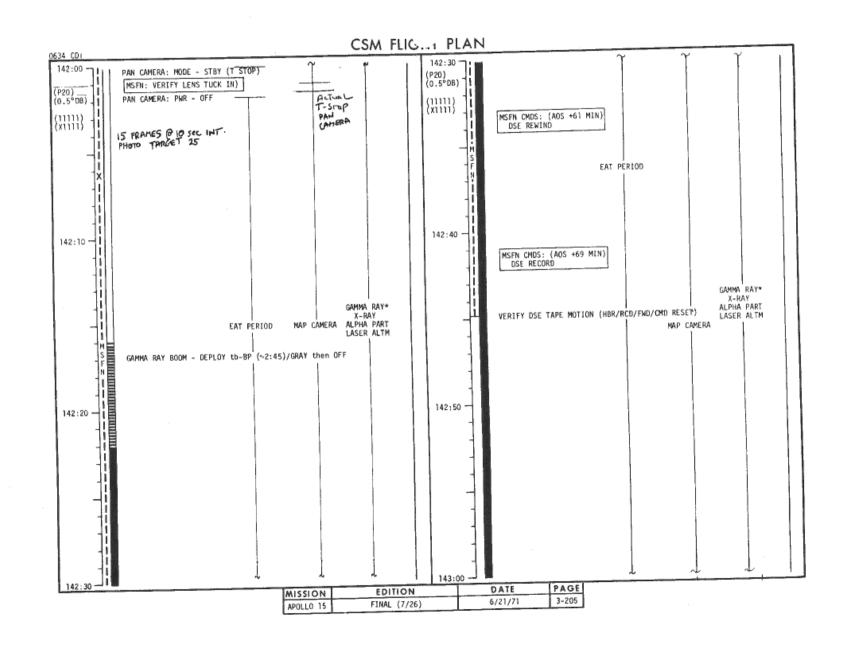


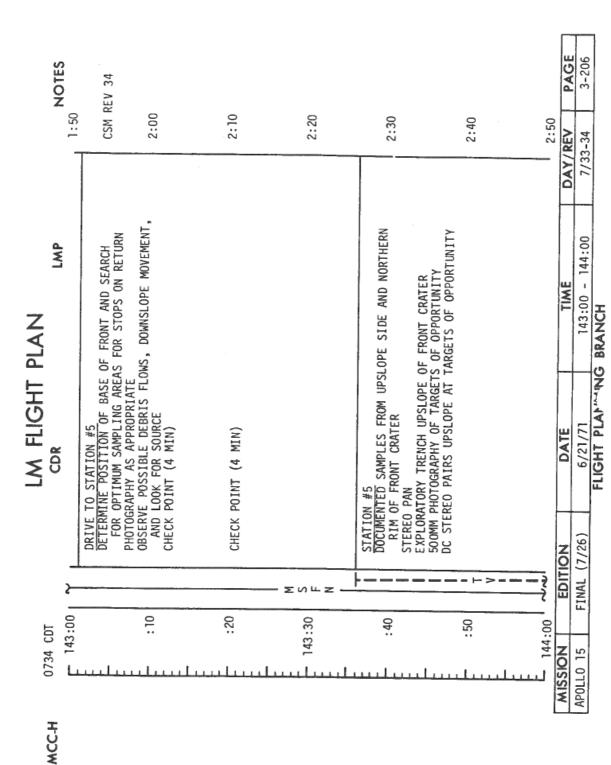
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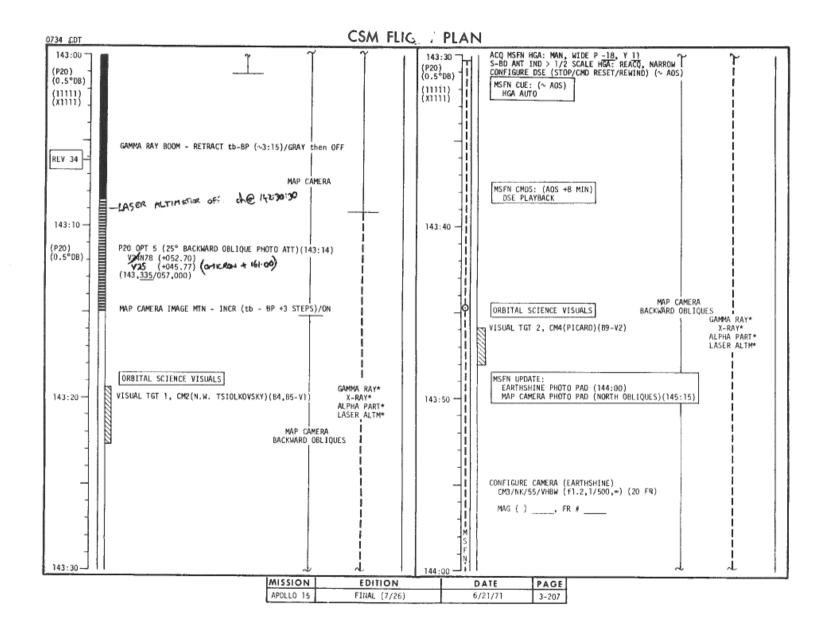




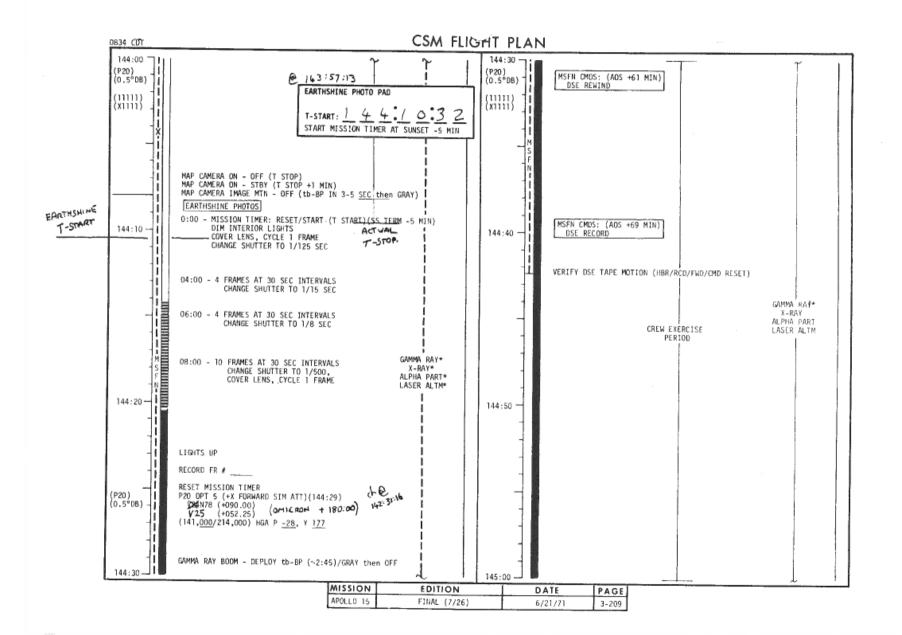
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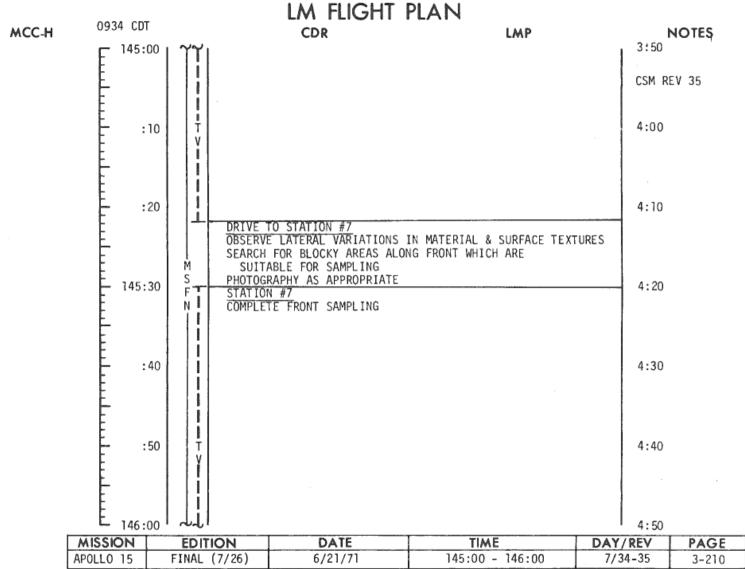




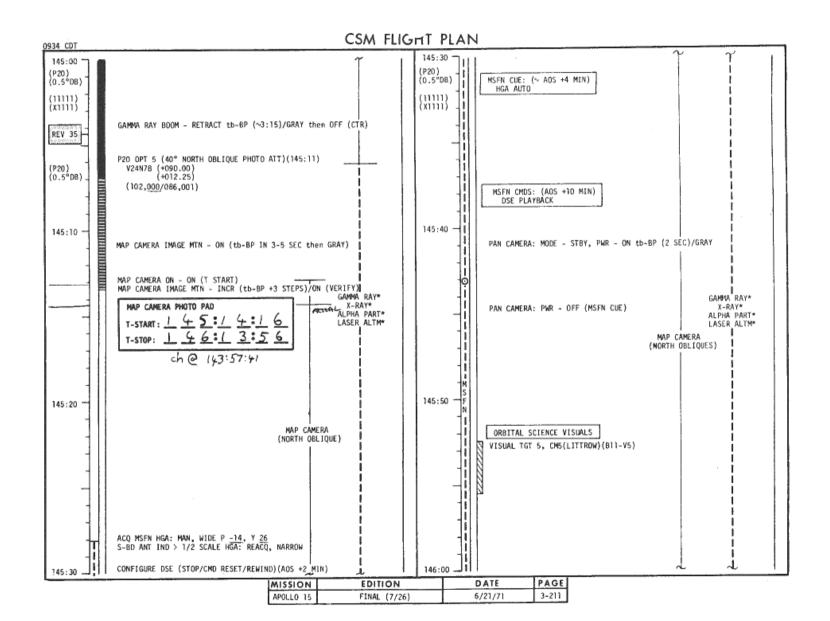


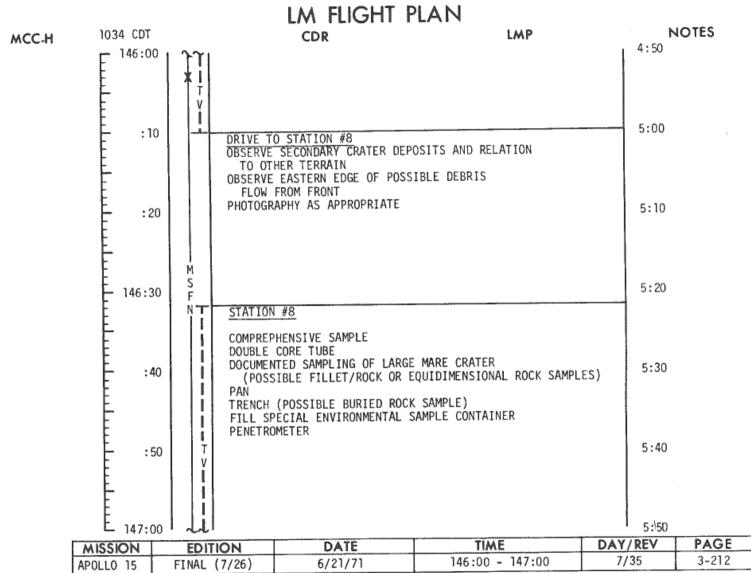
NOTES	~					PAGE	3-208	
9.60	3:00	3:20	3:30	3:40	3:50	REV	7/34	
LM FLIGHT PLAN		TURES		1175		DAY/REV	//	
		DRIVE TO STATION #6 OBSERVE LATERAL VARIATIONS IN MATERIAL & SURFACE TEXTURES SEARCH FOR BLOCKY AREAS ALONG FRONT WHICH ARE SUITABLE FOR SAMPLING PHOTOGRAPHY AS APPROPRIATE		STATION #6 DOCUMENTED SAMPLES OF FRONT MATERIAL PAN EXPLORATORY TRENCH 500MM PHOTOGRAPHY OF BLOCKS, OUTCROPS, ETC. DC STEREO PAIRS OF TARGETS OF OPPORTUNITY DESCRIPTION OF FRONT COMPARISON OF FRONT COMPARISON OF FRONT POSSIBLE CORE TUBE		TIME	144:00 - 145:00	3 BRANCH
				STATION #6 DOCUMENTED SAMPLES OF FRONT MATERIAL PAN EXPLORATORY TRENCH 500MM PHOTOGRAPHY OF BLOCKS, OUTCROPS, ETC. DC STEREO PAIRS OF TARGETS OF OPPORTUNITY DESCRIPTION OF FRONT COMPARISON OF FRONT COMPARISON OF FRONT		DATE	12/12/9	FLIGHT PLA NG
}	×	E OL Z			المهارا	EDITION	FINAL (7/26)	
0834 CDT		144:30	: ₄₀	% 	L 145:00	MISSION	APOLLO 15	
MCC-H							*	<u>.</u>



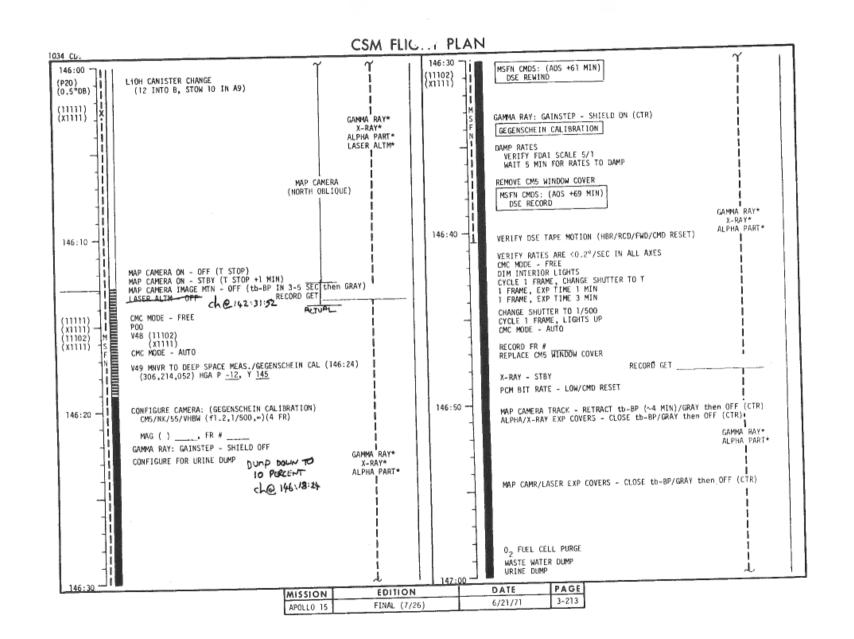


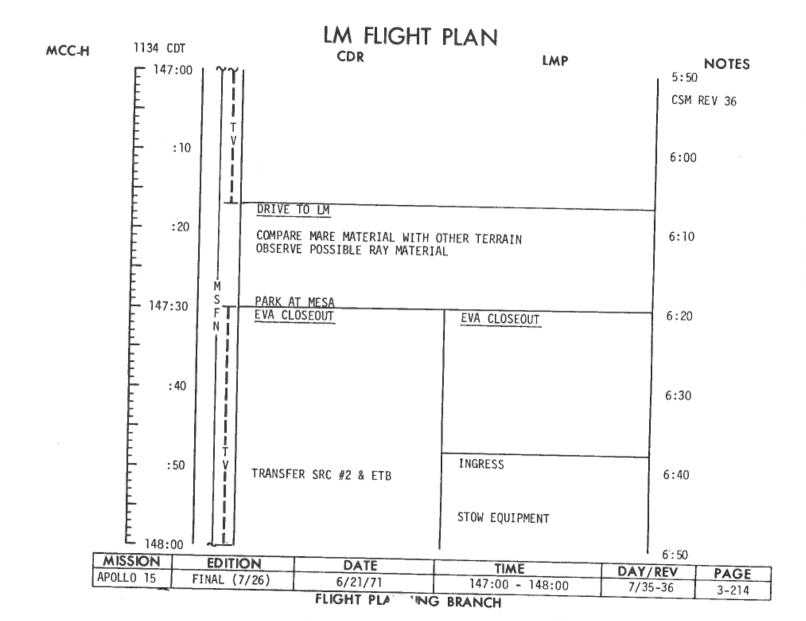
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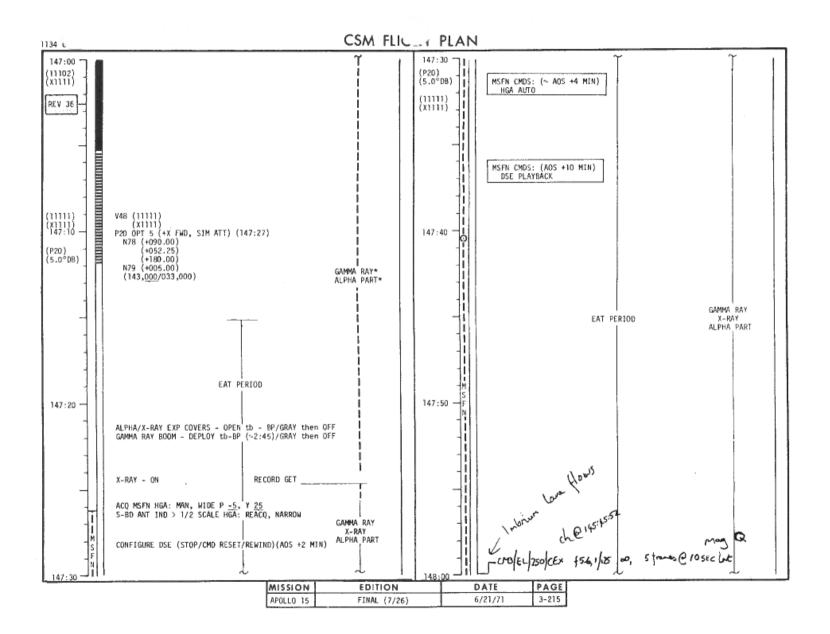


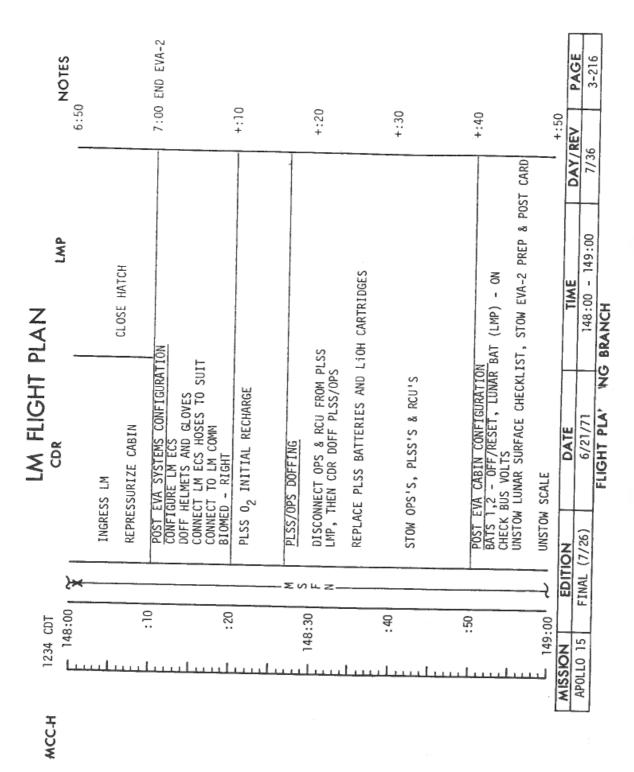


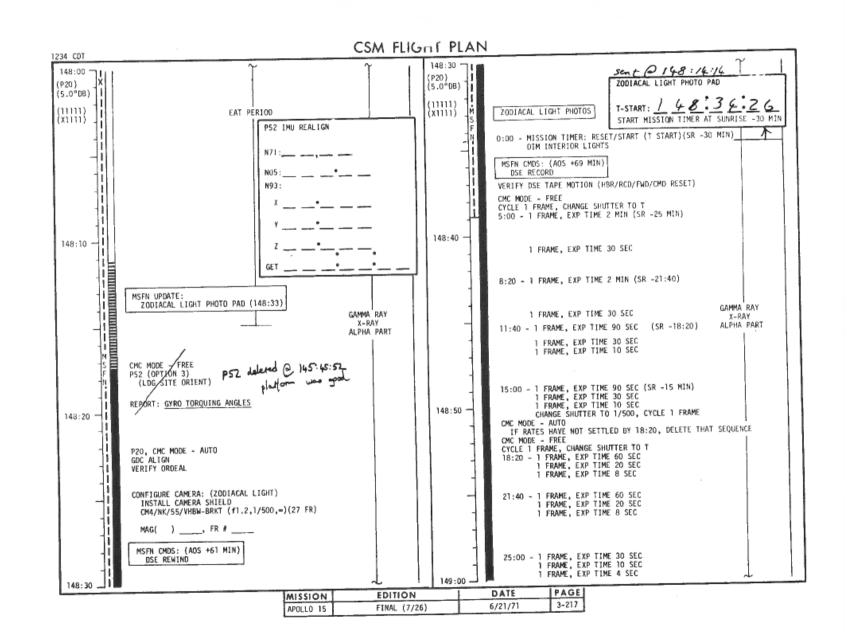
FLIGHT PLA' "NG BRANCH

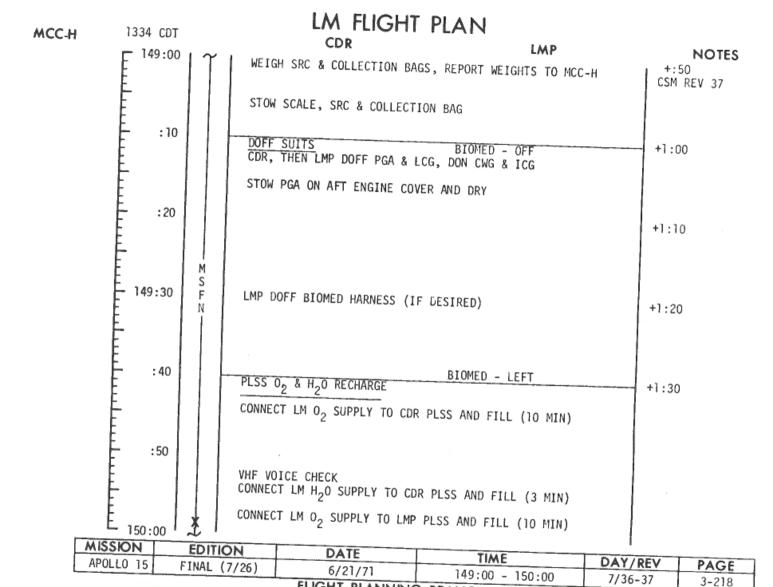




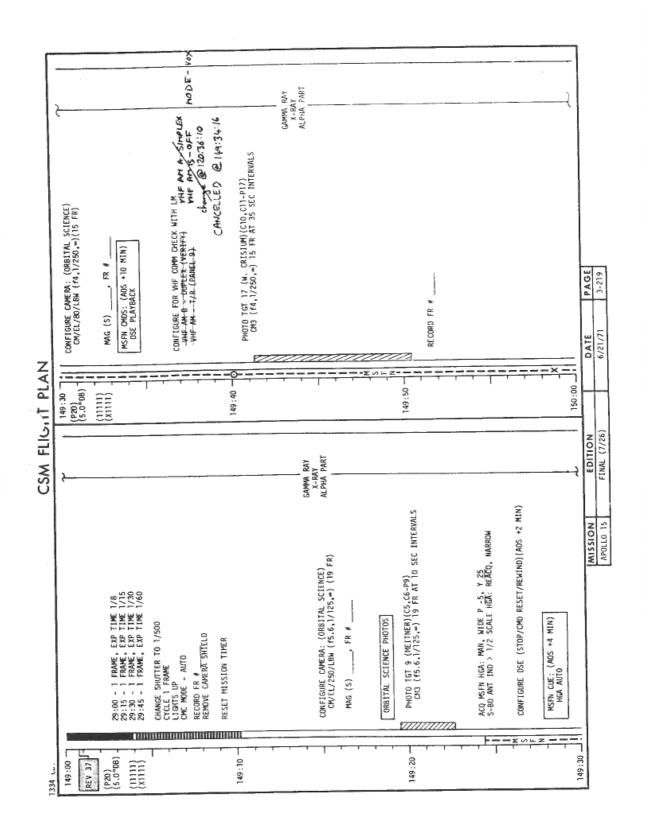


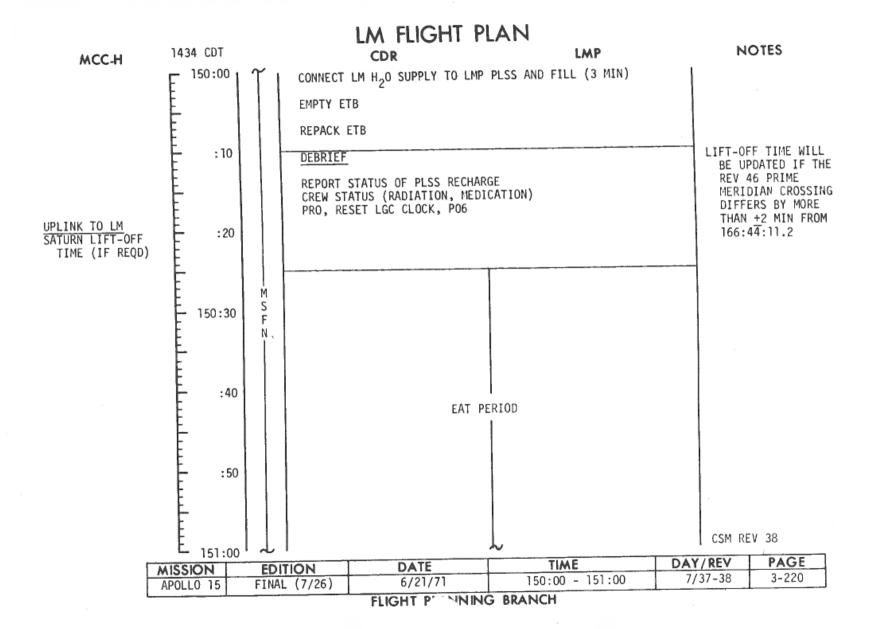


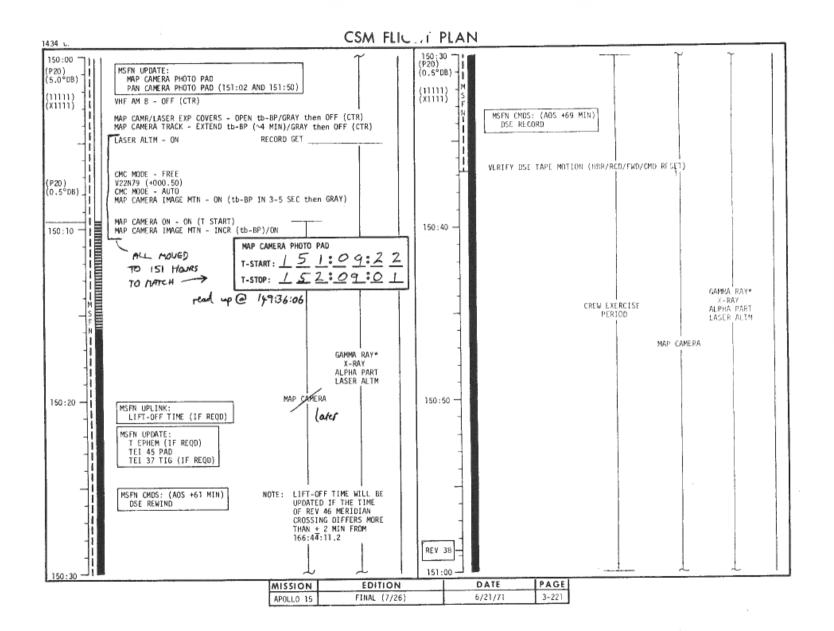


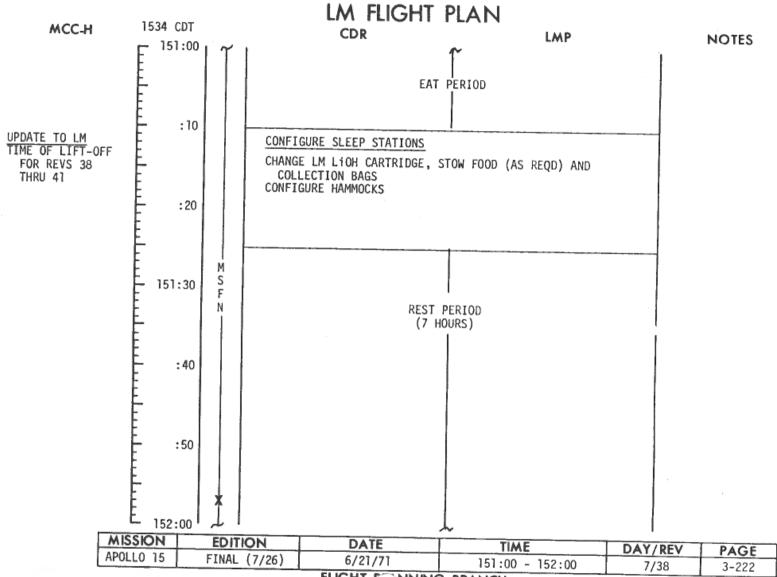


FLIGHT PLANNING BRANCH

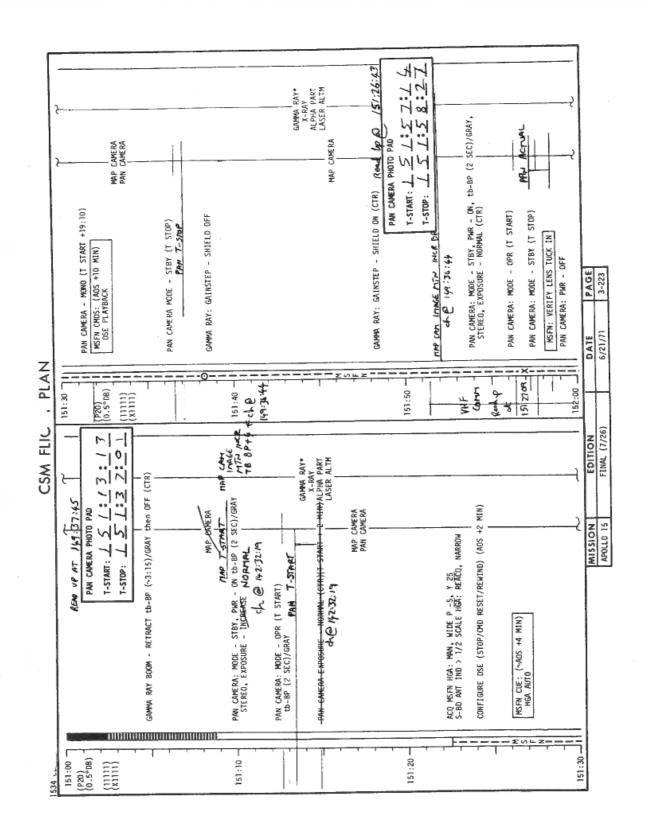


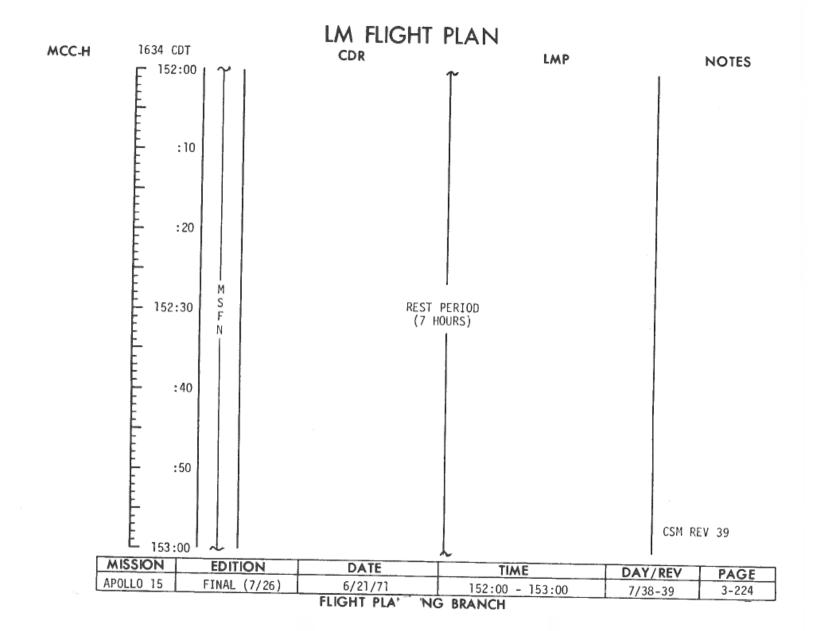


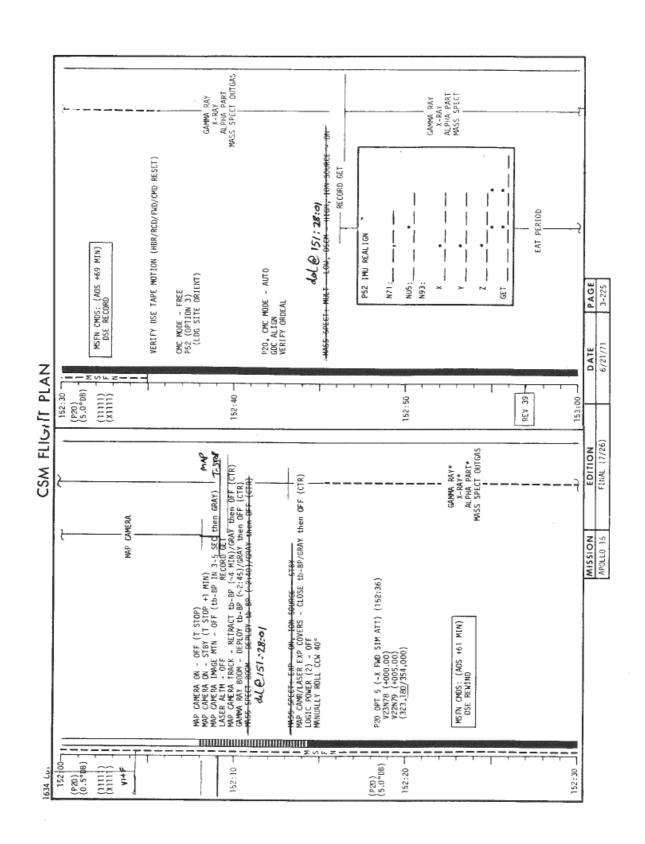


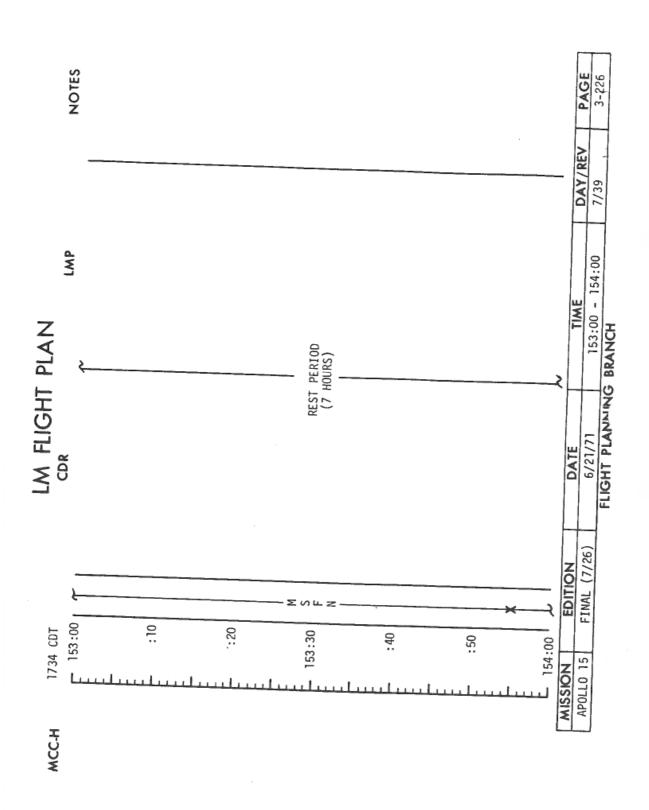


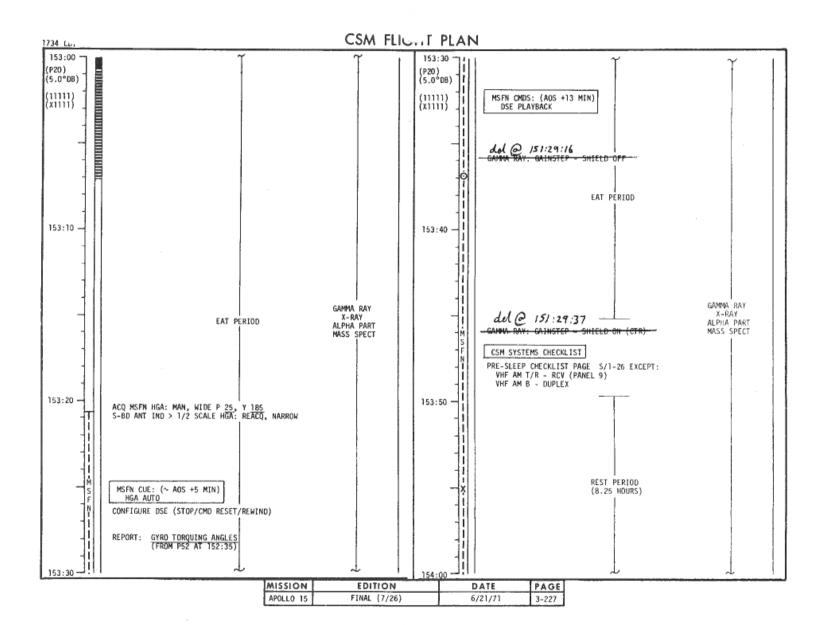
FLIGHT NNING BRANCH

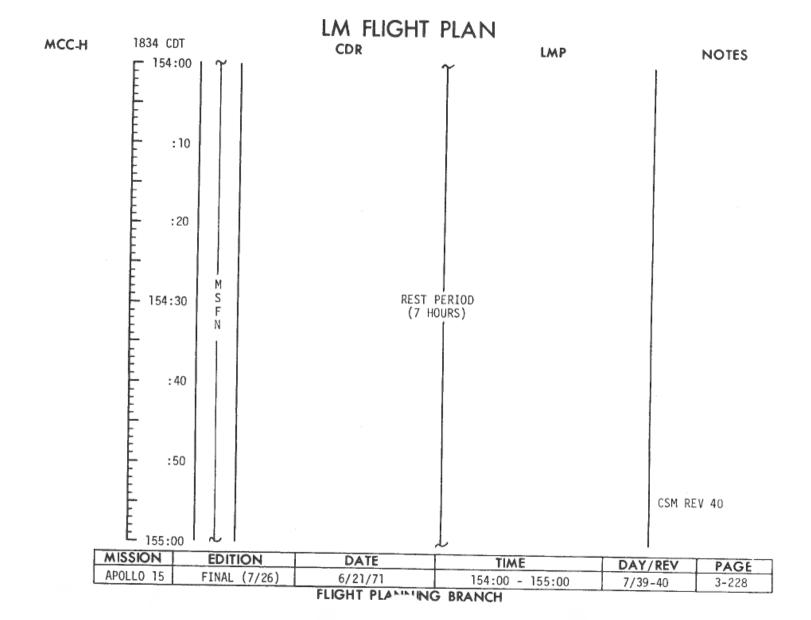


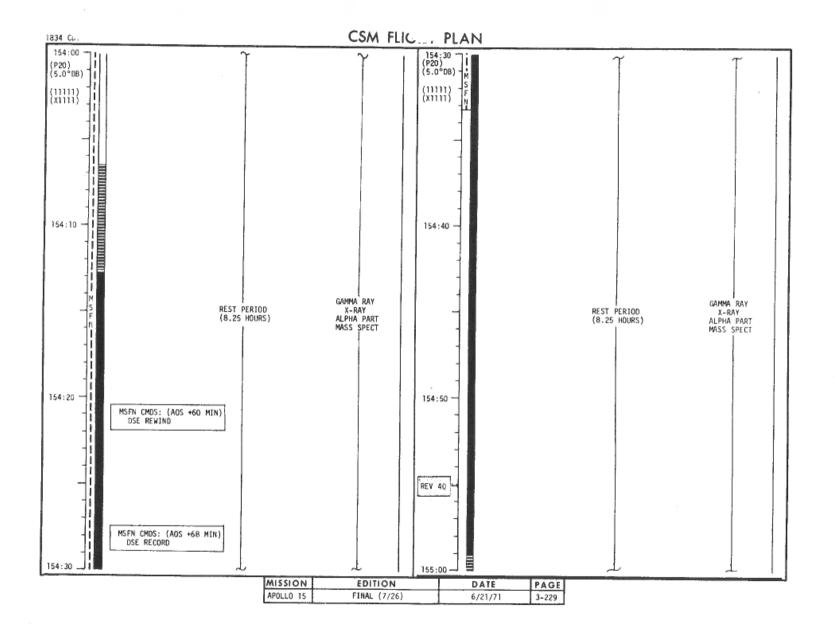


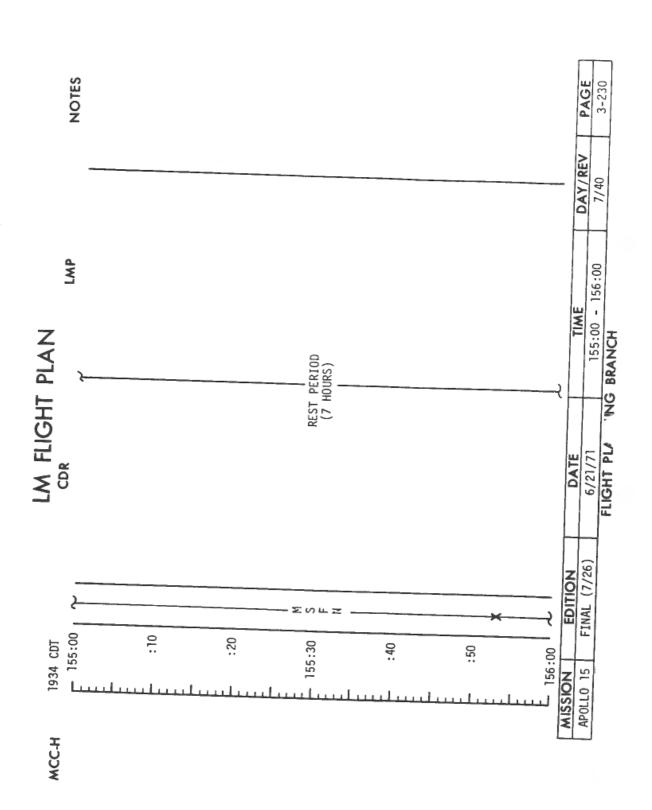


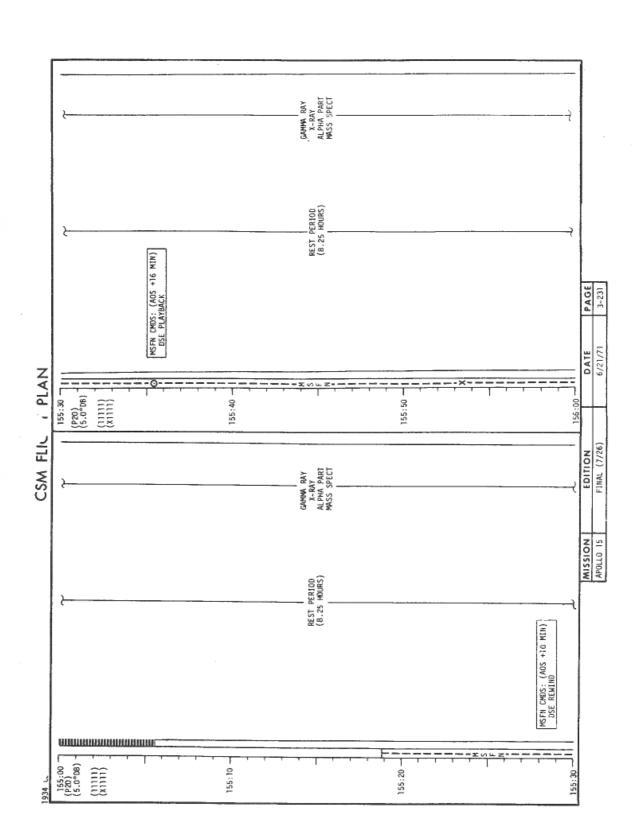


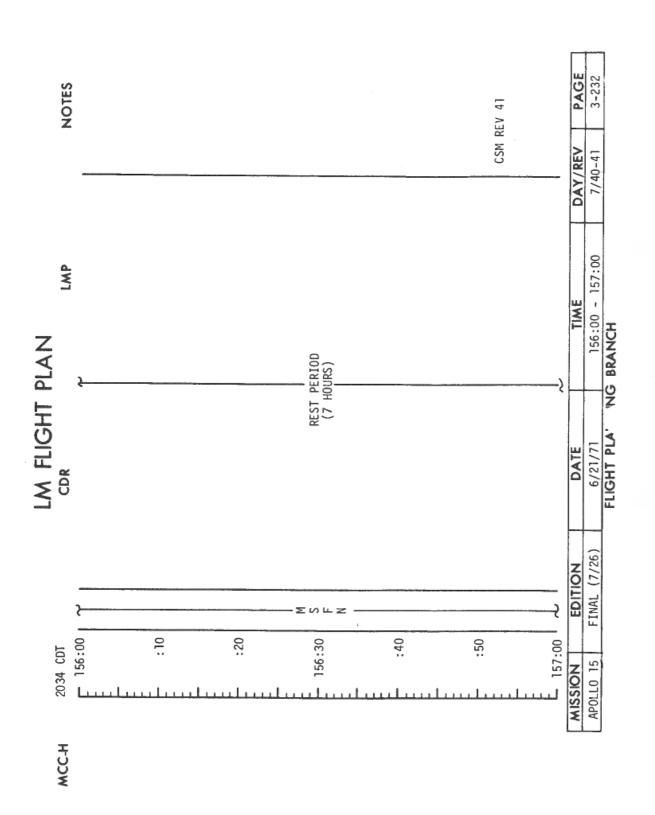


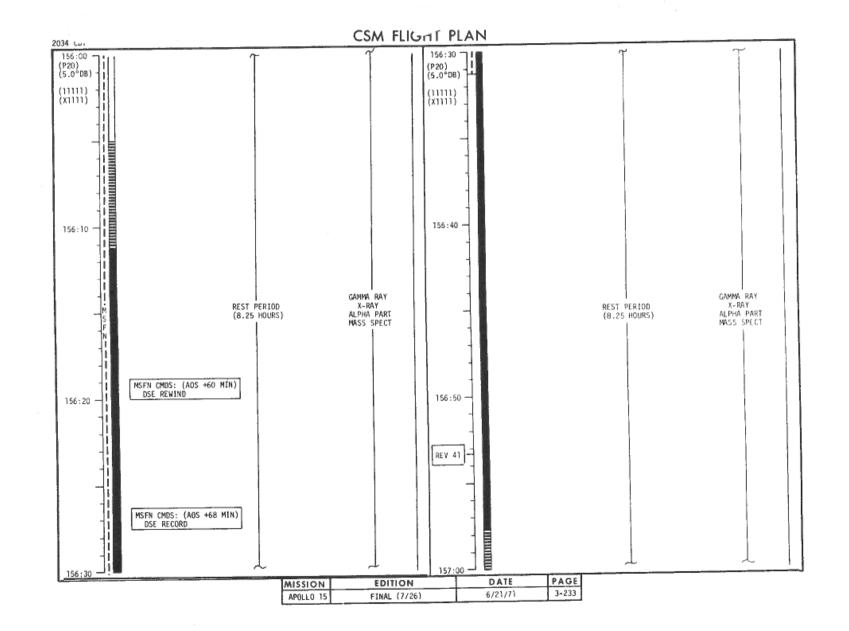


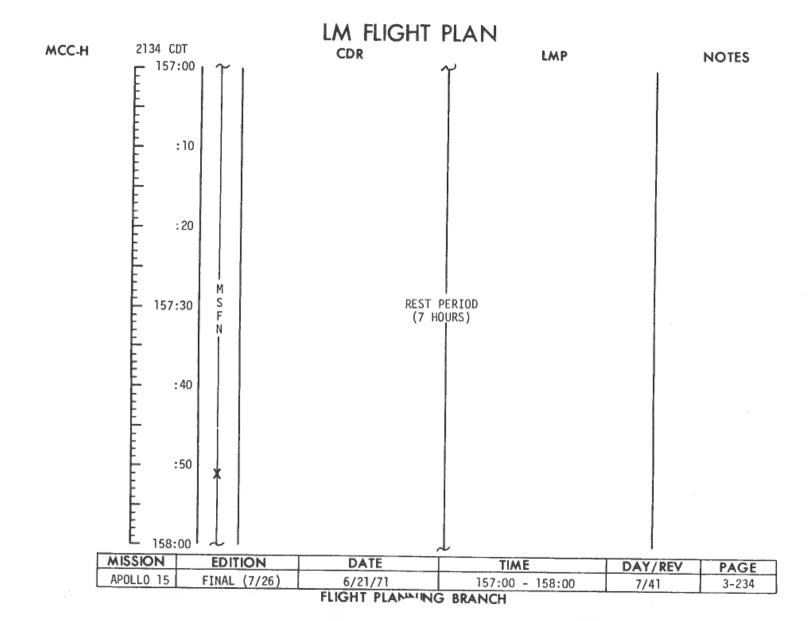


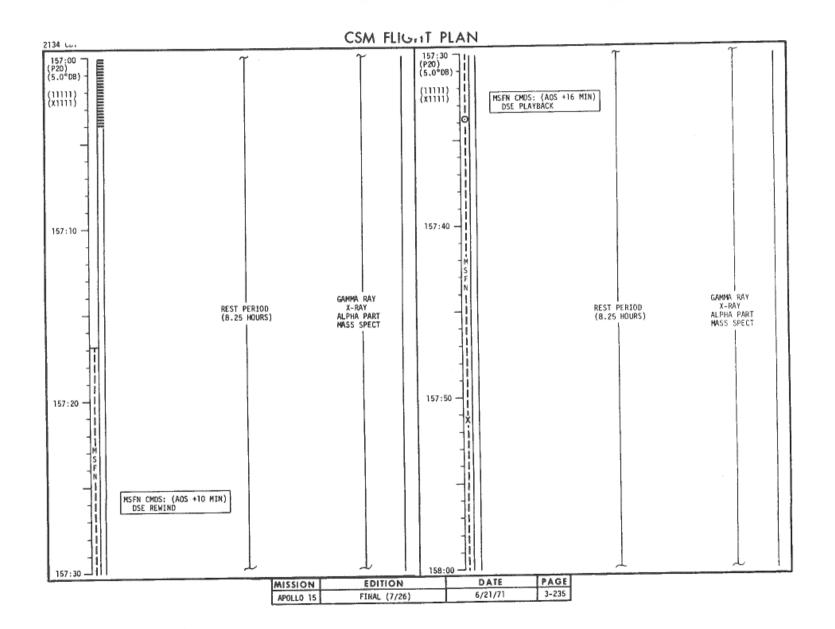


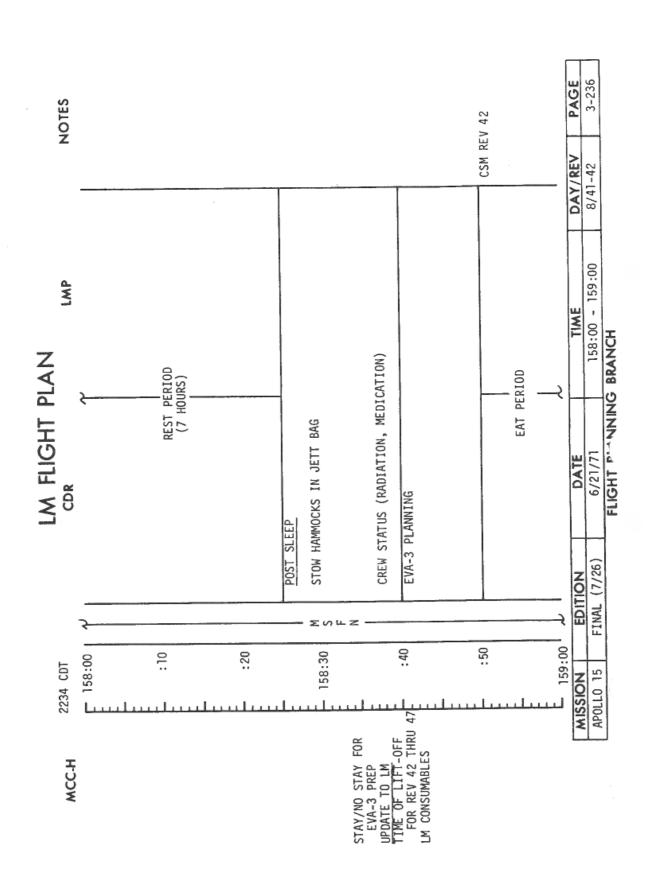


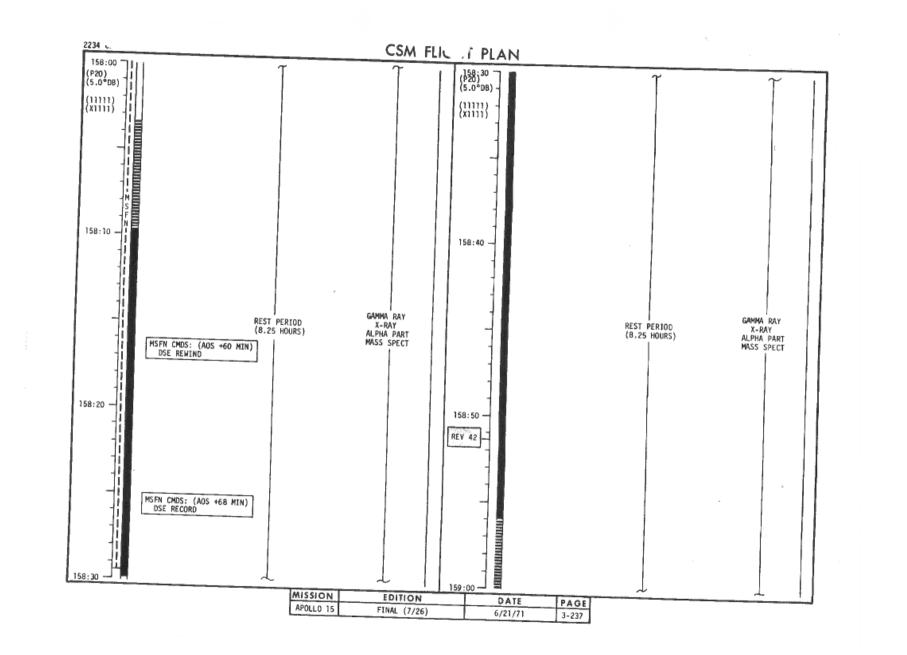


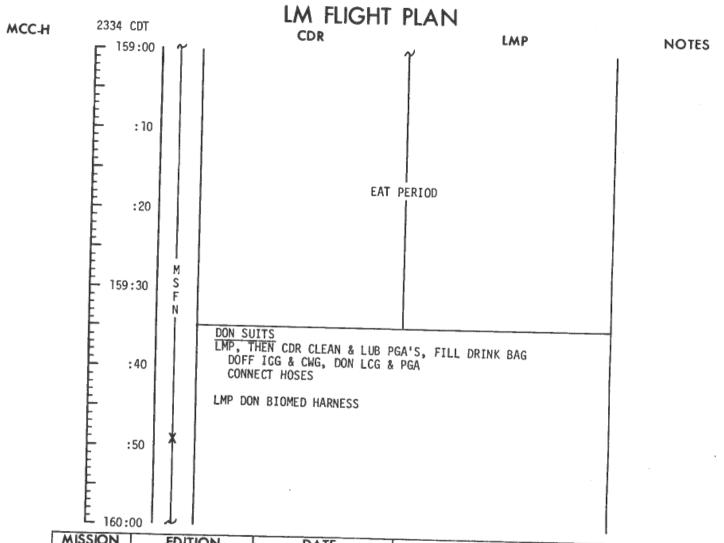






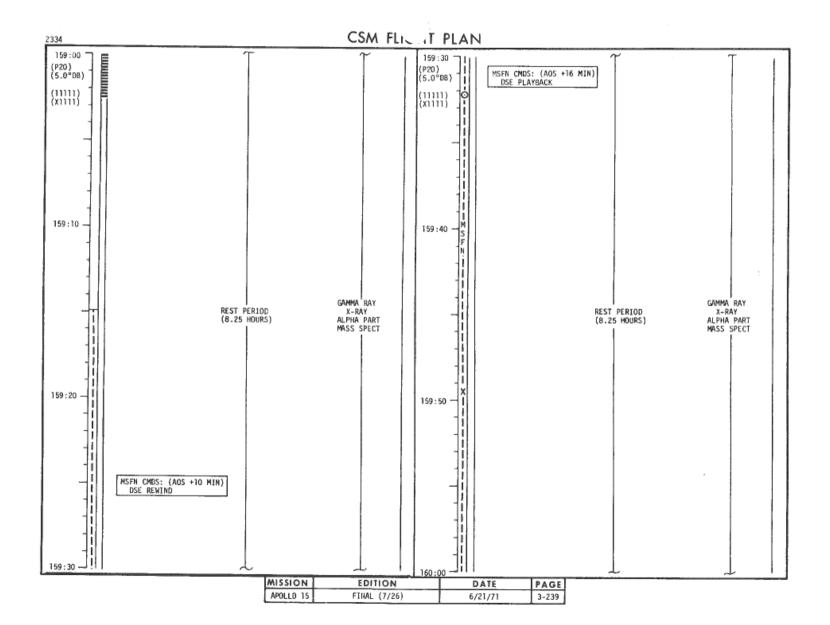


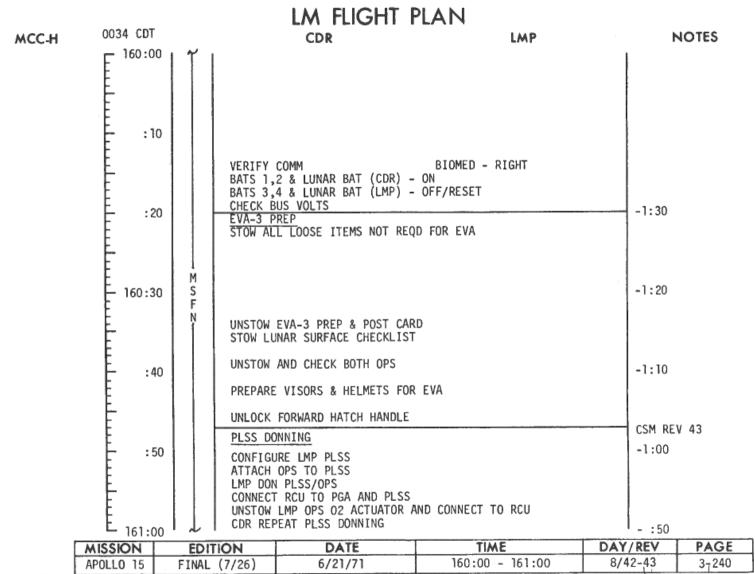




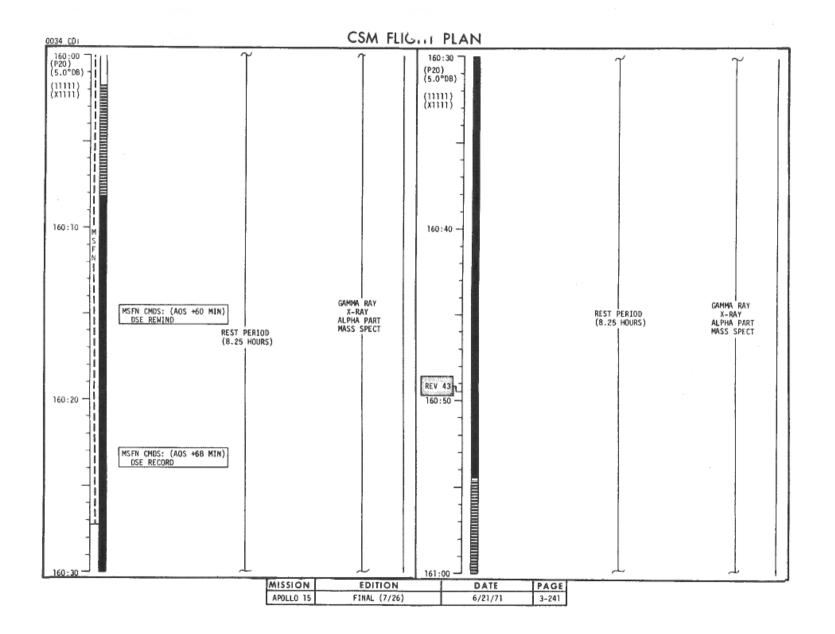
MISSION	EDITION	DATE	TIME	DAYIBEN	
APOLLO 15	FINAL (7/26)			DAY/REV	PAGE
	121112 (7720)	6/21/71	159:00 - 160:00	8/42	3-238

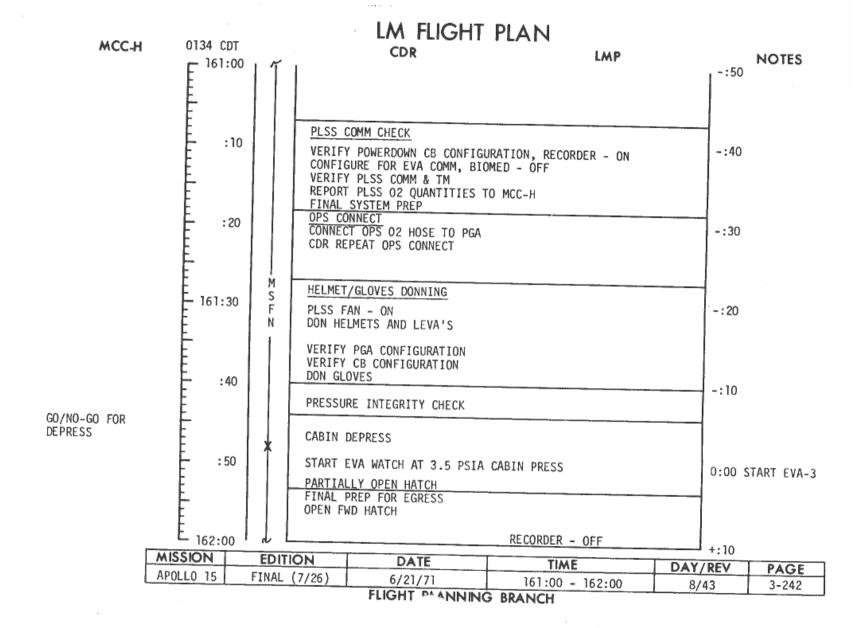
FLIGHT PLAY NG BRANCH

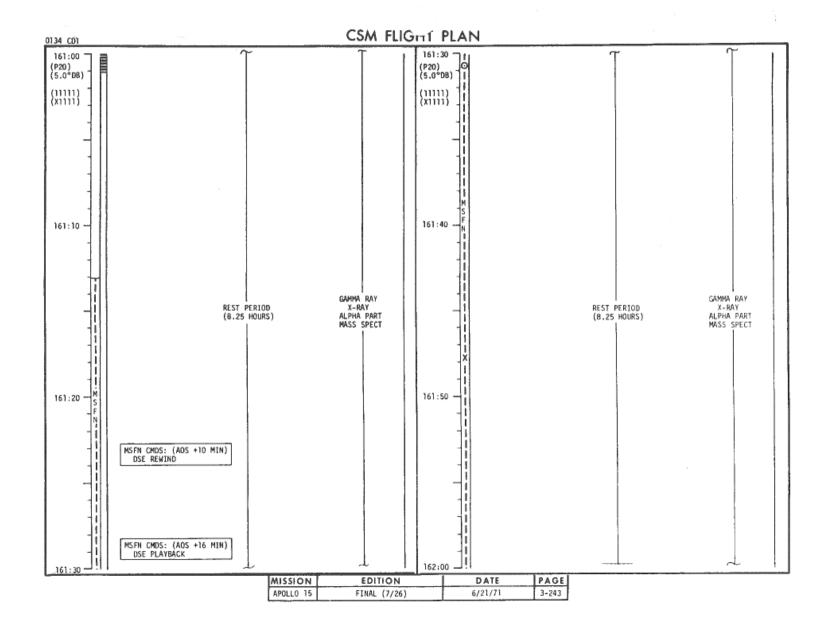




FLIGHT PLA VING BRANCH







LM FLIGHT PLAN

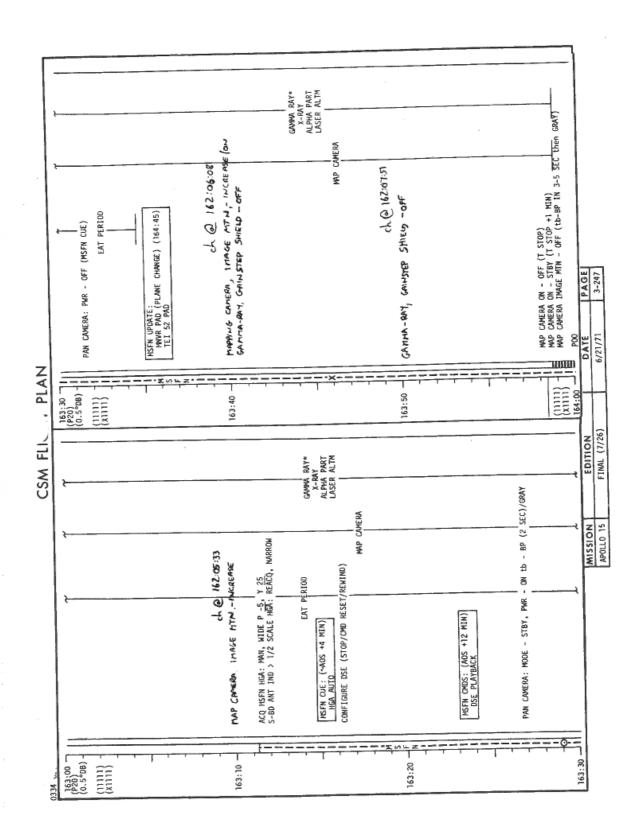
			EN TEIOTH TEAT			
MCC-H	0234 CDT		CDR	LMP		NOTES
	F 162:00	T EGRESS		RELEASE COR PLSS ANTEN	0:10	
	E.	EQUIPM	ENT TRANSFER			
	E		CTIVATION	EGRESS		
	:10		E LMP PLSS ANTENNA ENT PREP	EQUIPMENT PREP	0:20	
	<u>-</u>	T V				
	:20				0:30	
	<u>.</u> :					
	162:30	S LRV NA	/ INITIALIZATION		0:40	
	-		TO STATION #9			
				L TO RILLE RIM MATERIAL		
	:40		MENTARY SAMPLE STOP (PROCK SAMPLE	5 MIN)	0:50	
	-				CSM F	REV 44
	:50				1:00	
	E					
	100.00	STATIO	N #9 E AND DESCRIBE RILLE	AND FAR WALL		
	163:00				1:10	
	MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
	APOLLO 15	FINAL (7/26)	6/21/71	162:00 - 163:00	8/43-44	3-244

FLIGHT PLAP "ING BRANCH

LM FLIGHT PLAN

	O224 CDT		LM FLIGHT	FLAN		
WCC-H			CDR	LMP		NOTES
	F 163:00	COMPRE	PHOTOGRAPHY HENSIVE SAMPLE (DOUBLE) CORE TUBE		1:10	
	:10	DOCUME PENETR	NTED SAMPLING OF CRAT OMETER LE PAN OF EDGE OF CRA		1:20	
	:20	T V	-		1:30	
	163:30	M S I S I I I I I I I			7:40	
	:40				1:50	
	:50	STATION 500MM I DOCUMEN T PAN V	PAN NTED SAMPLE FROM CRAT	ER ON RILLE RIM	2:00	
1	MISSION	EDITION	O STATION #11	711.5	2:10	
		FINAL (7/26)	DATE	TIME	DAY/REV	PAGE
	711 0220 13	1 THAT (//20)	6/21/71 FLIGHT PLAN	163:00 - 164:00	8/44	3-246

FLIGHT PLAN IG BRANCH



LM FLIGHT PLAN

MCC-H	0434 CDT	CDR LAB	
	F 164:00	CONTINUE DESCRIPTION OF RILLE AND RILLE RIM MATERIAL PHOTOGRAPHY AS APPROPRIATE	2:10 NOTES
	:05	STATION #11 OBSERVE AND DESCRIBE RILLE AND FAR RILLE WALL, COMPARE TO PREVIOUS OBSERVATIONS 500MM PHOTOGRAPHY	
	:10	DOCUMENTED SAMPLES OF RILLE RIM AND CRATER AT EDGE OF RILLE PAN COMPARE RILLE RIM MATERIAL TO OTHER TERRAIN	2:20
	164:15	S T F V N II	
	:20		2:30
	:25	DRIVE TO STATION #12 OBSERVE CHANGES IN MATERIAL BETWEEN RILLE RIM, MARE, AND NORTH COMPLEX OBSERVE CHARACTERISTICS OF CRATER CHAIN ORIGINATING IN CHAIN CRATER OBSERVE POSSIBLE SECONDARY CRATERS	
MI	SSION	EDITION DATE	2:40

MISSION	EDITION	DATE		2:40		
APOLLO 15			TIME	DAY/REV	PAGE	
NI OLLO 13	FINAL (7/26)	6/21/71	164:00 - 164:30	8/44	3-248	
		FLICHT DLAK SIG	D. D. D. L. L. C.	0) ++	L 3-240	

FLIGHT PLAY NG BRANCH

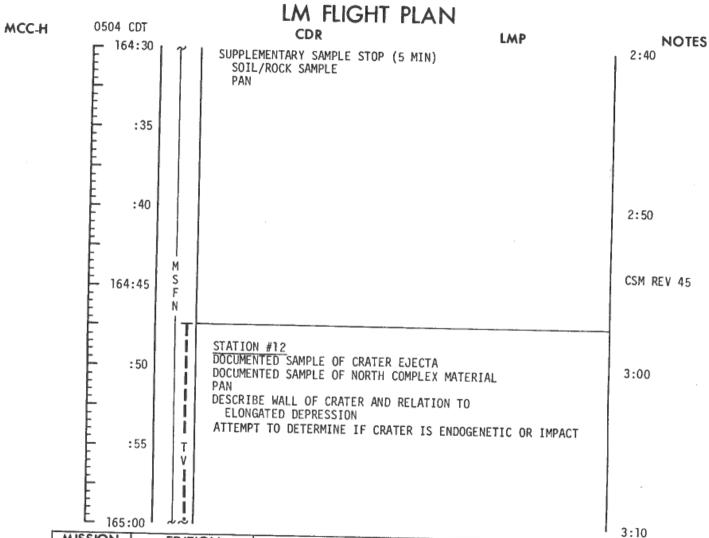
CSM FLIGHT PLAN 0434 CDT 164:00 (11111) (X1111) GAMMA RAY EXP - OFF X-RAY - OFF, a RAY/X DR - a OFF LASER ALTM - OFF RECORD GET MASS SPECT: EXP - OFF P52 IMU REALIGN MAP CAMERA TRACK - RETRACT tb-BP (~4 MIN)/GRAY then OFF (CTR)
ALPHA/X-RAY EXP COVERS - CLOSE tb-BP/GRAY then OFF (CTR) MSFN UPLINK: CSM S.V. PLANE CHANGE TGT LOAD DESIRED ORIENT (PLANE CHANGE) N93: MAP CAMR/LASER EXP COVERS - CLOSE tb-8P/GRAY then OFF (CTR) LOGIC POWER (2) - OFF cb SCS CONTR DIR 1 MNB - CLOSE cb SCS CONTR DIR 2 MNA - CLOSE ENABLE ALL JETS V49 MNVR TO P52 ATT (164:15) (180,214,045) HGA P <u>-10</u>, Y <u>206</u> 164:10 MSFN CMDS: (AOS +61 MIN) DSE REWIND P52 (OPTION 3) (LDG SITE ORIENT) REPORT: GYRO TORQUING ANGLES 164:20 P52 (OPTION 1) (PLANE CHANGE ORIENT) MSFN CMDS: (AOS +69 MIN) DSE RECORD VERIFY DSE TAPE MOTION (LBR/RCD/FWD/CMD RESET) GDC ALIGN VERIFY ORDEAL 164:30 -MISSION EDITION DATE PAGE

APOLLO 15

+ FINAL (7/26)

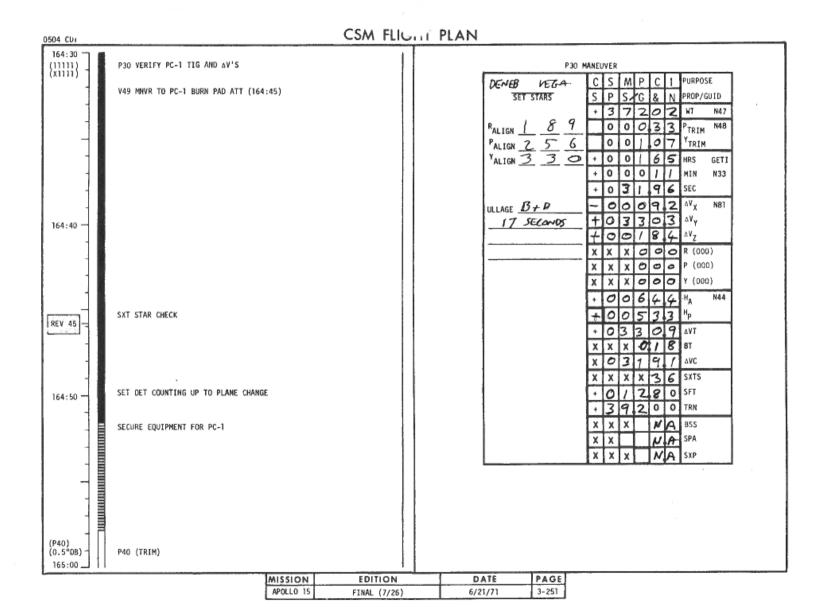
6/21/71

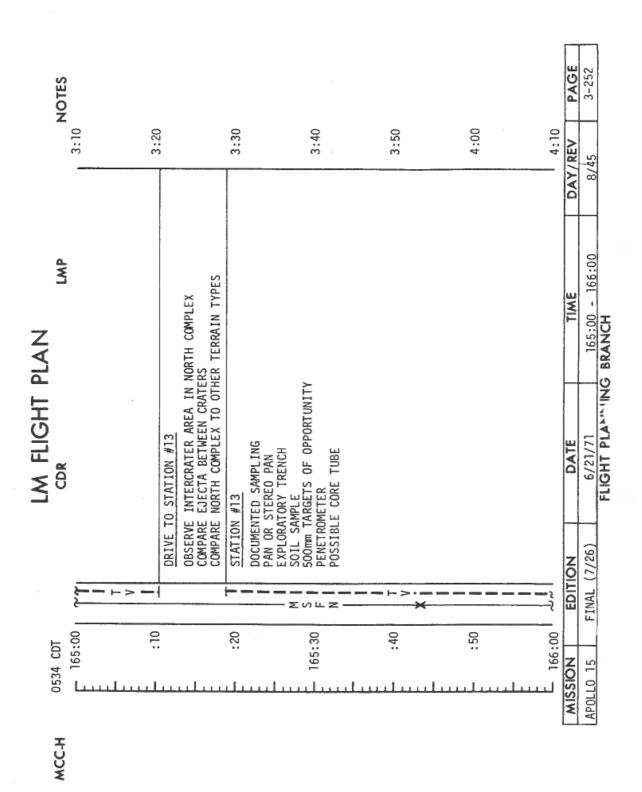
3-249

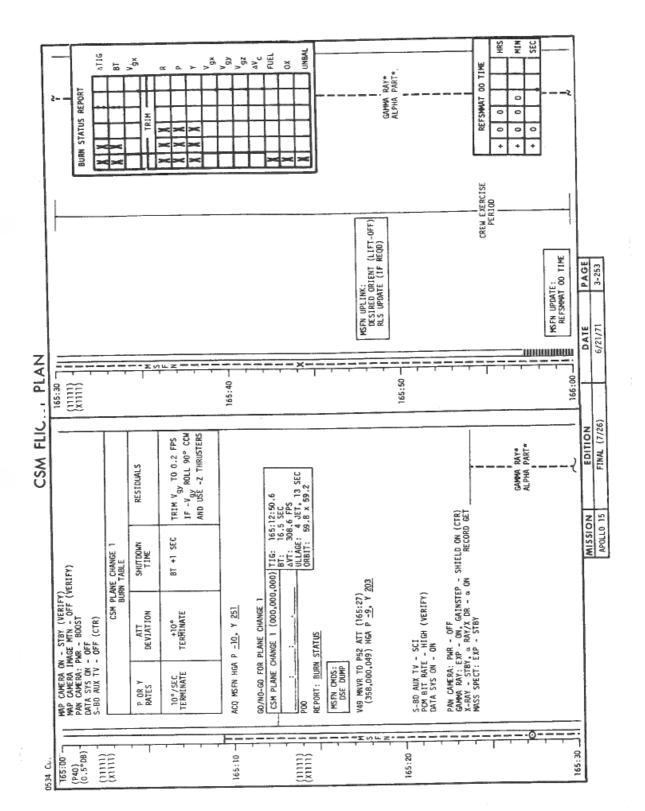


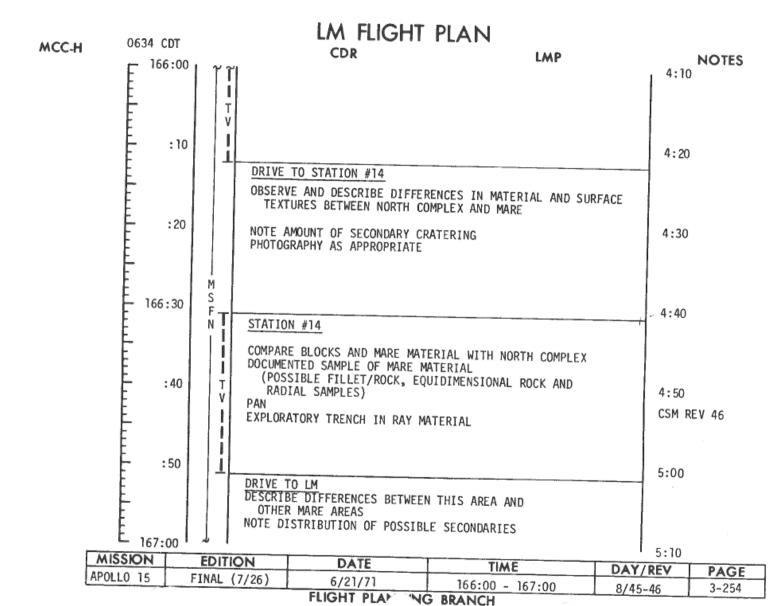
WISSION	EDITION	DATE	TIME	DAY/REV	DAGE	ı
APOLLO 15	FINAL (7/26)	6/21/71	164:30 - 165:00	8/44-45	9AGE 3-250	
		ELICUT DI ANIMINI		0/44-45	3-230	ı

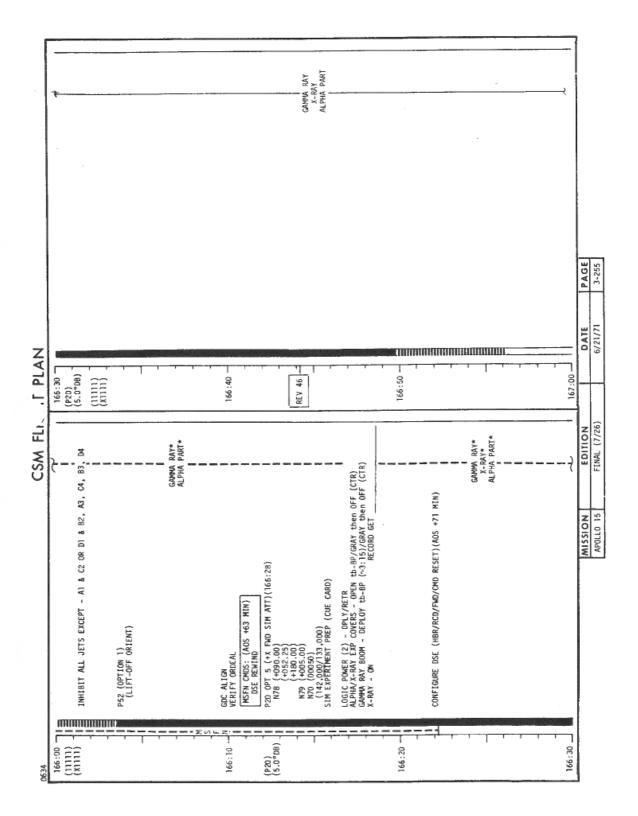
FLIGHT PLAY VING BRANCH

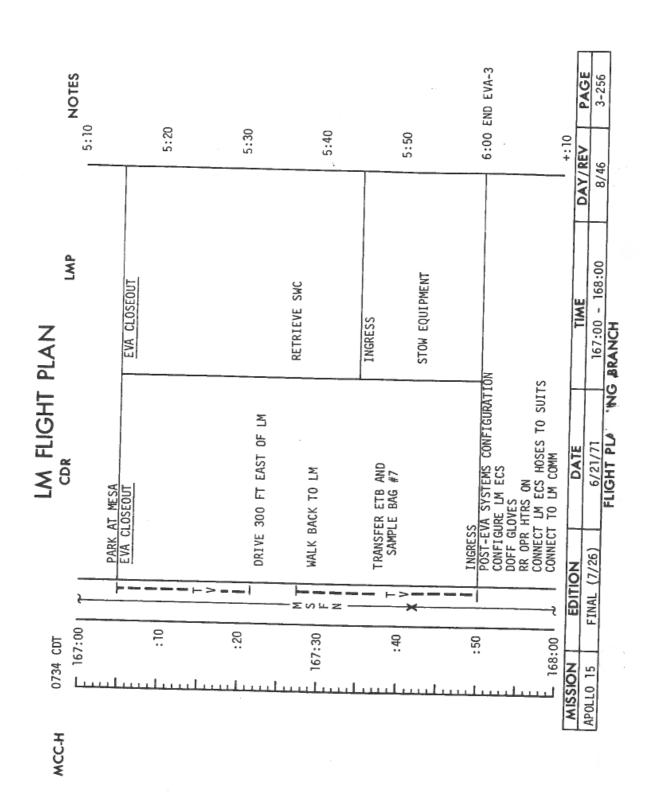


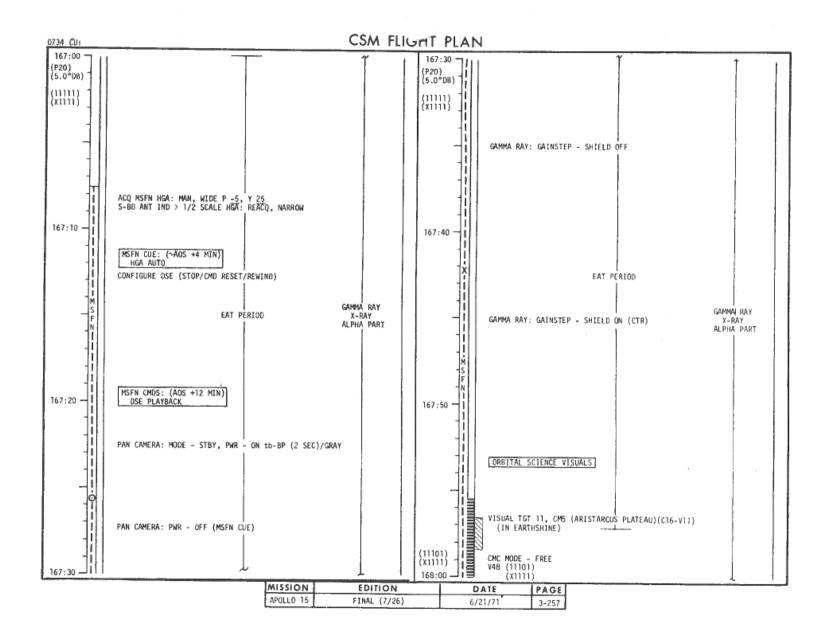


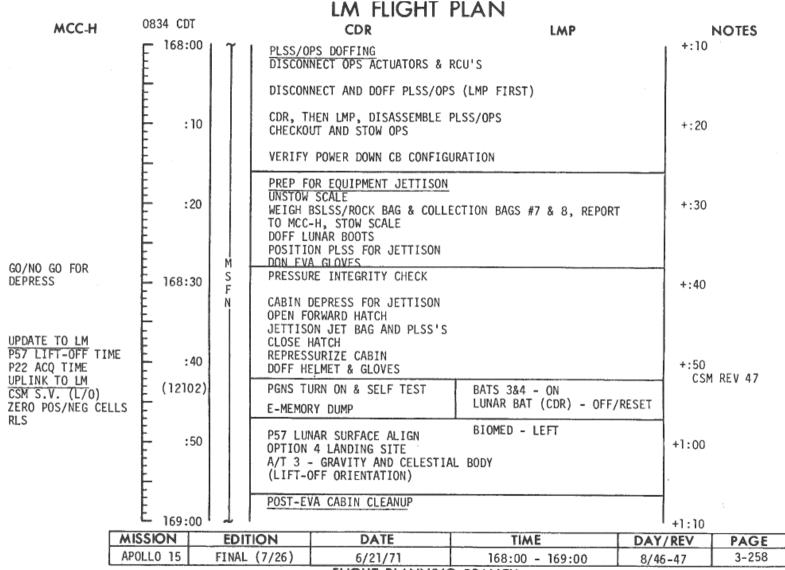




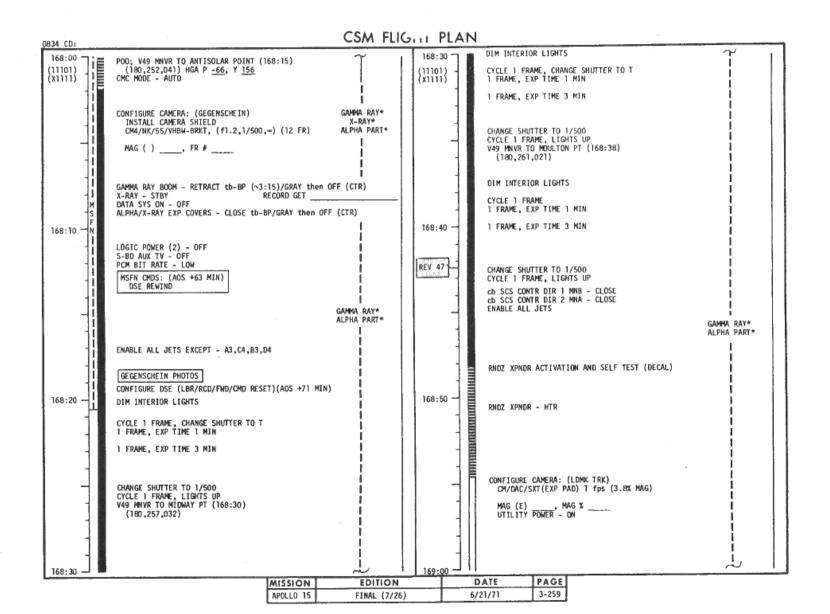


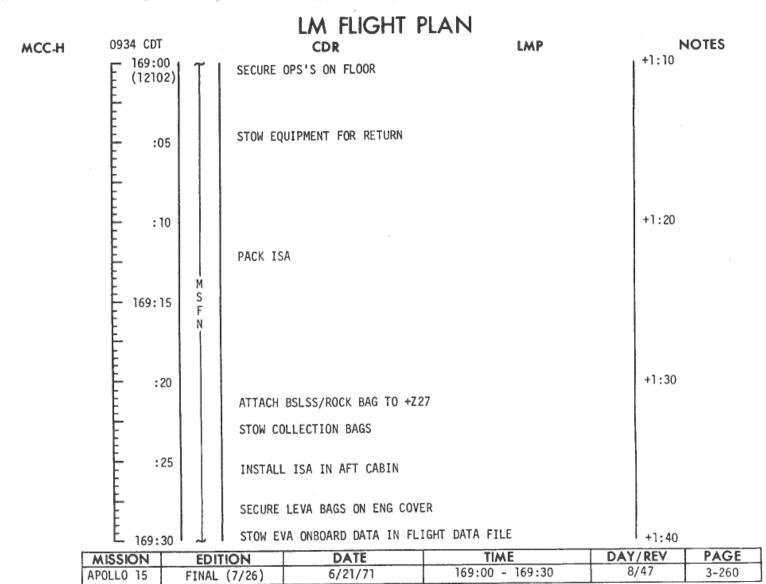




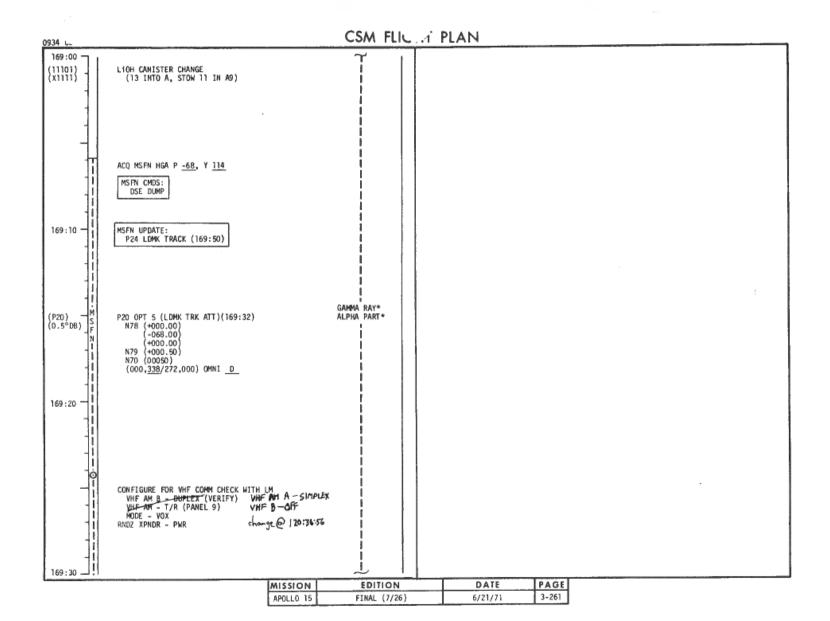


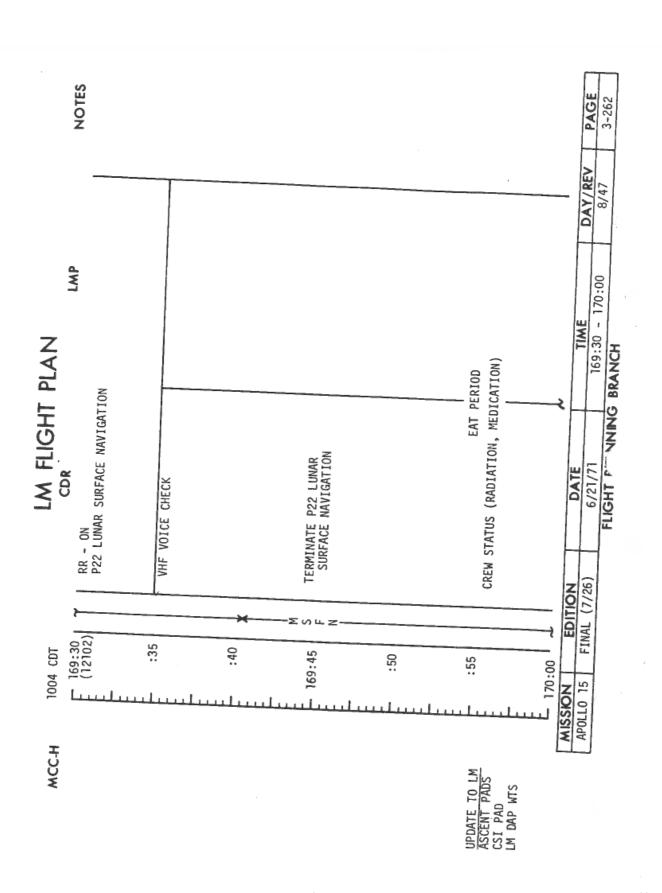
FLIGHT PLANNING BRANCH

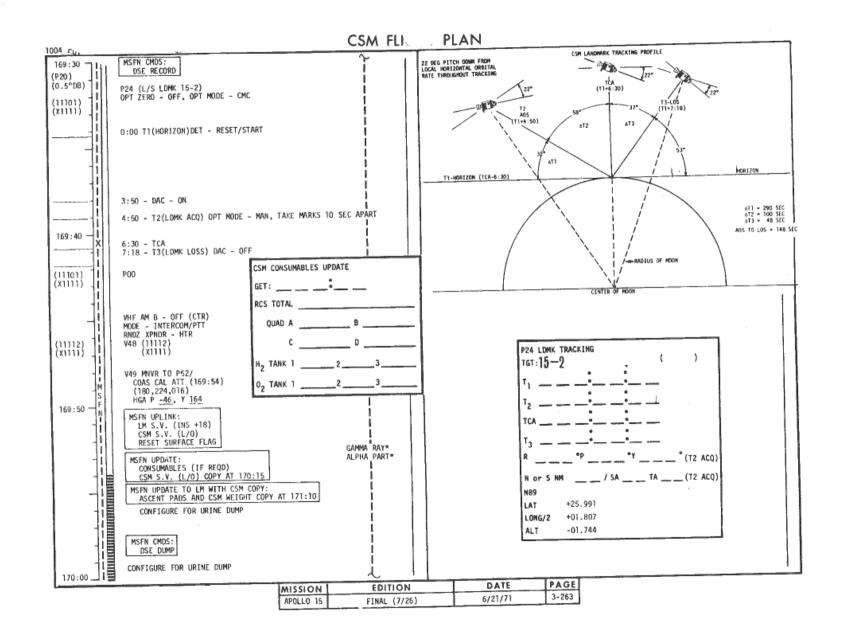


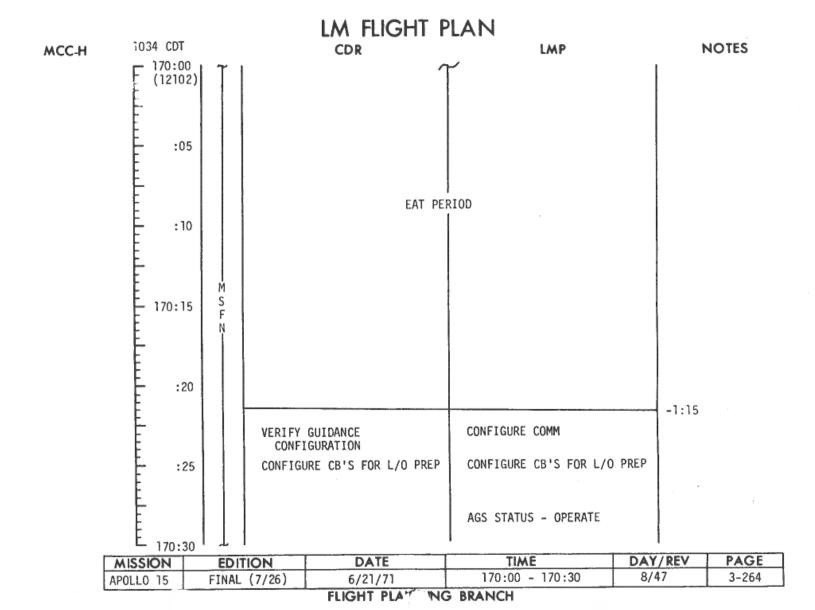


FLIGHT PLAPTING BRANCH

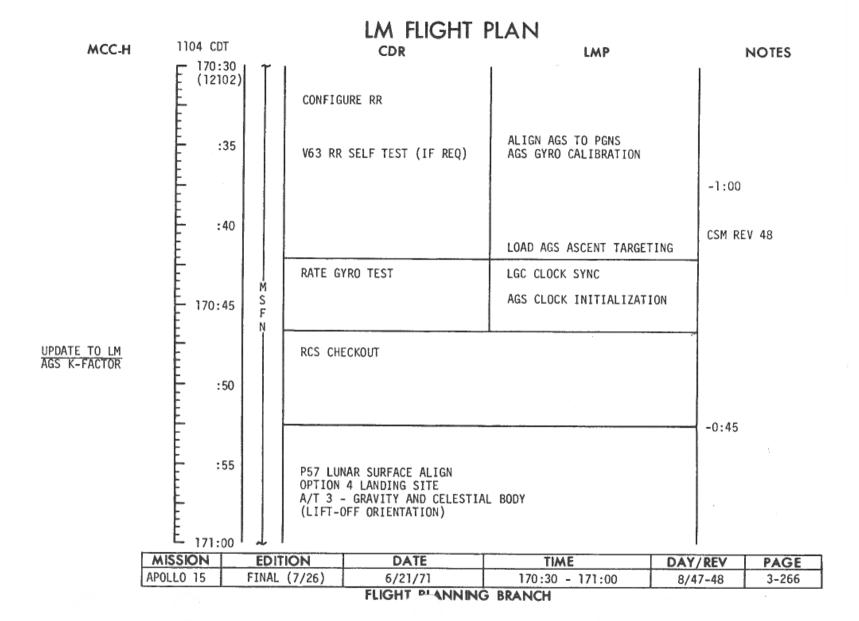


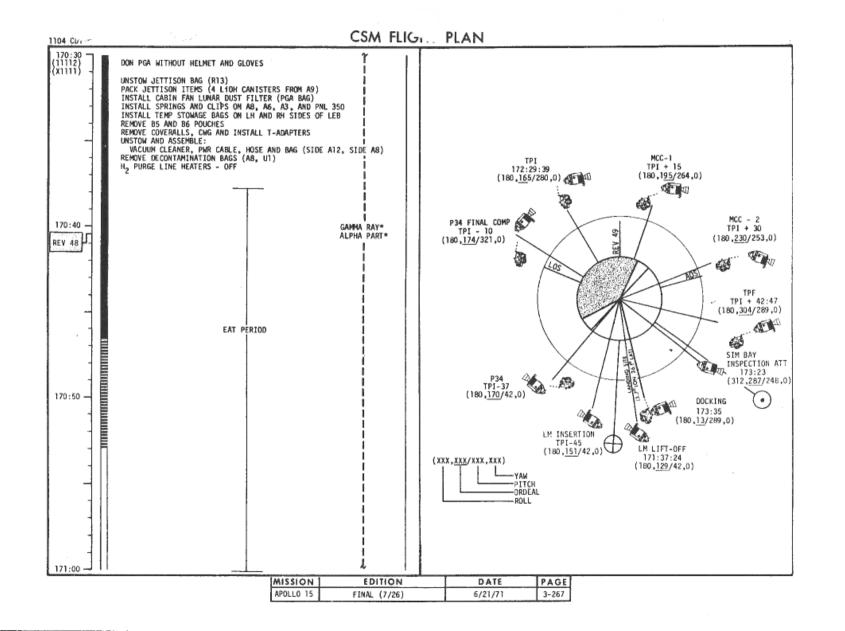




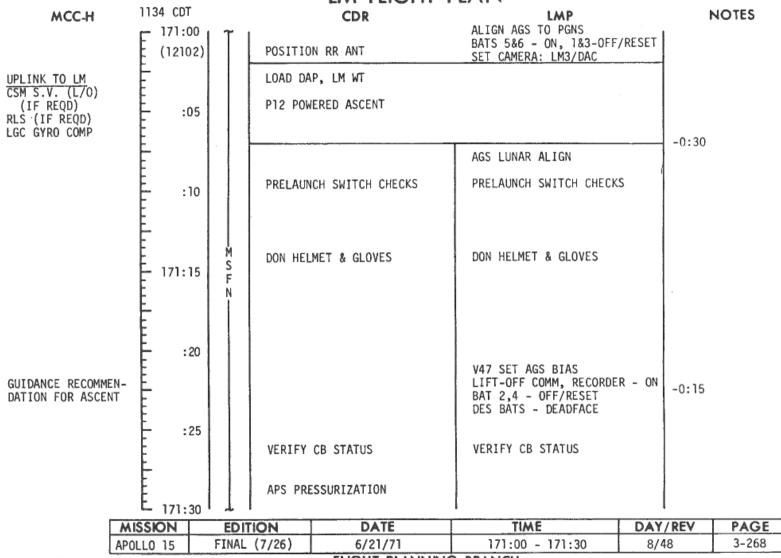


CSM FLIGHT PLAN 1034 CDT P52 IMU REALIGN H₂ PURGE LINE HEATERS - ON 170:00 715 (11112) (X1111) P52 (OPTION 3) (LIFT-OFF ORIENT) REPORT: GYRO TORQUING ANGLES P52 (COAS CALIB) USE STAR NO. 41 COAS CALIB - N92 SHAFT: ____ 170:10 GDC ALIGN VERIFY ORDEAL GAMMA RAY* CSM S.V. (L/0) P27 UPDATE ALPHA PART* CONFIGURE CAMERAS: (DOCKING) PURP V CM2/DAC/18/CEX-BRKT,MIR (T8,1/250,~) 6 fps (40% MAG) : :-; GET INDEX 304 01 INDEX MAG (C) MAG % ____ 02 MSFN CMDS: 03 DSE RECORD 04 VERIFY DSE TAPE MOTION (LBR/RCD/FWD/CMD RESET) 05 MSFN ENABLES MSFN S-BAND RELAY 07 CM/EL/80/CEX (f8,1/250,FOCUS) (10 FR) 170:20 -10 MAG (Q) ____, FR # ____ 11 12 13 CM4/TV-BRKT (f44, PEAK) 11 MIN (Z00M - 75MM) 14 15 H₂ AND O₂ FUEL CELL PURGE. WASTE WATER DUMP URINE DUMP 17 20 21 22 23 24 170:30 _ PAGE DATE EDITION MISSION 3-265 6/21/71 FINAL (7/26) APOLLO 15

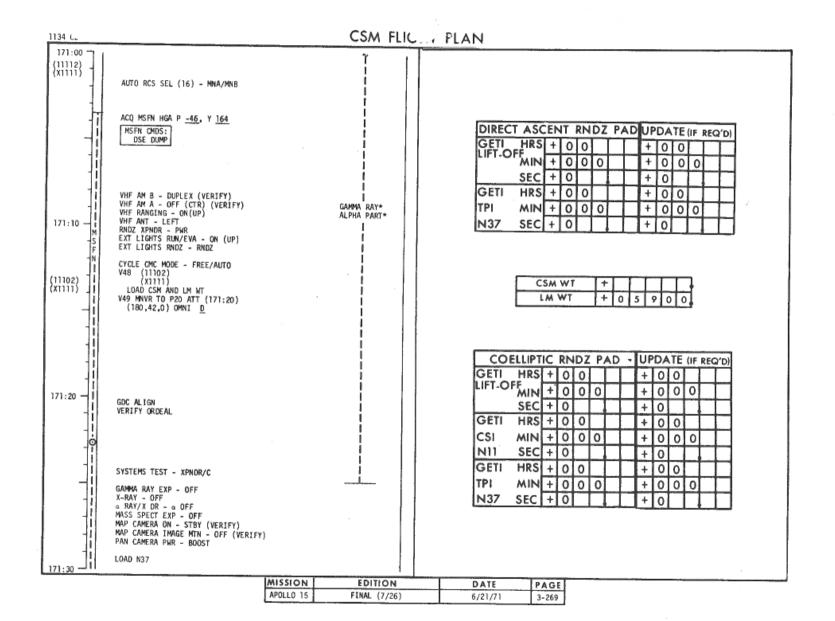


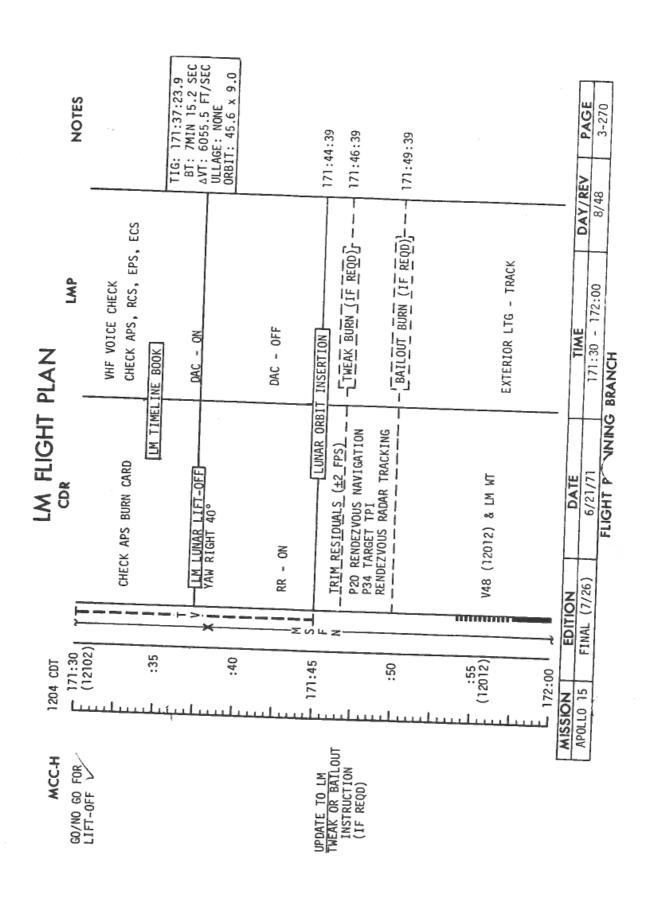


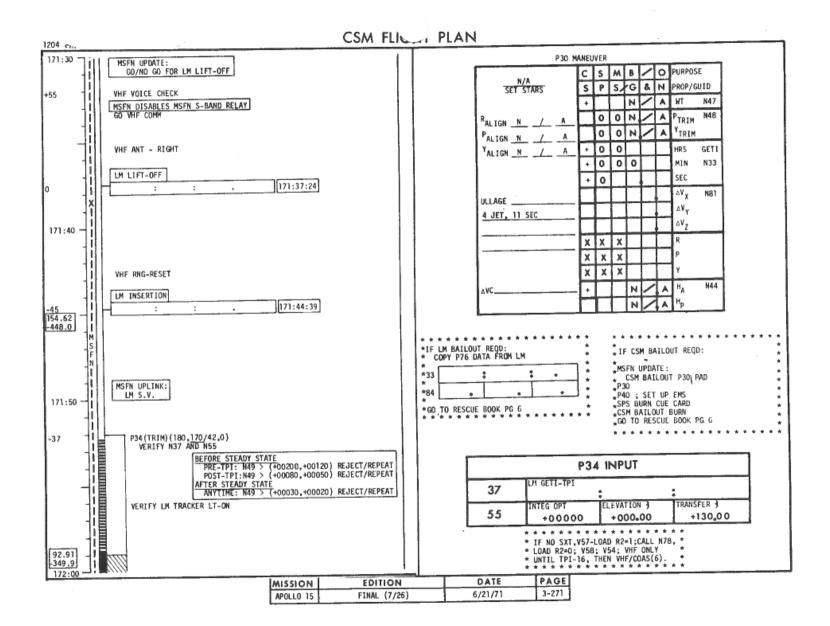
LM FLIGHT PLAN

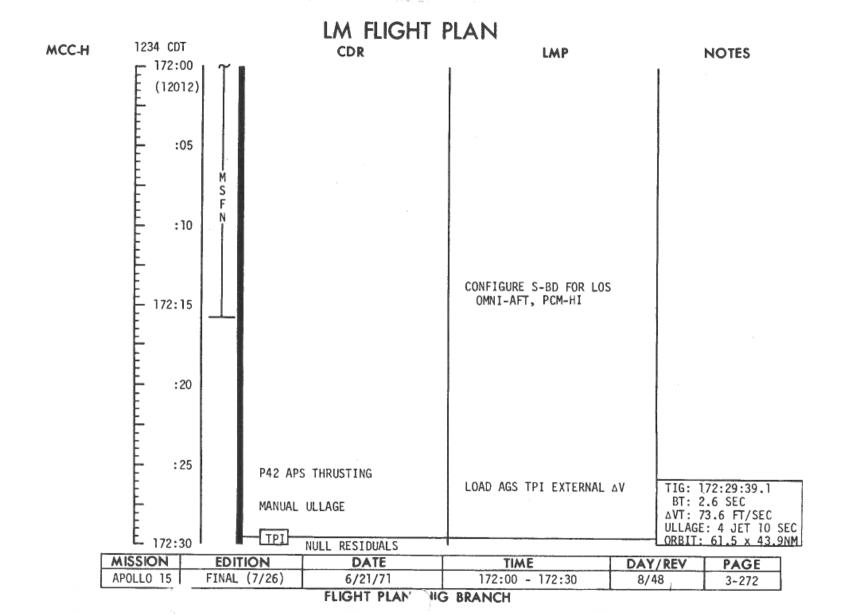


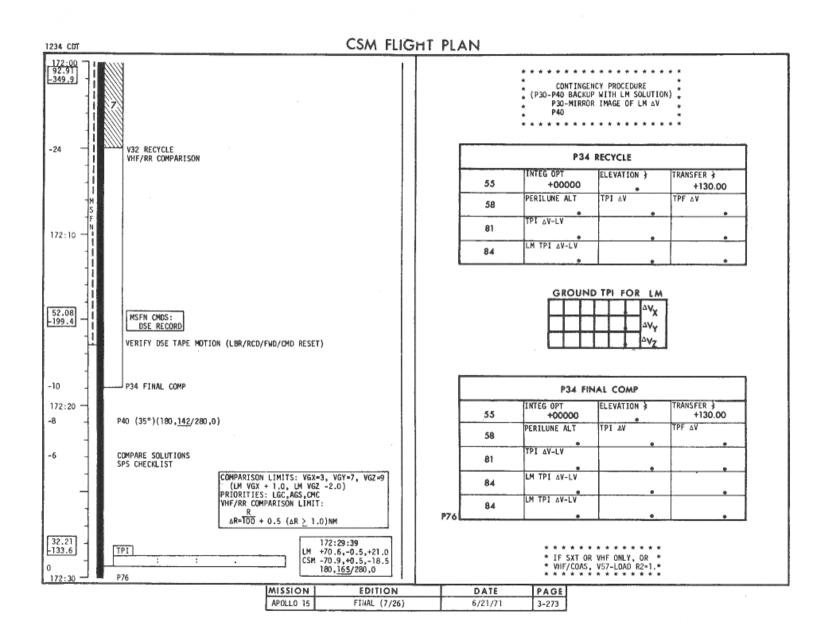
FLIGHT " 'NNING BRANCH





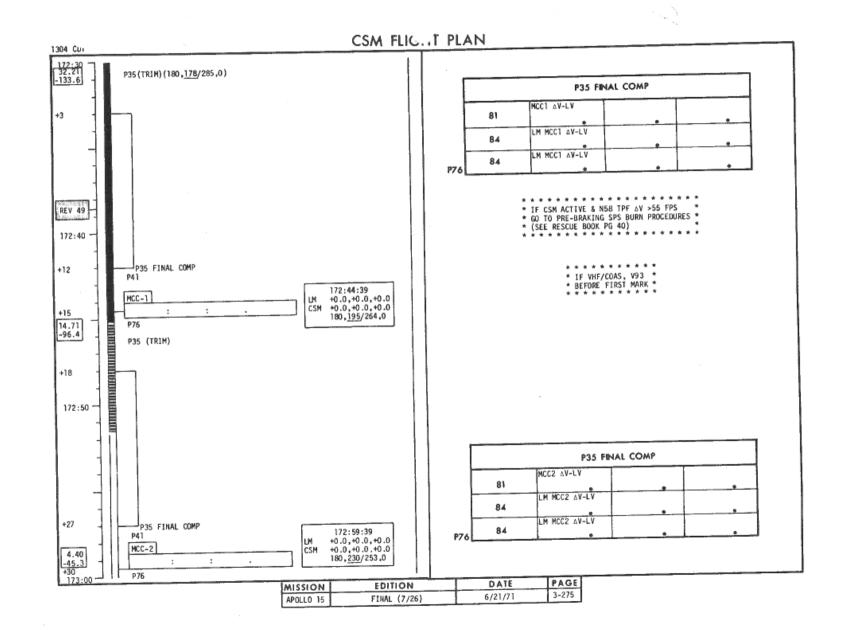


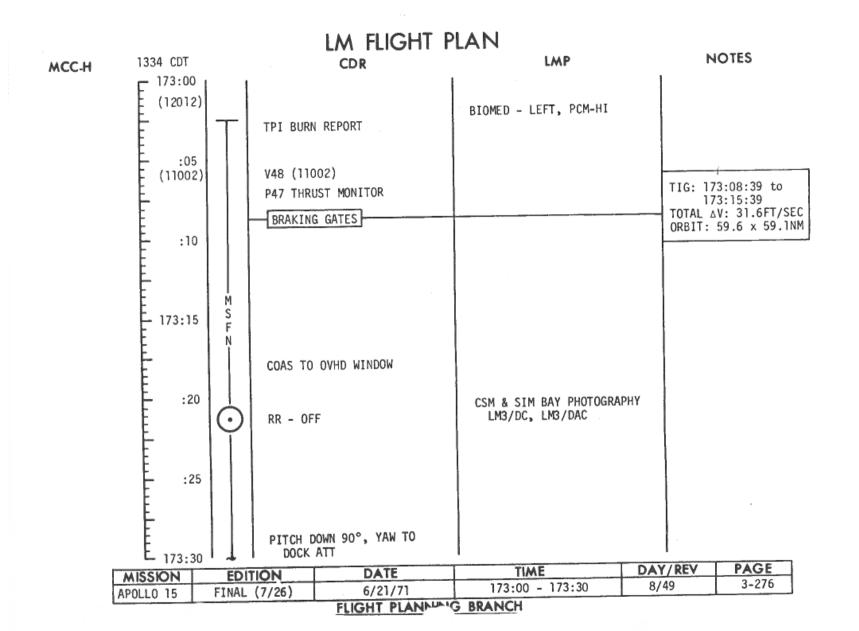


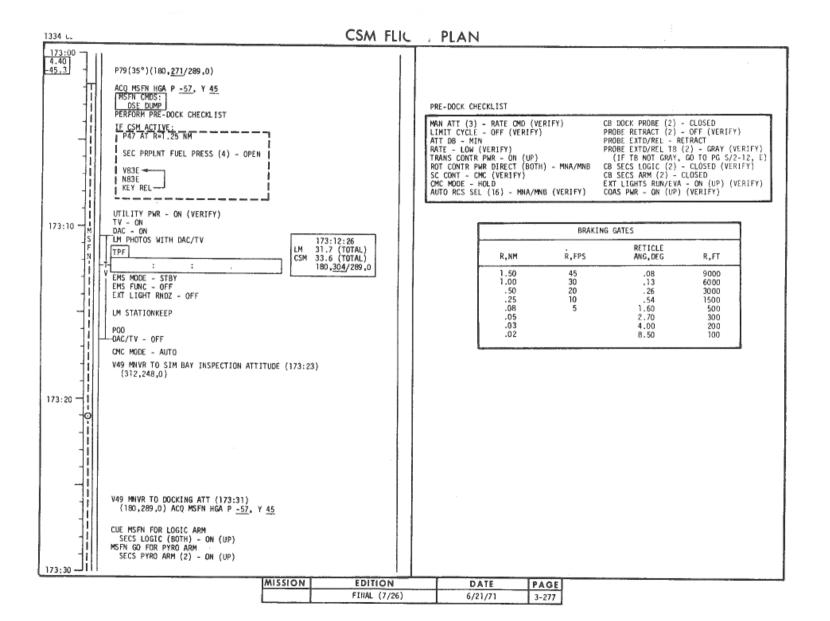


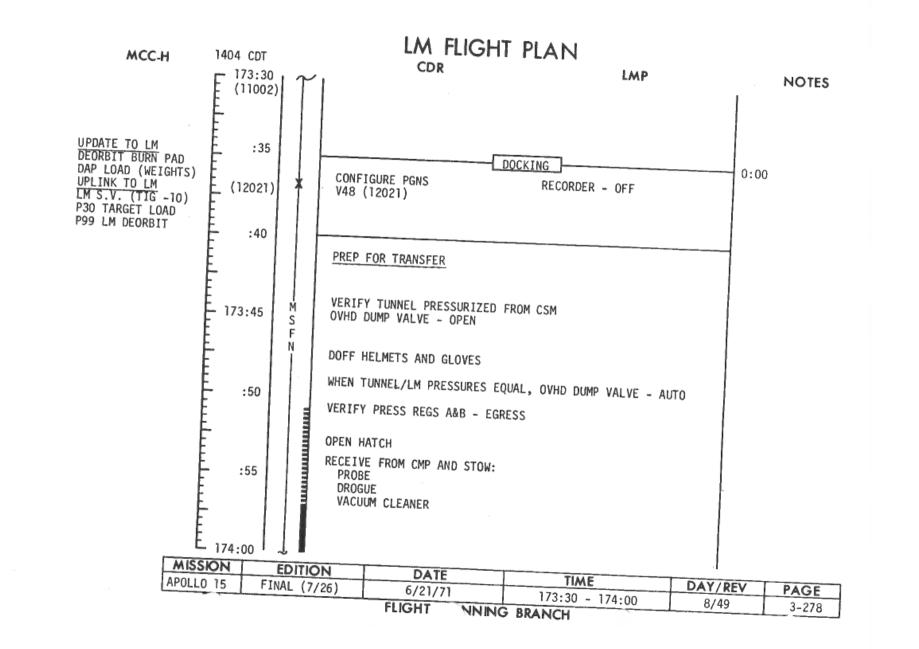
LM FLIGHT PLAN 1304 CDT MCC-H CDR LMP **NOTES** 172:30 P35 TARGET MCC-1 RENDEZVOUS RADAR TRACKING (12012):35 REV 49 FINAL MCC-1 COMPUTATION P41 RCS THRUSTING LOAD AGS MCC-1 EXTERNAL AV TIG: 172:44:39 MCC-1 ΔVT: NOM ZERO 172:45 NULL RESIDUALS EXTERIOR LTG - OFF P35 TARGET MCC-2 RENDEZVOUS RADAR TRACKING :50 :55 FINAL MCC-2 COMPUTATION P41 RCS THRUSTING LOAD AGS MCC-2 EXTERNAL AV TIG: 172:59:39 MCC-2 ΔVT: NOM ZERO NULL RESIDUALS □ 173:00 I MISSION EDITION DATE TIME DAY/REV PAGE 6/21/71 APOLLO 15 FINAL (7/26) 172:30 - 173:00 8/48-49 3-274

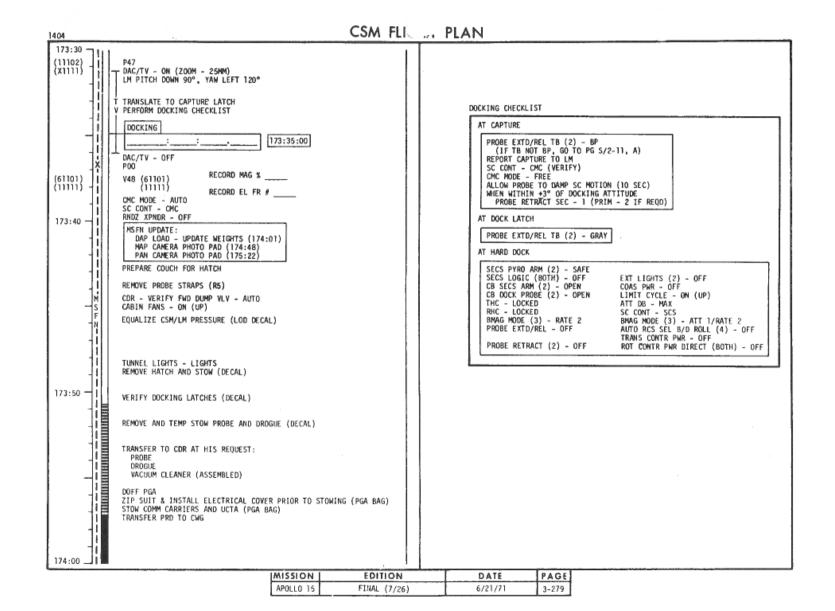
FLIGHT PLADINING BRANCH







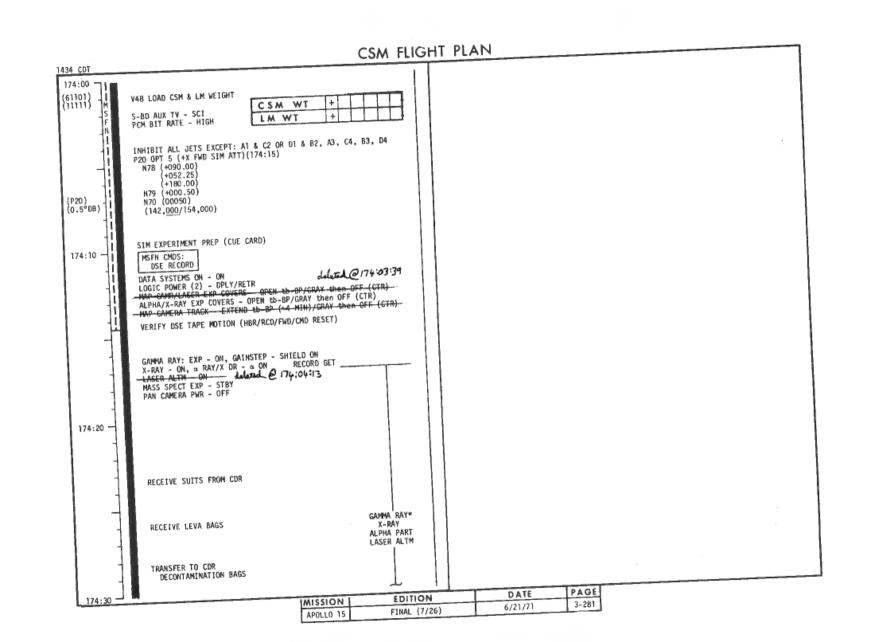


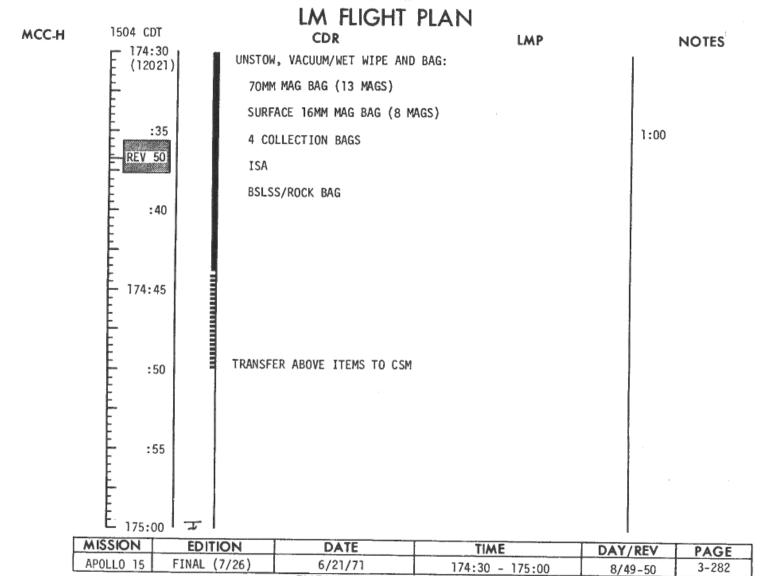


LM FLIGHT PLAN NOTES LMP CDR 1434 CDT MCC-H T 174:00 VACUUM PGA'S (12021) 0:30 :05 DOFF SUITS :10 174:15 :20 TRANSFER SUITS TO CMP PLACE EV GLOVES INSIDE HELMETS, HELMETS INSIDE LEVA'S, IN LEVA BAG :25 TRANSFER LEVA BAGS TO CMP RECEIVE DECONTAMINATION BAGS FROM CMP

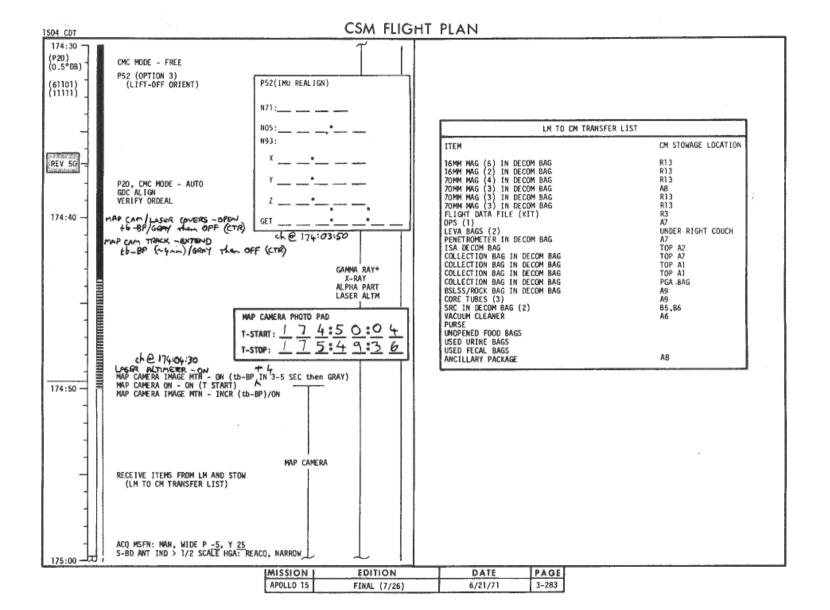
174.2	RECEIVE	DECONTAMINATION BAC	3 11011 0111		
<u> </u>		DATE	TIME	DAY/REV	PAGE
MISSION	EDITION	DATE			3-280
	FINAL (7/26)	6/21/71	174:00 - 174:30	8/49	3-200
APOLLO 15	FINAL (7/20)		IG RPANCH		

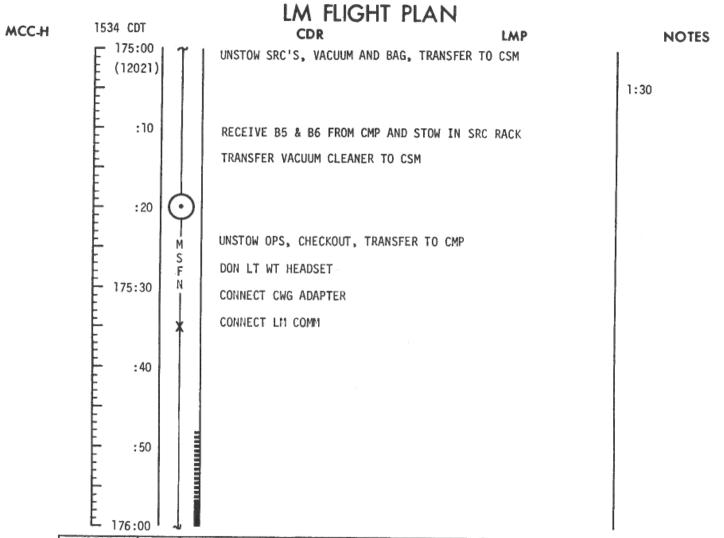
VING BRANCH FLIGHT PL





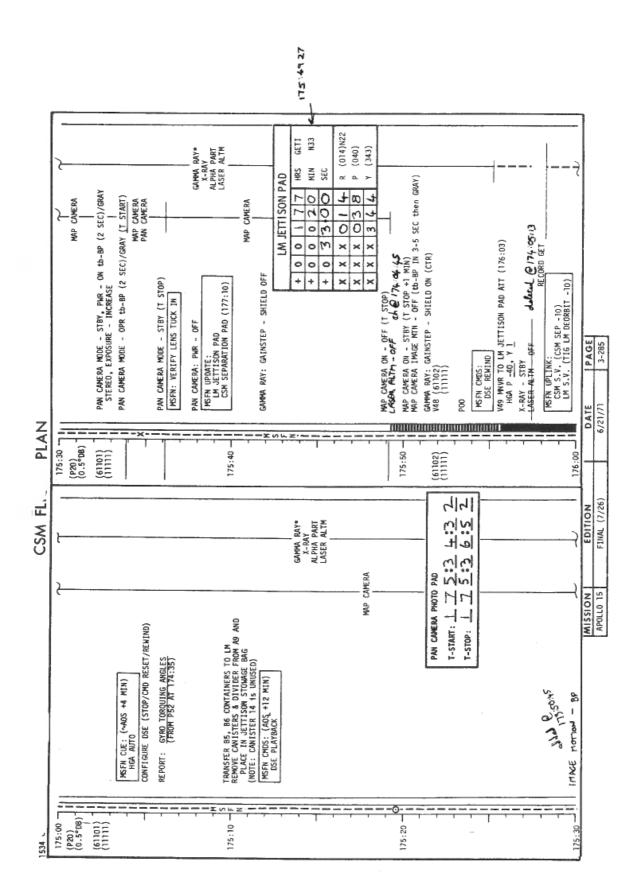
FLIGHT PLANNING BRANCH

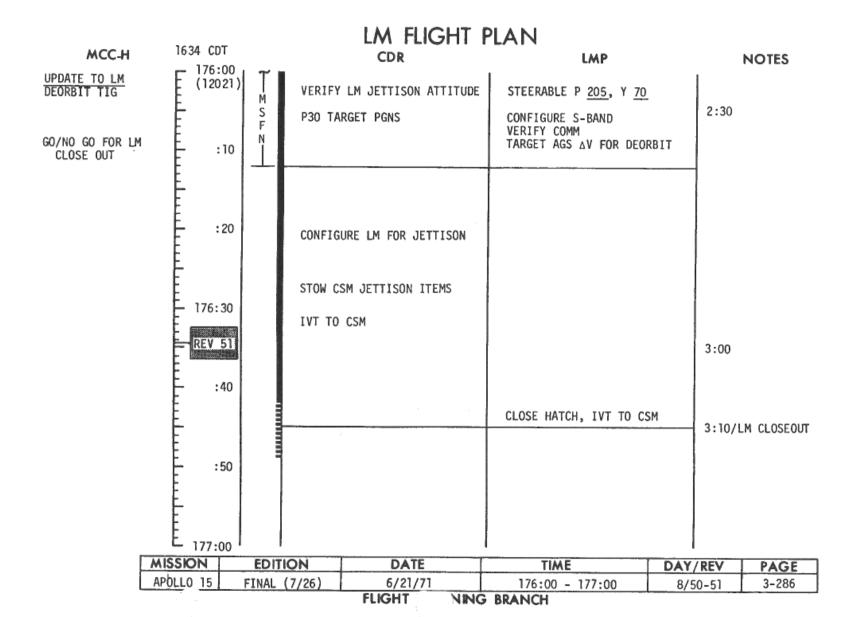


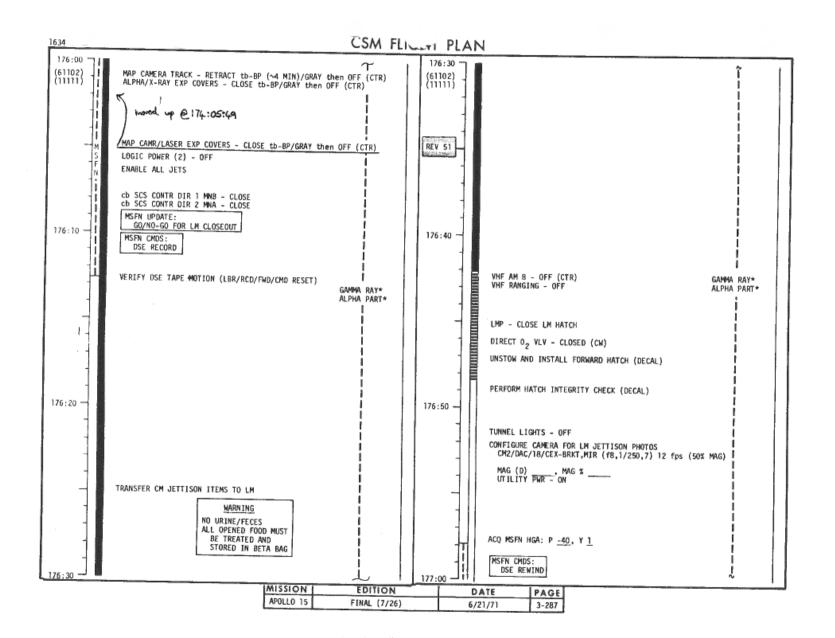


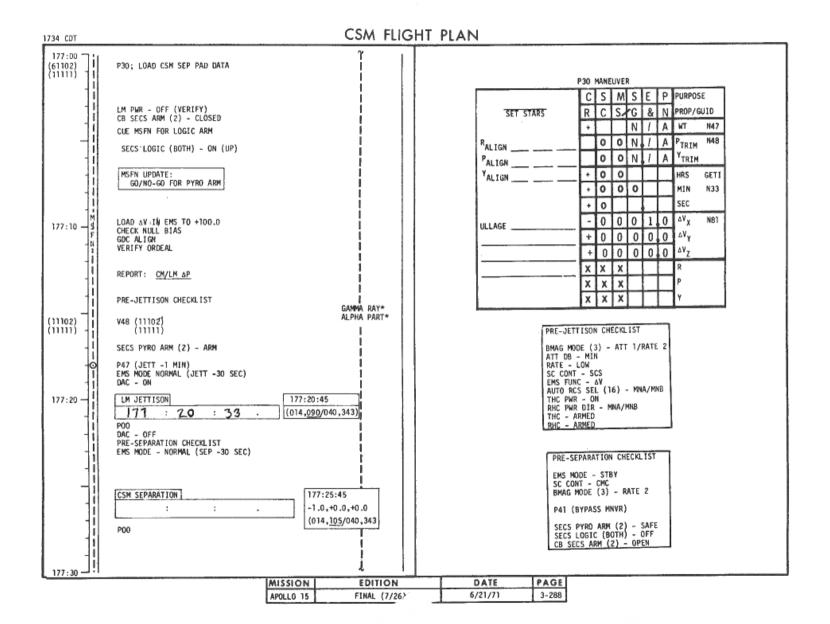
MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
APOLLO 15	FINAL (7/26)	6/21/71	175:00 - 176:00	8/50	3-284

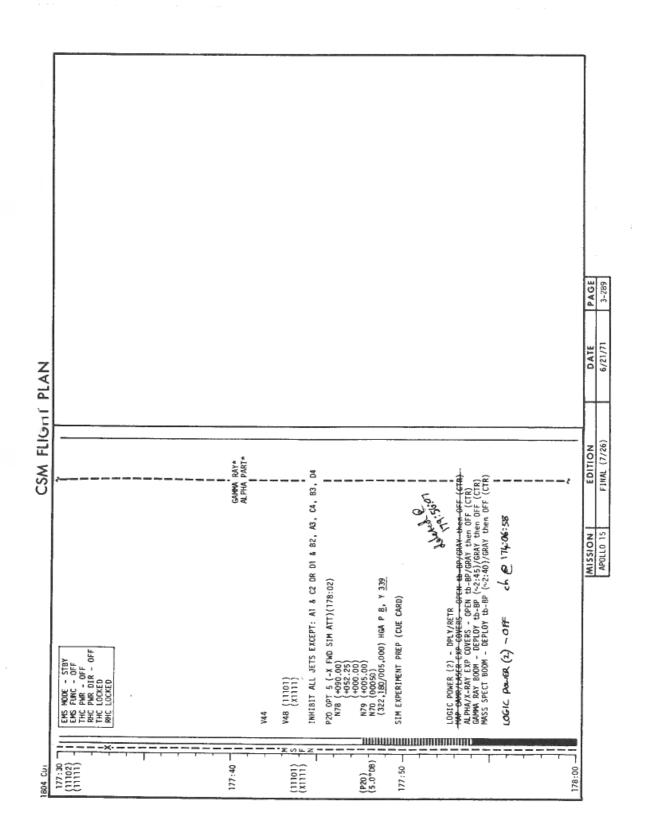
FLIGHT PLANNING BRANCH

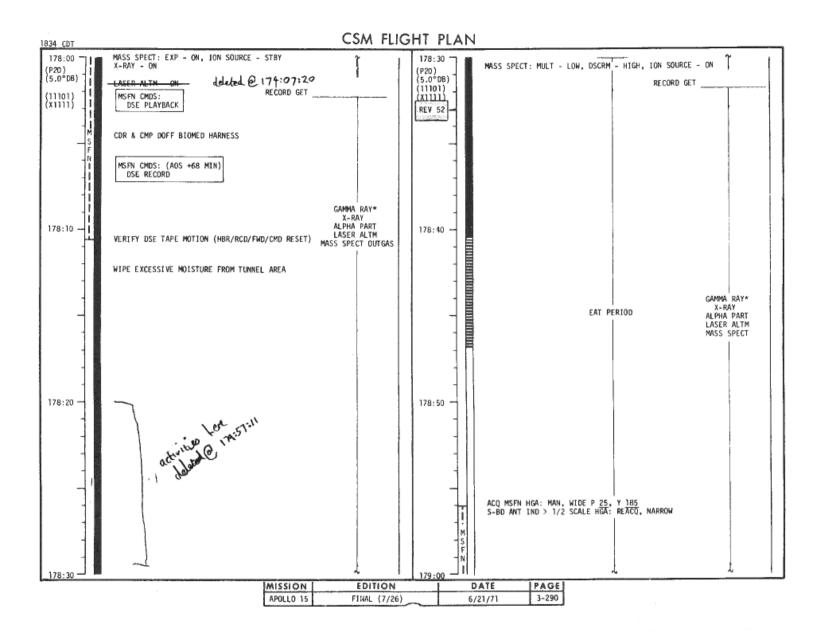


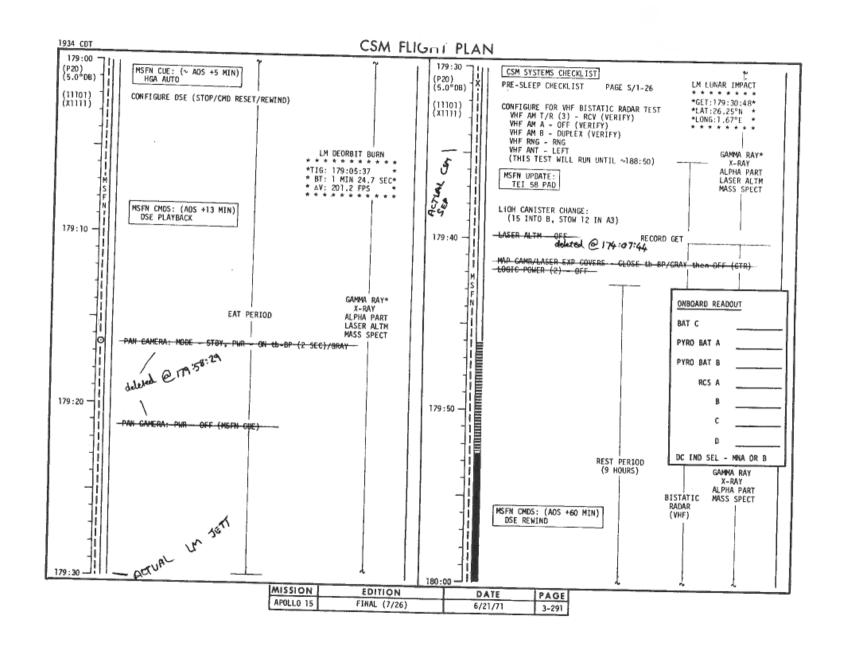


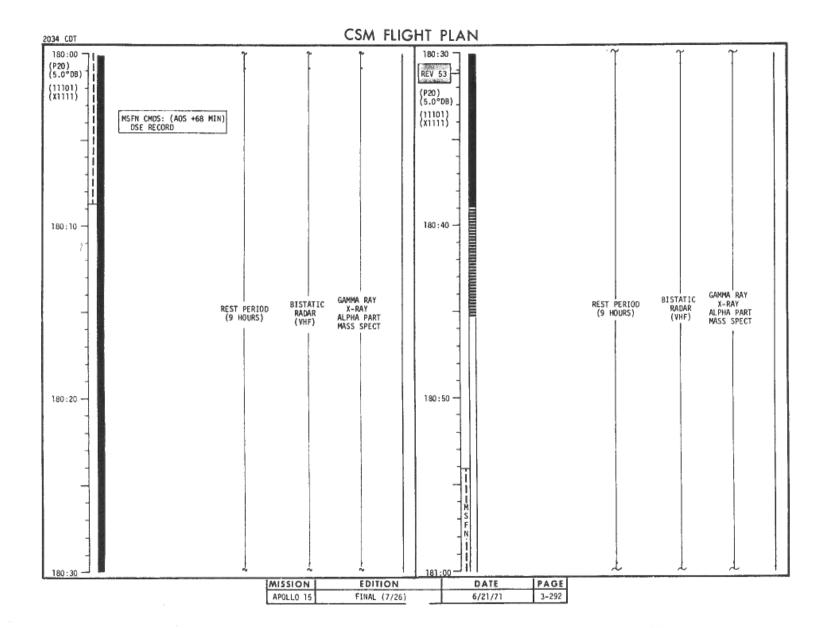


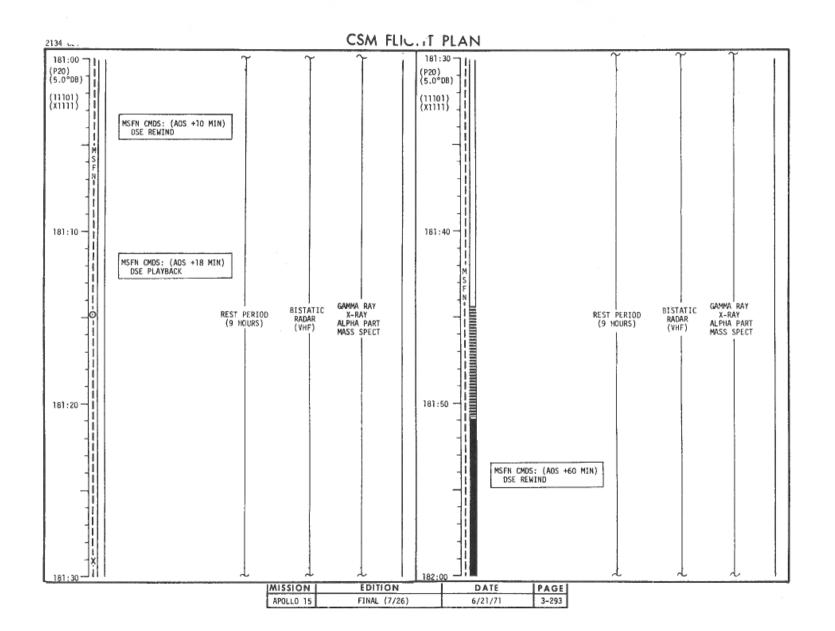


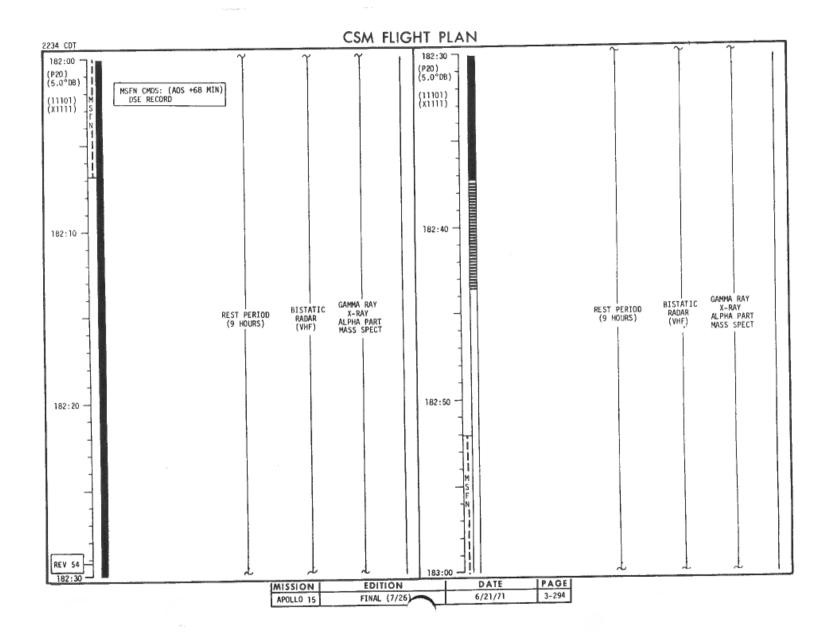


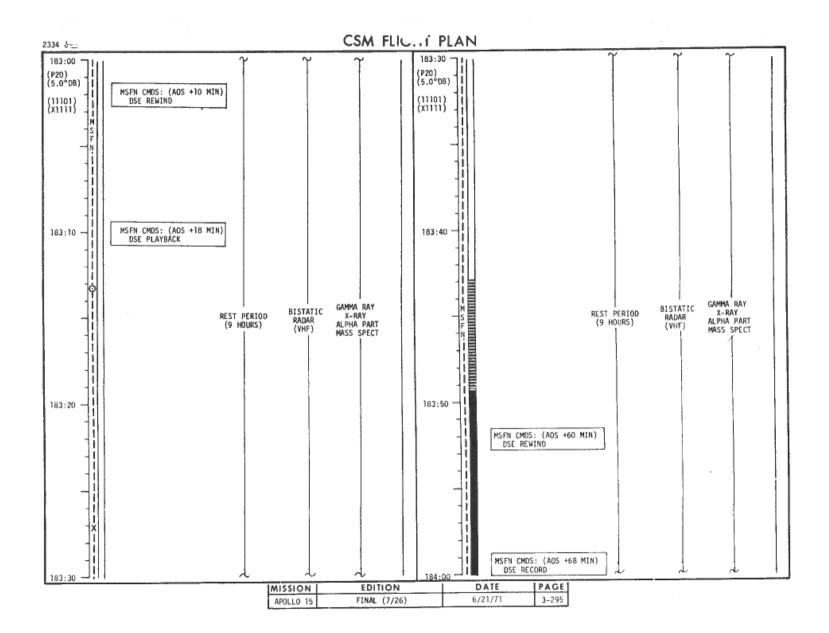


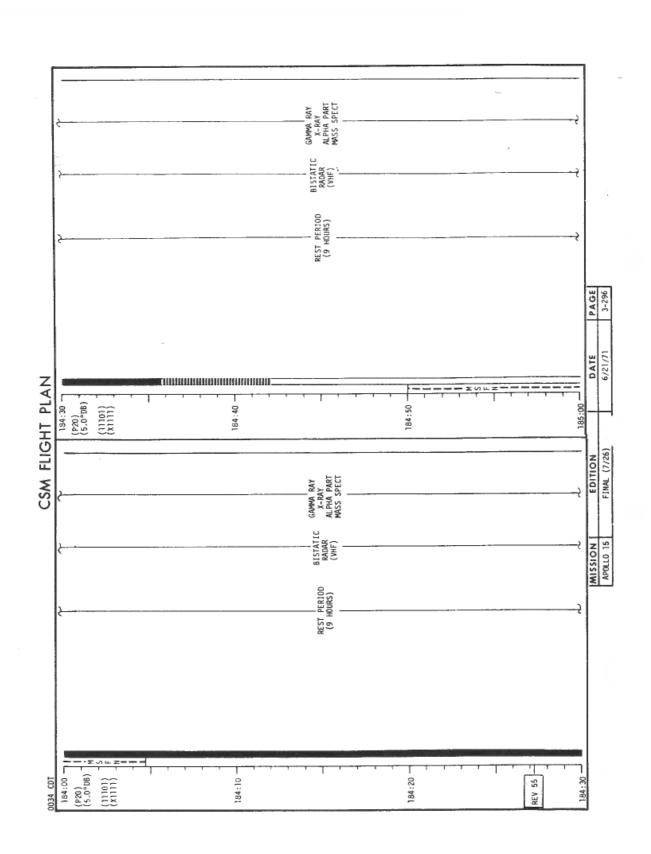


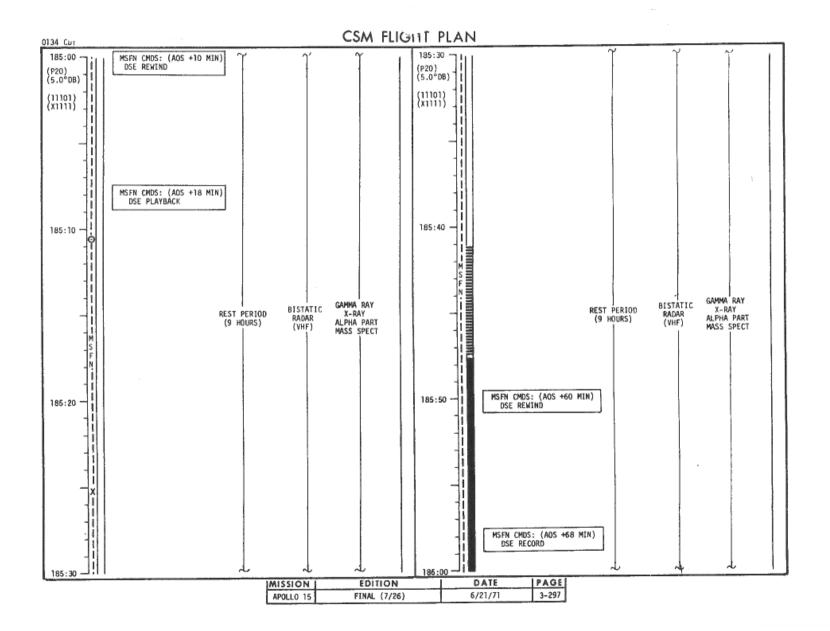


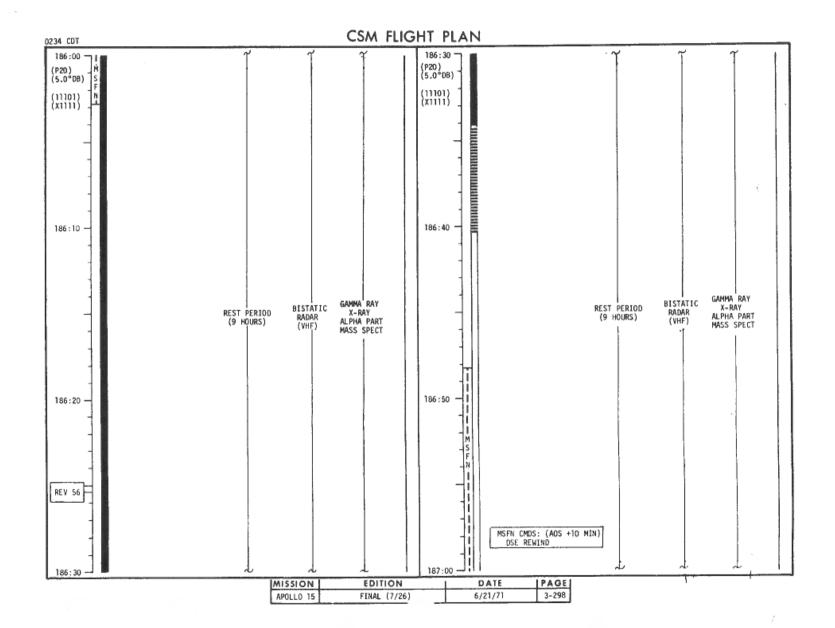


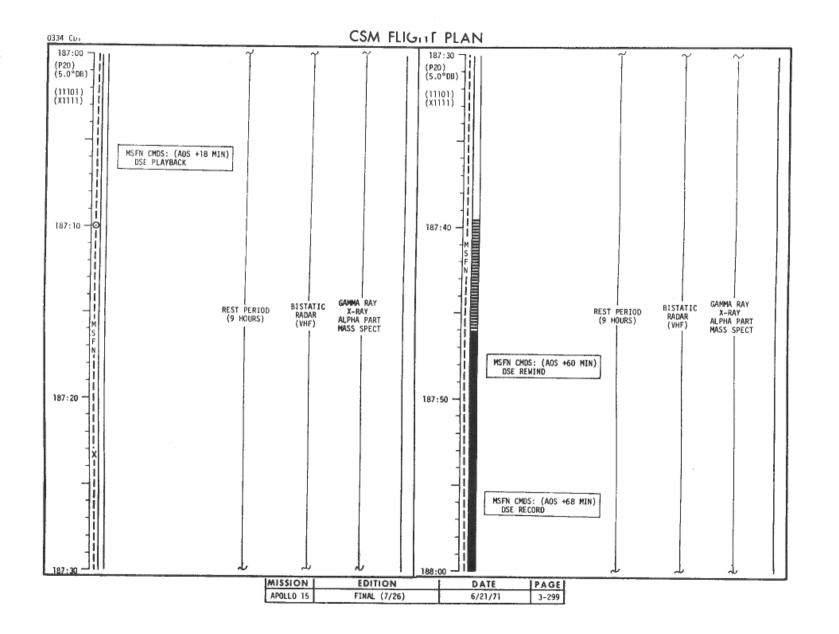


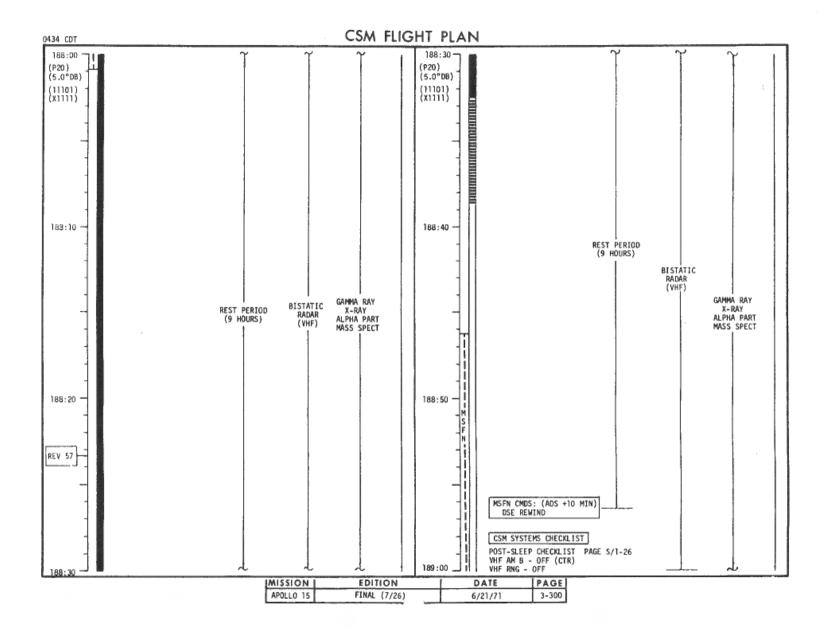


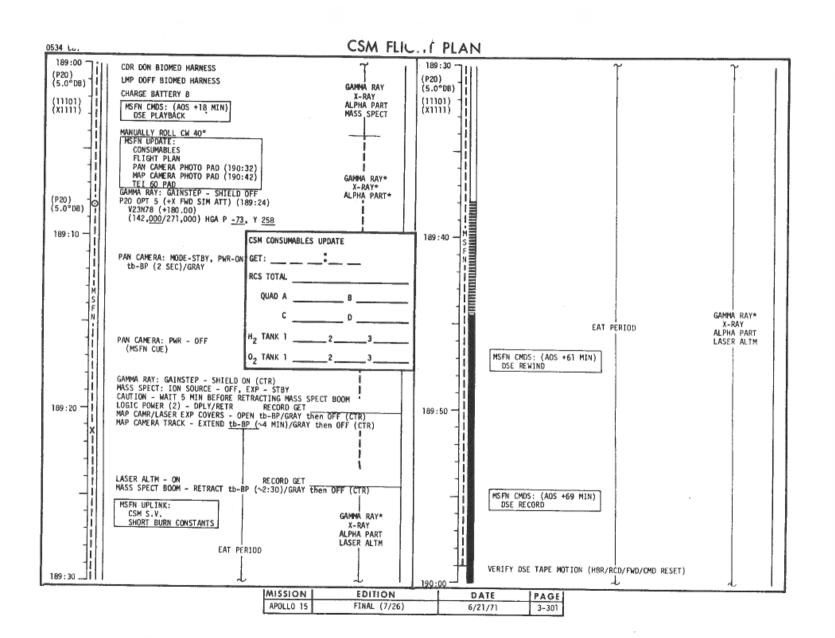


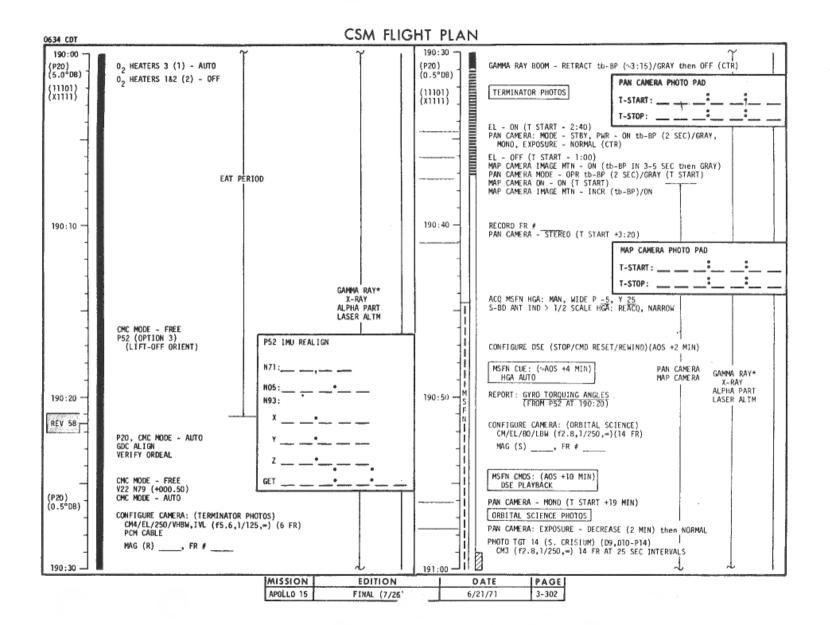


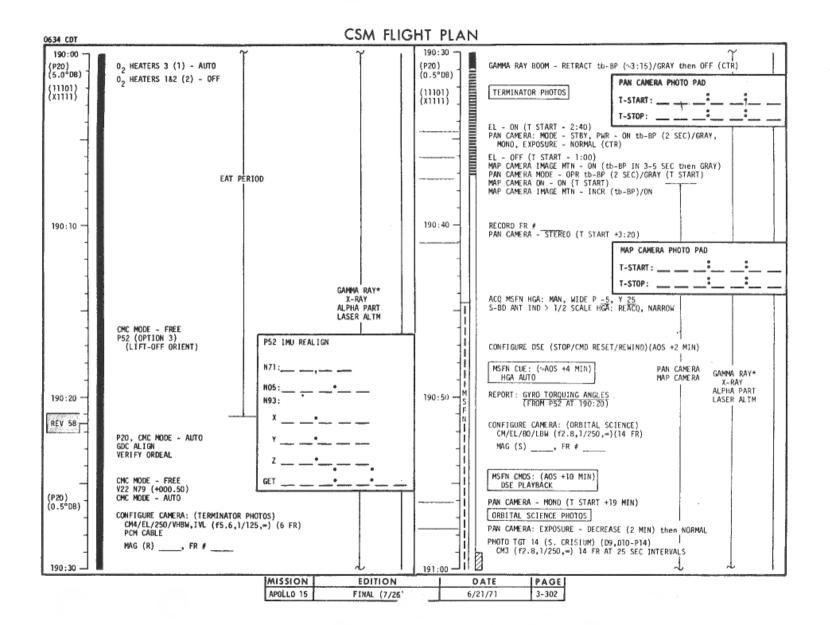


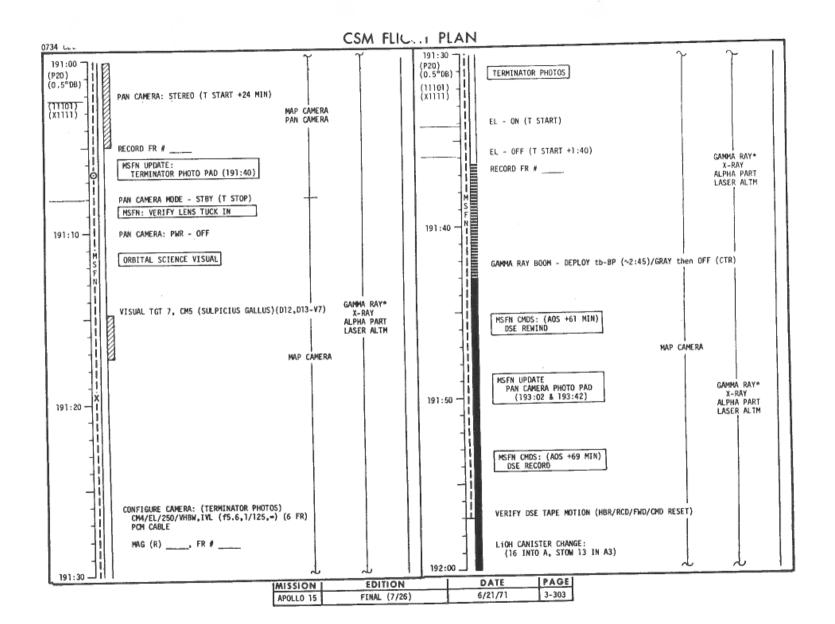


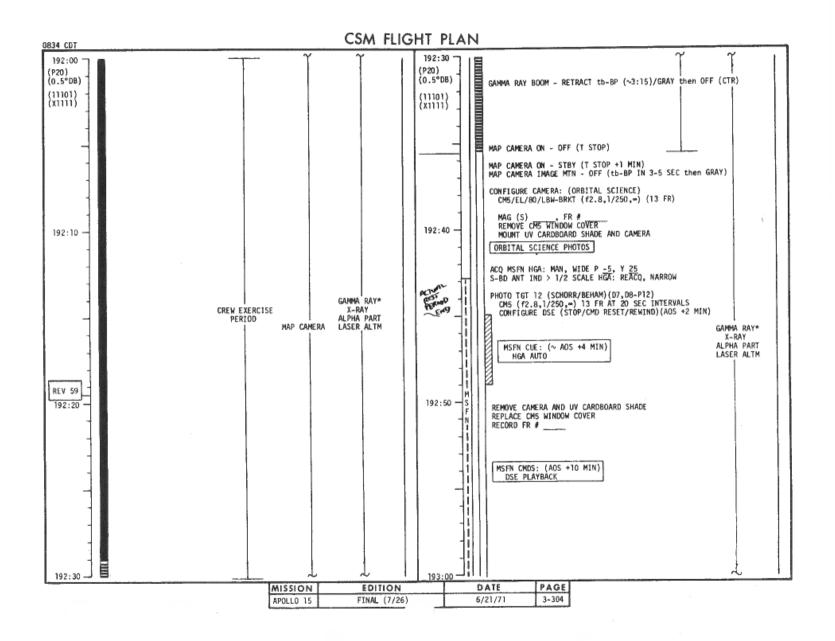


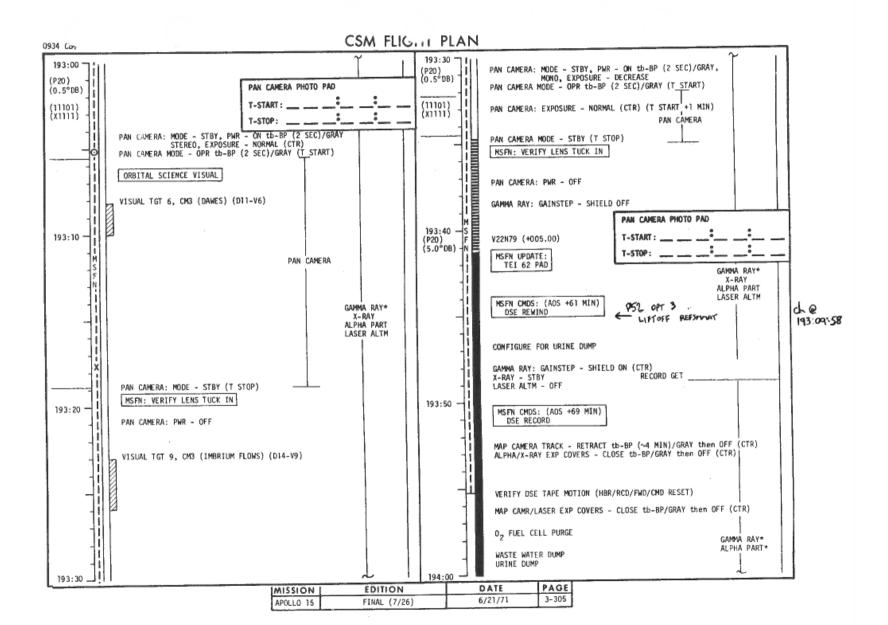


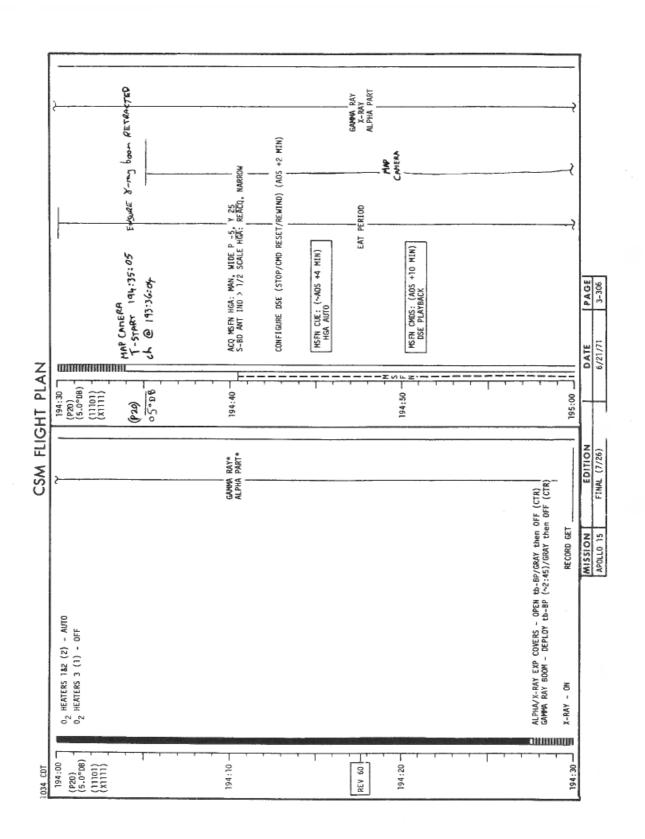






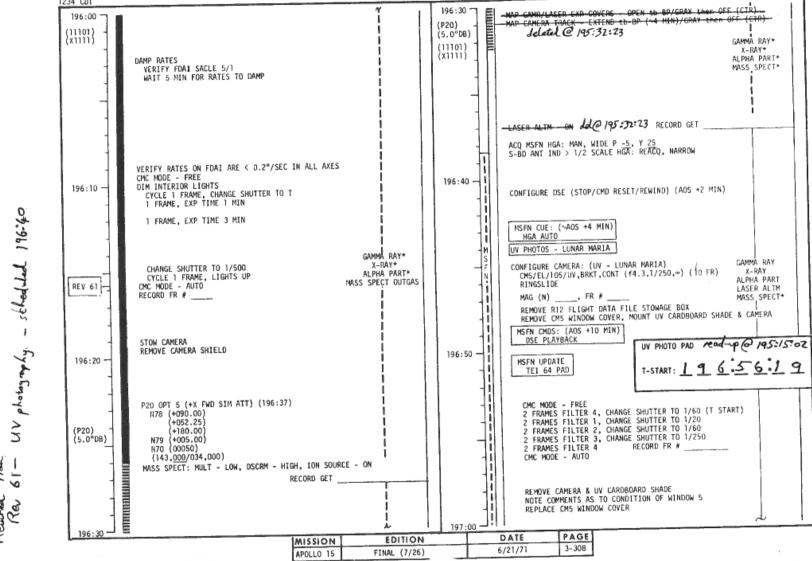






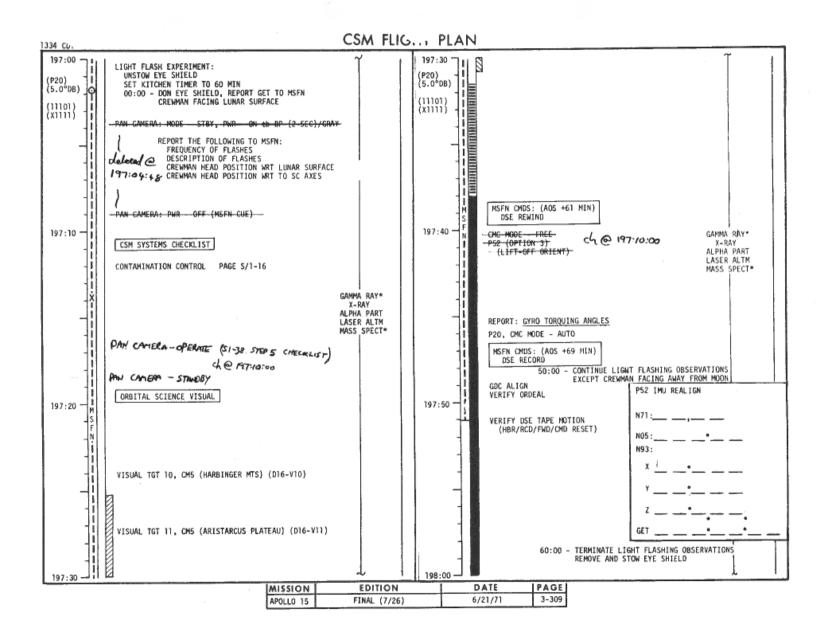
CSM FLIGHT PLAN PAN CAMERA NOBE -STANDBY PLIK-ON, STERED EXP-MORNAL HISFIR UPDATE: 195:30 UV PHOTO PAD (196:51) (P2G) (5.0°08) (P20) CONFIGURE CAMERA: (GEGENSCHEIN)
INSTALL CAMERA SHIELD
CN4/NK/55/YHBW - BRKT, (fl.2,1/500,-)(12 FR) (5.0°DB) (11101) (X1111) (11101) (xiiii) MAG () _____, FR # _____ GAMMA RAY 195:04:13 15:31 ac X-RAY PAN CAMERA MODE - OPERATE MAP CAMERA ALPHA PART deploy T-510P 195736:50 che 193:36204 TERMINATOR PHOTO PAD LASER OFF, RETRACT CANERA OFF (CTR) CHC MODE - FREE PCL ONE CONSIG 40 (11101) 195.31:14 CMC MODE - AUTO (TERMINATION - 2 MIN) (Ritti) 149 HAVE TO DEEP SPACE MEASUREMENT & MIDMAY PT ATT (195:48) (169,257,032) HGA P -61, Y 188 195 175: 10-MSFNONDS: (ACS +61 MIN) DSE RENIND RAMMA RAY! A BOY ALDINA PART MRSS SPECT MOOR - DEPLOY, 10-89 (~2:40)/GPAY then OFF (CTF) MAP CANNER RAY X-RAY ALPHA PART PANCAMERA MOSE STAND BY FUR-ON STERES-ROND CAMERA GEGENSCHE IN PHOTOS DAMP RATES VERIFY FOAT SCALE - 5/1 WALT 5 MEN FOR RATES TO DAMP 195:18:23 PAN CHERN SELF TEST - SELF TEST. BP for 30 secs
THEN SELF TEST
TO HEATER MSFN CHOS: (AOS +69 M1N) OSE RECORD 195:50 ---MAPCAM ING MEN-8P 195:20 -MASS SPECT: EXP - ON, 10H SOURCE - STBY RECORD GET ch @ 193136104 195:21,00 ON PURTUR OF VERIFY RATES ON FDAI ARE < 0.2°/SEC IN ALL AXES VERIFY DSE TAPE NOTION (HBR/RCD/FWD/CMD RESET) CMC MODE - FREE CMC MXDE - FREE DIM INTERIOR LIGHTS CYCLE 1 FRAME, CHANGE SHUTTER TO T 1 FRAME, EXP TIME 1 MIN GANMA RAY® K-RAY* ALPHA PART* MASS SPECT DUTGAS 1 FRAME, EXP TIME 3 NIN CHANGE SHUTTER TO 1/500 CYCLE 1 FRAME, LIGHTS UP CMC MDDE - AUTO V49 MNVR TO MOULTON PT ATT (196:04) (169,261,021) 196:00. 195:30 DATE PAGE EDITION MISSION 3-307 6/21/71 APOLLO 15 FINAL (7/26)

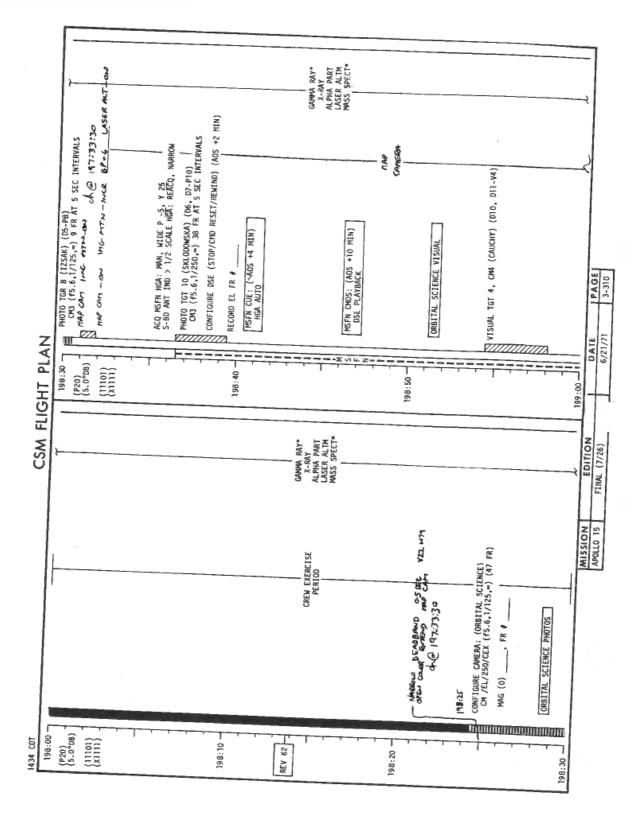
PAN CAM changes @ 194194:58



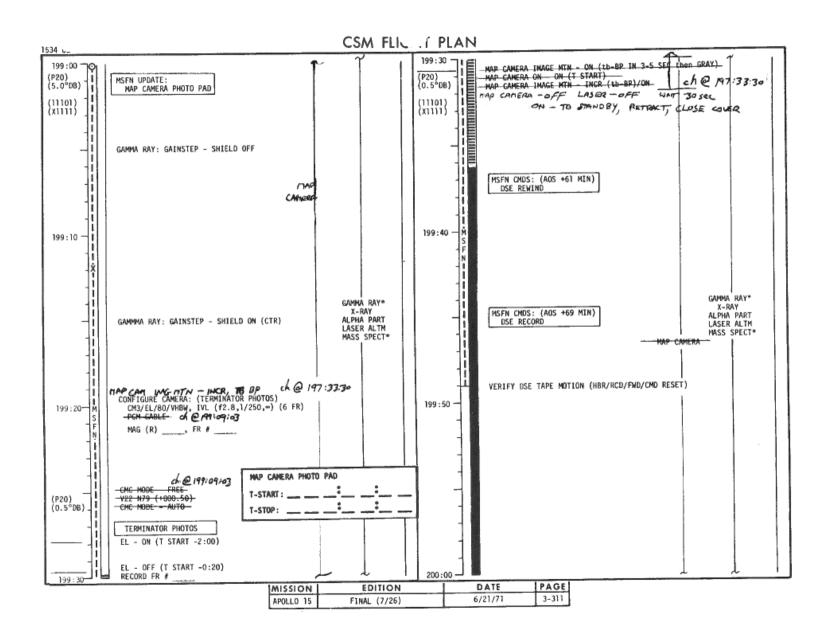
CSM FLIGHT PLAN

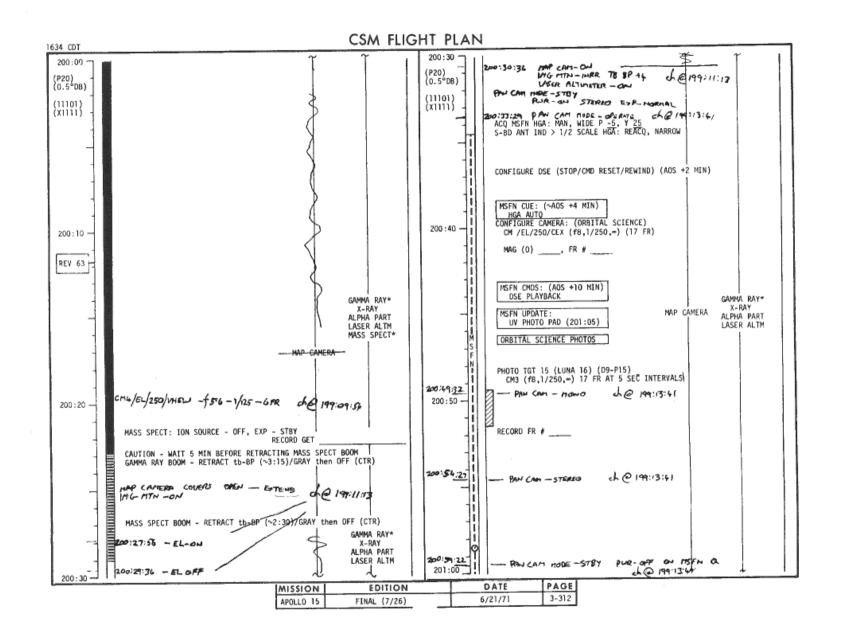
1234 CDT

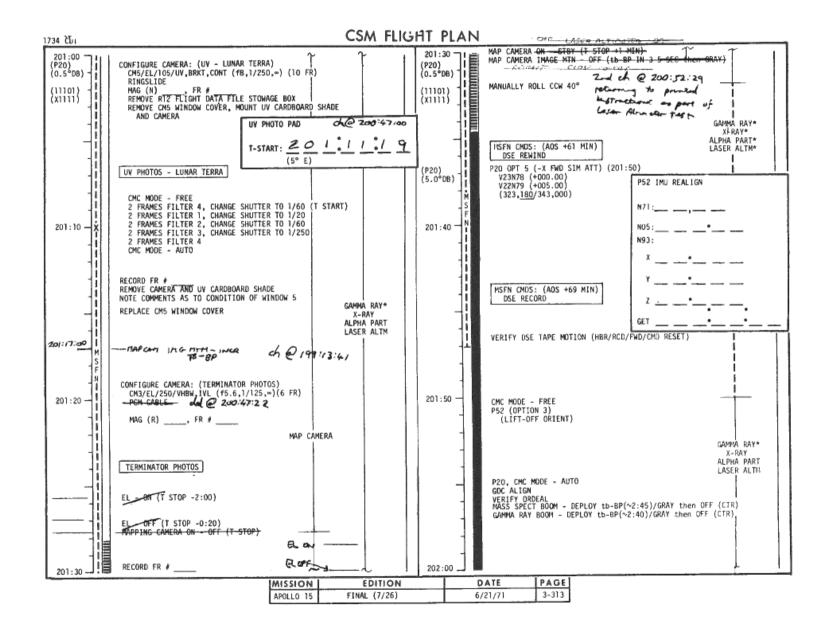


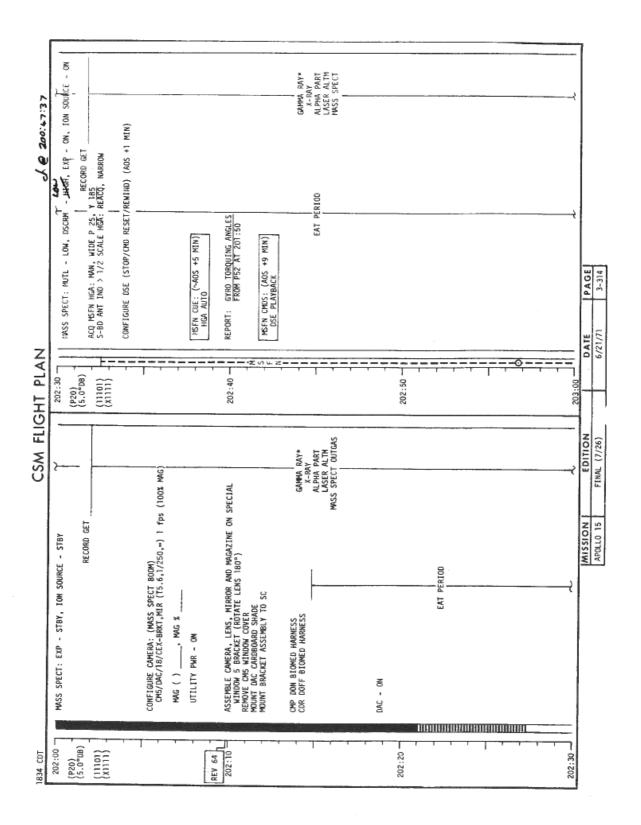


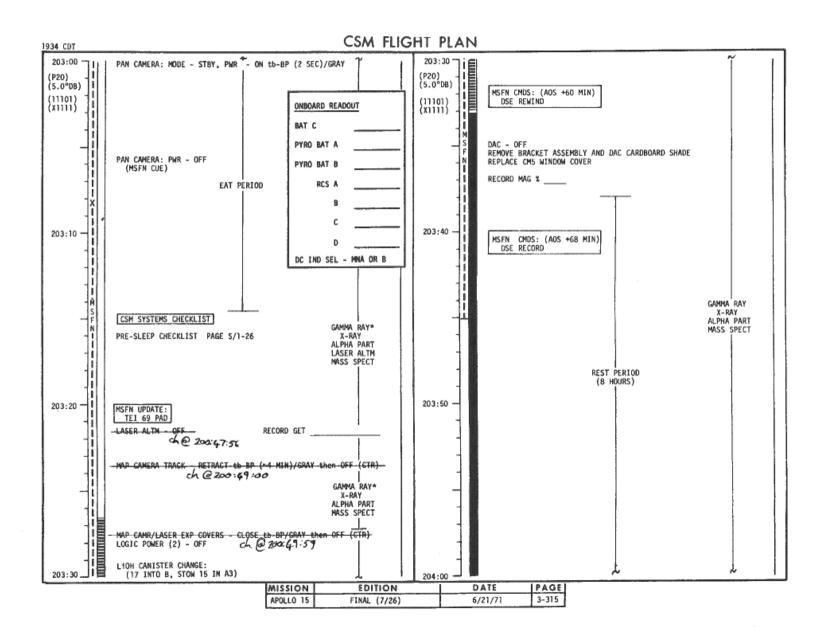
Reused Flight Phones exactse - Science photos + viruals
Reus 62 - Creu exactse - Science Photos + viruals
Termostar Photos

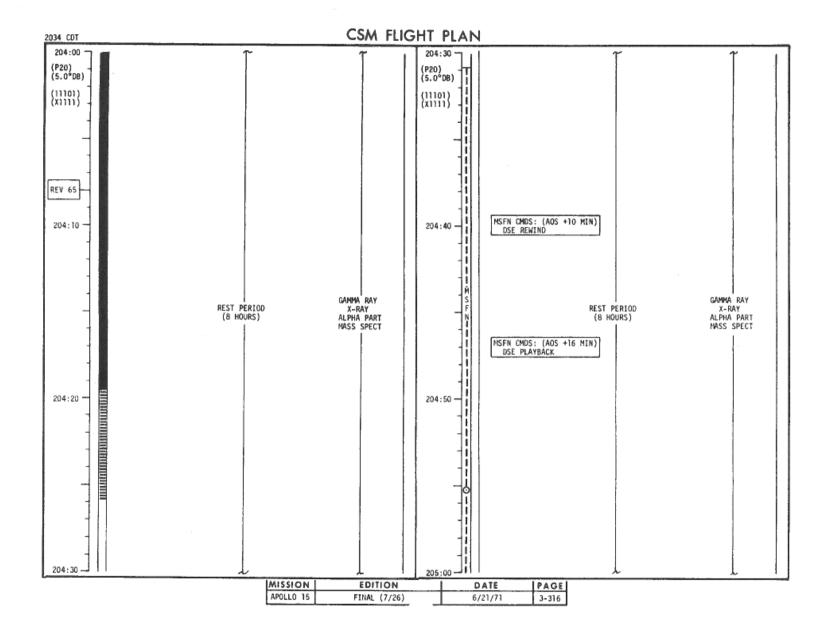


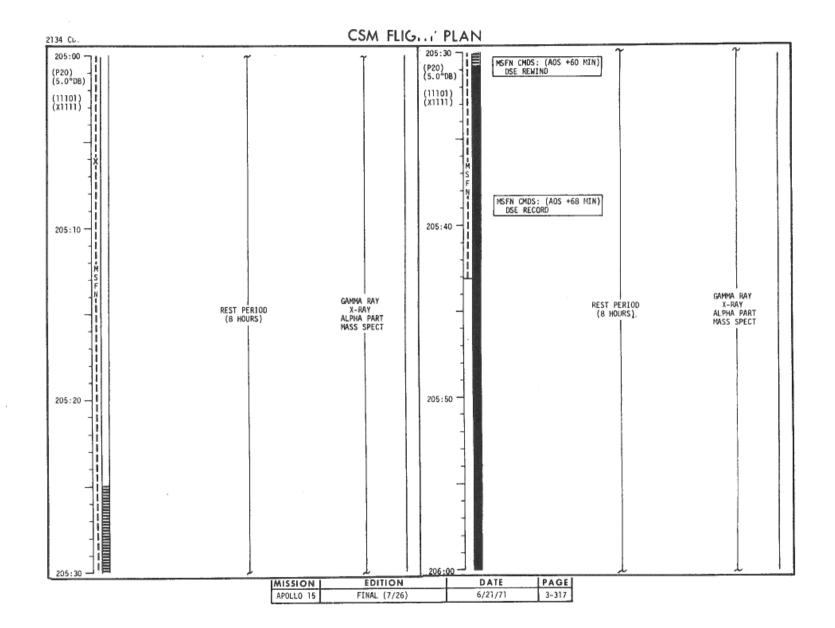


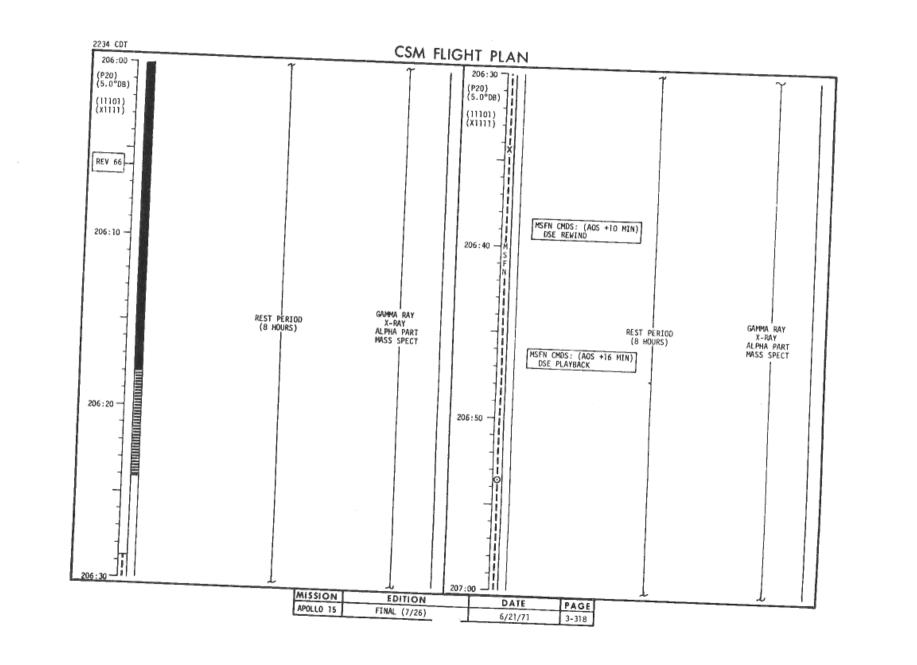


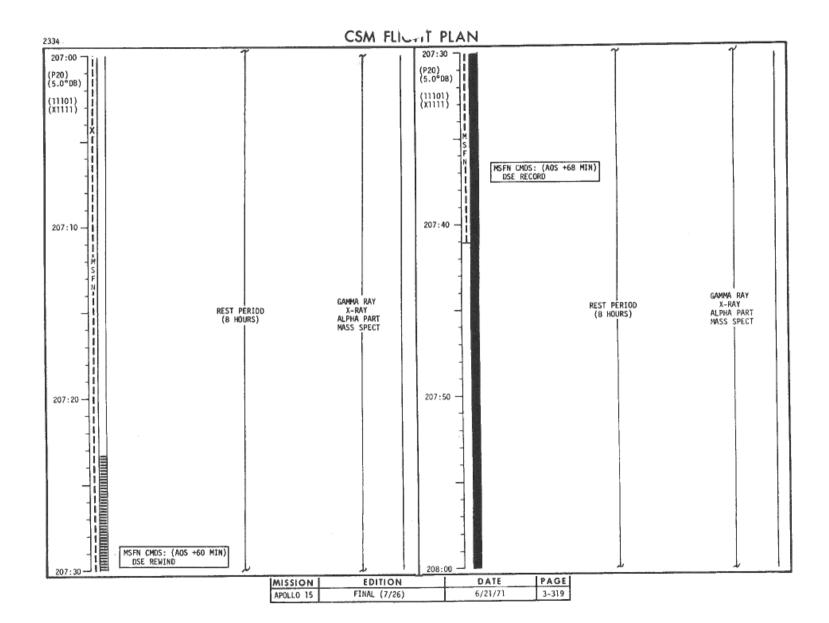


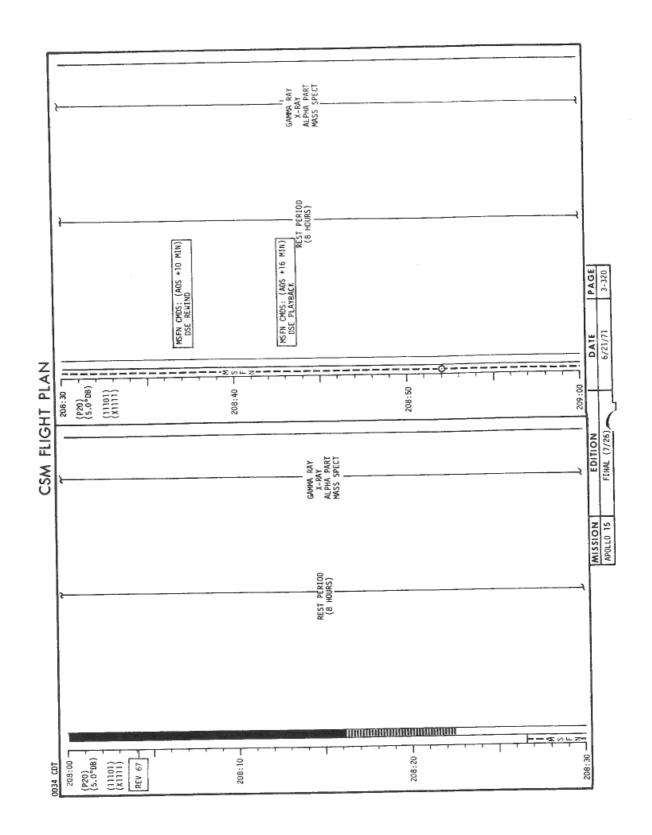


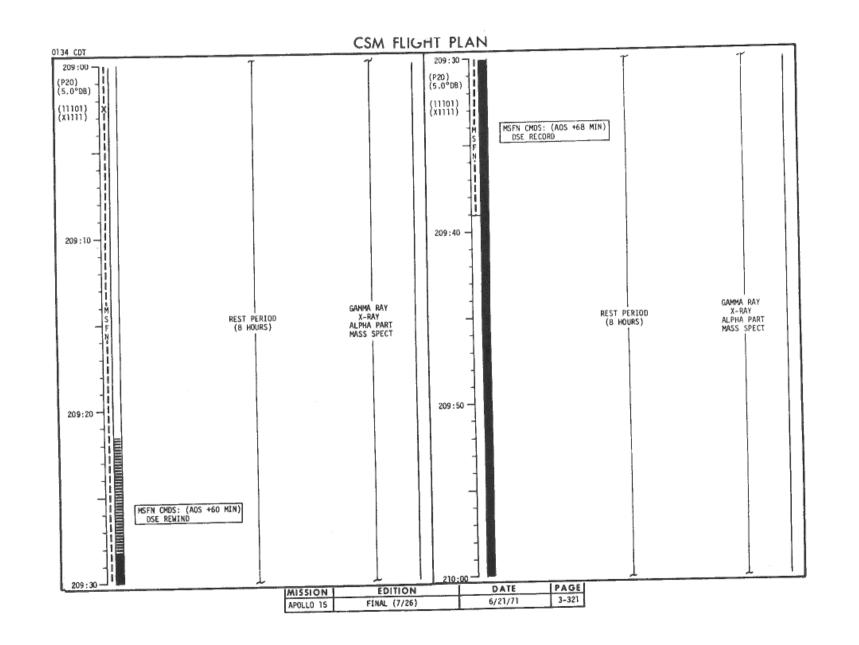


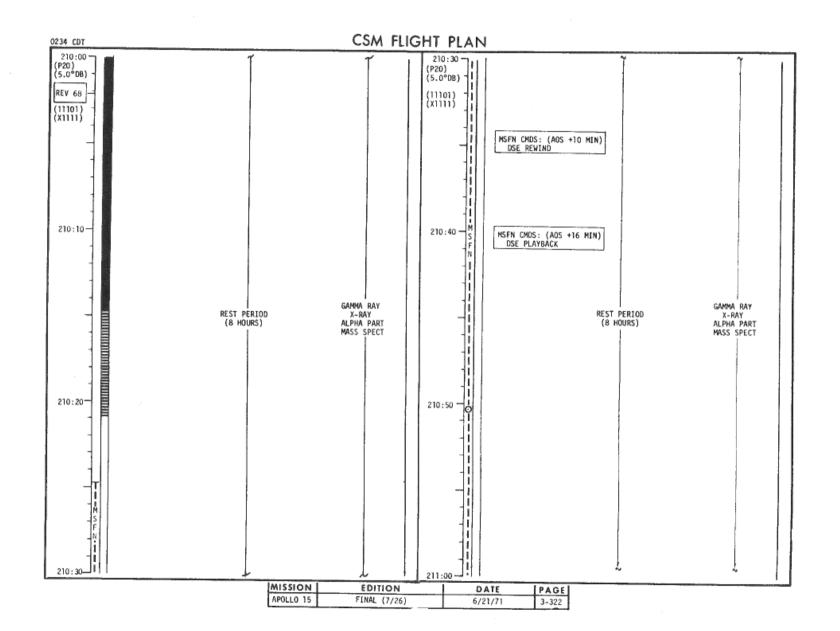


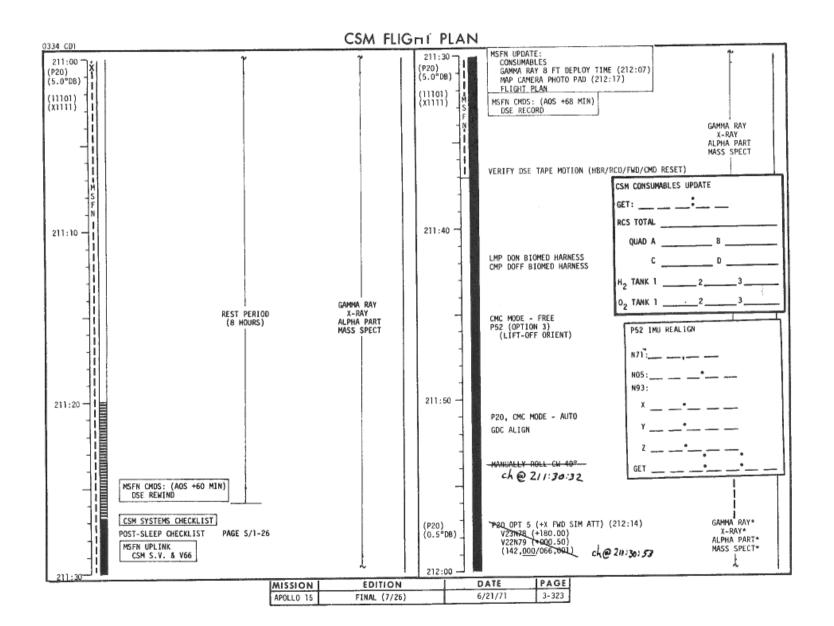


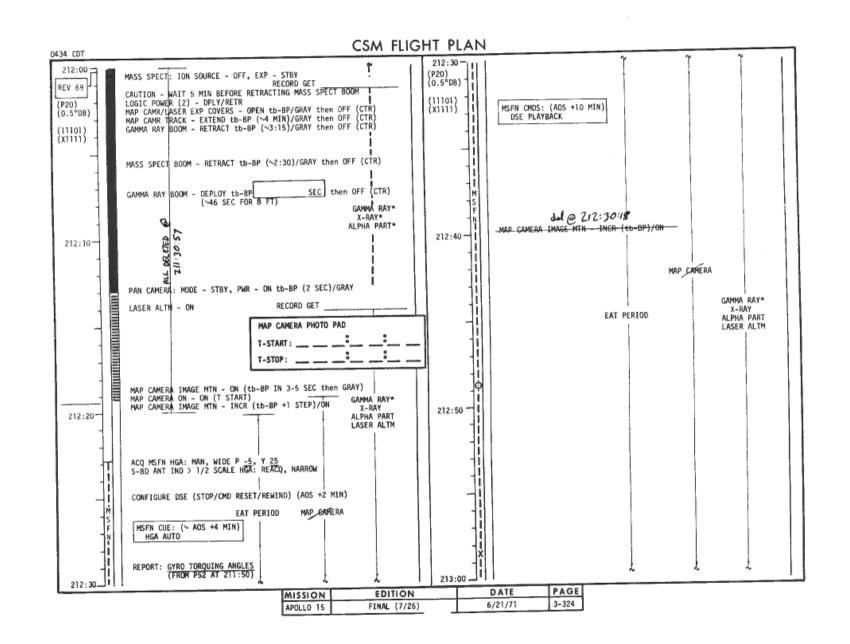


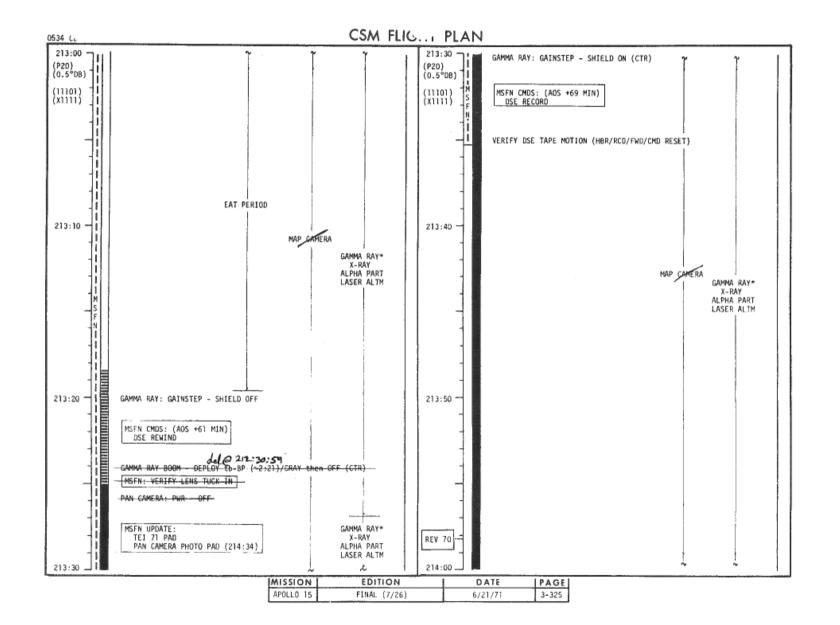


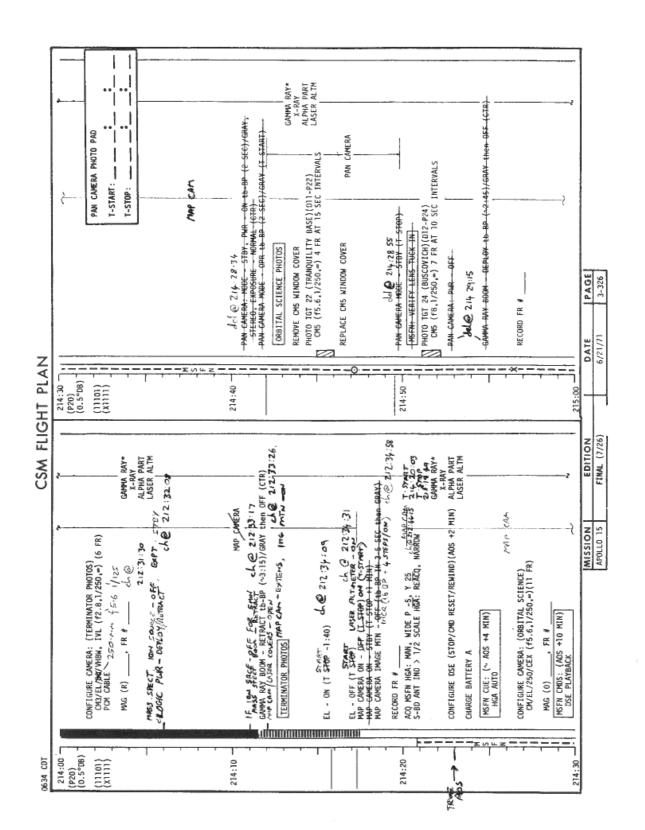


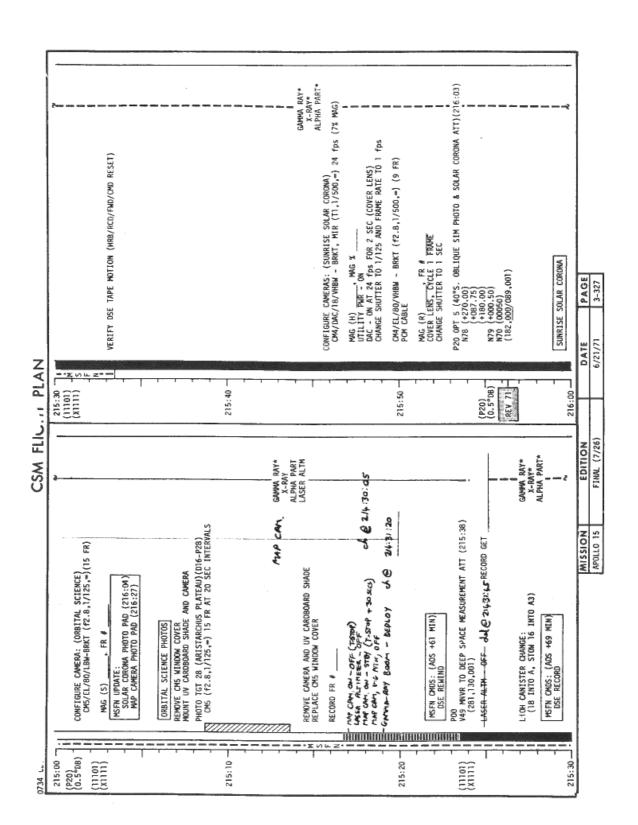


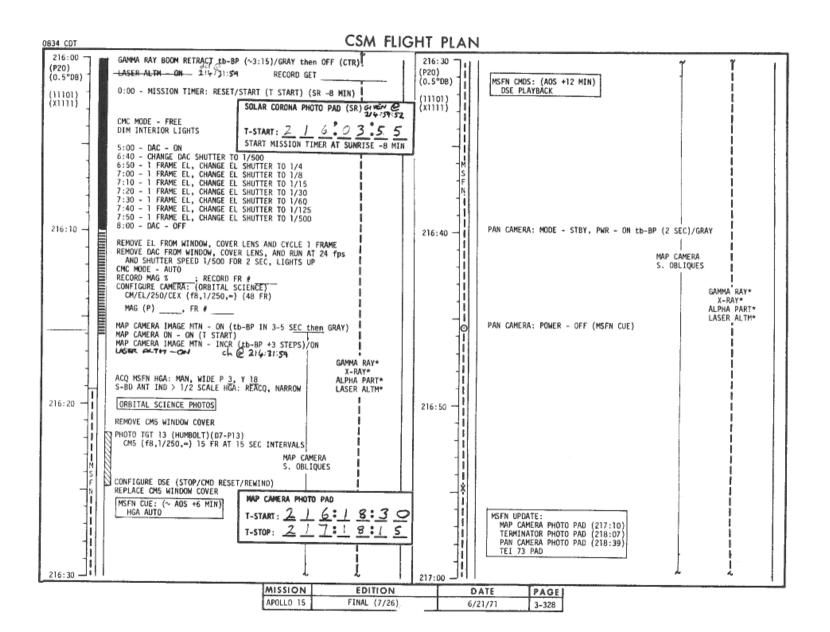


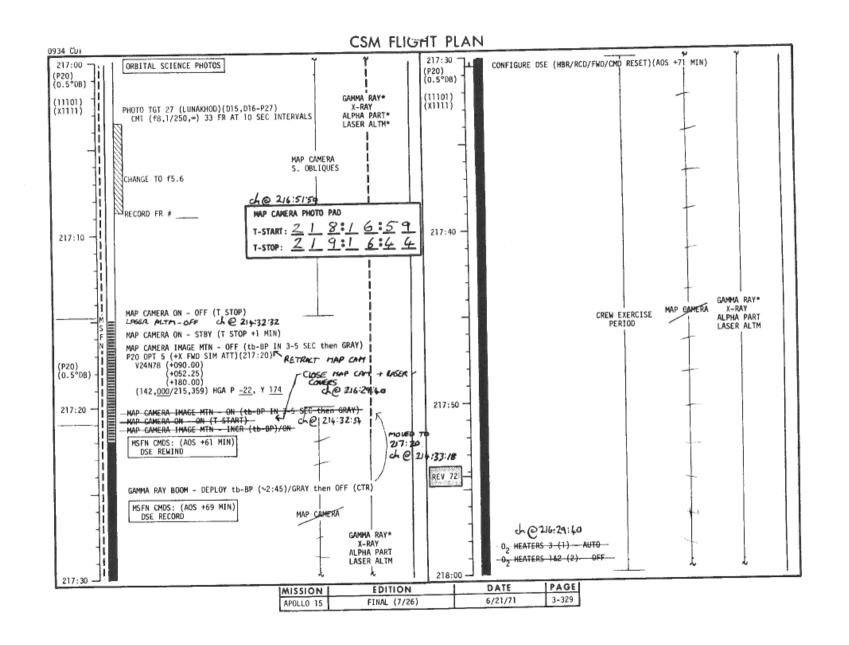


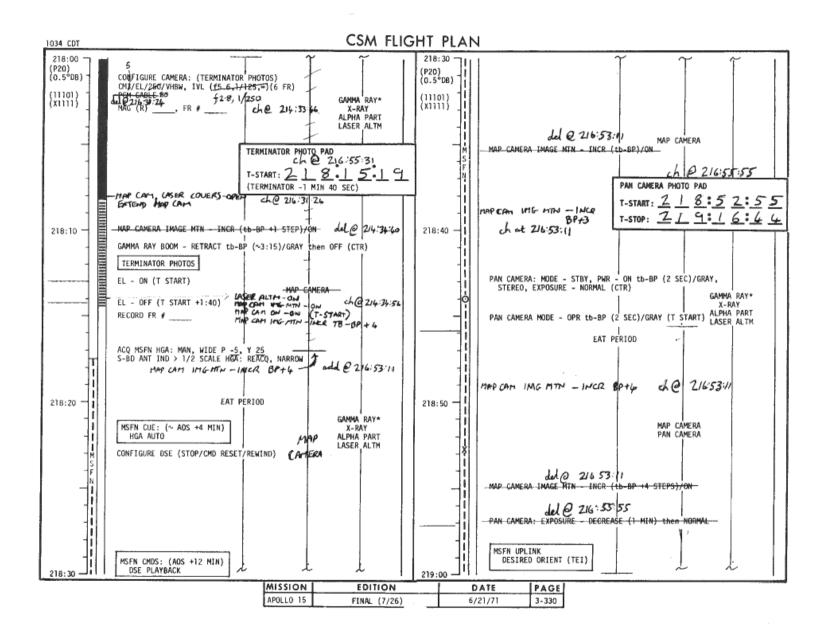


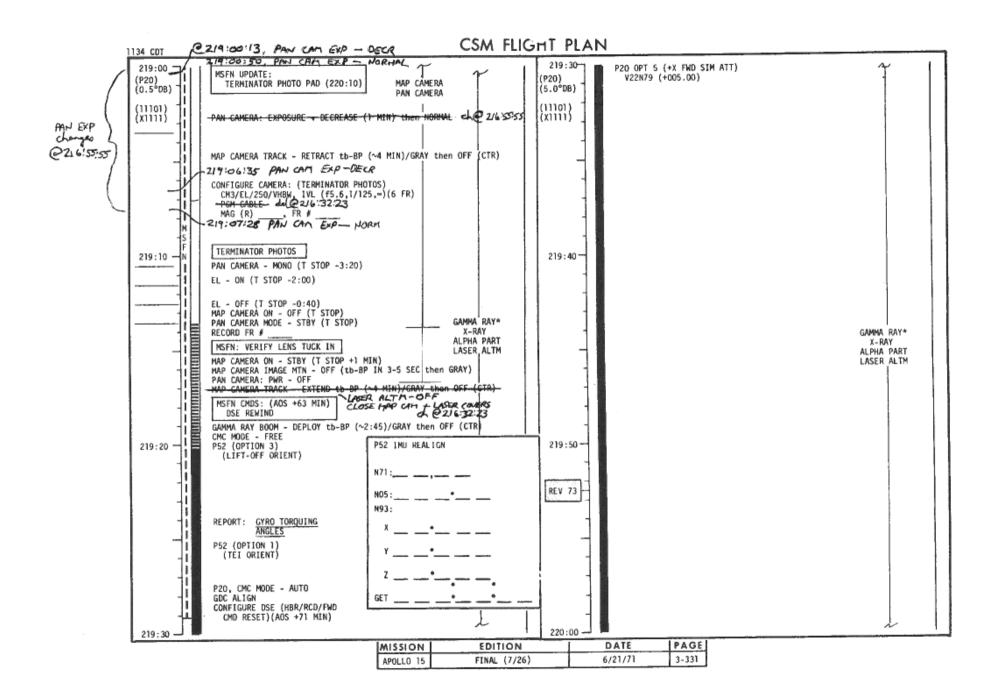


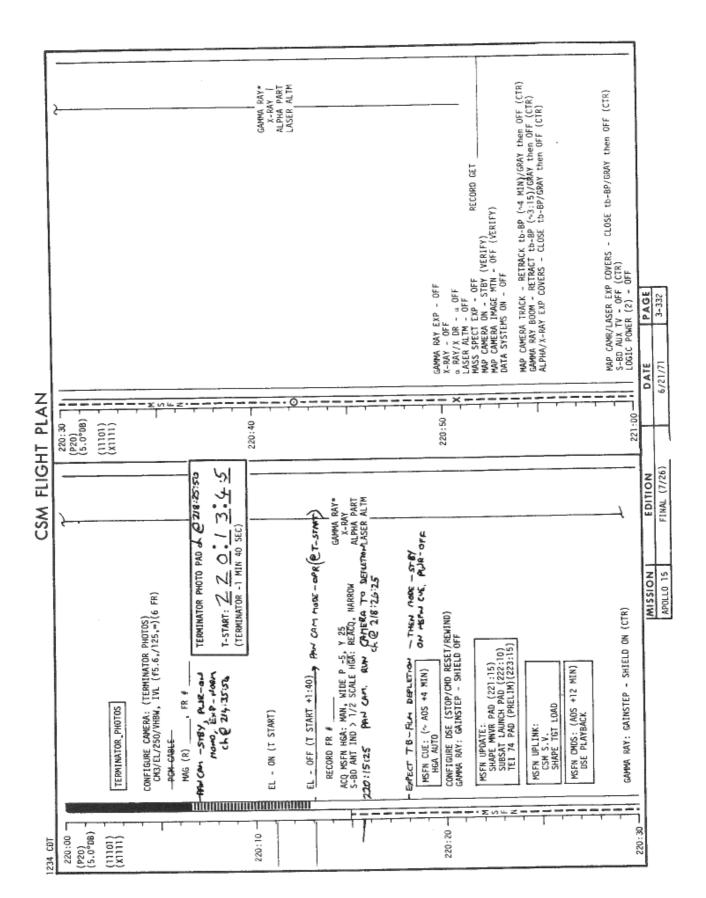


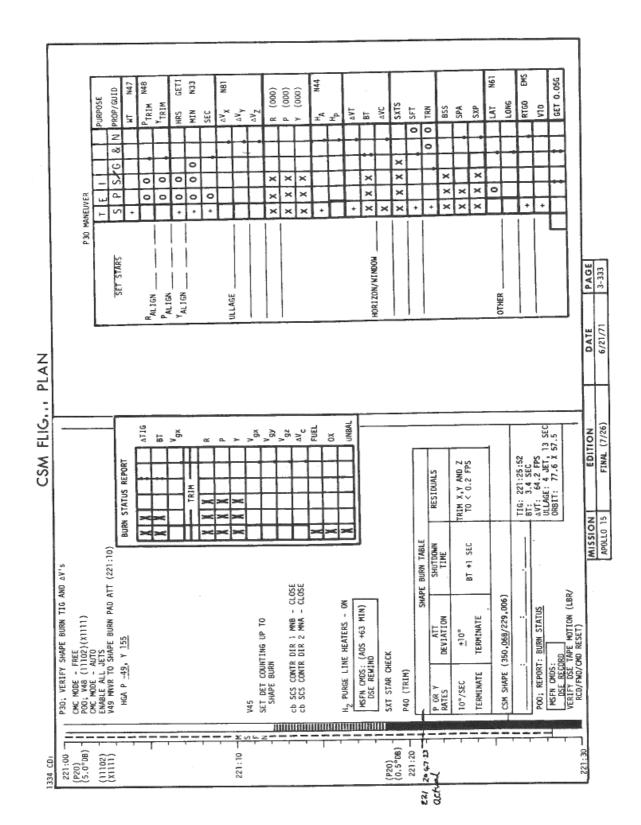


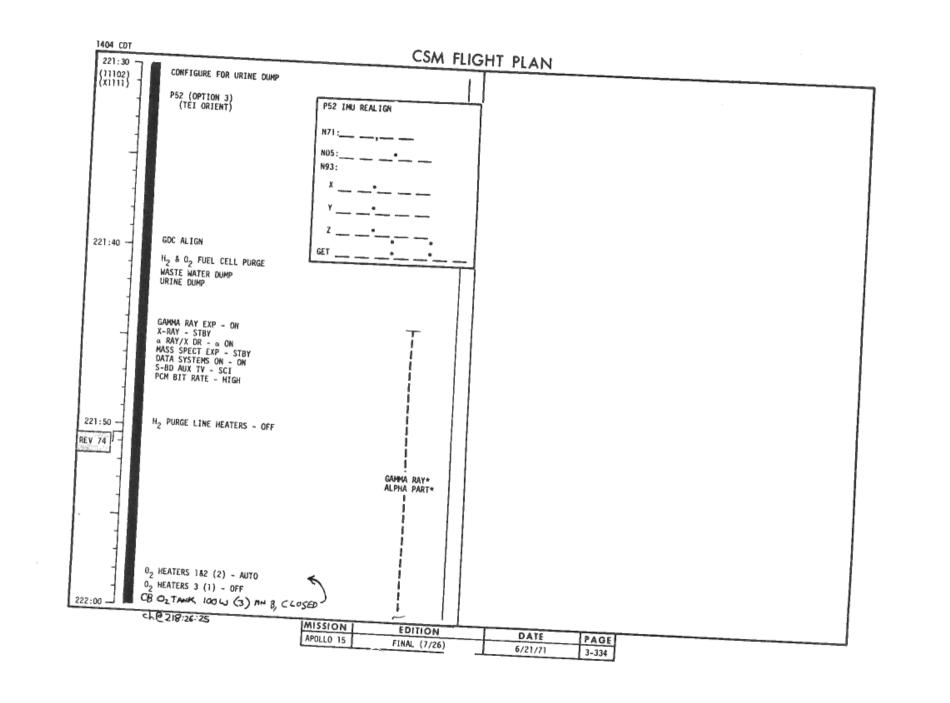


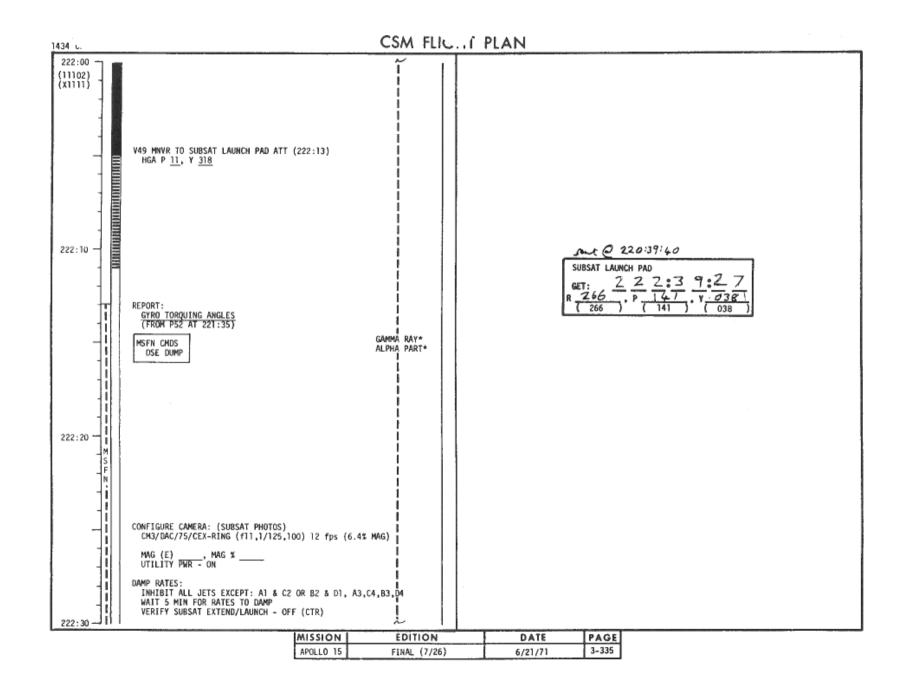


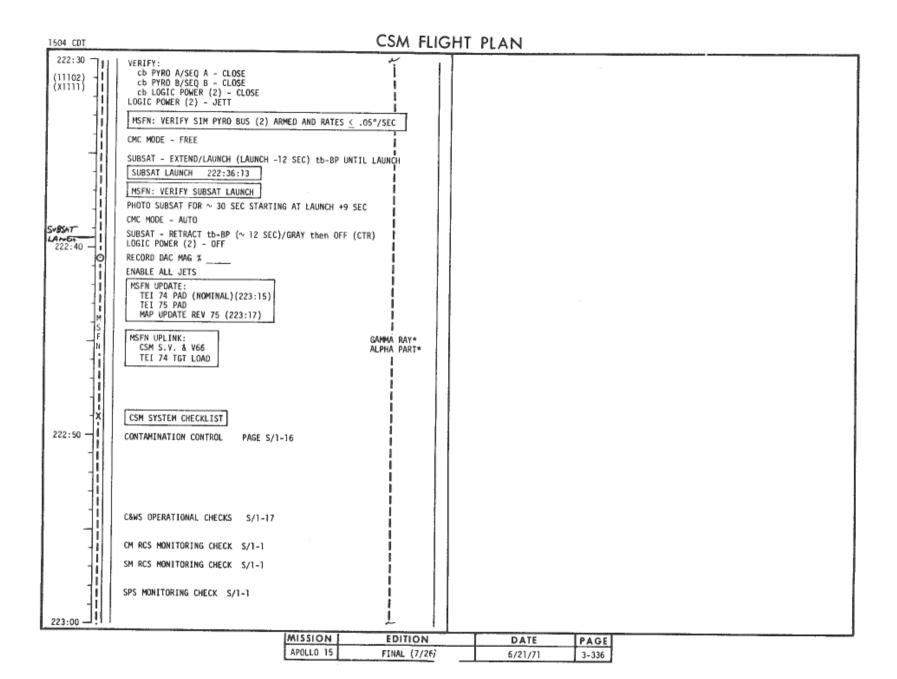


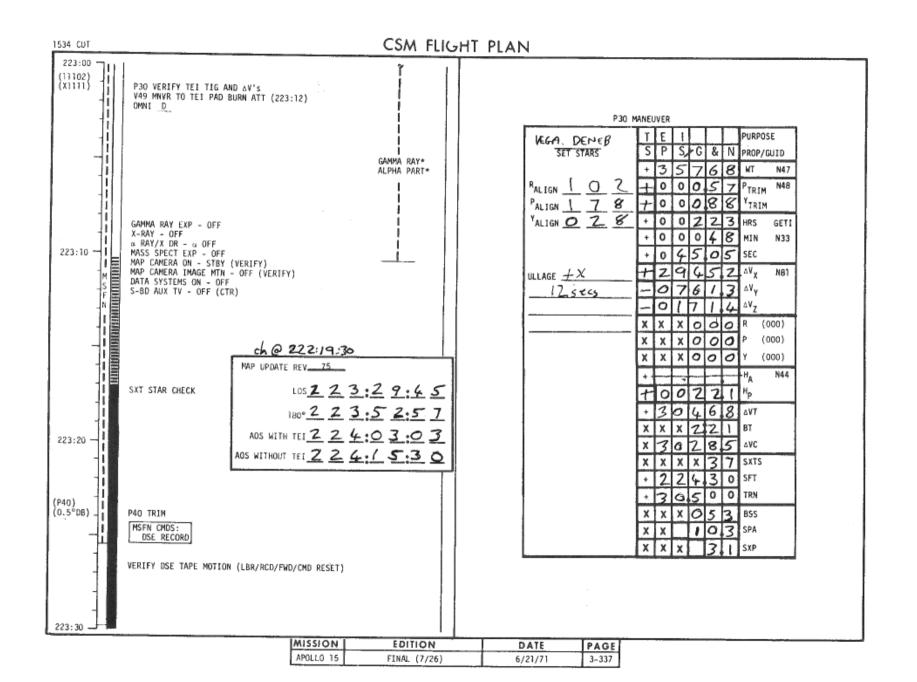


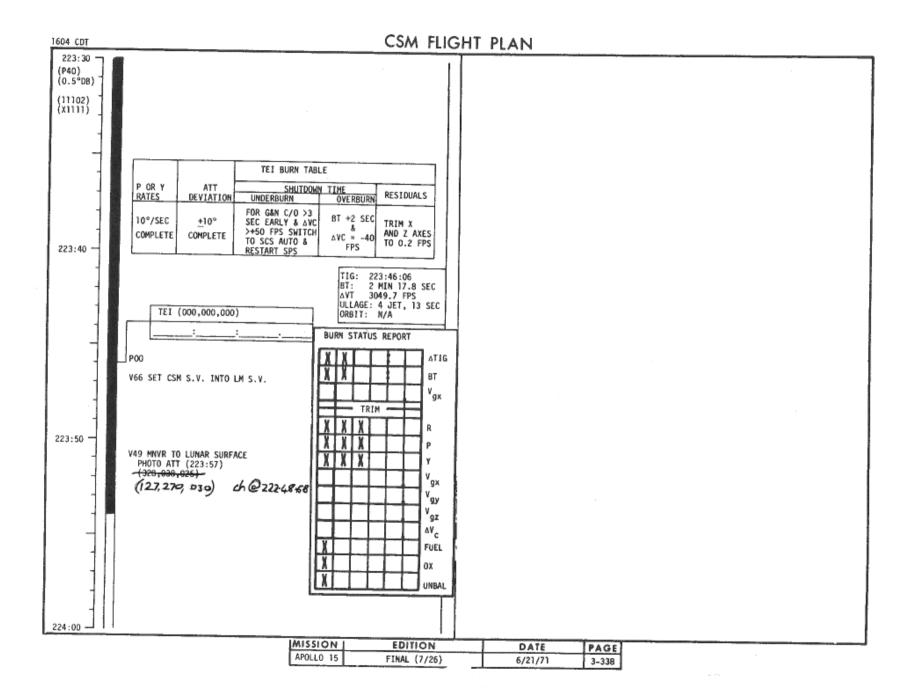




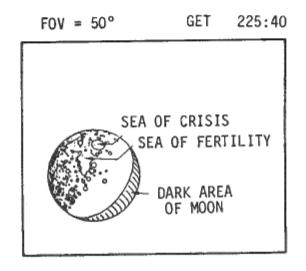






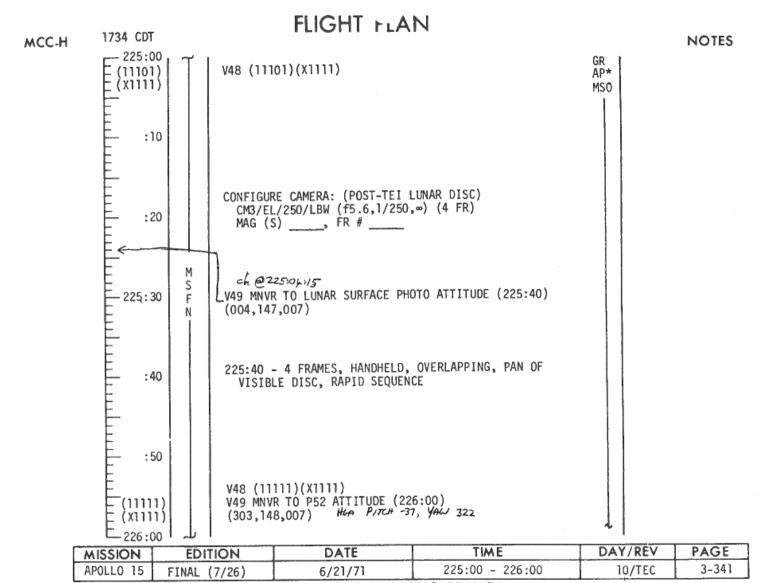


GAMMA RAY* ALPHA PART* MASS SPECT OUTGAS VH9 MANEUUNIN TO 127, 295, 030 HGA P23 Y229, GAMMA RAY: GAINSTEP - SHIELD ON (CTR) GAMMA RAY: GAINSTEP - SHIELD OFF NISUAL TOT 12 & 13 EASTERN MARIA MSFN UPLINK: DESIRED ORIENT (PTC) PAGE 3-339 NSFN CMDS: DSE DUMP 6/21/71 DATE CSM FLIGHT PLAN .52:00 224:50 224:40 224:30 -(11102) (X1111) 0 EDITION FINAL (7/26) T-START: 2 2 4:0 3:0 GAMMA RAY* ALPHA PART* MAP CAMERA TRACK - RETRACT tb-BP (~4 MIN)/GRAY then OFF (CTR)-- MAP CAMR/LASER EXP-COVEERS - CLOSE th-BP/GRAY then OFF (CTR)-ACQ MSFN OWNING C ALC 222:49:49
REPORT: BURN STATUS
INHIBIT ALL JETS EXCEPT: A1 & C2 OR B2 & D1, A3, C4, B3, D4
CD SCS CONTR DIR 1 MBL - OPEN
RHC PAR DIR (2) - OFF OR MNA/MNB
S-6D AUX TV - SCI MAP CAMERA TRACK - EXTEND tb-BP (~4 MIN)/GRAY then OFF (CTR) MASS SPECT BOOM - DEPLOY tb-8P (~2:40)/GRAY then OFF (CTR) MAP CAMR/LASER EXP COVERS - OPEN tb-8P/GRAY then OFF (CTR) MAP CAMERA ON - ON (T START)
MAP CAMERA INAGE MIN - INCR (tb-BP + STEPS)/OFF
2 - A 222:49:49 CA. @ 222:49:49 X-RAY - STBY
A RAY/N B - a ON
ARAY/NS SPECT EXP - STBY
WAS CAMERA INGE MTM ON (tb-8P IN 3-5 SEC then GRAY) MAP CAMERA PHOTO PAD MAP CAMERA GAMMA RAY: GAINSTEP - SHIELD ON (CTR), EXP - ON X-RAY - STBY APOLLO 15 MISSION MASS SPECT: EXP - ON, ION SOURCE - STBY RECORD GET A @ 222:49:49 MONITOR FOR FILM DEPLETION (tb-BP) PCM BIT RATE - HIGH LOGIC POWER (2) - DPLY/RETR MSFN UPDATE: MAP CAMERA PHOTO PAD DATA SYSTEMS ON - ON 224:30 224:20 -224:10 -(11102) (11111X) 224:00 \$ 222:49:49 1634 CDT deletion

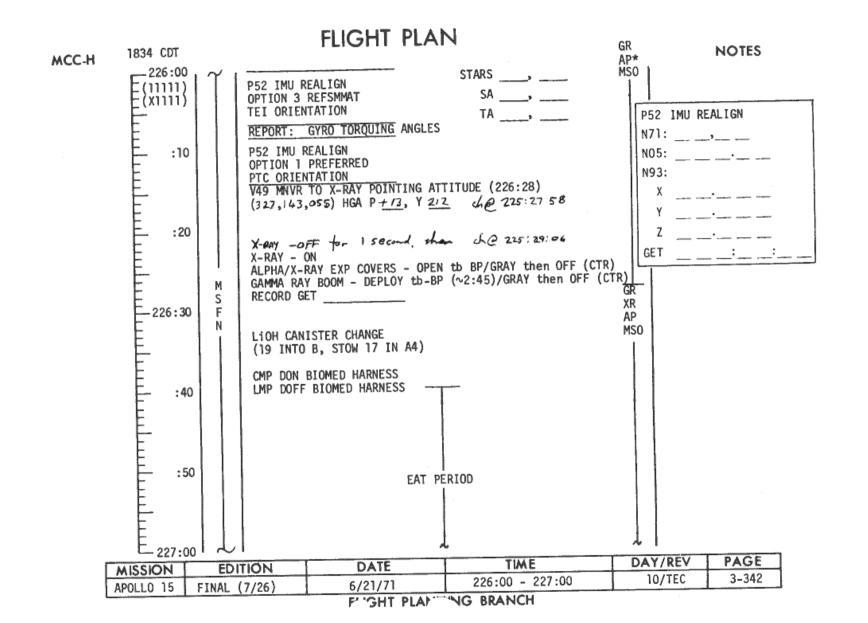


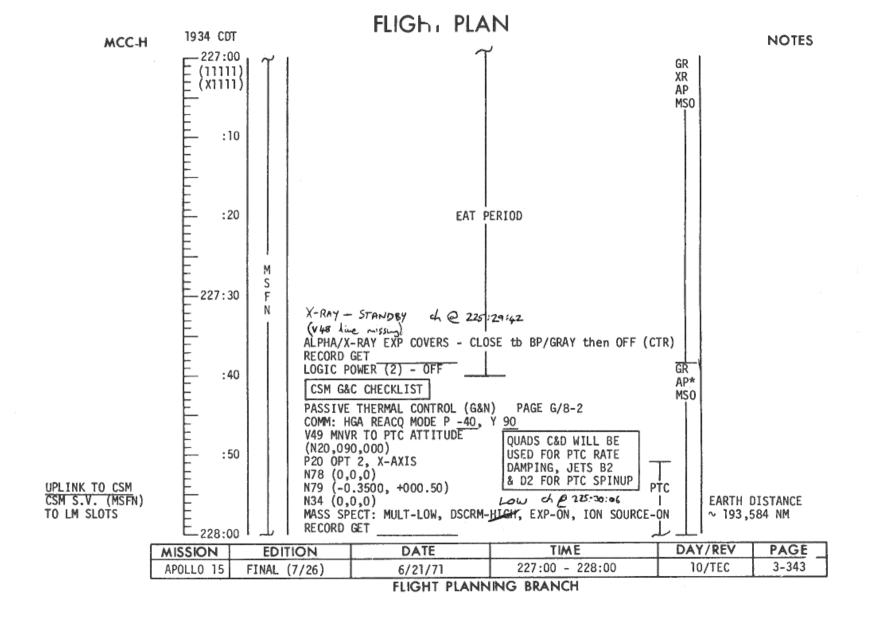
FINAL(7/26)

6/21/71



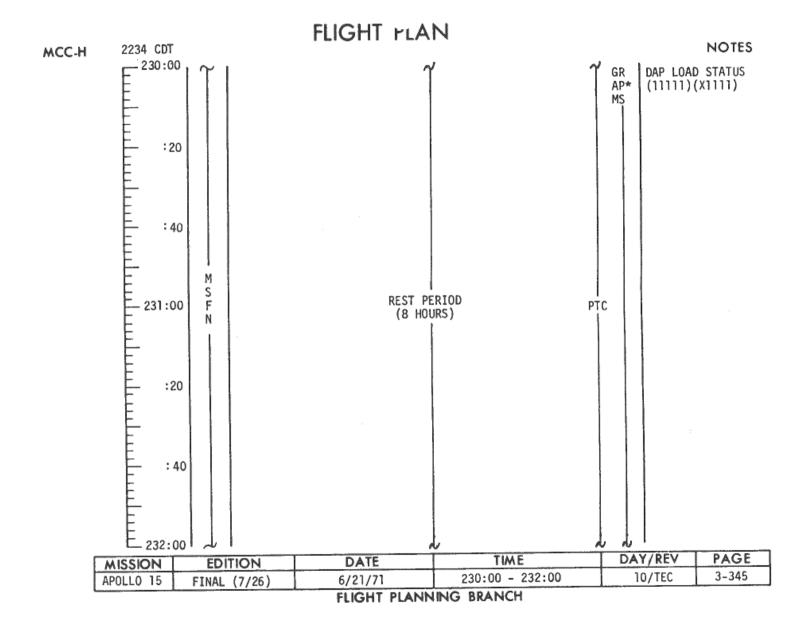
FLIGHT PLANNING BRANCH

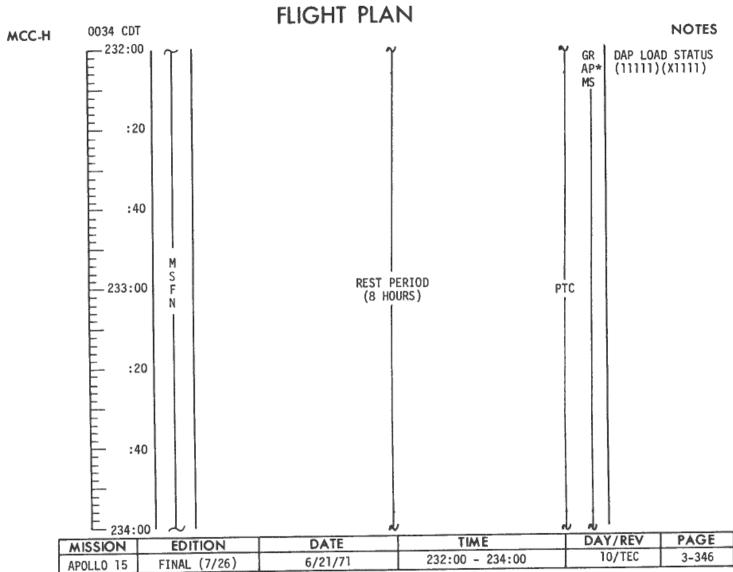




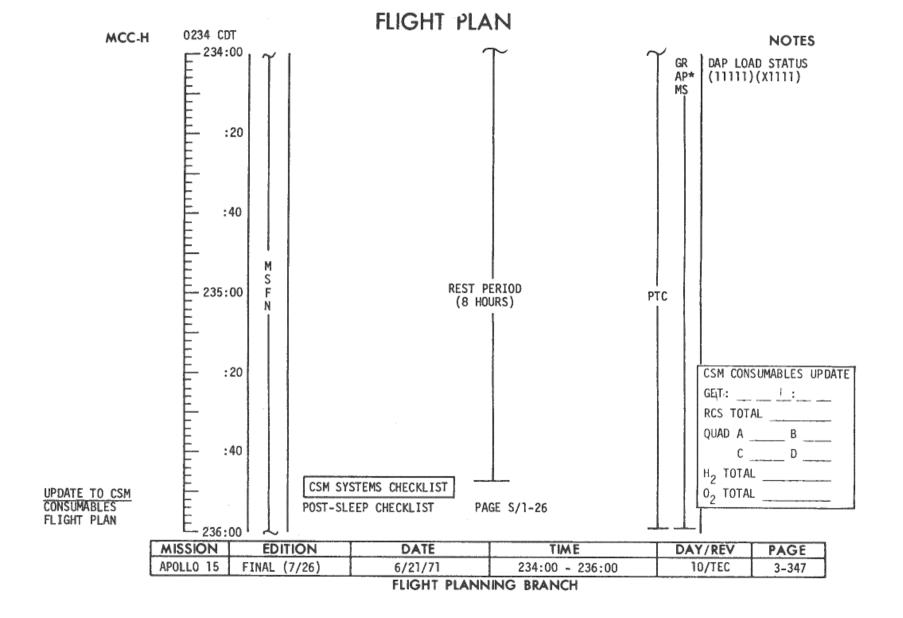
FLIGHT PLAN 2034 CDT MCC-H NOTES F-228:00 CSM SYSTEMS CHECKLIST GR DAP LOAD STATUS AP* (11111)(X1111)PRE-SLEEP CHECKLIST PAGE S/1-26 MS COMM - HGA ONBOARD READOUT : 20 BAT C PYRO BAT A PYRO BAT B RCS A _____ : 40 DC IND SEL - MNA OR B REST PERIOD ΡŤC - 229:00 (8 HOURS) :20 : 40 - 230:00 [|] DAY/REV MISSION DATE TIME EDITION PAGE 228:00 - 230:00 10/TEC APOLLO 15 FINAL (7/26) 6/21/71 3-344

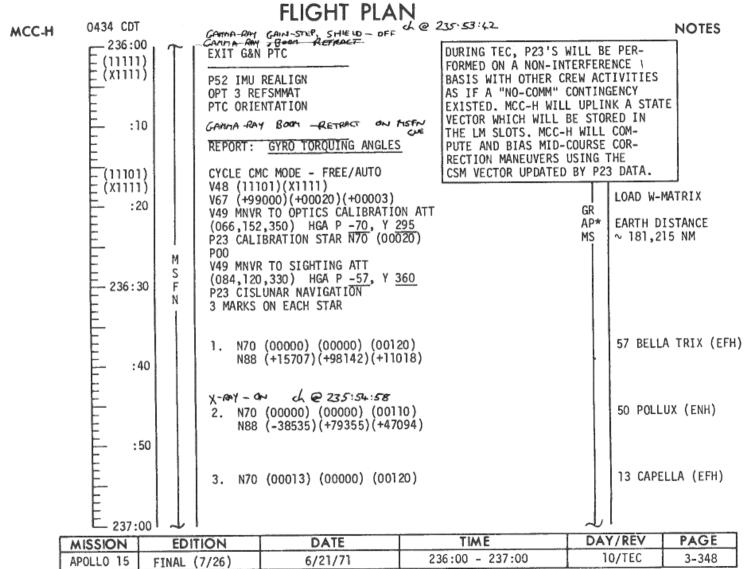
FLIGHT PLATING BRANCH





FLIGHT PLANNING BRANCH





FLIGHT PIANNING BRANCH

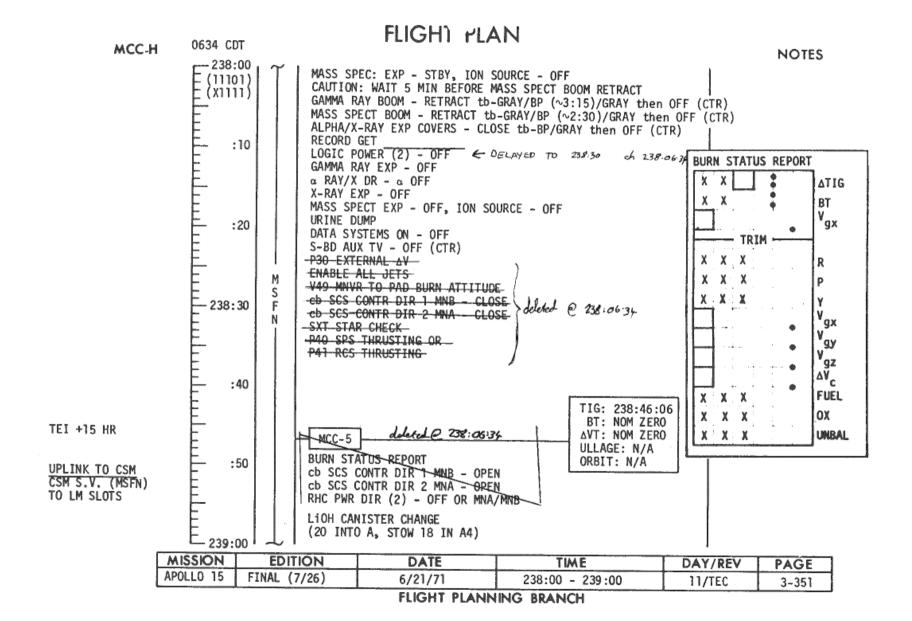
FLIGHT PLAN NOTES 0534 CDT MCC-H E (11101) E (X1111) V49 MNVR TO X-RAY POINTING ATTITUDE (237:15) () HGA P Y XR LOGIC POWER (2) - DEPLOY/RETR AP ALPHA/X-RAY EXP COVERS - OPEN tb BP/GRAY then OFF (CTR) MS :10 :20 UPLINK TO CSM -237:30 EAT PERIOD CSM S.V. (MSFN) TO LM SLOTS MCC-5 TGT LOAD EARTH DISTANCE UPDATE TO CSM MCC-5 MNVR PAD ~ 178,796 NM :40 :50 CONFIGURE FOR URINE DUMP | VERIFY REPRESS PKG 02 PRESSURE - 865-935 PSI 238:00 l DAY/REV PAGE TIME DATE MISSION EDITION 11/TEC 3-349 237:00 - 238:00 FINAL (7/26) 6/21/71 APOLLO 15

MCC-5 BURN TABLE

	_		_			
		KES I DUAL S	TRIM X AXIS ONLY	TO 0.2 FPS	TRIM X & Z	AXIS TO 0.2 FPS
	SHUTDOWN	TIME	BT + 1 SEC		BT + 1 SEC	0 = 2 \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \
1	ATT	DE VIALION	±10° COMPLETE		±10° TERMINATE	
	P OR Y RATES		10°/SEC COMPLETE	100/000	TERMINATE	
MANEUVER			CONTROL	1.0 CO.	TL CONTROL	

FINAL(7/26)

6/21/71



POSTPONED TO 247:36 CORONA WINDOW CALIBRATION UV PHOTOS - TRANSEARTH COAST V49 MNVR TO CORONA WINDOW CALIBRATION ATT (239:05) V49 MNVR TO EARTH UV PHOTO ATT (239:25) (090,011,025) HGA P -69, Y 191 239:08 (204,178,324) OMNI D 210,242,322 ch € 245:40:32 dre to postporenet (057,005,025) P-48 Y 238 ch@ 238:69:15 CONFIGURE CAMERA CONFIGURE CAMERA: (UV) CM4/EL/80/VHBW-BRKT (f2.8,1/500,∞) (5 FR) CM5/EL/105/UV, BRKT, CONT (f8,1/60,∞) (8 FR) MAG (R) , FR # RINGSLIDE MAG (N) , FR # INHIBIT JETS A3, C4, B3, D4 DAMP ROLES FOR 5 THINS, COIC - FREE REMOVE RT2 FLIGHT DATA FILE STOWAGE BOX - CYCLE 1 FRAME, CHANGE SHUTTER TO 1/125 DAMP RATES Lel @ 247:04:59 1 FRAME, CHANGE SHUTTER TO 1/60 -INHIBIT ALL JETS EXCEPT A1 & C2 or B2 & D1.A4, C3, B4, D3 1 FRAME, CHANGE SHUTTER TO 1/30 VERIFY FDAI SCALE 5/1 1 FRAME, CHANGE SHUTTER TO -1/500 1/125 WAIT 5 MIN FOR RATES TO DAMP CYCLE 1 FRAME ch @ 238.50.13 VERIFY RATES ARE <0.2°/SEC IN ALL AXES RECORD FR # REMOVE CM5 WINDOW COVER AND MOUNT UV CARDBOARD SHADE AND CAMERA CMC FREE ch @ 247:05:46 ENABLE ALL JETS CMC MODE-AUTO 2 FRAMES, FILTER 1, CHANGE SHUTTER TO B # FRAME \$, FILTER 2, EXP TIME 20 SEC and 1@2 sec exp. CHANGE SHUTTER TO 1/250 ch@ 11:40:01 2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500 2 FRAMES, FILTER 4 RECORD FR # CONFIGURE CAMERA: (UV COLOR) CM5/EL/105/CEX, BRKT, CONT (f8,1/250,∞) (1 FR) RINGLSIDE MAG (M) P , FR # _ 1 FRAME, FILTER 4 RECORD FR # REMOVE CAMERA AND UV CARDBOARD SHADE - CMC AUTO

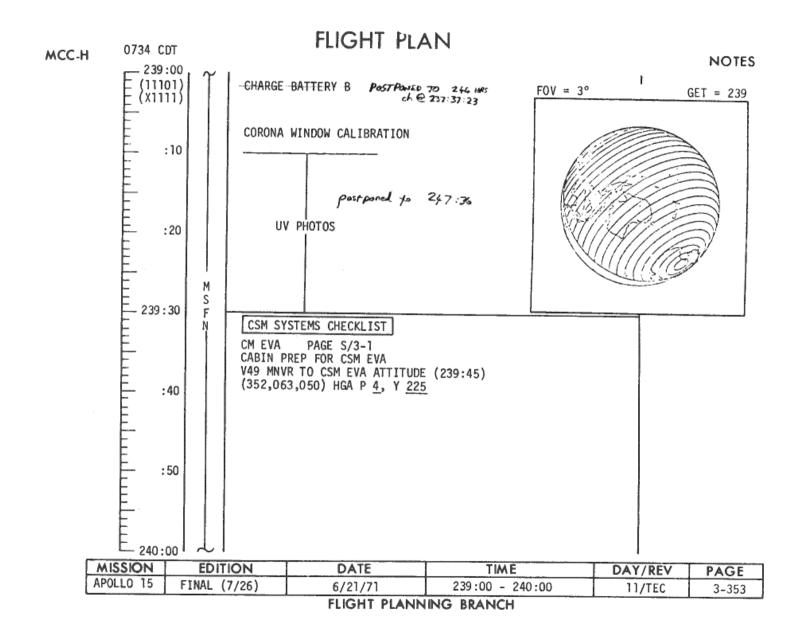
ch@

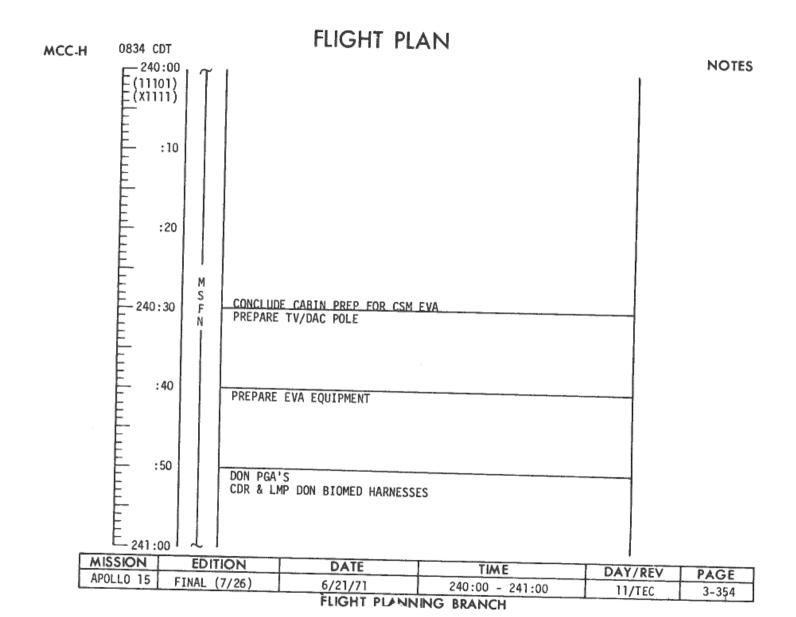
247:03.46

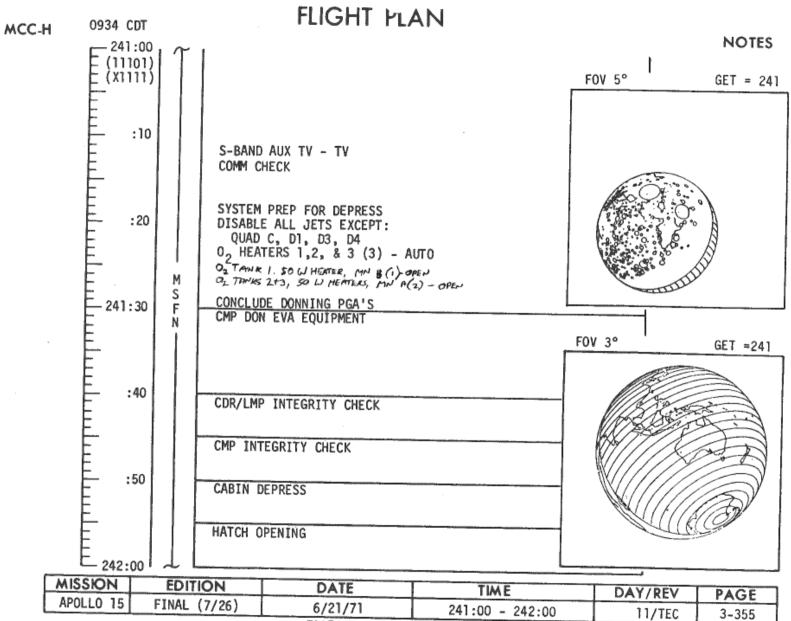
NOTE: COMMENTS AS TO CONDITION OF WINDOW 5

REPLACE CM5 WINDOW COVER

ENABLE ALL JETS







NOTES												CSM EVA LENGTH 1 HR. MAX		PAGE	3-356	
1										I		CSM EVA		DAY/REV	11/TEC	
														Δ 		
				ETTES			a							TIME	242:00 - 243:00	FLIGHT PI + NNING BRANCH
) ;	SS	IZATION	INSTALL TV/DAC, ADJUST	RETRIEVE MAPPING CAMERA CASSETTES		REST	PAN CAMERA CASSELLE			rv/DAC		TOSING	EPRESS	DATE	6/21/71	FLIGHT PI AN
	CMP EGRESS	FAMILIARIZATION	INSTALL	RETRIEVE		REST	RETRIEVE		REST	REMOVE TV/DAC	INGRESS	HATCH CLOSING	CABIN REPRESS	EDITION	FINAL (7/26)	
	_		<u> </u>		>	- :	Eυ							Γ	FINAL	
1034 CDT	242:00		l	ىبلى	м М	ىلى	111	242:30		тт 8	ببليب	ग्गाग ह	ر پ الىسلىد غ	NOISSIM	APOLLO 15	
MCC-H														_		_

FLIGHT FLAN 1134 CDT MCC-H **NOTES** ___ 243:00 (11101) (X1111) :10 OPS DEPLETION :20 CONCLUDE CABIN REPRESS DOFF PGA'S CMP & LMP DOFF BIOMED HARNESSES CLEANUP PROCEDURES 02 HEATERS 3 (1) - OFF - 243:30 :40 :50 STOW EQUIPMENT - 244:00 l MISSION EDITION DATE TIME DAY/REV PAGE APOLLO 15 FINAL (7/26) 6/21/71 243:00 - 244:00 11/TEC 3-357

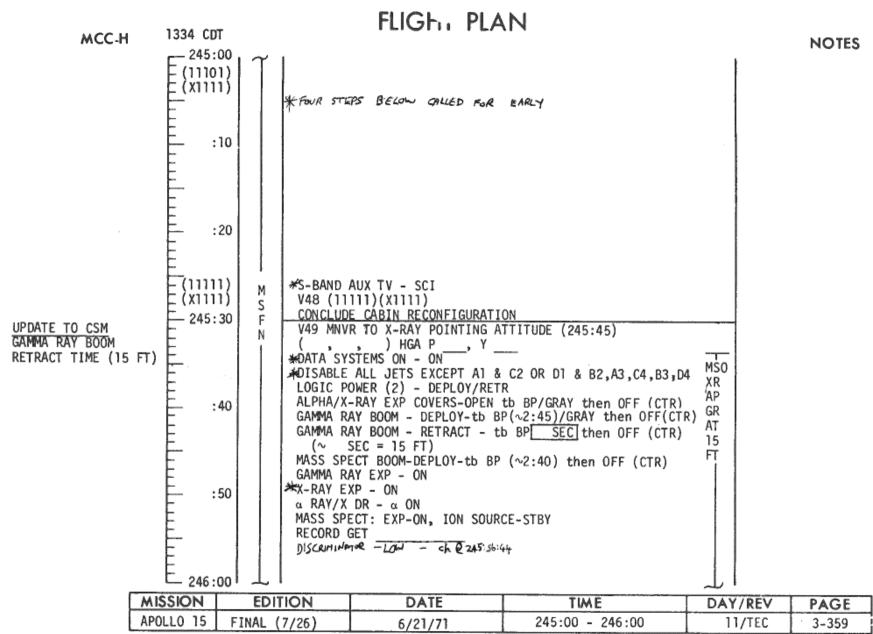
FLIGHT PLANNING BRANCH

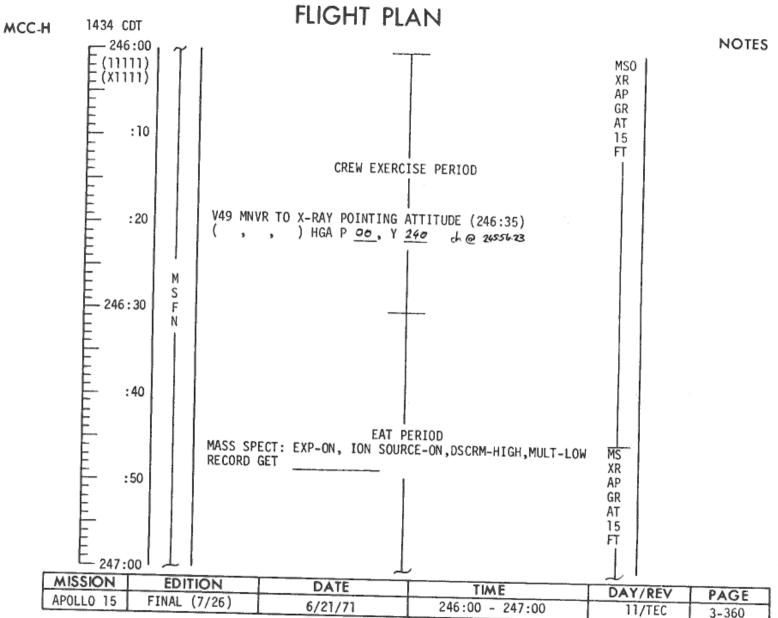
FLIGHT PLAN MCC-H 1234 CDT = 244:00 = (11101) = (X1111) **NOTES** WASTE WATER DUMP & 02 FUEL CELL PURGE CHARGE BATT B. DELAYED FROM 239 HRS Ch @ 237:37:23 :10 :20 244:30 :40 CONCLUDE EQUIPMENT STOWAGE RECONFIGURE CABIN :50

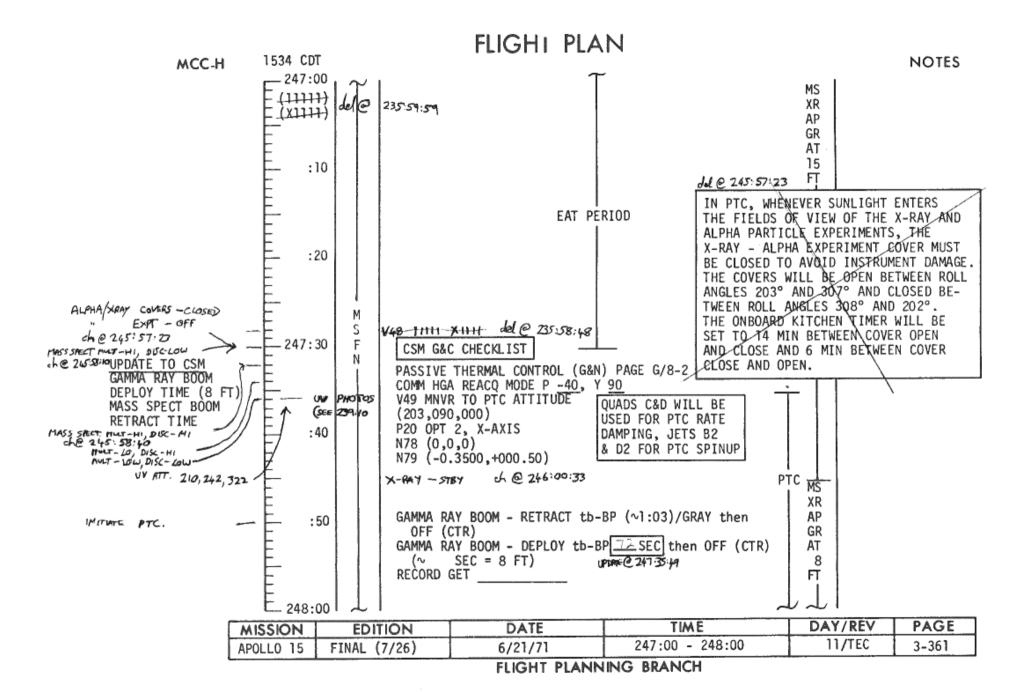
MISSION EDITION			1	
ADOLLO DE	DATE	TIME	DAY/REV	0.05
APOLLO 15 FINAL (7/26)	6/21/71	244:00 - 245:00		PAGE
	FLIGHT P' 'V	ING BRANCH	11/TEC	3-358

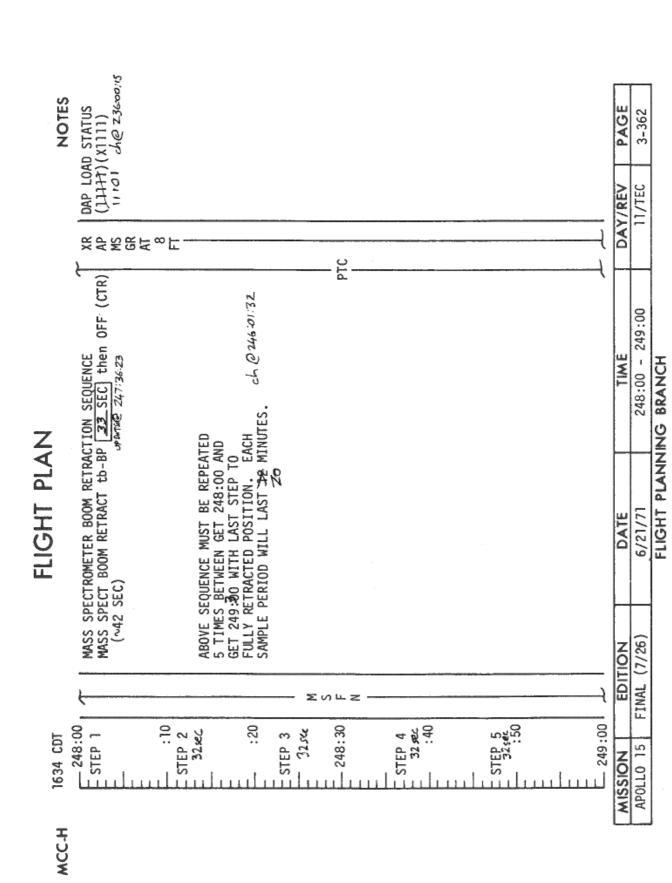
- 245:00 l

VING BRANCH

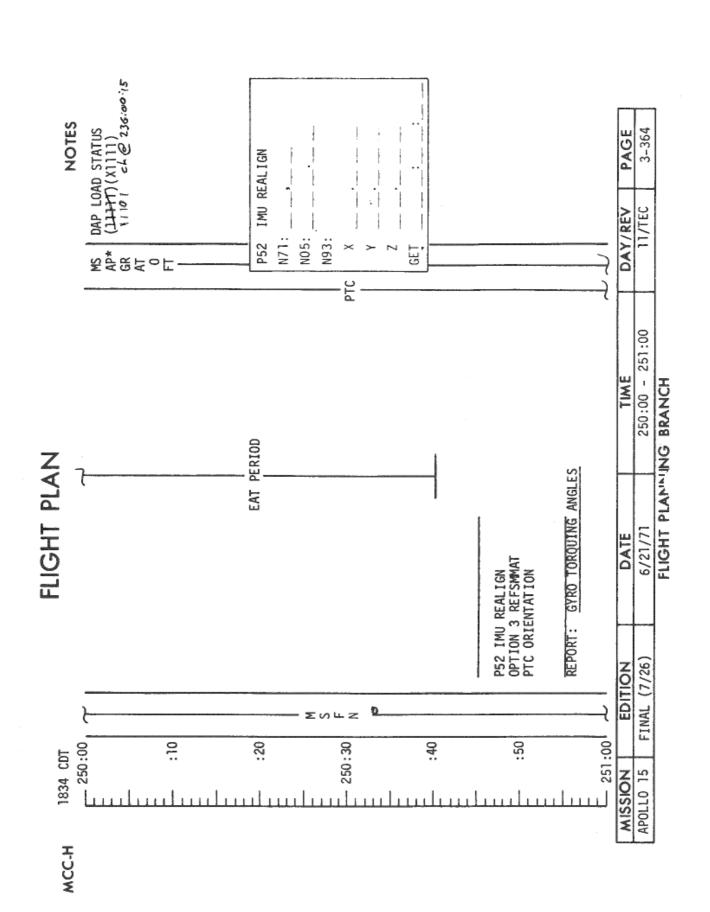








FLIGHT PLAN 1734 CDT MCC-H **NOTES** 249:00 CSM SYSTEMS CHECKLIST DAP LOAD STATUS XR (1111t) (X1111) 1/101 ch@ 236:00:15 AP CONTAMINATION CONTROL PAGE S/1-16 | MASS SPECT BOOM-DEPLOY tb-BP(~2:40)/GRAY then OFF (CTR) MOVETO 249:30 (ch @ 246:02 50) :10 FT :20 М CDR DOFF BIOMED HARNESS 249:30 PTC LMP DON BIOMED HARNESS ALPHA/X-RAY EXP COVERS-CLOSE tb-BP/GRAY then OFF (CTR) GAMMA RAY BOOM - RETRACT tb-BP/GRAY THEN OFF (CTR) MS AP* RECORD GET 63 secs deploy che 247:38:04 GR AT :40 0 FT EAT PERIOD :50 250:00 MISSION EDITION DATE TIME DAY/REV PAGE APOLLO 15 FINAL (7/26) 6/21/71 249:00 - 250:00 11/TEC 3-363



FLIGHT FLAN FLIGHT FLAN FLIGHT FLAN FLIGHT FLAN FLIGHT FLAN FLIGHT FLAN (21 INTO B, STOW 19 IN A4)

EXIT G&N PTC

P00

-V48 (11101) (X1111) - del @ 236:01:17

V49 MNVR TO OPTICS CALIBRATION ATT

(066,152,350) HGA P -69, Y 301

V49 MNVR TO SIGHTING ATT

P23 CISLUNAR NAVIGATION

3 MARKS ON EACH STAR

M € 236:01:38 - 111111)(X1111)

(N20,090,000)

CSM G&C CHECKLIST

P23 CALIBRATION STAR N70 (00020)

(084,141,330) HGA P -59, Y 001

1. N70 (00000) (00000) (00110) N88 (-38535)(+79355)(+47094)

2. N70 (00011) (00000) (00120)

N70 (00013) (00000) (00120)

COMM HGA REACQ MODE P -40, Y 90

V49 MNVR TO PTC ATTITUDE

PASSIVE THERMAL CONTROL (G&N) PAGE G/8-2

MCC-H

(11101)

(X1111)

:10

:20

251:30

:40

(11111)

 $\{x_{1111}\}$

:50

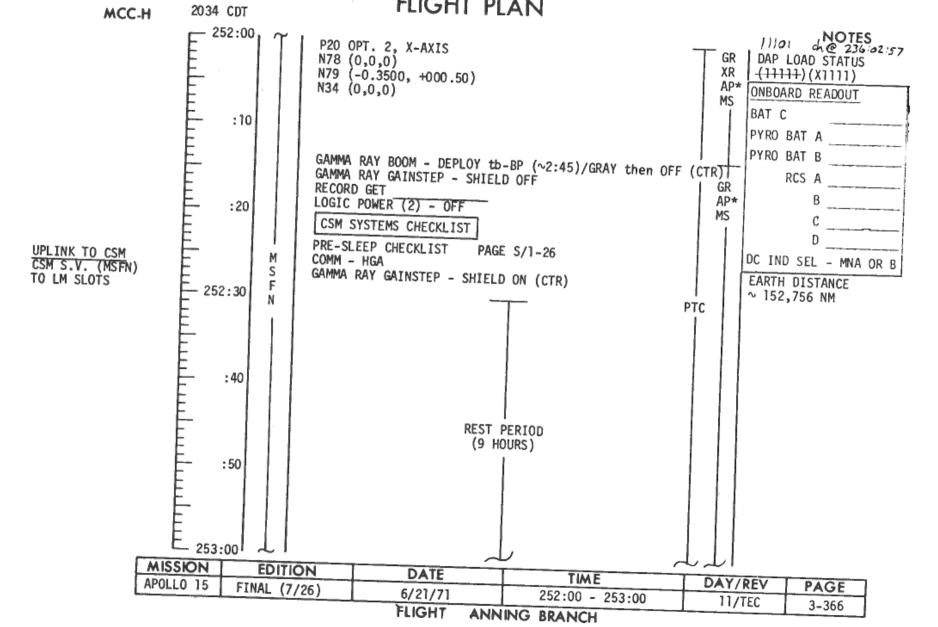
252:00

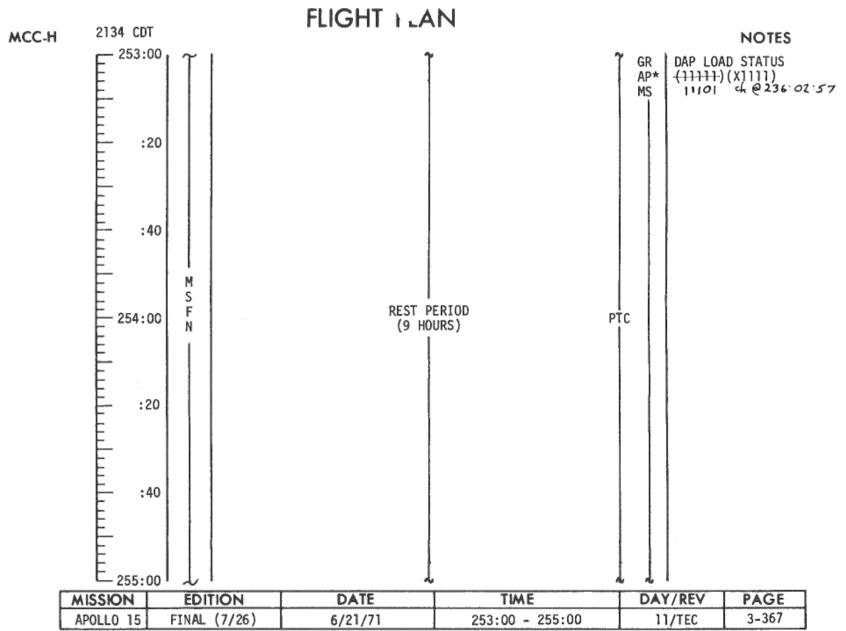
М

F

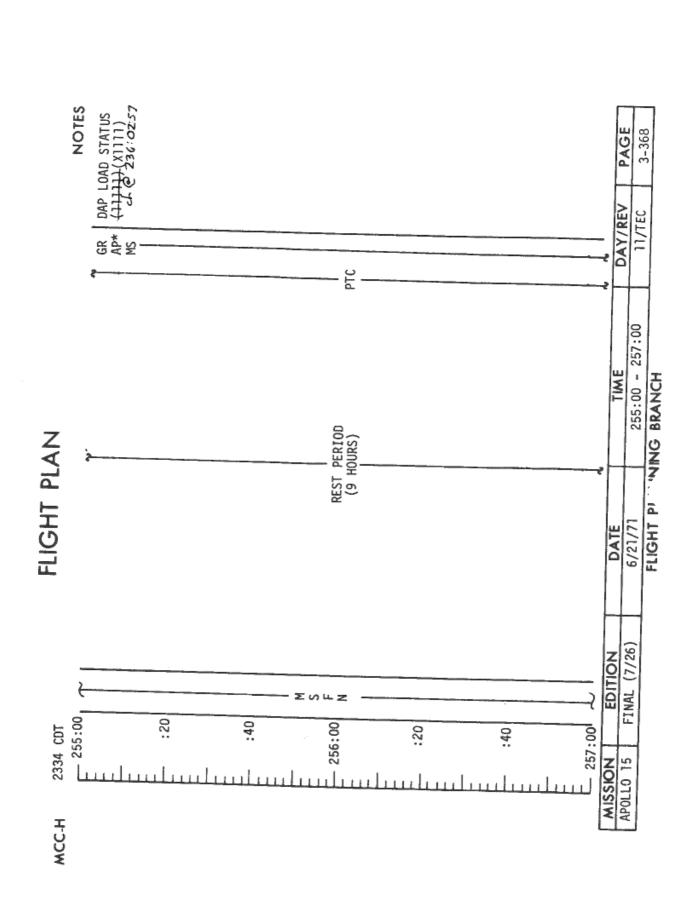
NOTES XR I DAP LOAD STATUS ΑP (11111)(X11111)11101 ch@ 236:00:15 MS GR AT FT EARTH DISTANCE ~ 154,731 NM 50 POLLUX (ENH) 11 ALDEBARAN (EFH) 13 CAPELLA (ÈFH) QUADS C&D WILL BE USED FOR PTC RATE DAMPING, B2 & D2 FOR PTC SPINUP

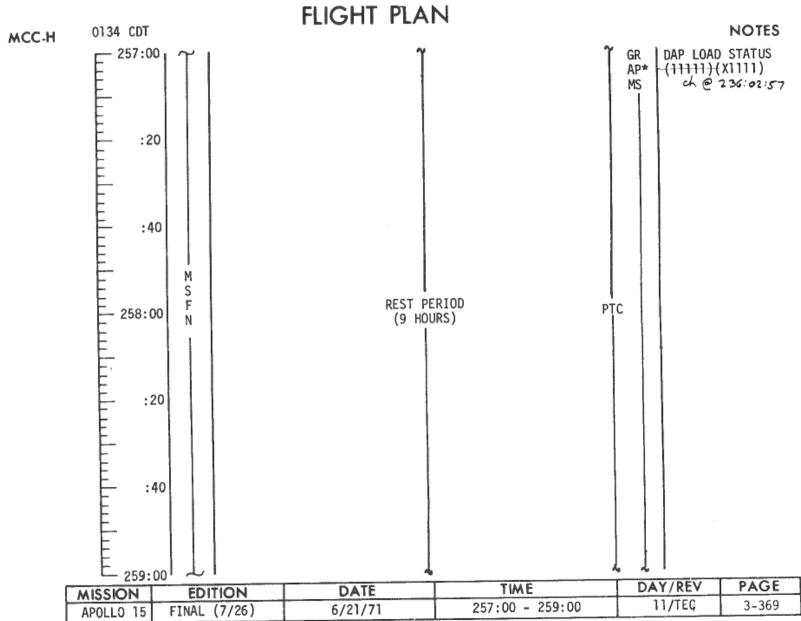
MISSION	EDITION	DATE	TIME	DAY/REV	PAGE	
APOLLO 15	FINAL (7/26)	6/21/71	251:00 - 252:00	11/TEC	3-365	



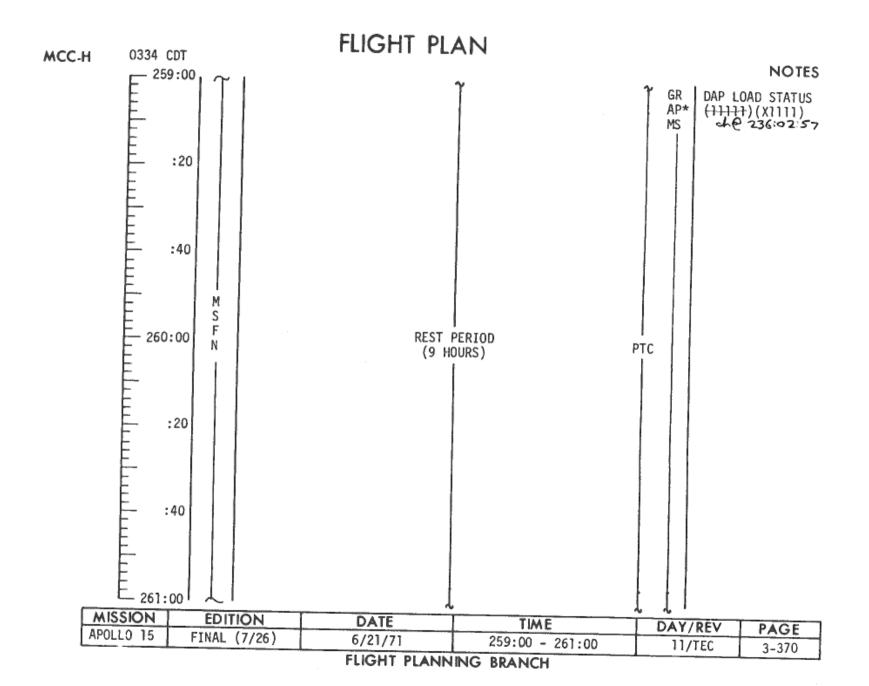


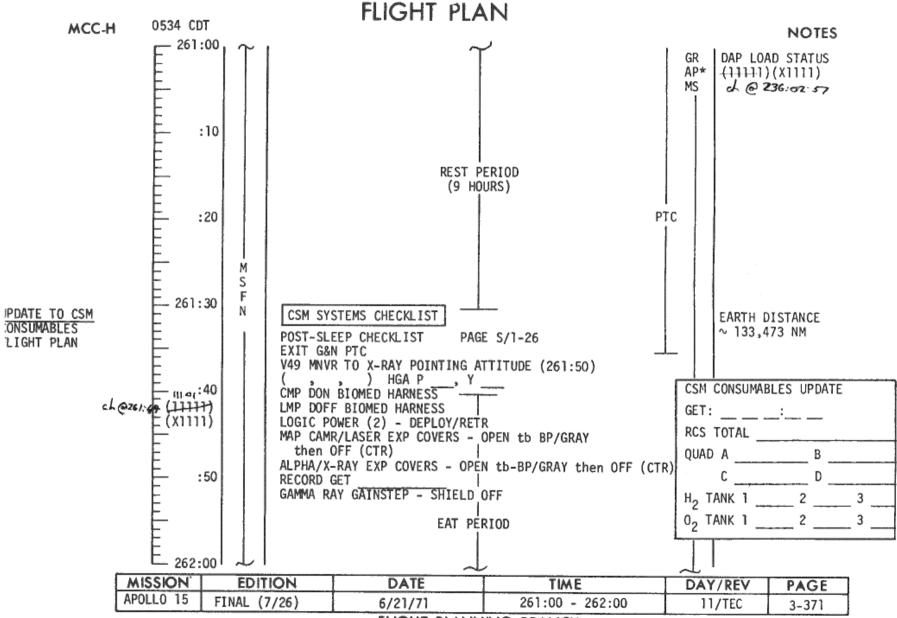
FLIGHT PLANNING BRANCH

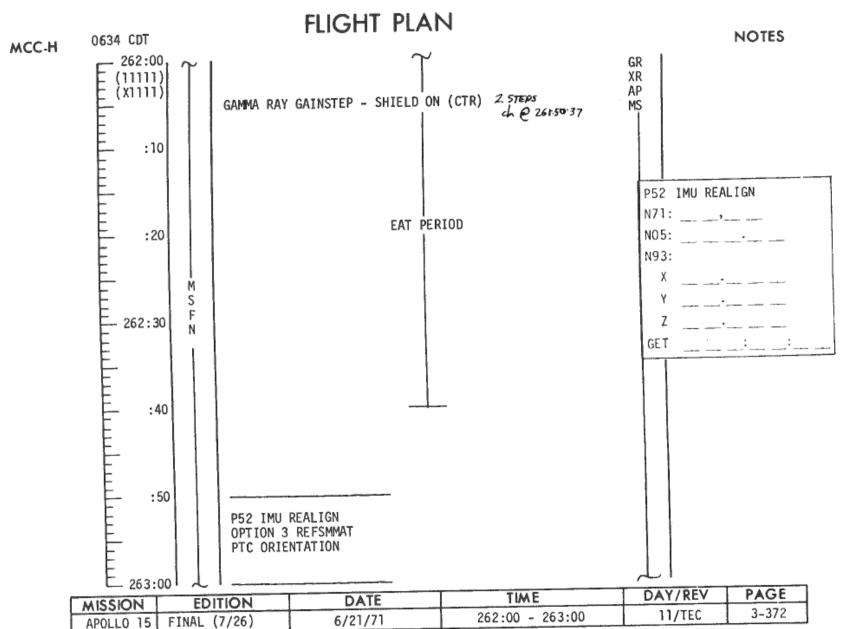




FLIGHT PLANNING BRANCH







FLIGHT PLANNING BRANCH

FLIGHT PLAN NOTES REPORT: GYRO TORQUING ANGLES XR LIOH CANISTER CHANGE (22 INTO A, STOW 20 IN A4) AΡ CYCLE CMC MODE - FREE/AUTO

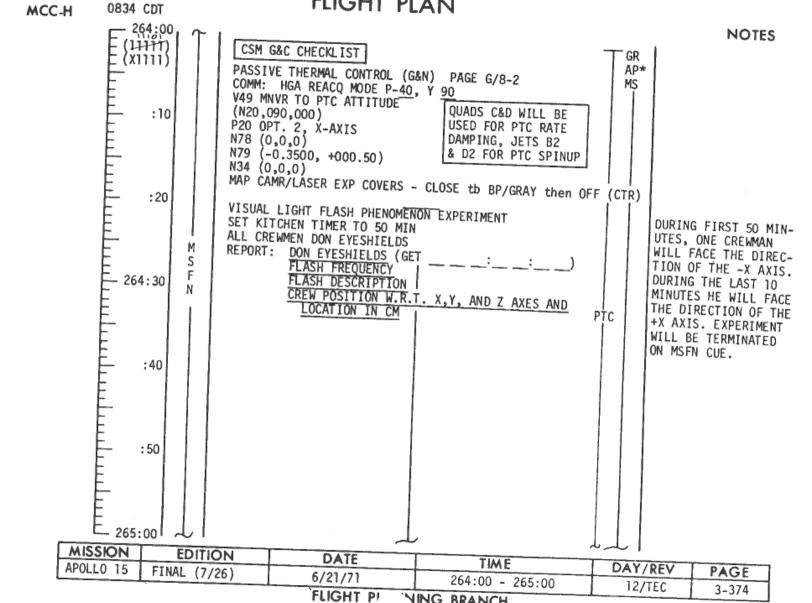
0734 CDT

MCC-H

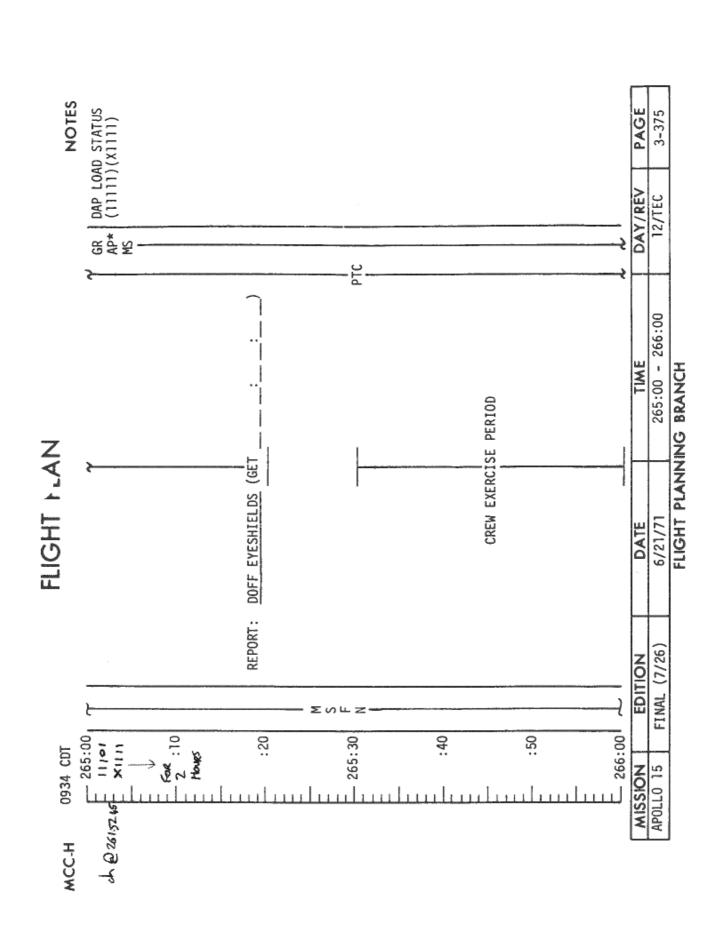
E (1111 (X1111)(11101) (x1111)V48 (11101)(X1111) :10 V49 MNVR TO OPTICS CALIBRATION ATTITUDE (066,152,350) HGA P -68, Y 307 P23 CISLUNAR NAVIGATION OPTICS CALIBRATION STAR N70 (00020) :20 P00 EARTH DISTANCE V49 MNVR TO SIGHTING ATTITUDE ~ 128,773 NM (082,166,330) HGA P -59, Y 000 P23 CISLUNAR NAVIGATION S 3 MARKS ON EACH STAR 263:30 142 ZETA TAURI (EFH) 1. N70 (00000) (00000) (00120) N88 (+09762)(+92766)(+36044) :40 11 ALDEBARAN (EFH) 2. N70 (00011) (00000) (00120) :50 13 CAPELLA (EFH) 3. N70 (00013) (00000) (00120) ALPHA/X-RAY EXP COVERS - CLOSE the BP/GRAY then OFF (JHHT) RECORD GET (X11111)changed @ 261:51:13 V48 (11111) (X1111) _ 264:00

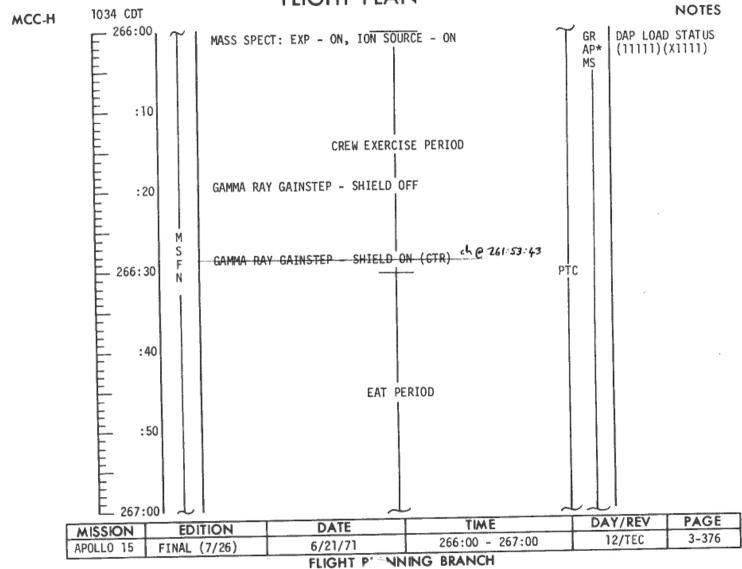
2011	• •				
MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
APOLLO 15	FINAL (7/26)	6/21/71	263:00 - 264:00	12/TEC	3-373

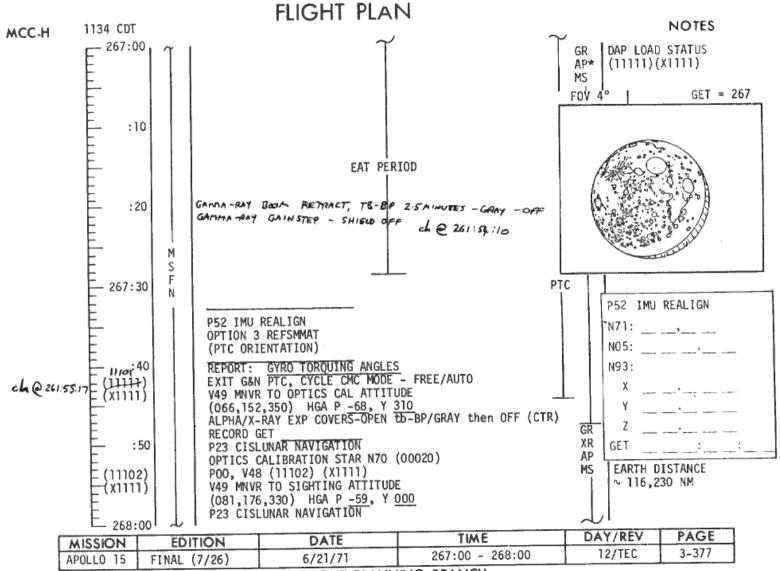
FLIGHT PLANNING BRANCH



'VING BRANCH





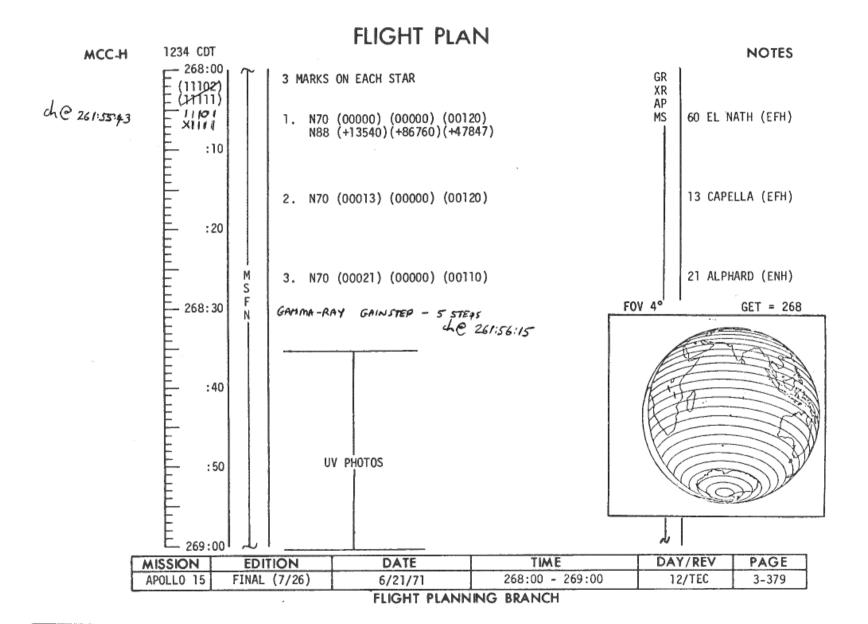


FLIGHT PLANNING BRANCH

UV PHOTOS - TRANSEARTH COAST

```
V49 MNVR TO EARTH UV PHOTO ATT (268:50)
               (217,045,332) OMNI D
          CONFIGURE CAMERA: (UV)
               CM5/EL/105/UV, BRKT, CONT (f4.3,1/60,∞) (8 FR)
               RINGSLIDE
               MAG (N)
                           , FR #
               REMOVE RT2 FLIGHT DATA FILE STOWAGE BOX
               INHIBIT ALL JETS EXCEPT AT & C2 or B2 & D1, A4, C3, B4, D3
               VERIFY FDAI SCALE 5/1
               WAIT 5 MIN FOR RATES TO DAMP
               VERIFY RATES ARE <0.2°/SEC IN ALL AXES
          REMOVE CM5 WINDOW COVER AND MOUNT UV CARDBOARD SHADE AND CAMERA
               2 FRAMES, FILTER 1, CHANGE SHUTTER TO B
             FRAMES, FILTER 2, EXP TIME 20 SEC and 1@ 2 sec exp.
                CHANGE SHUTTER TO 1/250
                                                                   da@ 11:40:01
              2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500
              2 FRAMES, FILTER 4
              RECORD FR #
          CONFIGURE CAMERA: (UV COLOR)
              CM5/EL/105/CEX, BRKT, CONT (f8,1/250,∞) (1 FR)
               RINGSLIDE
ch@241:57:49
              MAG (M)P
                        , FR # __
              1 FRAME, FILTER 4
               RECORD FR #
              REMOVE CAMERA AND UV CARDBOARD SHADE
          NOTE: COMMENTS AS TO CONDITION OF WINDOW 5
          REPLACE CM5 WINDOW COVER
          ENABLE ALL JETS
```

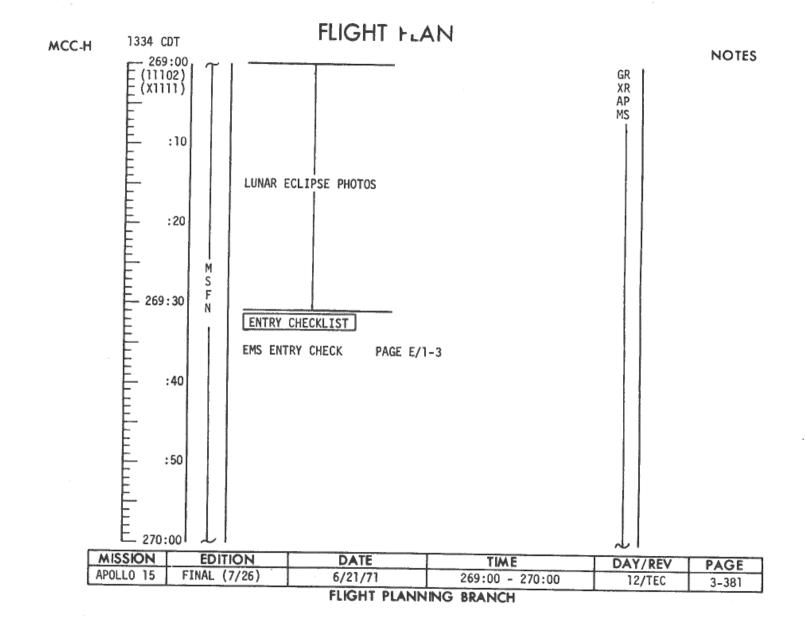
3-378



LUNAR ECLIPSE PHOTOS

CONFIGURE CAMERA: (LUNAR ECLIPSE) INSTALL CAMERA SHIELD CM4/NK/55/VHBW, BRKT (fl.2,1/500,∞) (8 FR) MAG () ____, FR # ____ CM2/EL/250/CEX (f5.6,1/500,∞) (8 FR) PCM CABLE MAG (P) ____, FR # ____ V49 MNVR TO LUNAR ECLIPSE ATT (269:00) (040.012.041) HGA P -22, Y 226 PHOTO SEQUENCE (for moon entering earth's umbra) 00:00 - MISSION TIMER, RESET/START (T START) DISABLE JETS A3, C4, B3, D4 DIM INTERIOR LIGHTS COVER EL LENS, CYCLE 1 FRAME CHANGE EL SHUTTER TO 1 SEC 05:00 - 1 FRAME EL CHANGE EL SHUTTER TO B 08:00 - 1 FRAME EL, EXP TIME 2 SEC CHANGE to 80mm LENS AND f2.8 CHANGE EL SHUTTER TO 1 SEC

11:00 - 1 FRAME EL CHANGE EL SHUTTER TO B 14:00 - 1 FRAME EL, EXP TIME 2 SEC 17:00 - 1 FRAME EL, EXP TIME 10 SEC CYCLE NK 1 FRAME, CHANGE NK SHUTTER TO T 19:00 - 1 FRAME EL, EXP TIME 2 MIN 1 FRAME NK, EXP TIME 2 SEC 20:00 - 1 FRAME NK, EXP TIME 4 SEC 21:00 - 1 FRAME NK, EXP TIME 8 SEC 22:00 - 1 FRAME NK, EXP TIME 15 SEC 23:00 - 1 FRAME NK, EXP TIME 30 SEC 24:00 - 1 FRAME NK, EXP TIME 60 SEC CHANGE NK SHUTTER TO 1/500 SEC CYCLE NK 1 FRAME CHANGE EL SHUTTER TO 1/500 COVER EL LENS, CYCLE 1 FRÂME LIGHTS UP, ENABLE JETS RECORD EL FR # RECORD NK FR #



SXT PHOTO TEST

CONFIGURE CAMERA

```
CM/DAC/SXT/VHBW (EXP 1/500) 24 fps (5% MAG)
                                                                          P52 (NO MARKS)
               MAG (H)
                           , MAG %
                                                                            N70 (00000)
da @ 261:58:18
              PCM CABLE
               UTILITY POWER - ON
                                                                            REMOVE DAC
             DISABLE JETS A3, B3, C4, D4
             P52 (NO MARKS)
                                                                          MOUNT DAC ON SXT
               N70 (00000)
               N88 (+78378)(+47837)(+39604)
             VERIFY THRU SXT THAT OPTICS BORESIGHTED ON STAR
             G&N PWR (AC - PNL 5) - OFF
             MOUNT DAC ON SXT
             DIM INTERIOR LIGHTS, DAC - ON AT 24 fps FOR 2 SEC
             CHANGE TO TIME & 1/60
              1 FRAME - 60 SEC EXP TIME
              1 FRAME - 20 SEC EXP TIME
              1 FRAME - 5 SEC EXP TIME
                                                                          RECORD MAG %
              1 FRAME - 1 SEC EXP TIME
            CHANGE TO 24 fps & 1/500
            RUN DAC FOR 2 SEC, LIGHTS UP
                                                                         ENABLE JETS
            V49 MNVR TO SXT PHOTO TEST ATT (270:15)
              (090,000,034) HGA P -54, Y 194
```

```
N70 (00000)

N88 (+78378)(+47837)(+39604)

REMOVE DAC

VERIFY THRU SXT THAT OPTICS BORESIGHTED ON STAR

MOUNT DAC ON SXT

DIM INTERIOR LIGHTS, DAC - ON AT 24 fps FOR 2 SEC

CHANGE TO T1 & 1/60

1 FRAME - 60 SEC EXP TIME
1 FRAME - 20 SEC EXP TIME
1 FRAME - 5 SEC EXP TIME
1 FRAME - 1 SEC EXP TIME
1 FRAME - 1 SEC EXP TIME
CHANGE TO 24 fps & 1/500

RUN DAC FOR 2 SEC, LIGHTS UP

RECORD MAG %

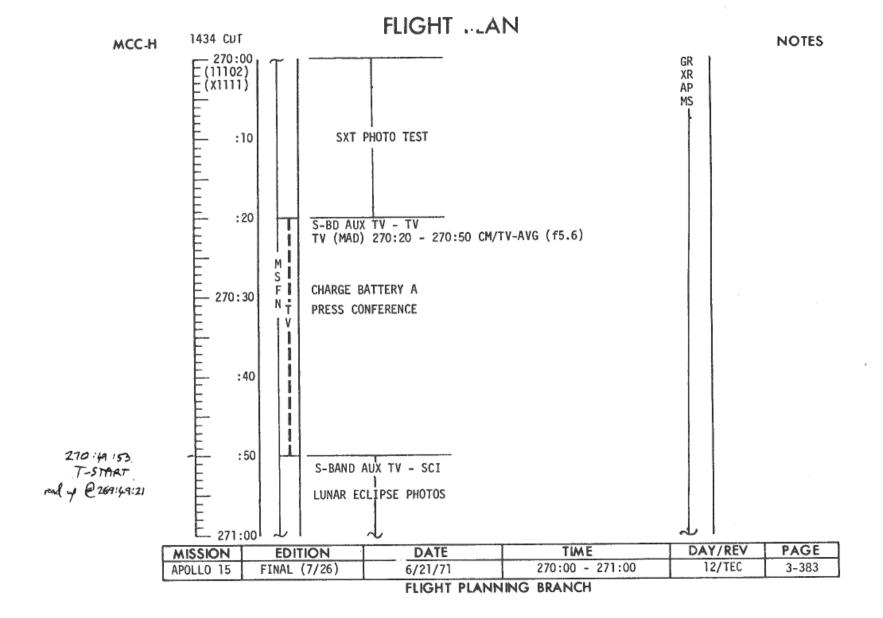
REMOVE AND STOW DAC

G&N PWR - AC1 OF AC2 (PNL 5)

ENABLE JETS

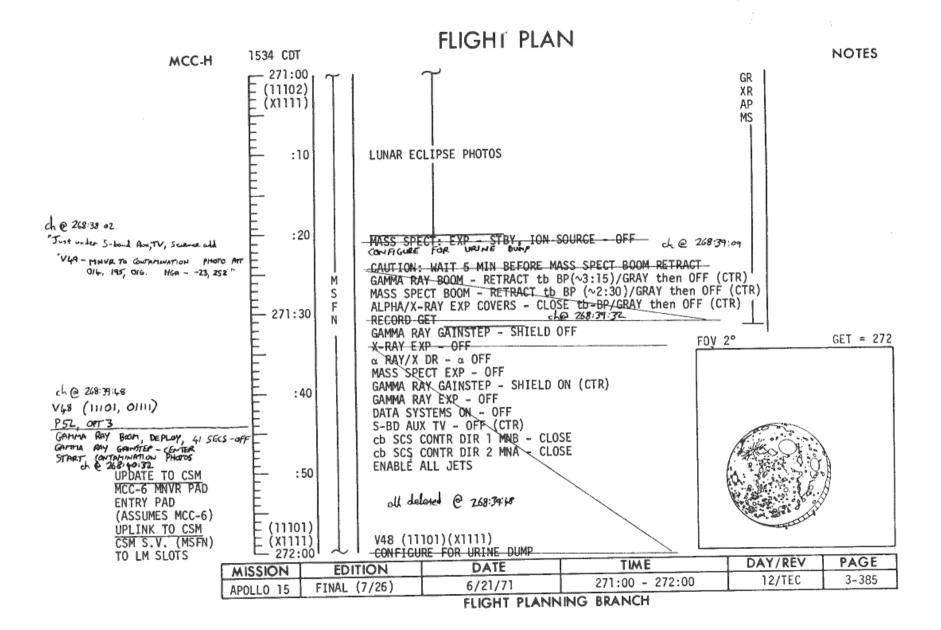
V49 MNVR TO LUNAR ECLIPSE ATT (270:50)

(040,011,041) HGA P -22, Y 227
```



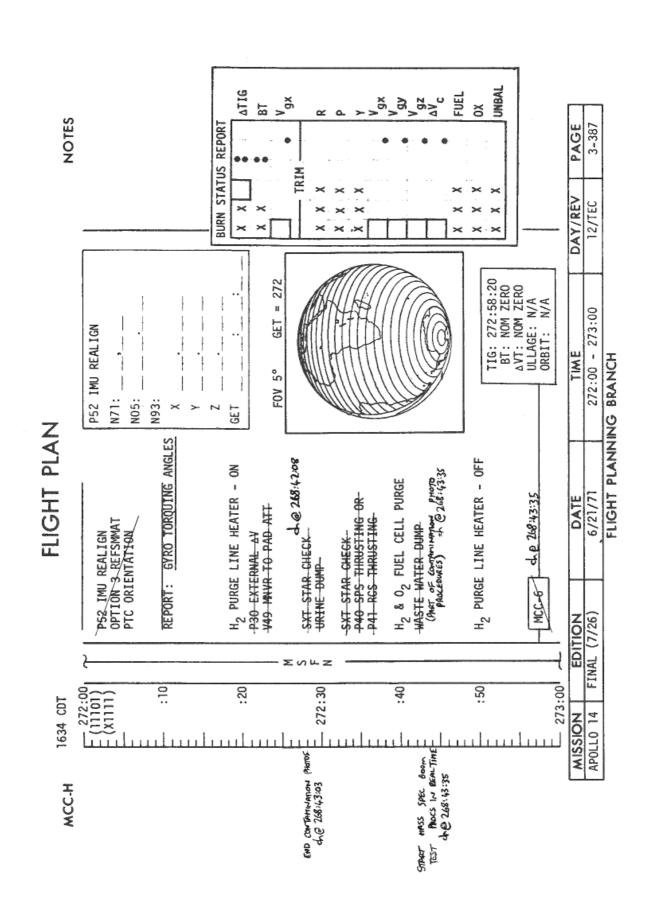
LUNAR ECLIPSE PHOTOS

CONFIGURE CAMERA: (LUNAR ECLIPSE) 11:00 - 1 FRAME NK, EXP TIME 2 SEC INSTALL CAMERA SHIELD 1 FRAME EL, EXP TIME 2 MIN CM4/NK/55/VHBW, BRKT (f1.2,1/500,∞) (8 FR) 13:00 - 1 FRAME EL, EXP TIME 10 SEC MAG () ____, FR # 16:00 - 1 FRAME EL, EXP TIME 2 SEC CHANGE EL SHUTTER TO 1 SEC CM2/EL/250/CEX (f5.6,1/500,∞) (8 FR) 19:00 - 1 FRAME EL PCM CABLE CHANGE TO 250mm LENS AND f5.6 MAG (P) _____, FR # ____ CHANGE EL SHUTTER TO B 22:00 - 1 FRAME EL, EXP TIME 2 SEC V49 TRIM TO LUNAR ECLIPSE ATT (270:50) CHANGE EL SHUTTER TO 1 SEC (040,011,041) HGA P -22, Y 227 25:00 - 1 FRAME EL CHANGE EL SHUTTER TO 1/500 SEC PHOTO SEQUENCE (for moon leaving earth's umbra) COVER EL LENS, CYCLE 1 FRAME CHANGE NK SHUTTER TO 1/500 SEC 00:00 - DET, RESET/START (T START) T-STARY 270:49:53 CYCLE NK 1 FRAME DISABLE JETS A3, C4, B3, D4 LIGHTS UP, ENABLE JETS DIM INTERIOR LIGHTS RECORD EL FR # CYCLÈ NK 1 FRAME, CHANGE NK SHUTTER TO T RECORD NK FR # 06:00 - 1 FRAME NK, EXP TIME 60 SEC STOW CAMERAS 07:00 - 1 FRAME NK, EXP TIME 30 SEC REMOVE CAMERA SHIELD 08:00 - 1 FRAME NK, EXP TIME 15 SEC 09:00 - 1 FRAME NK, EXP TIME 8 SEC 10:00 - 1 FRAME NK, EXP TIME 4 SEC COVER EL LENS, CYCLE I FRAME CHANGE EL SHUTTER TO B



MCC-6 CANCELLED BURN TABLE

IALS	TIS ONLY FPS	Z AXIS FPS	
RESIDUALS	TRIM X AXIS ONLY TO 0.2 FPS	TRIM X & Z AXIS TO 0.2 FPS	
SHUTDOWN	BT + 1 SEC AND $\Delta V_c = 0$	BT + 1 SEC AND $\Delta V_{c} = 0$	
ATT DEVIATION	+10° COMPLETE	+10° TERMINATE	
P OR Y RATES	10°/SEC COMPLETE	10°/SEC TERMINATE	
MANEUVER	CORRIDOR	IP CONTROL	



GR, XR, AP, MSO - ON (273:05)

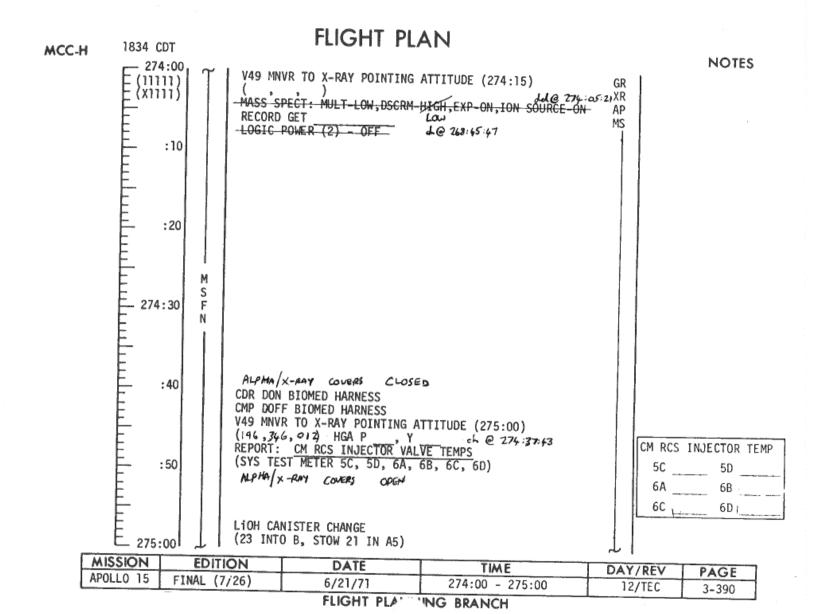
DISABLE ALL JETS EXCEPT A1 & C2 OR B1 & D2, A3, C4, B3, D4 cb SCS CONTR DIR 1 MNB - OPEN cb SCS CONTR DIR 2 MNA - OPEN RHC PWR DIR (2) - OFF OR MNA/MNB

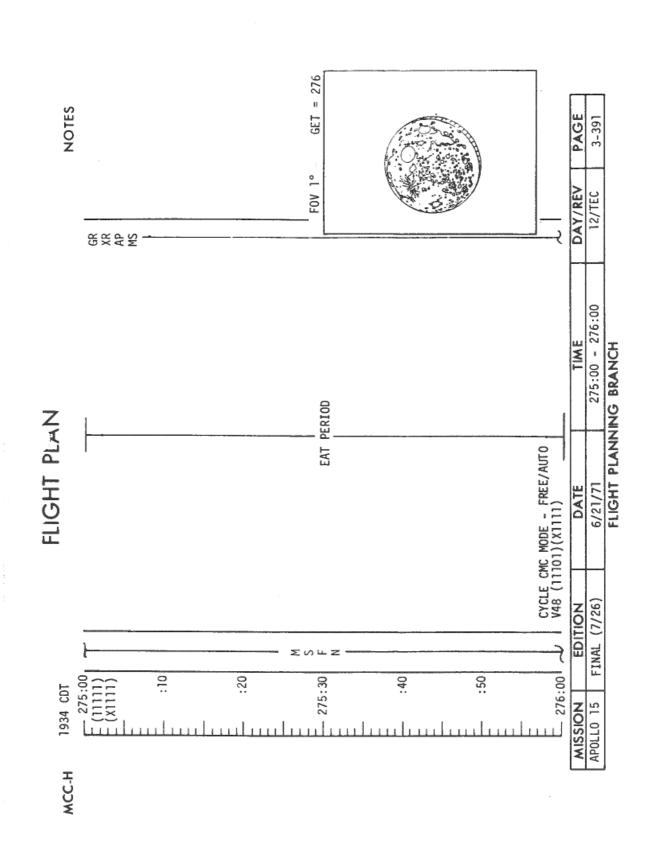
S-BD AUX TV - SCI DATA SYSTEMS ON - ON GAMMA RAY EXP - ON RAY/X DR - \alpha ON X-RAY EXP - ON

ALPHA/X-RAY EXP COVERS - OPEN tb - BP/GRAY then OFF (CTR) GAMMA RAY BOOM - DEPLOY tb - BP (~2:45)/GRAY then OFF (CTR) MASS SPECT BOOM - DEPLOY tb - BP (~2:40)/GRAY then OFF (CTR) MASS SPECT: EXP - ON, ION SOURCE - STBY RECORD GET

FLIGHT PLAN 1734 CDT MCC-H NOTES 273:00 E (11101) E (X1111) ch @ 268:44:25 UPLINK TO CSM BURN STATUS REPORT X-RAY-ON, CONGES -OPEN, RECORD GET GR CSM S.V. (MSFN) XR TO LM SLOTS GR, XR, AP, MSO ON V49 MNVR TO OPTICS CALIBRATION ATTITUDE AP MSO (066,152,350) HGA P -67, Y 314 P23 CISLUNAR NAVIGATION :10 OPTICS CALIBRATION STAR N70 (00020) V49 MNVR TO SIGHTING ATTITUDE EARTH DISTANCE (079,185,330) HGA P -59, Y 359 ~ 102,349 NM P23 CISLUNAR NAVIGATION 3 MARKS ON EACH STAR :20 1. N70 (00000) (00000) (00120) N88 (+13540)(+86760)(+47847) 60 EL NATH (EFH) BETA TAURI 273:30 2. N70 (00013) (00000) (00120) 13 CAPELLA (EFH) :40 3. N70 (00021) (00000) (00110) 21 ALPHARD (ENH) - GARNA-PANY BOOM, DEPLOY Ju @ 268:45:05 :50 (11111) V48 (11111)(X1111) (xiiii) E 274:001 PAGE MISSION TIME DAY/REV EDITION DATE FINAL (7/26) 6/21/71 273:00 - 274:00 12/TEC APOLLO 15 3-389

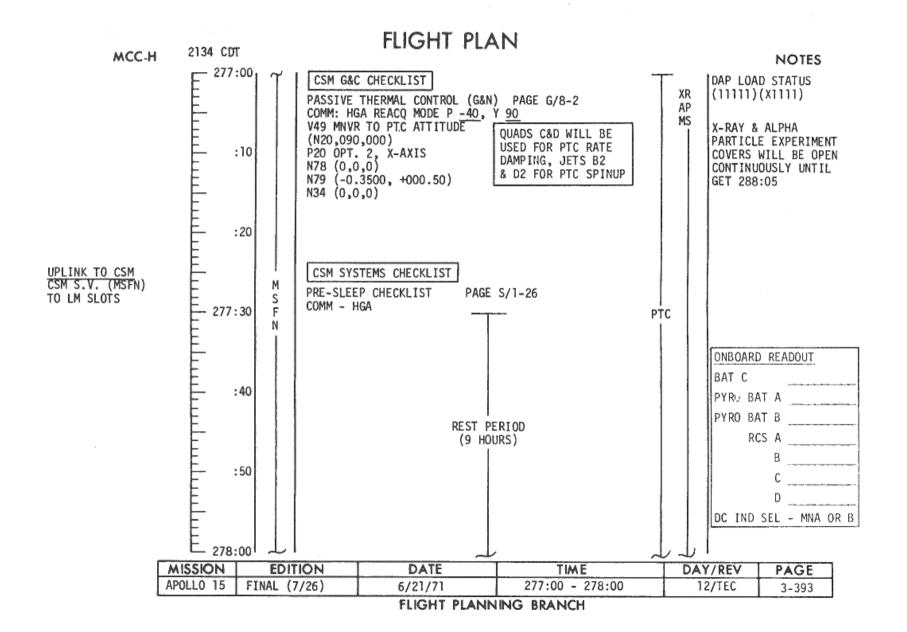
FLIGHT PLANNING BRANCH

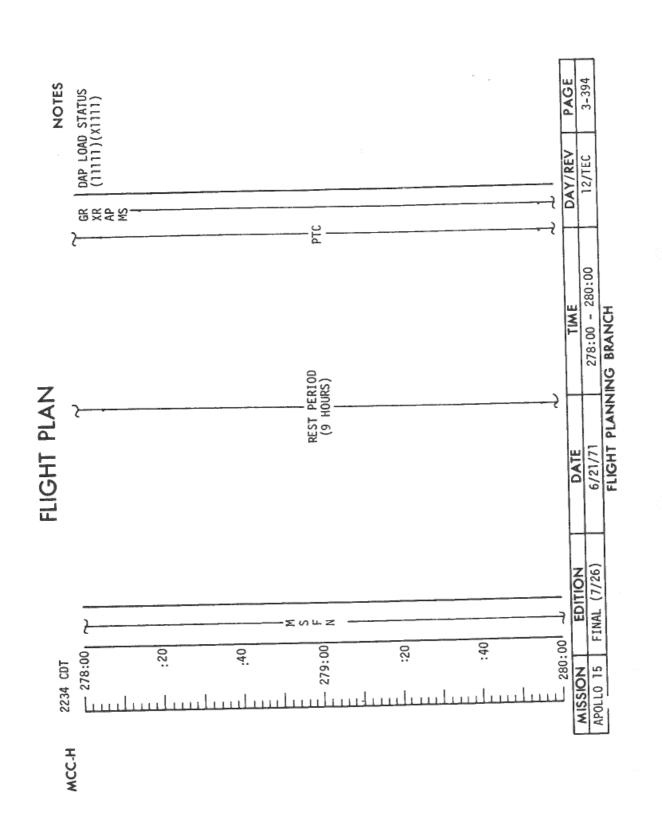


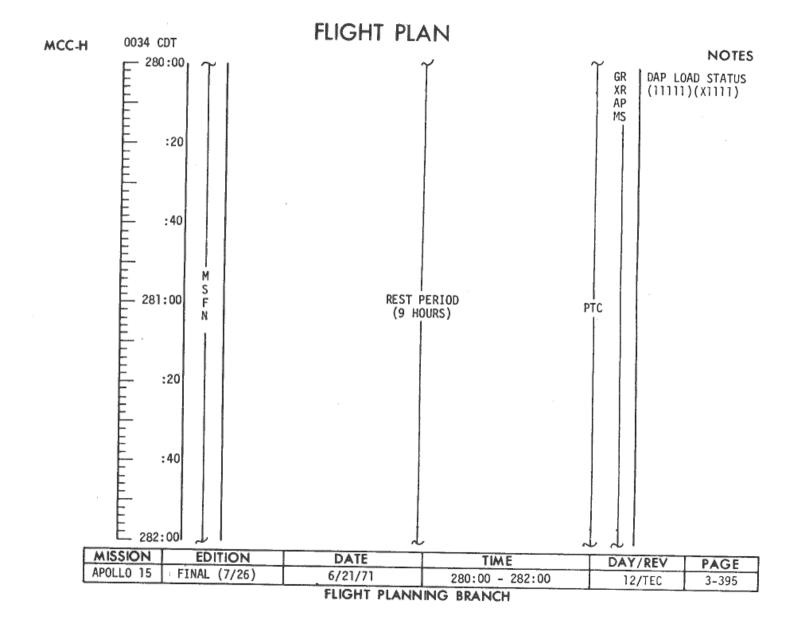


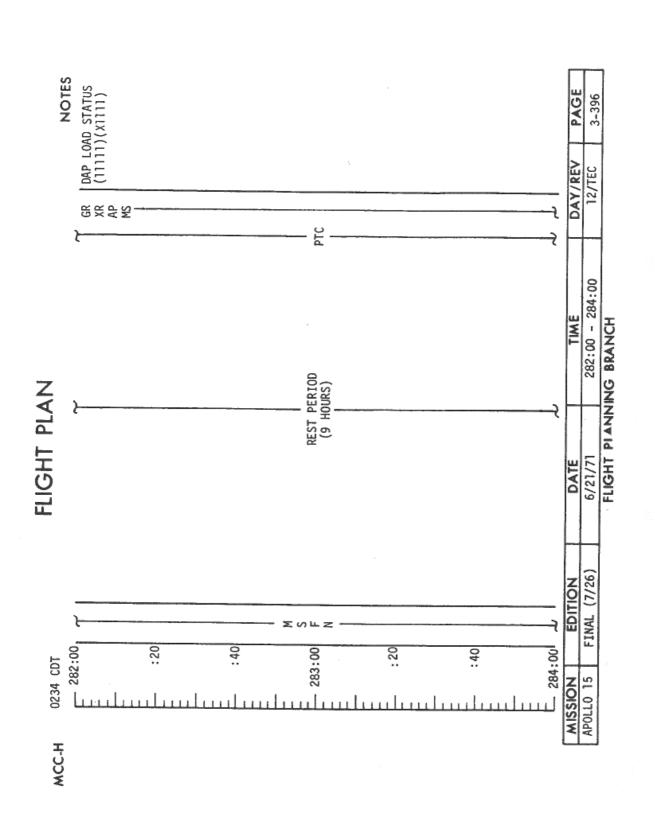
FLIGHT PLAN 2034 CDT MCC-H NOTES _ 276:00, V49 MNVR TO OPTICS CALIBRATION ATTITUDE GR (11101)(066,152,350) HGA P -66, Y 317 XR (x1111) P23 CISLUNAR NAVIGATION AP OPTICS CALIBRATION STAR N70 (00020) MS V49 MNVR TO SIGHTING ATTITUDE EARTH DISTANCE :10 (078,191,330) HGA P -59, Y 359 ~ 93,233 NM P23 CISLUNAR NAVIGATION 3 MARKS ON EACH STAR 1. N70 (00000) (00000) (00120) :20 60 EL NATH (EFH) N88 (+13540)(+86760)(+47847) 2. N70 (00013) (00000) (00120) 13 CAPELLA (EFH) 276:30 3. N70 (00021) (00000) (00110) 21 ALPHARD (ENH) FOV- 7° GET = 276:40 GAMMA RAY GAINSTEP-SHIELD OFF :50 (111111)GAMMA RAY GAINSTEP - SHIELD ON (CTR) E(X1111) ⊥ | V48 (11111)(X1111) $\overline{\lambda}$ DAY/REV PAGE MISSION EDITION DATE TIME APOLLO 15 FINAL (7/26) 6/21/71 276:00 - 277:00 12/TEC 3-392

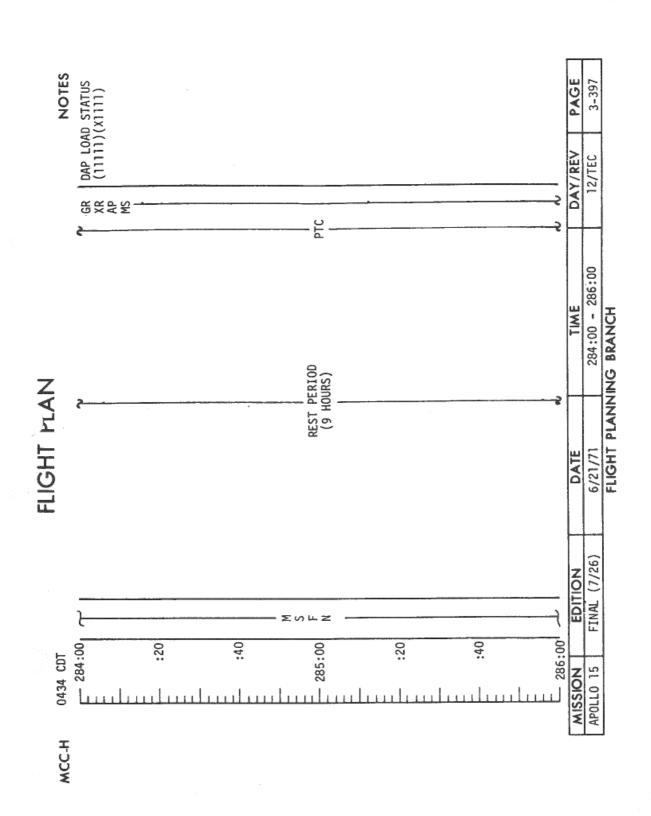
FLIGHT PI "'NING BRANCH

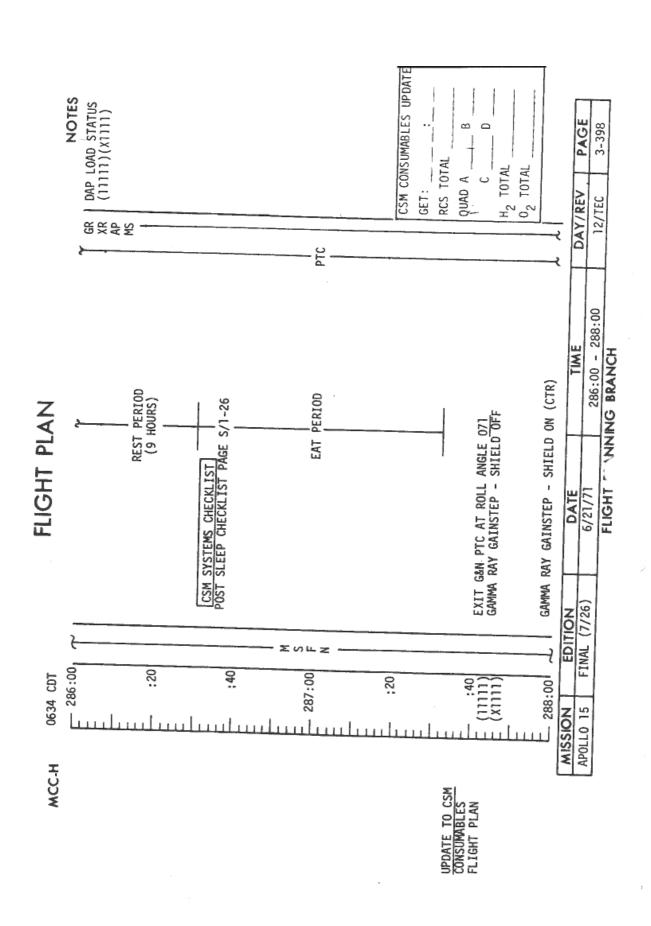


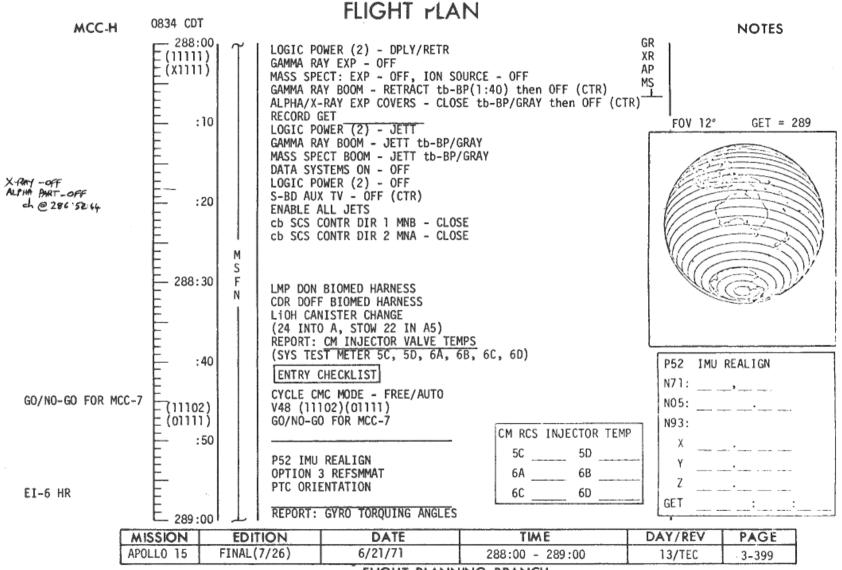




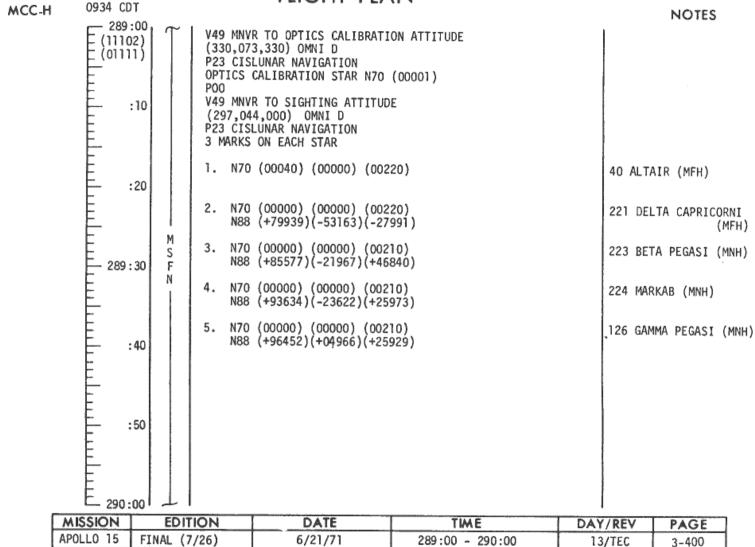








FLIGHT PLANNING BRANCH



FLIGHT PLATING BRANCH

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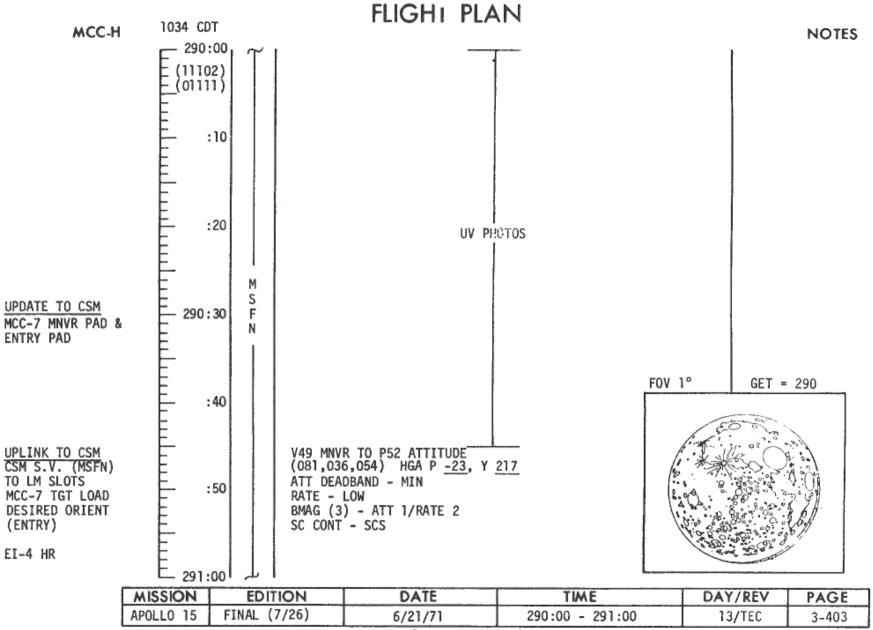
UV PHOTOS - TRANSEARTH COAST

V49 MNVR TO EARTH UV PHOTO ATT (290:15) (226,352,312) OMNI <u>D</u>	V49 MNVR TO MOON PHOTO CALIBRATION ATT (290:40) (002,333,000) HGA P 7, Y 250
CONFIGURE CAMERA: (UV) CM5/EL/105/UV, BRKT, CONT (f8,1/60,∞) (8 FR) RINGSLIDE MAG (N) FR # REMOVE R12 FLIGHT DATA FILE STOWAGE BOX	CONFIGURE CAMERA: (UV CALIBRATION) CM5/EL/105/UV, BRKT, CONT (f8,1/60,∞) (8 FR) RINGSLIDE MAG (N), FR #
DAMP RATES INHIBIT ALL JETS EXCEPT A1 & C2 or B2 & D1, A4, C3, B4, D3 VERIFY FDAI SCALE 5/1 WAIT 5 MIN FOR RATES TO DAMP	DAMP RATES WAIT 5 MIN FOR RATES TO DAMP VERIFY RATES ARE <0.2°/SEC IN ALL AXES 2 FRAMES, FILTER 1, CHANGE SHUTTER TO B
VERIFY RATES ARE <0.2°/SEC IN ALL AXES REMOVE CM5 WINDOW COVER, MOUNT UV CARDBOARD SHADE AND CAMERA	2 FRAMES, FILTER 2, EXP TIME 20 SEC CHANGE SHUTTER TO 1/250 2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500 2 FRAMES, FILTER 4
2 FRAMES, FILTER 1, CHANGE SHUTTER TO B 1 FRAMES, FILTER 2, EXP TIME 20 SEC and 10 250 CHANGE SHUTTER TO 1/250 2 FRAMES, FILTER 3, CHANGE SHUTTER TO 1/500 2 FRAMES, FILTER 4 RECORD FR #	RECORD FR # REMOVE CAMERA AND UV CARDBOARD SHADE NOTE: COMMENTS AS TO CONDITION OF WINDOW 5 REPLACE CM5 WINDOW COVER ENABLE ALL JETS
CONFIGURE CAMERA: (UV COLOR) CM5/EL/105/CEX,BRKT,CONT (f8,1/250,∞) (1 FR) RINGSLIDE MAG (M) P , FR # 1 FRAME, FILTER 4 RECORD FR #	
ch@ 286:53:29 (Q ALTUALLY USED)	

6/21/71

3-402

FINAL (7/26)



FLIGHT PLANNING BRANCH

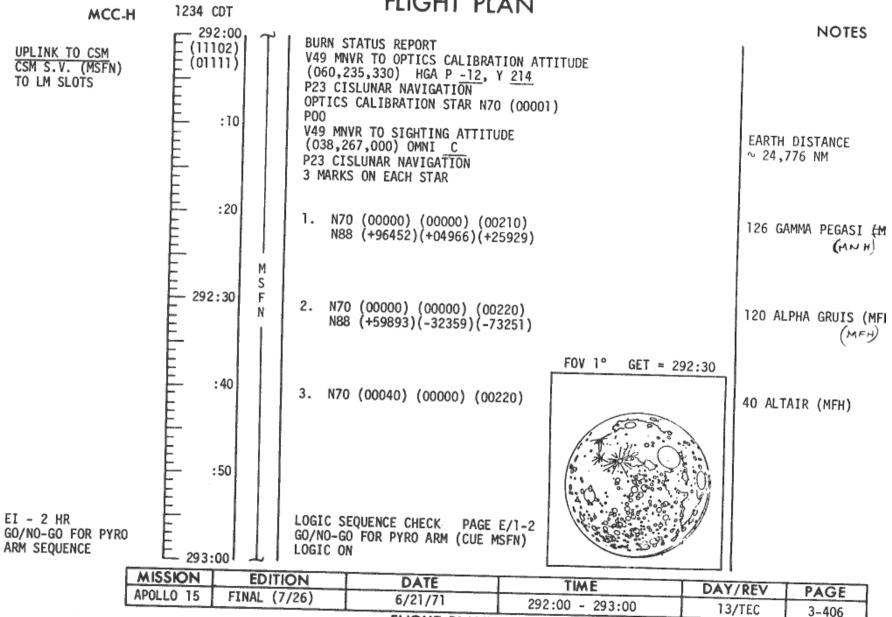
MCC-7 BURN TABLE

MANEUVER	P OR Y RATES	ATT DEVIATION	SHUTDOWN TIME	RESIDUALS
CORRIDOR	10°/SEC	±10°	BT + 1 SEC	TRIM X AXIS ONLY
CONTROL	COMPLETE	COMPLETE	AND $\Delta C_c = 0$	TO 0.2 FPS

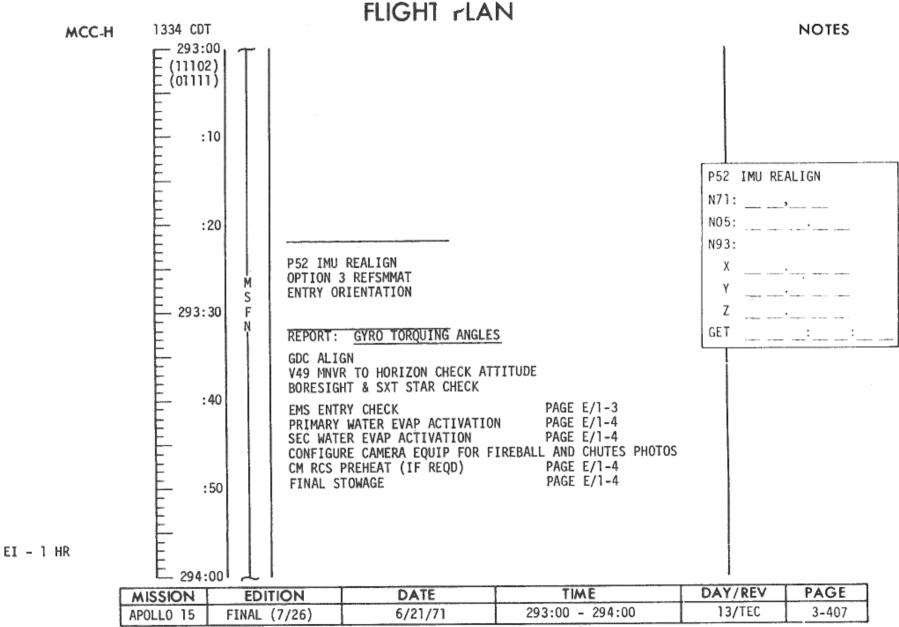
FLIGHT PLAN 1134 CDT MCC-H NOTES - 291:00, (11102)P52 IMU REALIGN STARS ___, ___ (01111) OPTION 3 REFSMMAT PTC ORIENTATION REPORT: GYRO TORQUING ANGLES :10 P52 IMU REALIGN P52 IMU REALIGN OPTION 1 PREFERRED ENTRY ORIENTATION SC CONT - CMC **BURN STATUS REPORT** N93: :20 BMAG (3) - RATE 2 ΔTIG P30 EXTERNAL AV V49 MNVR TO PAD BURN ATT ΧХ BT ۷_{gx} S - TRIM 291:30 GET $X \quad X \quad X$ $X \quad X \quad X$ SXT STAR CHECK X X X P40 SPS THRUSTING OR P41 RCS THRUSTING :40 gz ΔVc X X X FUEL :50 OX TIG: 291:58:20 BT: NOM ZERO X X X UNBAL ΔVT: NOM ZERO ULLAGE: N/A EI - 3 HR MCC-7 ORBIT: N/A 292:00

MISSION	EDITION	DATE	TIME	DAY/REV	PAGE
APOLLO 15	FINAL (7/26)	6/21/71	291:00 - 292:00	13/TEC	34405

FLIGHT PLANNING BRANCH



FLIGHT MANNING BRANCH



FLIGHT PLANNING BRANCH

